

## ACKNOWLEDGEMENTS

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TABLE OF CONTENTS

ABSTRACT . . . . . ii

ACKNOWLEDGEMENTS . . . . . iii

LIST OF FIGURES . . . . . vii

LIST OF TABLES . . . . . ix

LIST OF SYMBOLS . . . . . xi

SECTION 1. THE STACK STRUCTURE, LOADS AND MODELS

    I AND II . . . . . 1

    1.1 The Stack Structure . . . . . 2

    1.2 The Applied Loads . . . . . 2

    1.3.1 The ANSYS Element Used in Models I  
          and II . . . . . 2

    1.3.2 Model I . . . . . 5

    1.3.3 Model II . . . . . 13

SECTION 2. ANALYSIS RESULTS . . . . . 23

SECTION 3. CONCLUSIONS AND RECOMMENDATIONS . . . . . 40

    3.1 Analysis Conclusions . . . . . 41

    3.2 Modelling Recommendations . . . . . 46

    3.3 Stiffening Recommendations . . . . . 48

BIBLIOGRAPHY . . . . . 51

APPENDIX A . . . . . 52

A.1 Exact Calculation of Stack Self  
Weight Dead Load, Models I and II . . . . . 53

A.2 Stack Lateral Wind Load . . . . . 57

A.3 Model I Wind Loads . . . . . 58

A.4 Minimum Required Stack Plate  
Thickness . . . . . 60

A.5 Model II Lateral Wind . . . . . 61

A.6.1 Load Summary for Models I and II . . . . . 64

A.6.2 Model I Joint Loads . . . . . 65

A.6.2.1 Calculation of Load Case 1 (DL)  
Joint Forces . . . . . 65

A.6.2.2 Vertical Joint Loads due to Wind  
Moment, Tangential Horizontal Joint  
Loads due to Wind Shear . . . . . 69

A.6.3 Model II Joint Loads . . . . . 77

A.6.4 Input Data Check of Stack Properties  
at Appropriate Elevations . . . . . 79

A.6.5 Input Data Check of Model I Top Edge  
Joint Forces due to East+West  
(Load Case 2) . . . . . 81

A.6.6 Input Data Check of Model I Top Edge  
Joint Forces due to North+South  
Wind (Load Case 3) . . . . . 88

A.6.7 Input Data Check of Model II Joint  
Forces due to East+West Wind . . . . . 95

APPENDIX B . . . . . 120

    B.1 Model I Element Stresses for Load Case  
        1,2,3,1+2, and 1+3 for Elements About  
        and Adjacent to the Breach . . . . . 121

    B.2 Model II Element Stresses for Load  
        Case 1+2 for Elements About and  
        Adjacent to the Breach . . . . . 160

APPENDIX C . . . . . 180

    C.1 Model I Joint Deflections . . . . . 181

APPENDIX D . . . . . 194

    D.1 Model II ANSYS Analysis . . . . . 195



## LIST OF FIGURES

1.1.1	Stack Structure . . . . .	3
1.2.1	Stack Lateral Wind Load . . . . .	4
1.3.2.1	Model I (and II) Location on Stack Structure . . . . .	7
1.3.2.2	Model I Orientation of Figs. 1.3.2.3 and 1.3.2.4 . . . . .	8
1.3.2.3	Model I, Elevation . . . . .	9
1.3.2.4	Model I, Elevation . . . . .	10
1.3.2.5	Model I, Plan Cross Section . . . . .	11
1.3.3.1	Model II Orientation of Fig. 1.3.3.2 to Fig. 1.3.3.6 . . . . .	14
1.3.3.2	Half Elevation, Model II, EL 0' to 50' . .	15
1.3.3.3	Half Elevation, Model II, EL 0' to 50' . .	16
1.3.3.4	Half Elevation, Model II, EL 50' to 100' .	17
1.3.3.5	Half Elevation, Model II, EL 50' to 100' .	18
1.3.3.6	Total Elevation, Model II, EL 100' to 200'.	19
1.3.3.7	Model II Cross Section at EL 0', 50' and 80' . . . . .	20
1.3.3.8	Model II Cross Section at EL 120', 160' and 200' . . . . .	21
2.1	Model I and II Cross Sectional translations at EL = 50 Ft . . . . .	27
2.2	Joint Displacements of Cross Section at 100' and 200' Elevations, Model II . . .	28
2.3	Model II Breech Region Elements . . . . .	29
2.4	Model II Vertical (z) and Tangential (y) Translations in Global Coordinates . . .	30
2.5	Model II Isometric of Global x or radial joint translations . . . . .	31
2.6	Model II Moments at Element Centers . . . .	33
2.7	Model II Isometric of $M_x$ Moments . . . . .	34
2.8	Model II Isometric of $M_y$ Moments . . . . .	35
2.9	Model II Membrane Stresses in Local Element x Directions . . . . .	36

2.10	Model II Membrane Stresses in Local Element y Directions . . . . .	37
2.11	Model II Bending Stresses in Local Element x Directions . . . . .	38
2.12	Model II Bending Stresses in Local Element y Directions . . . . .	39
3.1.1	Analogous Frame Action About Breech Opening . . . . .	45
3.3.1	Stiffener Recommendations . . . . .	50
A.1.1	Stack Geometry Used to Determine Self Weight with Respect to Plate Thickness and Model Height . . . . .	54
A.5.1	Model II Dimensions Required to Determine Wind Forces at Model Elevations Shown Boxed In . . . . .	62
A.6.2.1	Model I Top Edge Vertical Line Loads Due to Lateral Wind and Self Weight Dead Load . . . . .	66
A.6.2.2	Lateral Shear Along Cylindrical Cross Section . . . . .	67
A.6.2.2.1	Description of Variables Used in Equations for Vertical Flexural and Horizontal Tangential Shear Joint Forces . . . . .	70
A.6.2.2.2	Model I Cross Sectional Joint Angles for East to West Wind . . . . .	73
A.6.2.2.3	Model I Cross Sectional Joint Angles for North to South Wind . . . . .	75

## LIST OF TABLES

1.3.2.1	Model I Top Edge Joint Forces . . . . .	12
1.3.3.1	Summary of Model II Joint Forces at Prescribed Elevations . . . . .	22
2.1a	Model I Top Edge Deflections . . . . .	25
2.1b	Model I Top Edge Deflections . . . . .	26
A.1.1	Heights and Outside Diameters . . . . .	55
A.1.2	Mean Diameters and Self Weights for Plate Thickness of 1, 1½ and 1¾ Inches . . . . .	55
A.1.3	Base Dead Load Stresses for t = 1, 1½ and 1¾ Inches . . . . .	56
A.1.4	Model I and Model II Dead Load Stresses . .	57
A.2.1	Exposure Factors with Respect to Height . .	58
A.2.2	Wind Pressures with Respect to Height . . .	58
A.3.1	Lateral Wind Shears and Overturning Moments for Model I Base and Top . . . . .	59
A.4.1	Summary of Maximum Axial and Flexural Stresses in Stack Plate at Base Elevations . . . . .	61
A.5.1	Wind Forces on Model II for q = 9.2 PSF at Elevations 50', 120', 160' and 200' . .	63
A.6.1.1	Models I and II loads at Respective Model Elevations . . . . .	
A.6.1.2	Radial Dimensions and Section Moduli for Models I and II at the Elevations where Joint Loads were applied . . . . .	65
A.6.2.1.1	Model I Top Edge Self Weight Joint Loads for Load Case I . . . . .	68
A.6.2.2.1	SR52 Input Data . . . . .	72
A.6.2.2.2	SR52 Input Data . . . . .	74
A.6.3.1	SR52 Input Data, Model II, East to West Wind, q = 9.2 PSF . . . . .	78
A.6.4.1	Stack Radii and Section Moduli with Respect to Stack Height, Base and Top Diameters, Plate Thickness and Elevation of Cross Section . . . . .	80

A.6.5.1	Model I: Joint Forces due to Wind Moment	
to	and Shear East to West Wind . . . . .	82
A.6.5.6		
A.6.6.1	Model I: Joint Forces due to Wind Moment	
to	and Shear North to South Wind . . . . .	89
A.6.6.6		
A.6.7.1	Model II: Joint Forces due to Wind Moment	
to	and Shear East to West Wind . . . . .	96
A.6.7.24		
B.1.1	50 Ft Tall Breached Steel Stack Model	
to	Center of Element Stresses . . . . .	122
B.1.38		
B.2.1	200 Ft Tall Breached Steel Stack Model	
to	Center of Element Stresses . . . . .	161
B.2.19		
C.1.1	Model I Joint Deflections . . . . .	182
to		
C.1.12		

## LIST OF SYMBOLS

A	area
$A_m$	mean diameter area
$A_{mb}$	mean diameter area at base
ARCL	arc length
ARCL <sup>1</sup>	angle of arc subtended by ARCL
$A_{TE}$	area at top edge
$C_e$	wind exposure factor
$C_g$	wind gust factor
$C_p$	wind external pressure factor
$C_n$	wind shape factor
D	diameter
$D_B$	base diameter
$D_i, D_{in}$	inside diameter
$D_m$	mean diameter
$D_{mb}$	mean/base diameter
$D_{mt}$	mean top diameter
$D_{mi}$	$(D_{mb} + D_{mt})/2$
$D_o, D_{out}$	outside diameter
DL	dead load
$D_T$	top edge diameter
E	modulus of elasticity
EL	elevation
$F_x$	shear force
$f_{bc}$	compression bending stress
$f_{bt}$	tensile bending stress
$f_{bx}$	bending stress at distance x
H	stack height
h	elevation, height
d	H-h
J1, J2, J3 ... Jn	joint
LL	live load
M	moment
p	$qC_eC_gC_pC_n$

$P_{DL}$	joint dead load
$P_j$	joint load
$q$	Montreal reference wind, 1/100 wind = 9.2 PSF
$R$	radius
$R_i$	inside radius
$R_m$	mean radius
$R_o$	outside radius
$S$	section modulus
$t$	plate thickness
Tang.	tangential force
$U_x, U_y, U_z$	joint displacements in nodal coordinate system
$V$	total shear force
Vol	volumn
$Y$	yield stress
$x, y, z$	local directions
$\alpha$	angle
$\beta$	angle
$\phi_1, \phi_2$	angle
$\sigma_a$	axial stress
$\sigma_b$	bending stress
$\sigma_{DL}$	dead load stress
$\sigma_t$	tangential shear stress
$\theta$	angle
$\Delta\theta$	change in angle $\theta$
$\rho_s$	steel density

SECTION I  
THE STACK STRUCTURE, LOADS AND  
MODELS I AND II

### 1.1: THE STACK STRUCTURE

The stack structure analyzed may be seen in Fig. 1.1.1. Except for the plate thickness of the stack, all dimensions shown were established during preliminary discussions of the analysis problem. The calculation of the model plate thickness may be found in Section A.4 of the Appendix.

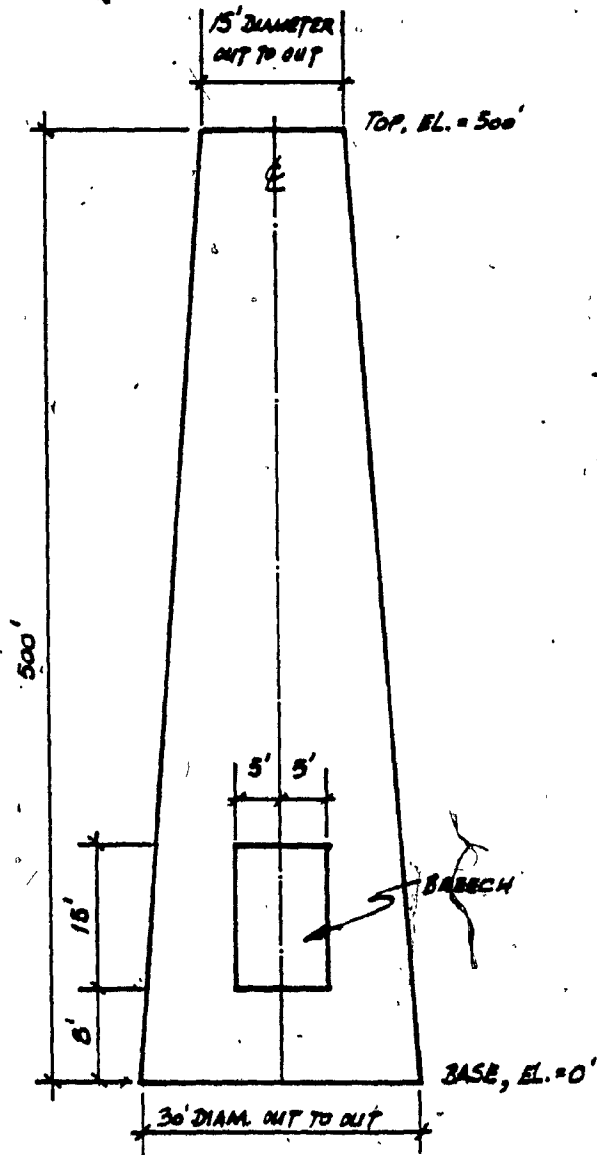
### 1.2: THE APPLIED LOADS

The loads initially considered were self weight dead load, lateral wind, and earthquake. Preliminary calculations in accordance to the National Building Code of Canada concluded that overturning moments due to wind governed over seismic. Subsequently, the static stress analysis presented herein was based on the structure dead load and wind applied North to South and East to West. Fig. 1.2.1 reflects the wind distribution utilized. The determination of this distribution may be found in Section A.2.

### 1.3.1: THE ANSYS ELEMENT USED IN MODELS I AND II

Prior to formulating the finite element Models I and II, an element meeting the requirements of the analysis had to be chosen from the ANSYS element library. The requirements were thin plate behavior, membrane and bending stress capability, and topology establishment described by an element thickness and a two dimensional mesh location at the mid-surface of the element plane.





STACK SPECIFICATIONS

HEIGHT: 500 FT.

DIAMETERS, TOP: 15 FT. OUT TO OUT.

BOTTOM: 30 " " " "

BREECH: 10' x 15', 8 FT. ABOVE BASE.

PLATE THICKNESS: 1/4 INCH.

YOUNG'S MODULUS, E; 29,000 KSI.

POISSONS RATIO,  $\nu$ ; 0.30.

NOTE: STACK ELEVATION

NOT TO SCALE.

FIG. 1.1.1 STACK STRUCTURE

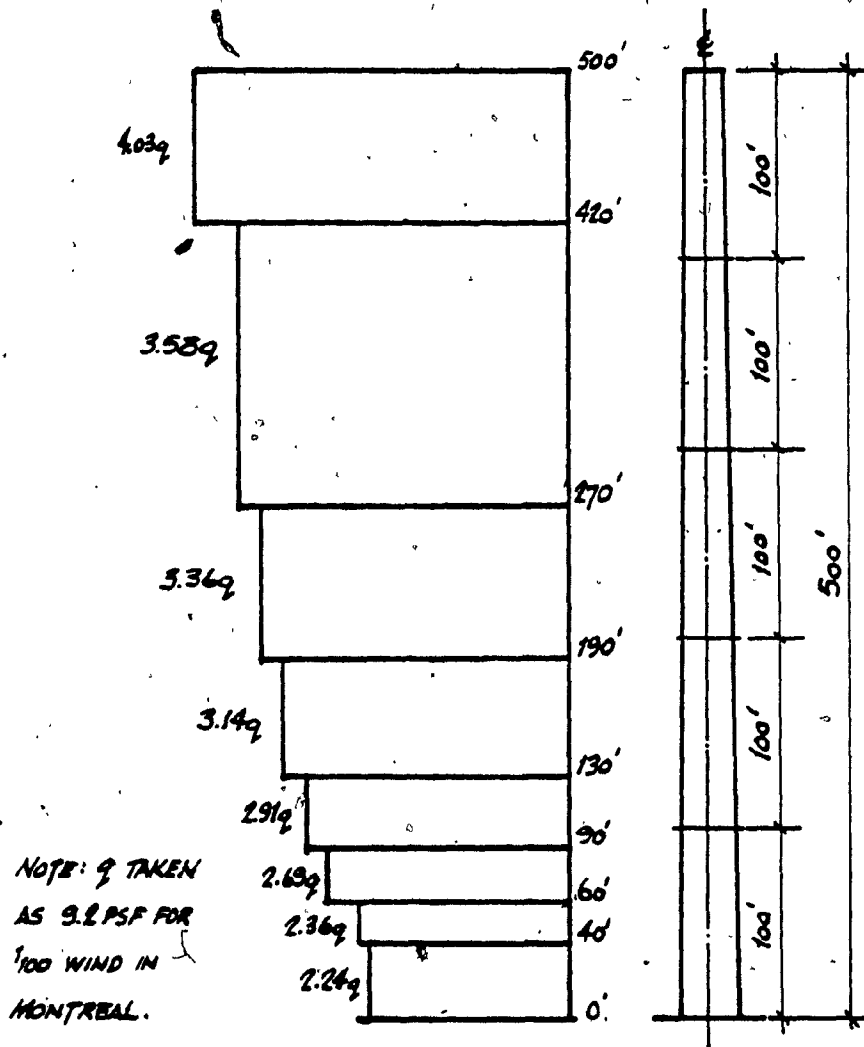


FIG. 1.2.1 STACK LATERAL WIND LOAD

The two ANSYS library elements satisfying these requirements were the Stiff 63 and Stiff 13 elements; the former being a four jointed quadrilateral, and the latter a three jointed triangular isoparametric elastic flat shell element. Each of these elements has six degrees of freedom ( 6 DOF) per joint; three orthogonal translations and three rotations. More specific information about these elements may be found in an ANSYS User's Manual.

#### 1.3.2: MODEL I.

Proceeding element selection, the following assumptions were made to formulate Model I.

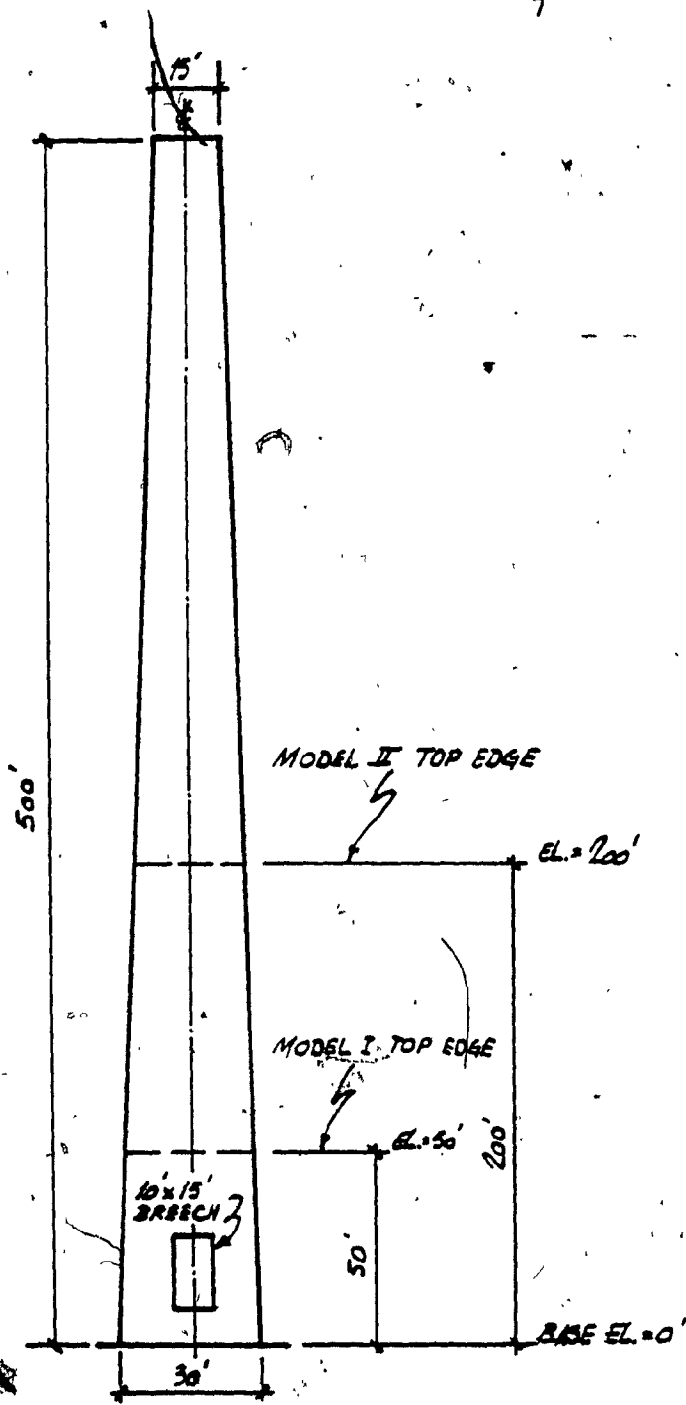
1. It was assumed that a minimum height of nearly two base diameters was sufficient elevation for the model top edge to negate the effects of the breach of the top edge displacements. A height of 50 Ft was subsequently chosen.
2. Because there was little difference between stack diameters at its base and elevation 50 Ft, it was considered acceptable to use a cylinder with a diameter equal to the mean of the stack base and 50 Ft elevation diameters.
3. Because the stack diameter at elevation 50 Ft yielded a large cross sectional area and moment of inertia, even with a relatively thin plate thickness, the effects of loads placed at the model top edge were considered insignificant, provided plate stresses could be maintained below acceptable allowable values.

4.° Since it was difficult to estimate the degree of constraint provided to the model top edge by the portion of the stack above the 50 Ft elevation, it was assumed that no constraint at the top edge would not adversely affect the model performance.

Fig. 1.3.2.1 shows the relation of Model I (and II) to the stack structure. Fig. 1.3.2.2 shows Model I in perspective and opened up as Figs. 1.3.2.3 and 1.3.2.4. Fig. 1.3.2.5 reflects the cylindrical cross section.

The forces applied to the Model I top edge joints to represent the applied load cases are shown in Table 1.3.2.1. These loads were determined during the early stages of the analysis to initiate the analysis solution as soon as possible. A data check conducted at a later date verified the applied joint forces with insignificant variations.

The principles used to determine the applied loads appear in Appendix A.6.2 for Model I, and A.6.3 for Model II. The data checks for Models I and II appear in A.6.6 and A.6.7.



NOTE: STACK ELEVATION NOT TO SCALE.

FIG. 1.3.2.1 MODEL I (AND II) LOCATION ON STACK STRUCTURE

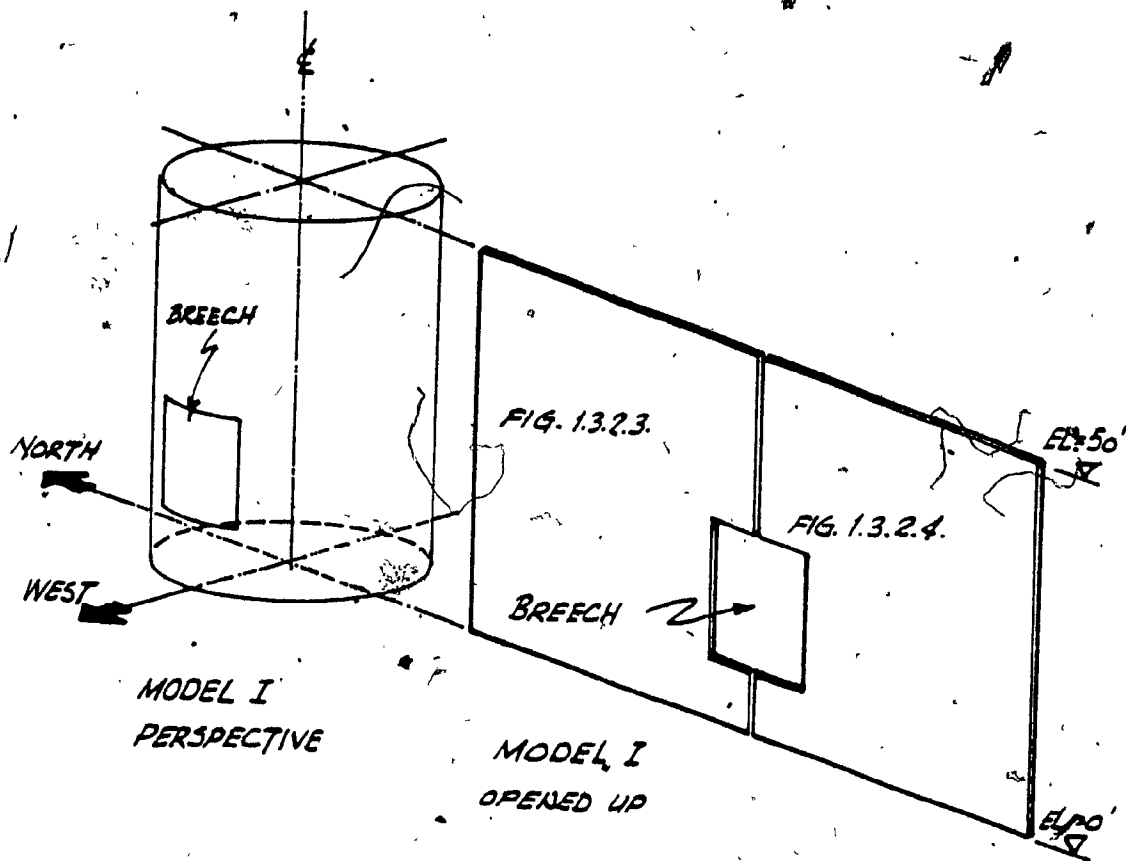


FIG. 1.3.2.2 MODEL I ORIENTATION OF FIGS. 1.3.2.3 AND 1.3.2.4

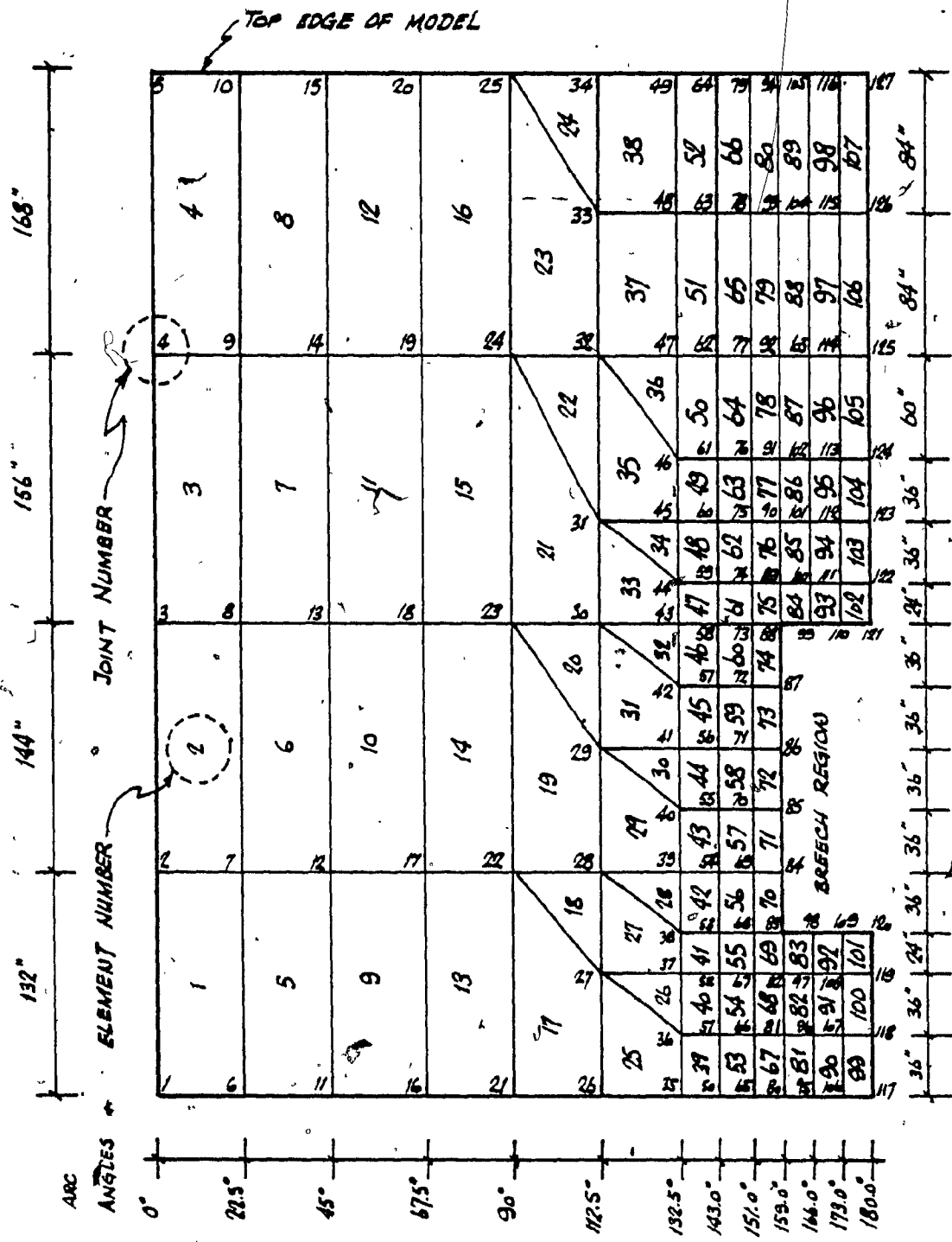


FIG. 1.3.2.3 MODEL I, ELEVATION

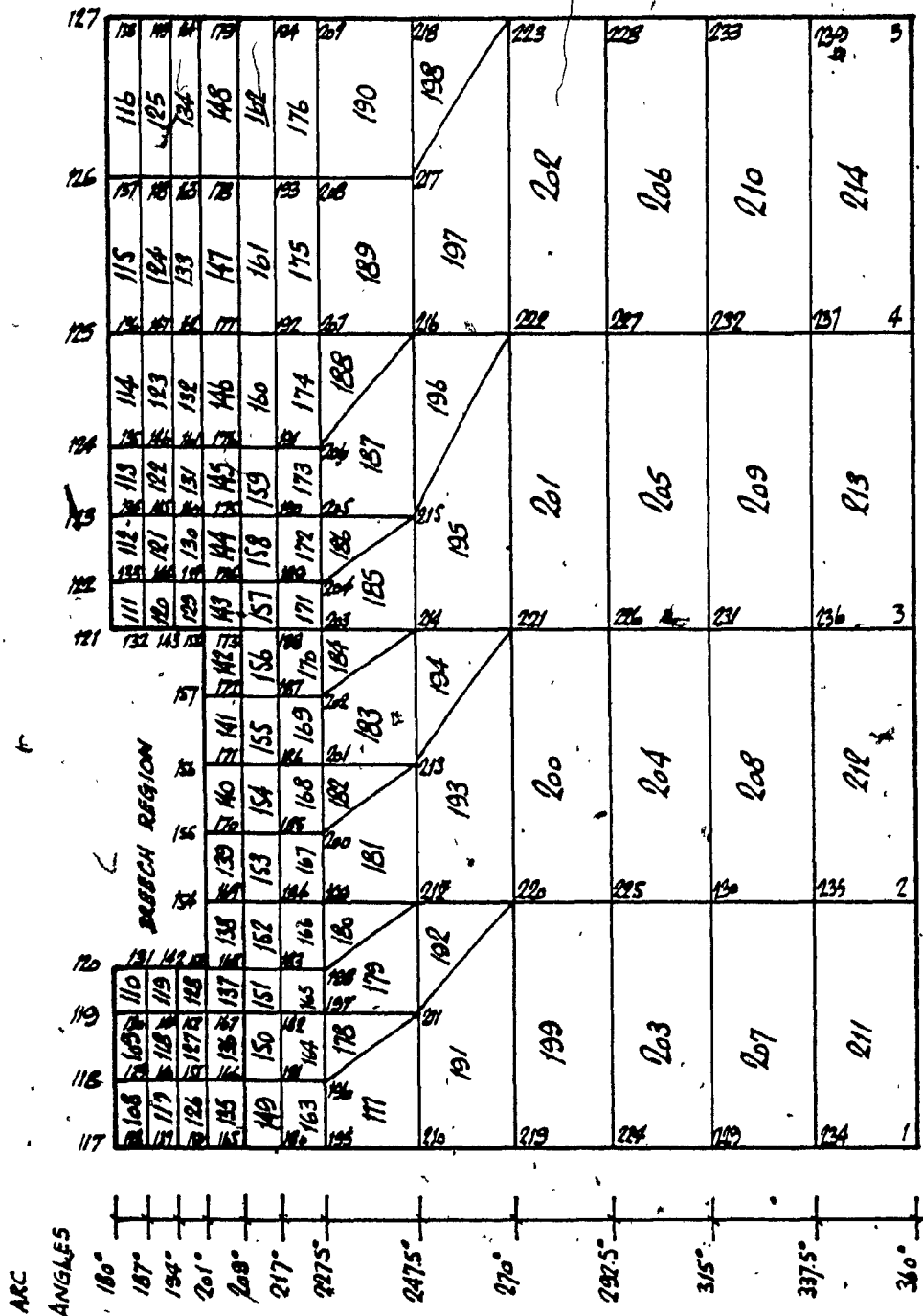


FIG. 1.3.2.4 MODEL I, ELEVATION



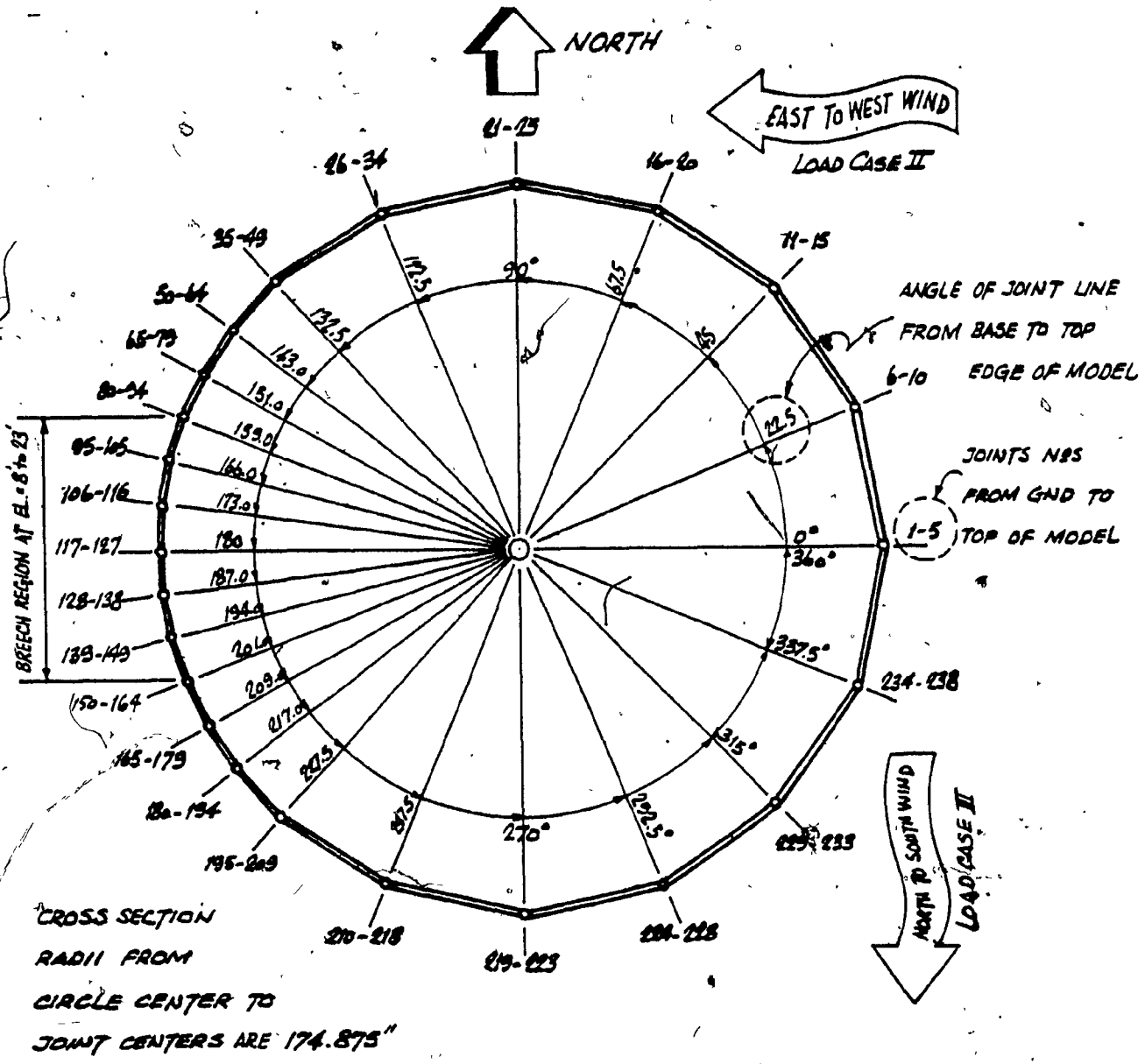


FIG. 1.3.2.5 MODEL I, PLAN CROSS SECTION

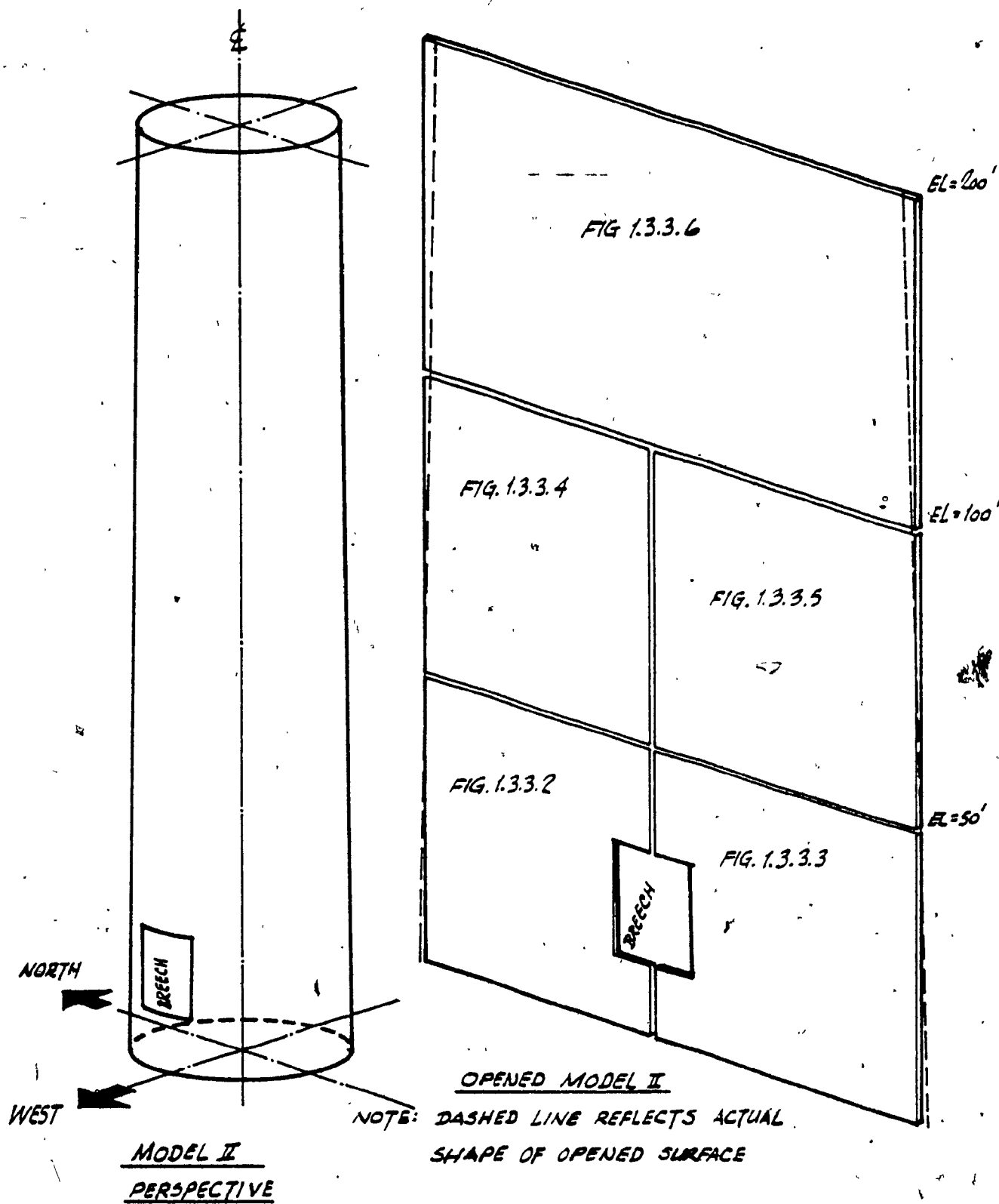
TABLE 1.3.2.1  
MODEL I TOP EDGE JOINT FORCES

Top Edge Joint No.	LOAD CASE 1		Joint Angle $\theta$ Deg.	LOAD CASE 2 East to West Wind			LOAD CASE 3 North to South Wind		
	Vertical Dead Load Lbs.	Flexural Mom. K-Ft		Shear Tangent Lbs.	Flexural Mom. K-Ft	Vertical Lbs.	Shear Tangent Lbs.	Flexural Mom. K-Ft	Vertical Lbs.
5	-97750	896	0	61460	0	0	0	0	-4045
10	"	764	22 1/2	56780	1663	23520	131	23520	-3746
15	"	448	45	43460	2897	43460	448	43460	-2897
20	"	131	67 1/2	23520	3746	56780	764	56780	-1664
25	"	0	90	0	4045	61460	896	61460	0
34	-92330	124	112 1/2	-22210	3557	53630	722	53630	1528
49	-66260	277	132 1/2	-28140	2120	30710	330	30710	1794
64	-40170	235	143	-20180	1030	15210	133	15210	1330
79	-34750	244	151	-19110	709	10590	75	10590	1273
94	-32580	260	159	-19130	499	7340	38	7340	1271
105	-30420	262	166	-18550	314	4630	16	4630	1235
116	"	274	173	-18980	168	2330	4	2330	1264
127	"	279	180	-19120	0	0	0	0	1273
138	"	274	187	-18980	168	2330	4	2330	1264
149	"	262	194	-18550	314	4630	16	4630	1235
164	-32580	244	201	-19130	499	7340	38	7340	1271
179	-34750	244	209	-19110	709	10590	75	10590	1273
194	-40170	235	217	-20180	1030	15210	133	15210	1333
209	-66260	277	227 1/2	-28140	2120	30710	330	30710	1794
218	-92330	123	247 1/2	-22210	3557	53630	722	53630	1528
223	-97750	0	270	0	4045	61460	896	61460	0
228	"	131	292 1/2	23520	3746	56780	764	56780	-1664
233	"	448	315	43460	2897	43460	448	43460	-2897
238	"	764	337 1/2	56780	1663	56780	131	23520	-3746
	1564x10 <sup>3</sup>	7214					7116		

### 1.3.3: MODEL II

Utilizing the same elements employed in Model I, Model II was developed to remove the assumptions used in forming Model I as a means of verifying the assumptions. Subsequently, Model II employed a 200 Ft height, tapered as required, and loaded with the Model I governing load combination of dead load and East to West Wind, with  $q = 9.2$  PSF.

Fig. 1.3.3.1 shows Model II in perspective and opened up as Figs. 1.3.3.2 to 1.3.3.6. Figs. 1.3.3.7 and 1.3.3.8 reflect the Model II cross sections. Table 1.3.3.1 summarizes the cross section joint forces defined with respect to angular location and elevation. Note that dead load was only applied at the 200 Ft elevation to represent the stack material above 200 Ft because the weight of the stack below 200 Ft was automatically determined by the ANSYS program.



MODEL II  
PERSPECTIVE

OPENED MODEL II  
NOTE: DASHED LINE REFLECTS ACTUAL  
SHAPE OF OPENED SURFACE

FIG. 1.3.3.1 MODEL II ORIENTATION OF  
FIG. 1.3.3.2 TO FIG. 1.3.3.6

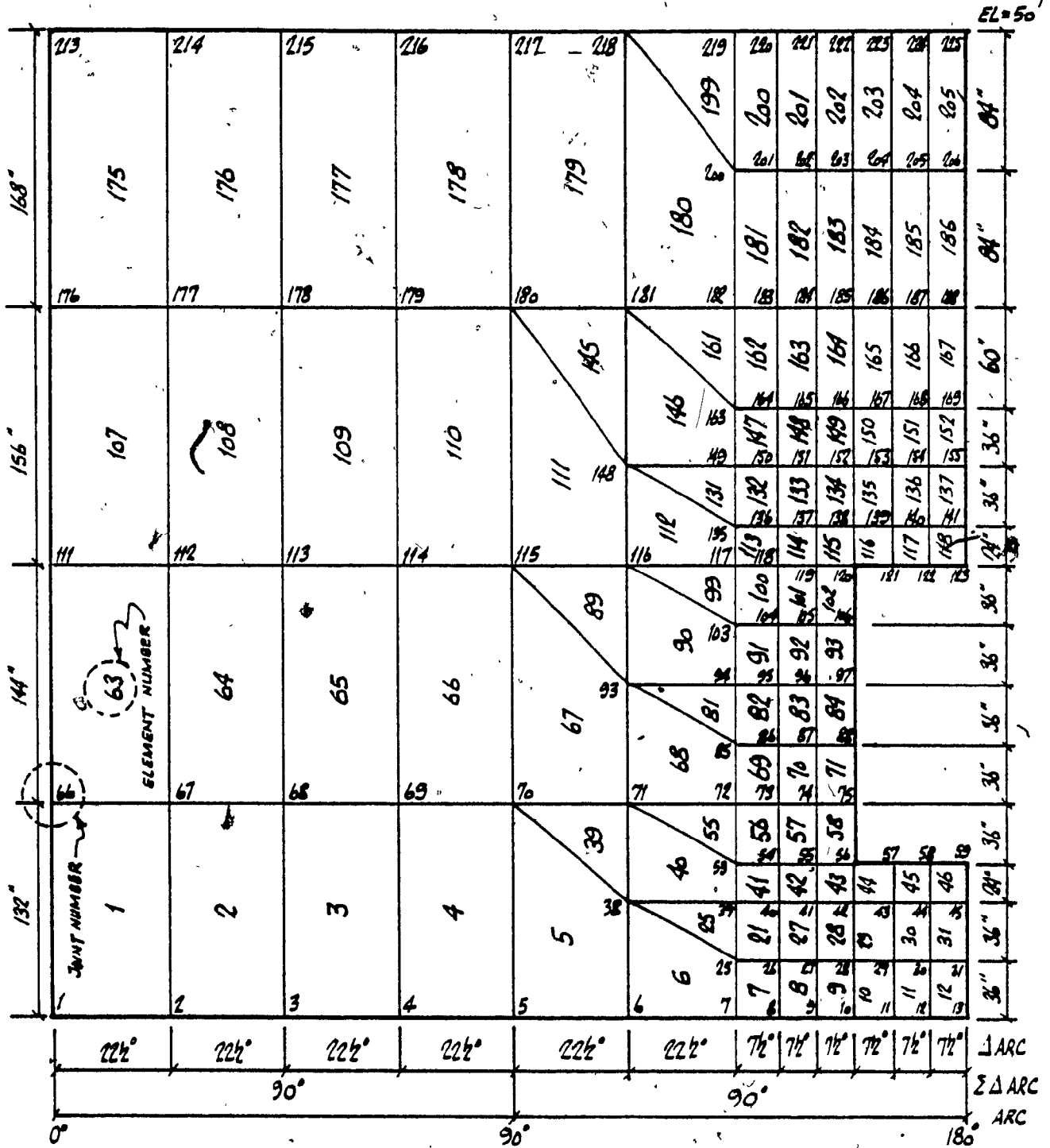


FIG. 1.3.3.2 HALF ELEVATION, MODEL II, EL 0' TO 50'

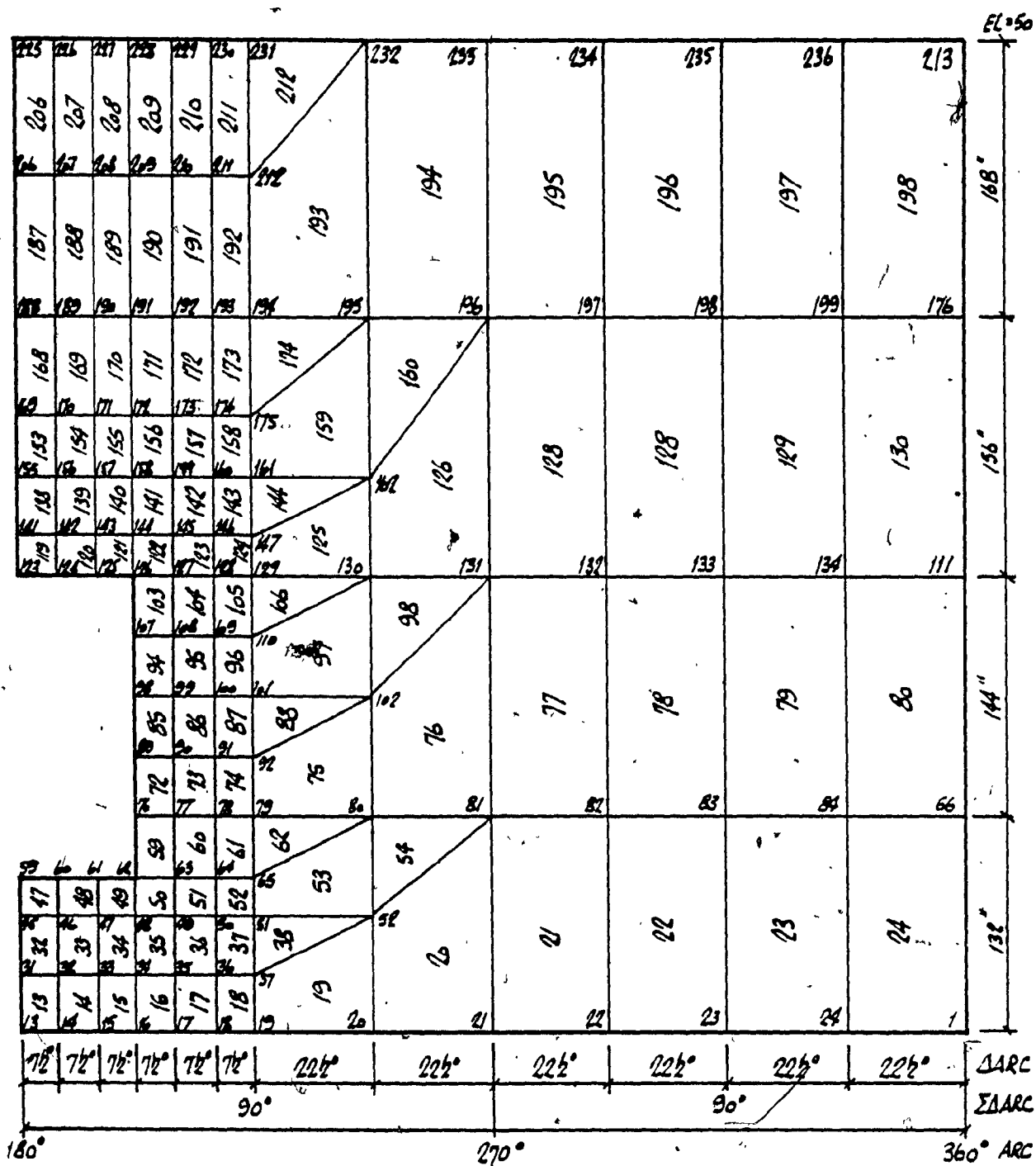


FIG. 1.3.3.3 HALF ELEVATION, MODEL II, EL 0' TO 50'

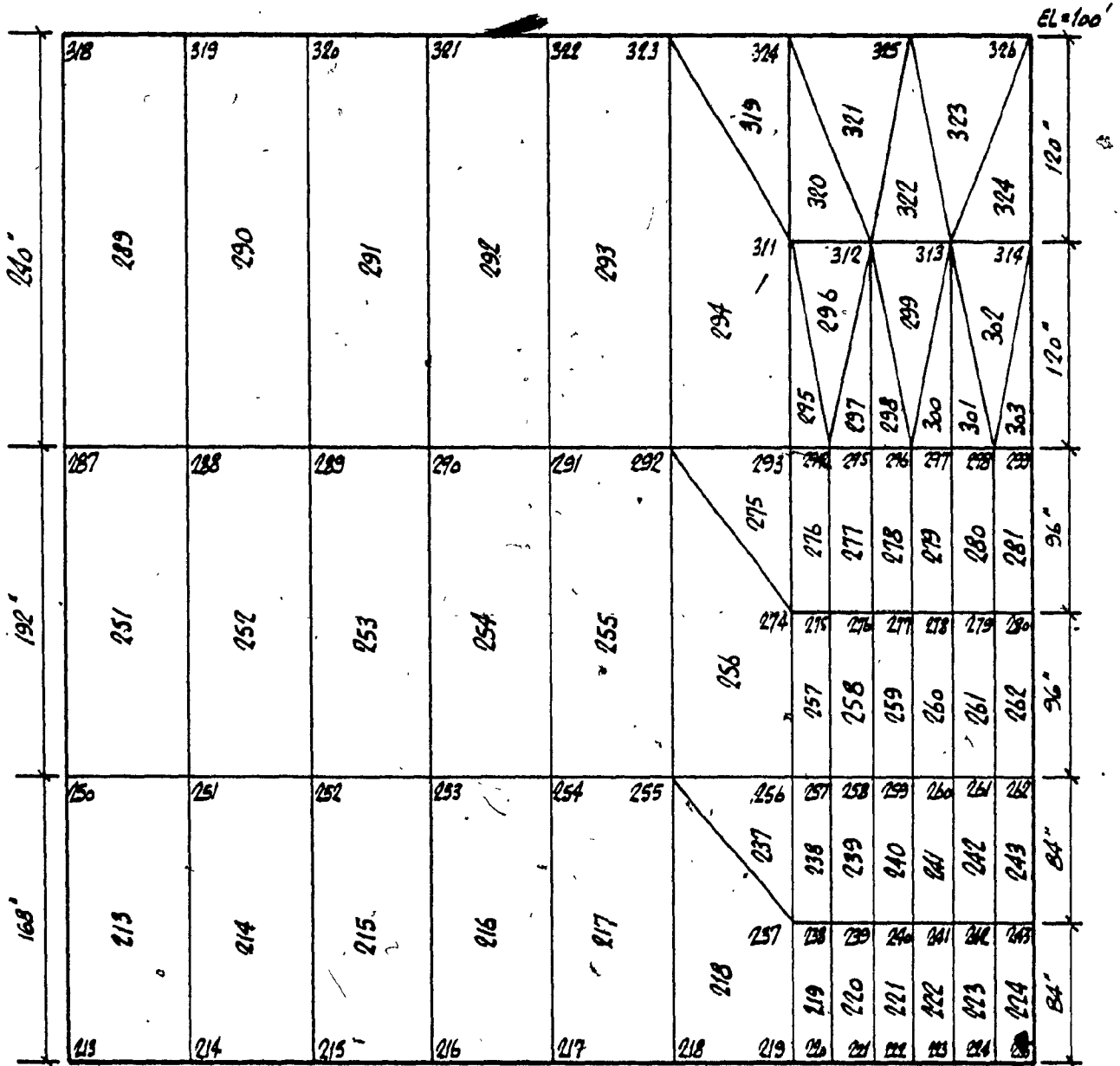


FIG. 1.3.3.4 HALF ELEVATION, MODEL II, EL 50' TO 100'

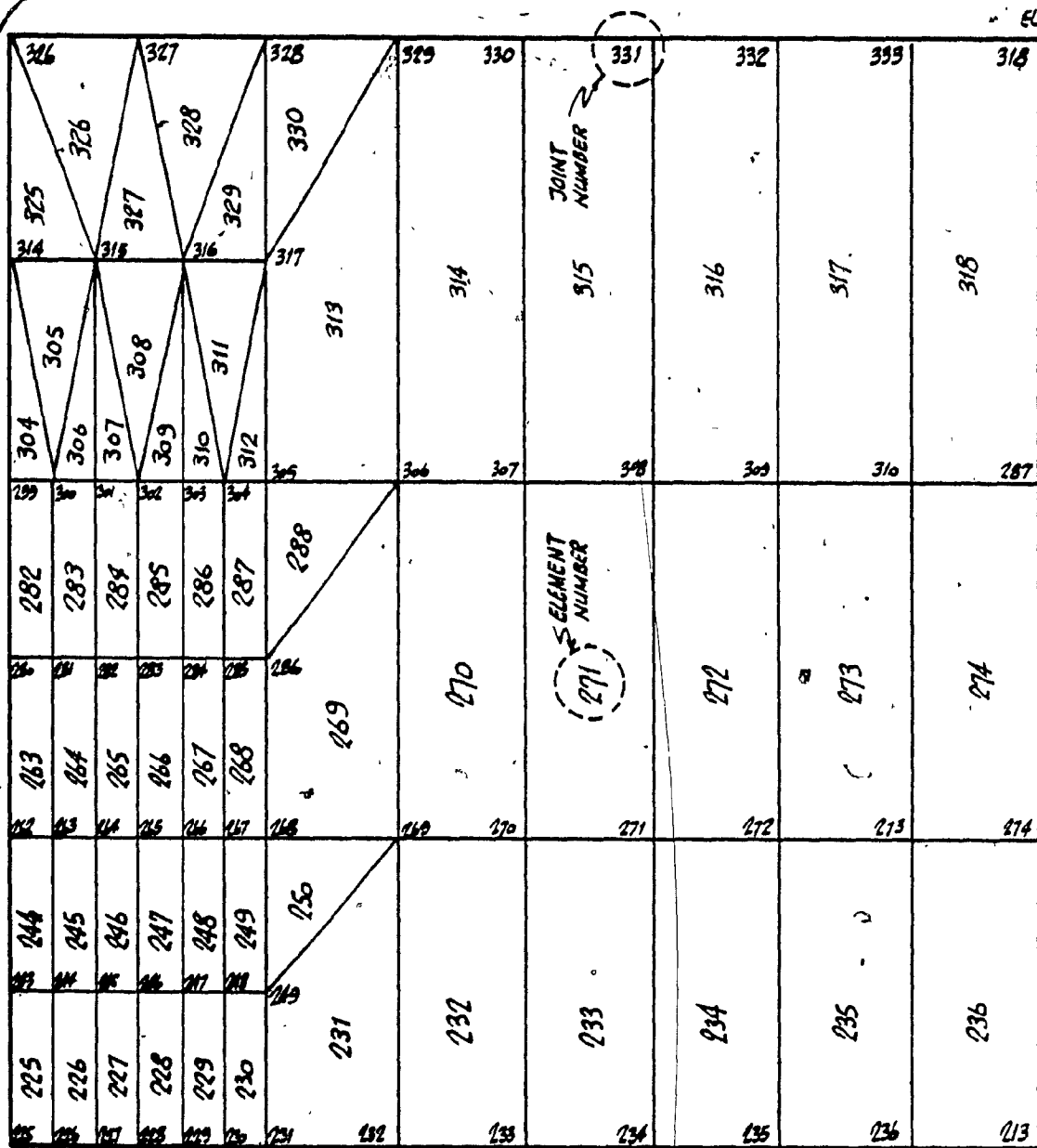


FIG. 1.3.3.5 HALF ELEVATION, MODEL II,  
EL 50' TO 100'



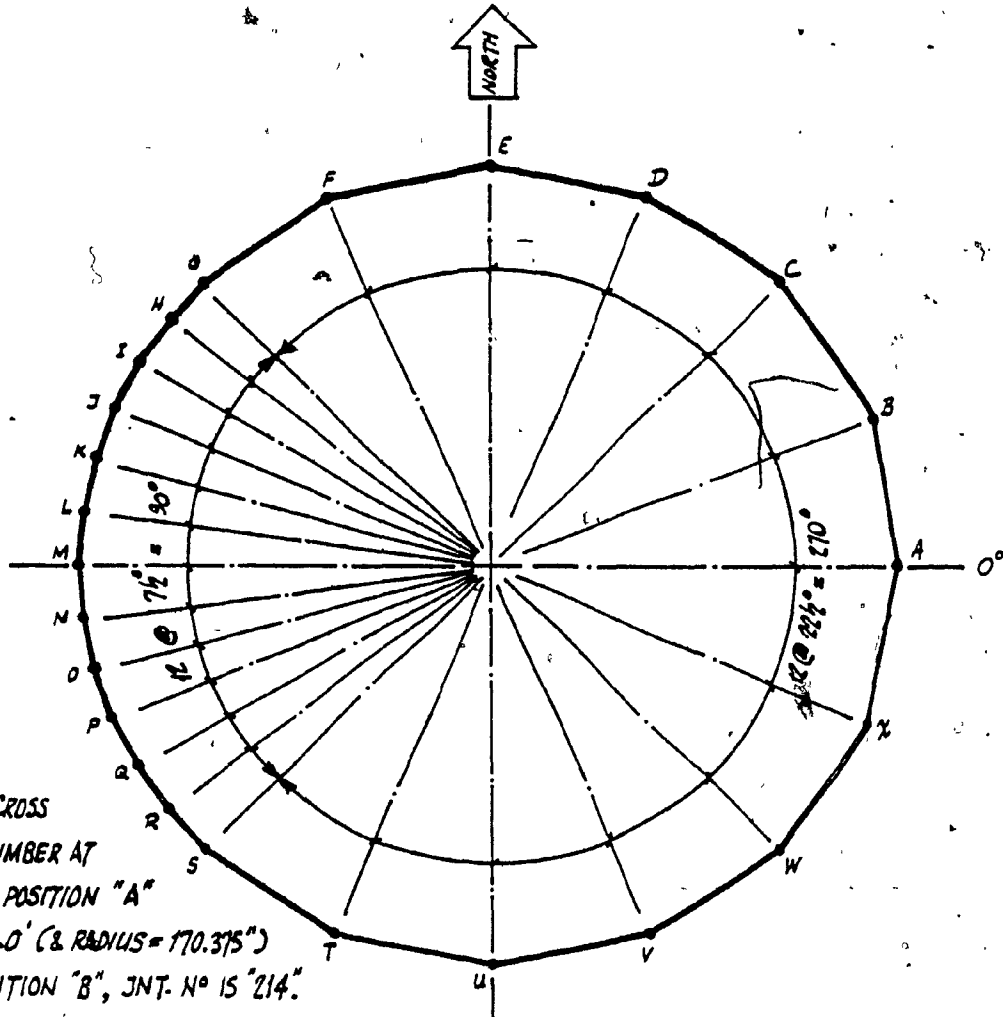
EL. 200'

ELEMENT NUMBER

JOINT NUMBER

398	399	400	401	402	403	404	405	406	407	408	409	410	411	412	413	378
395	396	397	398	399	400	401	402	403	404	405	406	407	408	409	410	
392	393	394	395	396	397	398	399	400	401	402	403	404	405	406	407	394
379	380	381	382	383	384	385	386	387	388	389	390	391	392	393	394	
376	377	378	379	380	381	382	383	384	385	386	387	388	389	390	391	381
363	364	365	366	367	368	369	370	371	372	373	374	375	376	377	378	378
350	351	352	353	354	355	356	357	358	359	360	361	362	363	364	365	350
347	348	349	350	351	352	353	354	355	356	357	358	359	360	361	362	
334	335	336	337	338	339	340	341	342	343	344	345	346	347	348	349	334
331	332	333	334	335	336	337	338	339	340	341	342	343	344	345	346	
318	319	320	321	322	323	324	325	326	327	328	329	330	331	332	333	318

FIG. 1.3.3.6 TOTAL ELEVATION, MODEL II, EL. 100' TO 200'



NOTE:

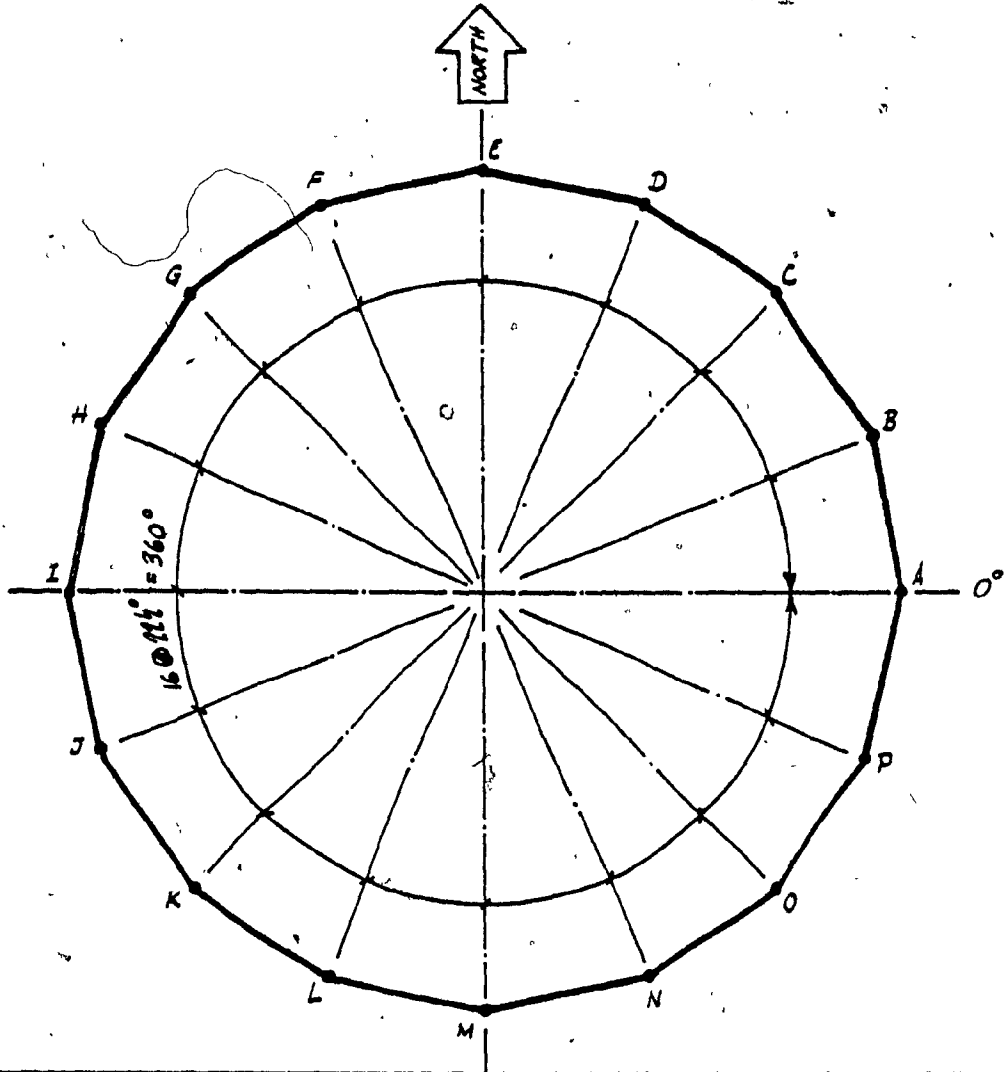
BY EXAMPLE, THE CROSS SECTION JOINT NUMBER AT CROSS SECTIONAL POSITION "A" AT ELEVATION = 50.0' (& RADIUS = 170.375") IS "213", FOR POSITION "B", JNT. N° IS "214".

		JOINT LOCATIONS											
ELEV.	RADIUS	A	B	C	D	E	F	G	H	I	J	K	L
0.0'	179.375"	1	2	3	4	5	6	7	8	9	10	11	12
50.0'	170.375"	213	214	215	216	217	218	219	220	221	222	223	224
80.0'	164.375"	287	288	289	290	291	292	293	294	295	296	297	298

		JOINT LOCATIONS											
ELEV.	RADIUS	M	N	O	P	Q	R	S	T	U	V	W	X
0.0'	179.375"	13	14	15	16	17	18	19	20	21	22	23	24
50.0'	170.375"	225	226	227	228	229	230	231	232	233	234	235	236
80.0'	164.375"	299	300	301	302	303	304	305	306	307	308	309	310

TABLE OF CROSS SECTIONAL JOINT NUMBERS WRT CROSS SECTION, ELEVATION (& RADIUS).

FIG. 1.3.3.7 MODEL II CROSS SECTION AT EL 0', 50' AND 80'



		JOINT LOCATIONS															
ELEV.	RADIUS	A	B	C	D	E	F	G	H	I	J	K	L	M	N	O	P
120.0'	157.775'	334	335	336	337	338	339	340	341	342	343	344	345	346	347	348	349
160.0'	150.575'	366	367	368	369	370	371	372	373	374	375	376	377	378	379	380	381
200.0'	143.375'	398	399	400	401	402	403	404	405	406	407	408	409	410	411	412	413

TABLE OF CROSS SECTIONAL JOINT NUMBERS W.R.T. CROSS SECTION ELEVATION (& RADIUS)

FIG. 1.3.3.8 MODEL II CROSS SECTION AT EL 120', 160' AND 200'

TABLE 1.3.3.1  
 -SUMMARY OF MODEL II JOINT FORCES AT PRESCRIBED ELEVATIONS

Elev.	50 Ft		80 Ft		120 Ft		160 Ft		200 Ft			
	LL Lbs.	Tang. Lbs.	LL Lbs.	Tang. Lbs.	LL Lbs.	Tang. Lbs.	LL Lbs.	Tang. Lbs.	DL Lbs.	LL Lbs.	Tang. Lbs.	DL+LL Lbs.
0	2704	0	5190	0	5595	0	5783	0	-58405	294448	0	236043
22½	2498	1023	4795	1435	5169	1480	5343	1462	"	272034	9838	213629
45	1912	1781	3670	2497	3956	2575	4089	2545	"	208206	17125	149801
67½	1035	2302	1986	3228	2141	3329	2213	3290	"	112680	22138	54275
90	0	2486	0	3487	0	3596	0	3554	"	0	23909	-58405
112½	-1035	2302	-1986	3228	-2141	3329	-2213	3290	"	-112680	22138	-171085
135	-1275	1340	-2447	1880	-3956	2575	-4089	2545	-58405	-208206	17125	-266611
142½	-715	512	-1373	717	-	-	-	-	-	-	-	-
150	-781	421	-1498	590	-	-	-	-	-	-	-	-
157½	-833	323	-1598	454	-5169	1480	-5343	1462	-58405	-272034	9838	-330439
165	-871	221	-1671	310	-	-	-	-	-	-	-	-
172½	-894	118	-1715	166	-	-	-	-	-	-	-	-
180	-901	0	-1730	0	-5595	0	-5783	0	-58405	-294448	0	-352853
187½	-894	118	-1715	166	-	-	-	-	-	-	-	-
195	-871	221	-1671	310	-	-	-	-	-	-	-	-
202½	-833	323	-1598	454	-5169	-1480	-5343	-1462	-58405	-272034	-9838	-330439
210	-781	421	-1498	590	-	-	-	-	-	-	-	-
217½	-715	512	-1373	717	-	-	-	-	-	-	-	-
225	-1275	1340	-2447	1880	-3956	-2575	-4089	-2545	-58405	-208206	-17125	-266611
247½	-1035	-2302	-1986	-3228	-2141	-3329	-2213	-3290	"	-112680	-22138	-171085
270	0	-2486	0	-3487	0	-3596	0	-3554	"	0	-23909	-58405
292½	1035	-2302	1986	-3228	2141	-3329	2213	-3290	"	112680	-22138	54275
315	1912	-1781	3670	-2497	3956	-2575	4089	-2545	"	208206	-17125	149801
337½	2498	-1023	4795	-1435	5169	-1480	5343	-1462	-58405	272034	-9838	213629

LL = Vertical Live Load, Tang. = Tangential Load, DL = Vertical Dead Load  
 due to Wind Shear due to Wind Shear

SECTION II  
ANALYSIS RESULTS

By comparing the results of Model I and Model II, it was apparent that Model II was more representative of the actual stack behavior. The single most revealing comparison towards this conclusion may be seen in Fig. 2.1 where Model I and Model II cross sectional translations at stack elevation 50 Ft. are plotted together. Because the dead load plus East to West Wind, the governing load combination for the breach region, yielded excessive if not unrealistic cross sectional translations for Model I, it was decided that Model I could be discounted henceforth. The tabular summary of Model I ANSYS analysis translations at the 50 Ft elevation cross section for the three applied loads and two load combinations may be seen in Tables 2.1a and 2.1b. The translations at the 50 Ft elevation for Model II may be found in the ANSYS analysis output in Appendix D. Note that Model II was loaded directly with the Model I governing load combination of dead load plus East to West Wind, with  $q = 9.2$  PSF.

Concentrating on Model II, Fig. 2.2 reflects the cross sectional translations at elevations 200 Ft and 100 Ft. Note that the 200 Ft elevation is the Model II top free edge. Fig. 2.3 reflects the elements investigated to determine the analysis results about the breach region. Figs. 2.4 and 2.5 display the translations in the plate about the breach region. Fig. 2.4, an elevation view, depicts vertical and tangential translations,  $U_z$  and  $U_y$ , respectively, and Fig. 2.5 the radial, or  $U_x$  translation. Note that all translations are in global coordinates.

TABLE 2.1a  
MODEL I TOP EDGE DEFLECTIONS

Joint No.	Dead Load			(E+W) Wind			(N+S) Wind		
	Ux	Uy	Uz	Ux	Uy	Uz	Ux	Uy	Uz
5	-0.019	~0.0	-0.025	-0.049	~0.0	0.017	~0.0	-0.033	~0.0
10	-0.005	0.006	-0.025	-0.037	0.017	0.016	-0.017	-0.030	0.007
15	0.028	0.002	-0.026	-0.007	0.025	0.011	-0.027	-0.021	0.012
20	0.043	-0.011	-0.026	0.017	0.023	0.006	-0.023	-0.012	0.015
25	-0.022	-0.014	-0.023	-0.011	0.021	0.001	-0.018	-0.004	0.017
34	-0.155	0.022	-0.021	-0.099	0.043	-0.003	-0.047	0.009	0.017
49	-0.210	0.087	-0.027	-0.125	0.083	-0.012	-0.079	0.030	0.013
64	-0.121	0.118	-0.036	-0.057	0.100	-0.021	-0.074	0.044	0.010
79	0.007	0.126	-0.05	0.034	0.102	-0.028	-0.059	0.054	0.007
94	0.160	0.114	-0.054	0.141	0.090	-0.035	-0.041	0.060	0.005
105	0.285	0.087	-0.060	0.227	0.068	-0.039	-0.026	0.065	0.003
116	0.374	0.047	-0.064	0.287	0.036	-0.043	-0.012	0.067	0.001
127	0.405	~0.0	-0.066	0.308	~0.0	-0.044	~0.0	0.068	~0.0
138	0.374	-0.047	-0.064	0.287	-0.036	-0.043	0.012	0.067	-0.001
149	0.285	-0.087	-0.060	0.227	-0.068	-0.039	0.026	0.065	-0.003
164	0.160	-0.114	-0.054	0.141	-0.090	-0.035	0.041	0.060	-0.005
179	0.007	-0.126	-0.045	0.034	-0.102	-0.028	0.059	0.054	-0.007
194	-0.121	-0.118	-0.036	-0.057	-0.100	-0.021	0.074	0.044	-0.010
209	-0.210	-0.087	-0.027	-0.125	-0.083	-0.012	0.079	0.030	-0.013
218	-0.155	-0.022	-0.021	-0.099	-0.043	-0.003	0.047	0.009	-0.017
223	-0.022	0.014	-0.023	-0.011	-0.021	0.001	0.018	-0.004	-0.017
228	0.043	0.011	-0.026	0.017	-0.023	0.006	0.023	-0.012	-0.015
233	0.028	-0.002	-0.026	-0.007	-0.025	0.011	0.027	-0.021	-0.012
238	-0.005	-0.006	-0.025	-0.037	-0.017	0.016	0.017	-0.030	-0.007

Note: All deflections in inches and in nodal coordinate system

Ux = Radial, Uy = Tangential, Uz = Vertical

TABLE 2.1b  
MODEL I TOP EDGE DEFLECTIONS

Joint No.	DL + [9.2(E+W)Wind]			DL + [9.2(N+S)Wind]		
	Ux	Uy	Uz	Ux	Uy	Uz
5	-0.470	0.0	0.131	-0.019	-0.304	-0.025
10	-0.345	0.162	0.122	-0.161	-0.270	0.039
15	-0.036	0.232	0.075	-0.220	-0.191	0.084
20	0.199	0.201	0.029	-0.169	-0.121	0.112
25	-0.123	0.179	-0.014	-0.188	-0.051	0.133
34	-1.066	0.418	-0.049	-0.587	0.105	0.135
49	-1.360	0.851	-0.137	-0.937	0.363	0.093
64	-0.645	1.038	-0.029	-0.802	0.523	0.056
79	0.320	1.064	-0.308	-0.536	0.623	0.014
94	1.457	0.942	-0.376	-0.217	0.666	-0.008
105	2.373	0.713	-0.419	0.046	0.685	-0.032
116	3.014	0.378	-0.460	0.264	0.663	-0.055
127	3.239	0.0	-0.471	0.405	0.626	-0.066
138	3.014	-0.378	-0.460	0.484	0.569	-0.073
149	2.373	-0.713	-0.419	0.524	0.511	-0.088
164	1.457	-0.942	-0.376	0.537	0.438	-0.100
179	0.320	-1.064	-0.303	0.550	0.371	-0.109
194	-0.645	-1.038	-0.229	0.560	0.287	-0.128
209	-1.360	-0.851	-0.137	0.517	0.189	-0.147
218	-1.066	-0.418	-0.049	0.277	0.061	-0.172
223	-0.123	-0.179	-0.014	0.144	-0.023	-0.179
228	0.199	-0.201	0.029	0.255	-0.099	-0.164
233	-0.036	-0.232	0.075	0.476	-0.195	-0.136
238	-0.345	-0.162	0.122	0.151	-0.282	-0.089

Note: All deflections in inches and in nodal coordinate system

Ux = Radial, Uy = Tangential, Uz = Vertical, DL = Vertical Dead Load



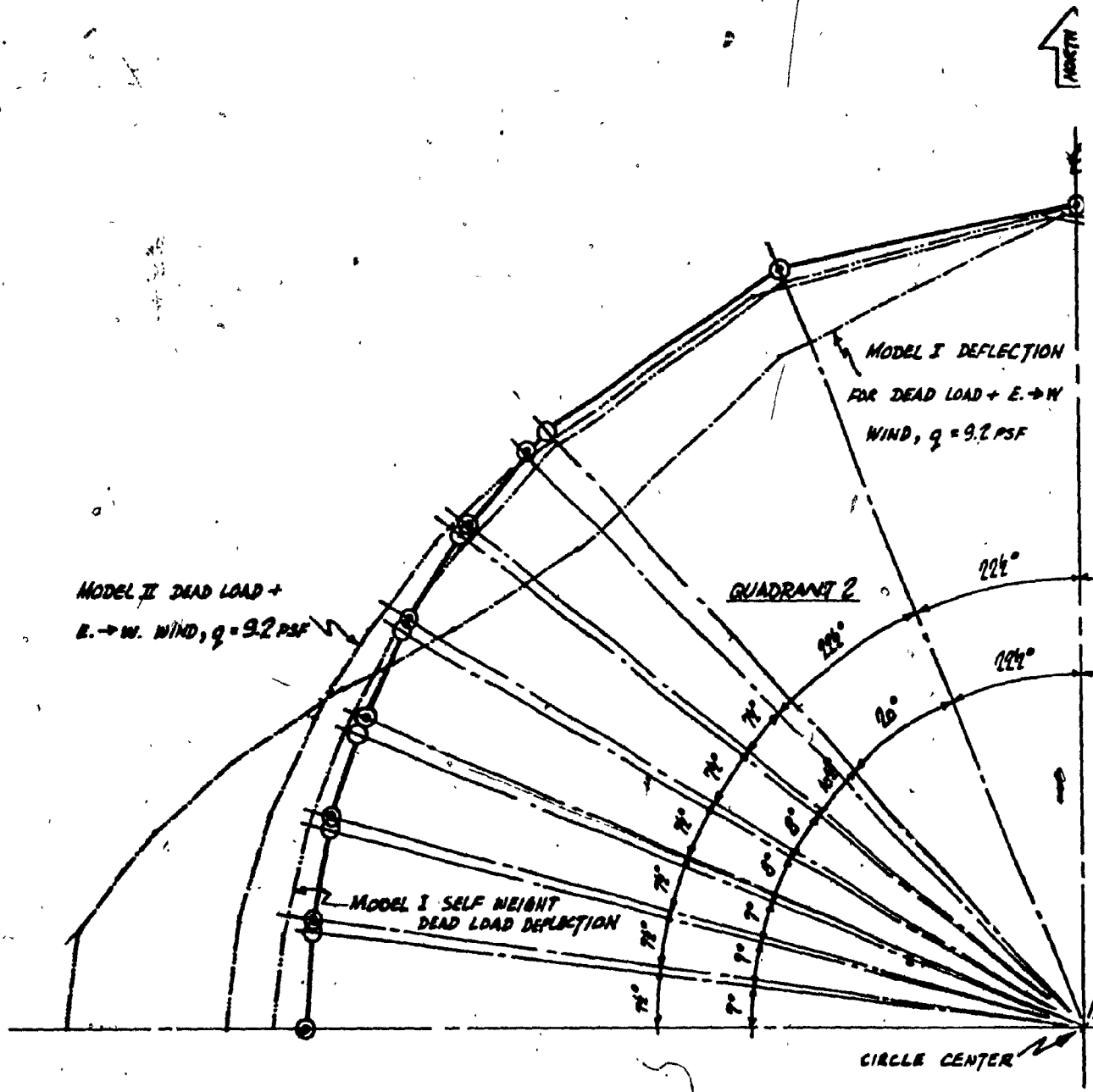
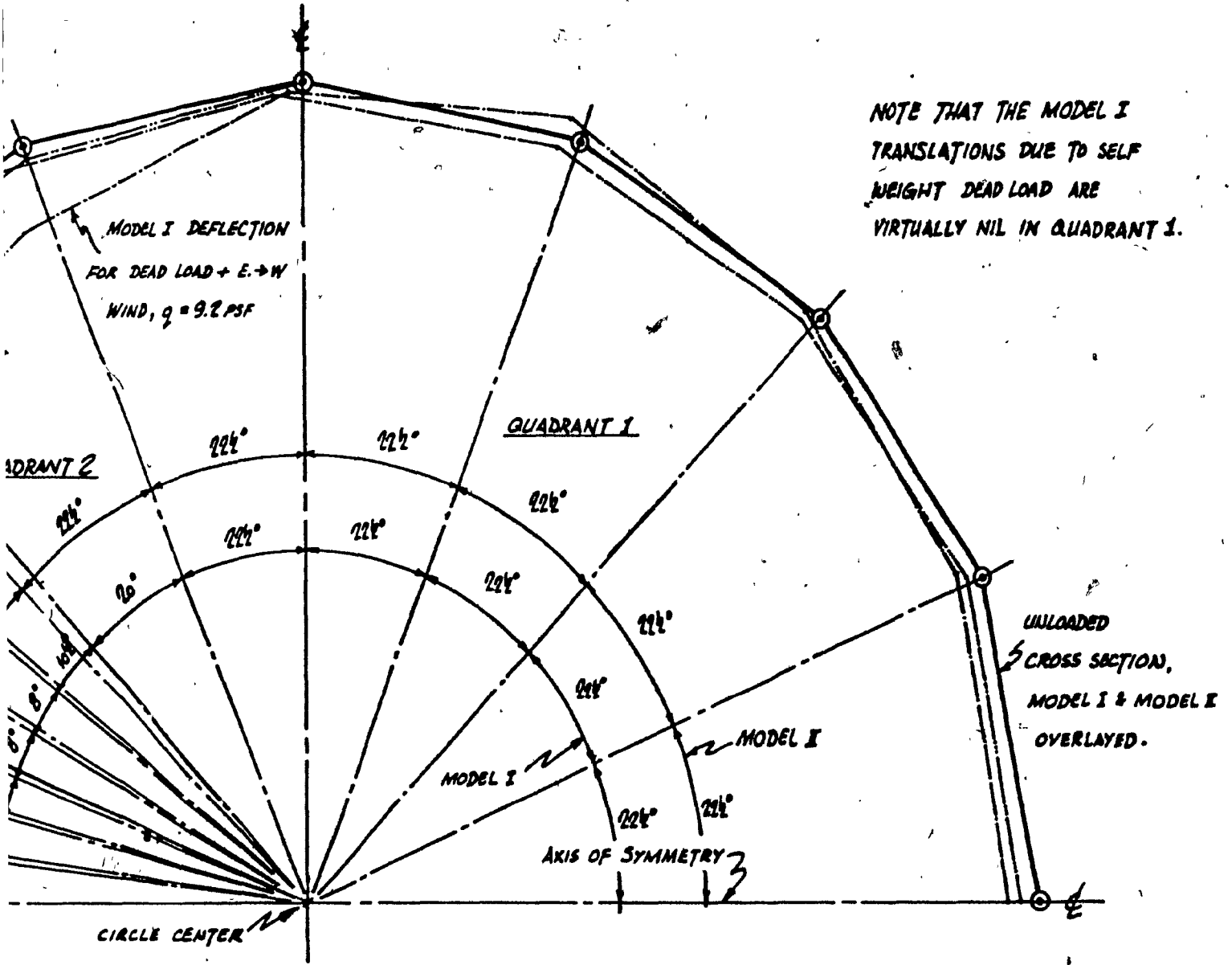


FIG. 2.1 MODEL I AND II CROSS

12



DEFLECTIONS PLOTTED AT 1" = 2" SCALE



NOTE THAT THE MODEL I TRANSLATIONS DUE TO SELF WEIGHT DEAD LOAD ARE VIRTUALLY NIL IN QUADRANT I.

MODEL I AND II CROSS SECTIONAL TRANSLATIONS AT EL. = 50 FT.

1 2 of 2

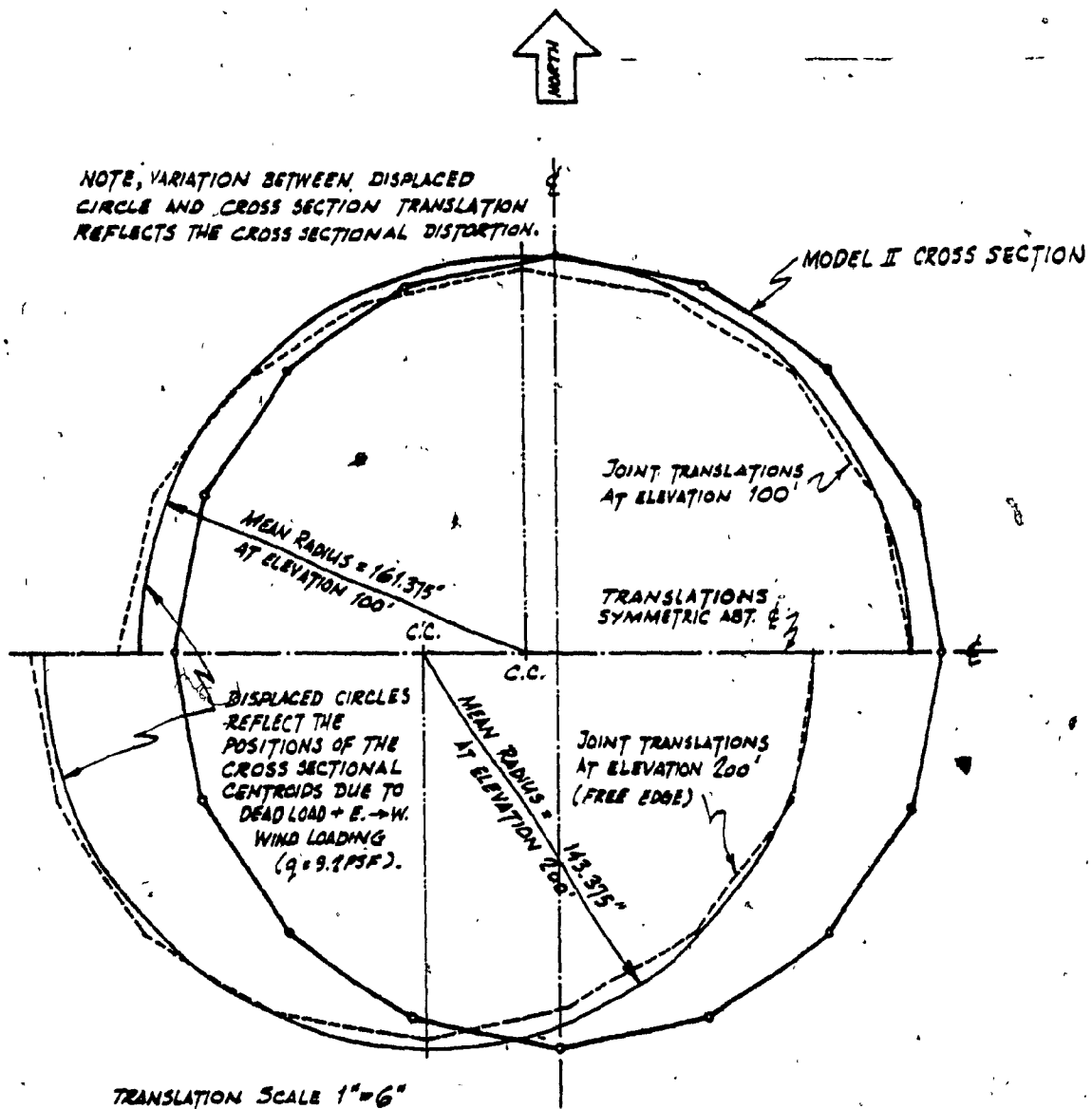


FIG. 2.2 JOINT DISPLACEMENTS OF CROSS SECTION AT 100' AND 200' ELEVATIONS, MODEL II

+	+	+	+	+	+
168	169	170	171	172	172
+	+	+	+	+	+
153	154	155	156	157	158
+	+	+	+	+	+
138	139	140	141	142	143
+	+	+	+	+	+
119	120	121	122	123	124
BREECH REGION ELEMENTS UNDER CONSIDERATION, ALL OUTPUT STRESSES AT ELEMENT CENTROIDS.			+	+	+
			103	104	105
			+	+	+
			94	95	96
			+	+	+
			85	86	87
+	+	+			
72	73	74			
+	+	+			
59	60	61			
+	+	+	+	+	+
47	48	49	50	51	52
+	+	+	+	+	+
32	33	34	35	36	37
+	+	+	+	+	+
13	14	15	16	17	18

FIG. 2.3  
MODEL II BREECH  
REGION ELEMENTS

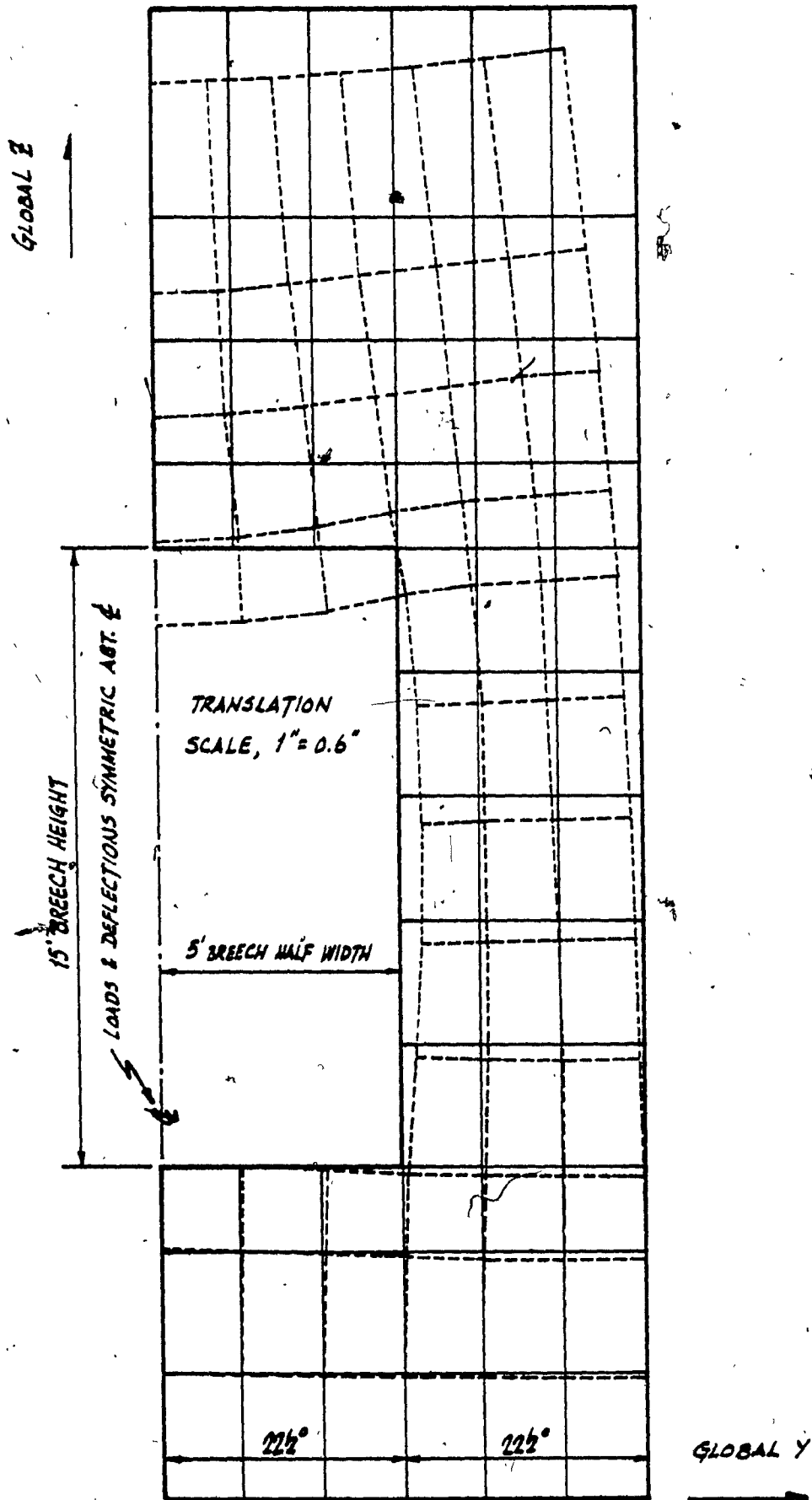


FIG. 2.4 MODEL II VERTICAL (Z) AND TANGENTIAL (Y) TRANSLATIONS IN GLOBAL COORDINATES

EL. 36'  
UNDEFLECTED MESH

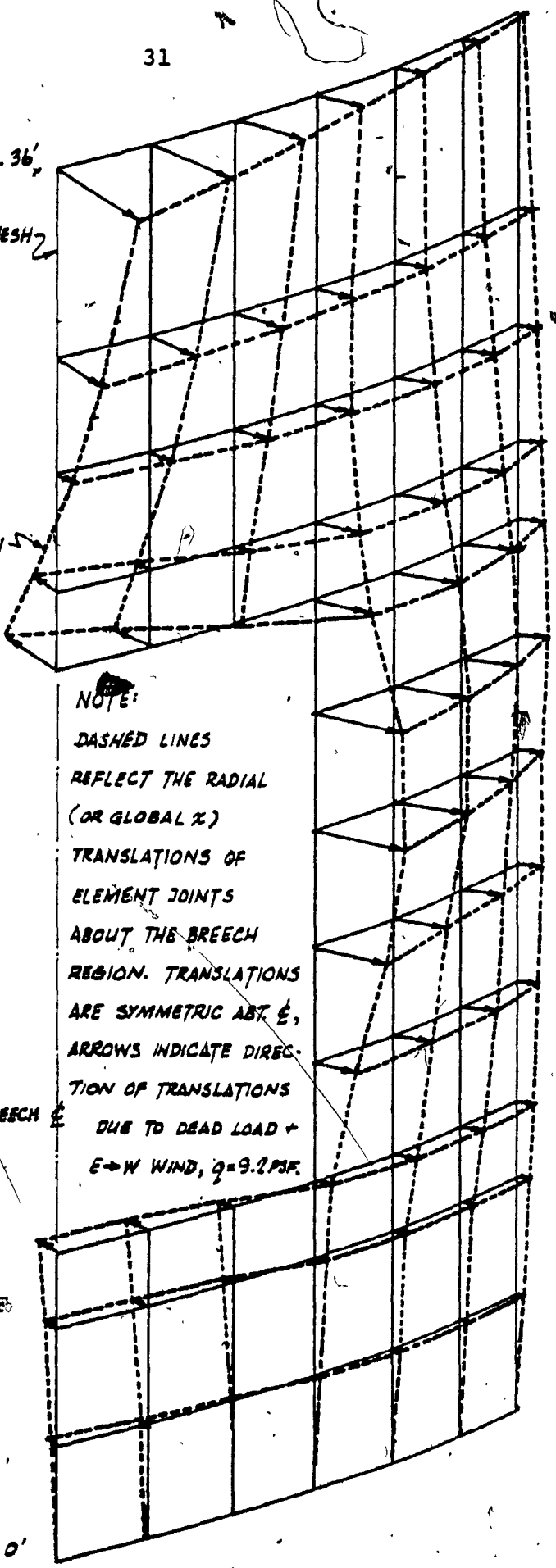
DEFLECTED MESH

NOTE:  
 DASHED LINES  
 REFLECT THE RADIAL  
 (OR GLOBAL X)  
 TRANSLATIONS OF  
 ELEMENT JOINTS  
 ABOUT THE BREECH  
 REGION. TRANSLATIONS  
 ARE SYMMETRIC ABT.  $\hat{E}$ ,  
 ARROWS INDICATE DIREC-  
 TION OF TRANSLATIONS  
 DUE TO DEAD LOAD +  
 E-W WIND,  $Q=9.2$  PSF.

BREECH  $\hat{E}$

FIG. 2.5  
MODEL II ISOMETRIC OF  
GLOBAL X OR RADIAL  
JOINT TRANSLATIONS

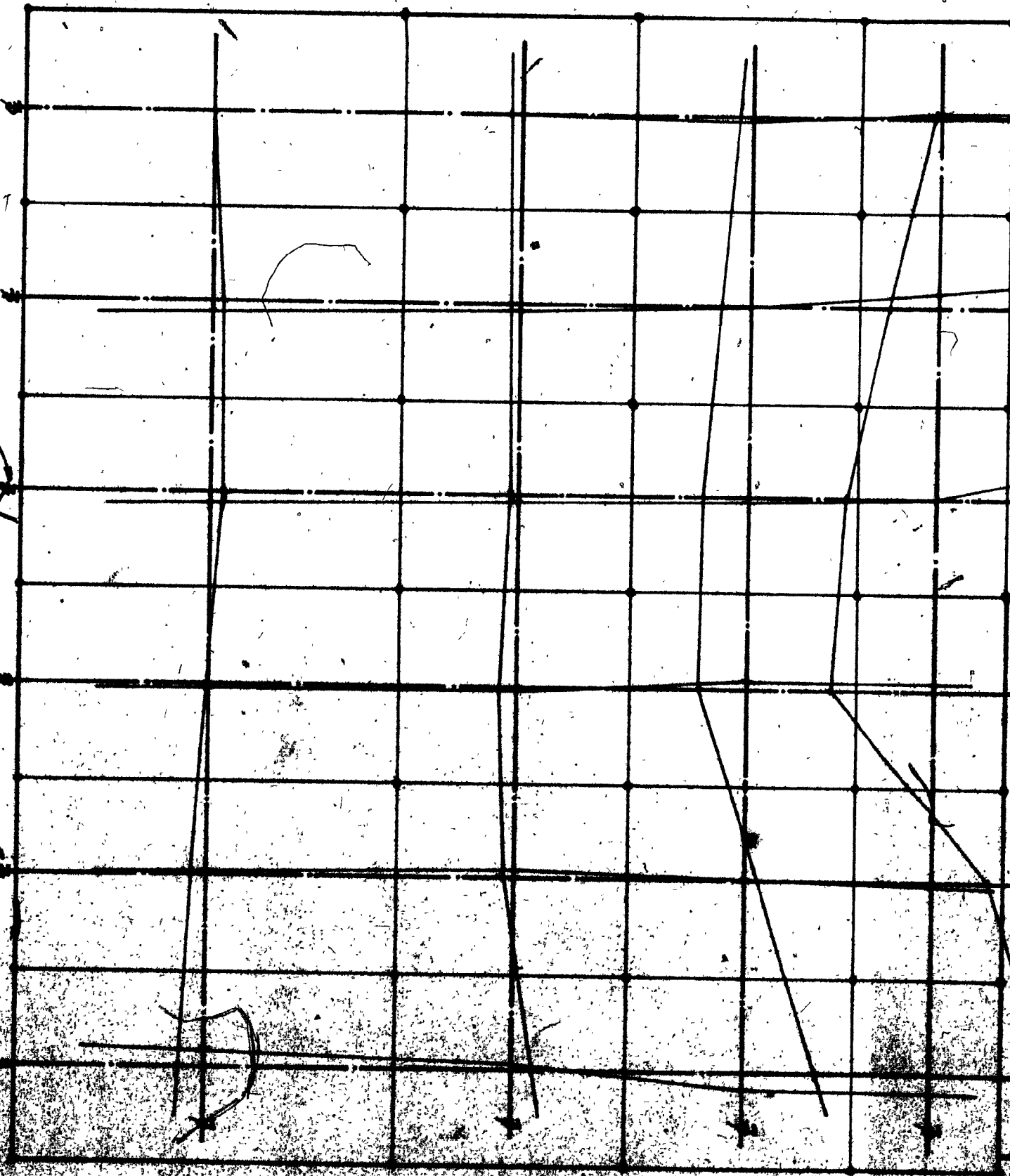
EL. 0'



Figs. 2.6, 2.7 and 2.8 reflect the bending moments in the plate material, where  $M_x$  and  $M_y$  yield the flexural stresses in x and y local element axis directions respectively. The moments are in #-In per inch of plate.

The membrane and bending stresses may be seen in Figs. 2.9 to 2.12. Tension is positive, and for the bending stresses, only the top surface values at each element centroid have been plotted. Stresses are in PSI in the local element x and y directions.

ELEMENT VERTICAL CENTER LINES 3



1.2



BRIDGE HALF WIDTH, 5'

MOMENTS PLOTTED AT SCALE 1" = 2000 LB-IN / IN. OF ELEMENT

BRIDGE SPAN, 15'

MOMENTS AS DEFINED ABOVE BY ANSYS PROGRAM.



$X_1, Y_1, Z_1$  ARE LOCAL ELEMENT AXIS

$X_2, Y_2, Z_2$  TENSION TOP SURFACE

ELEMENT C.G.

SIGN CONVENTION OF ANSYS PROGRAM  
MOMENTS PLOTTED ON ELEMENT TOP SURFACES AT ELEMENT CENTROIDS.

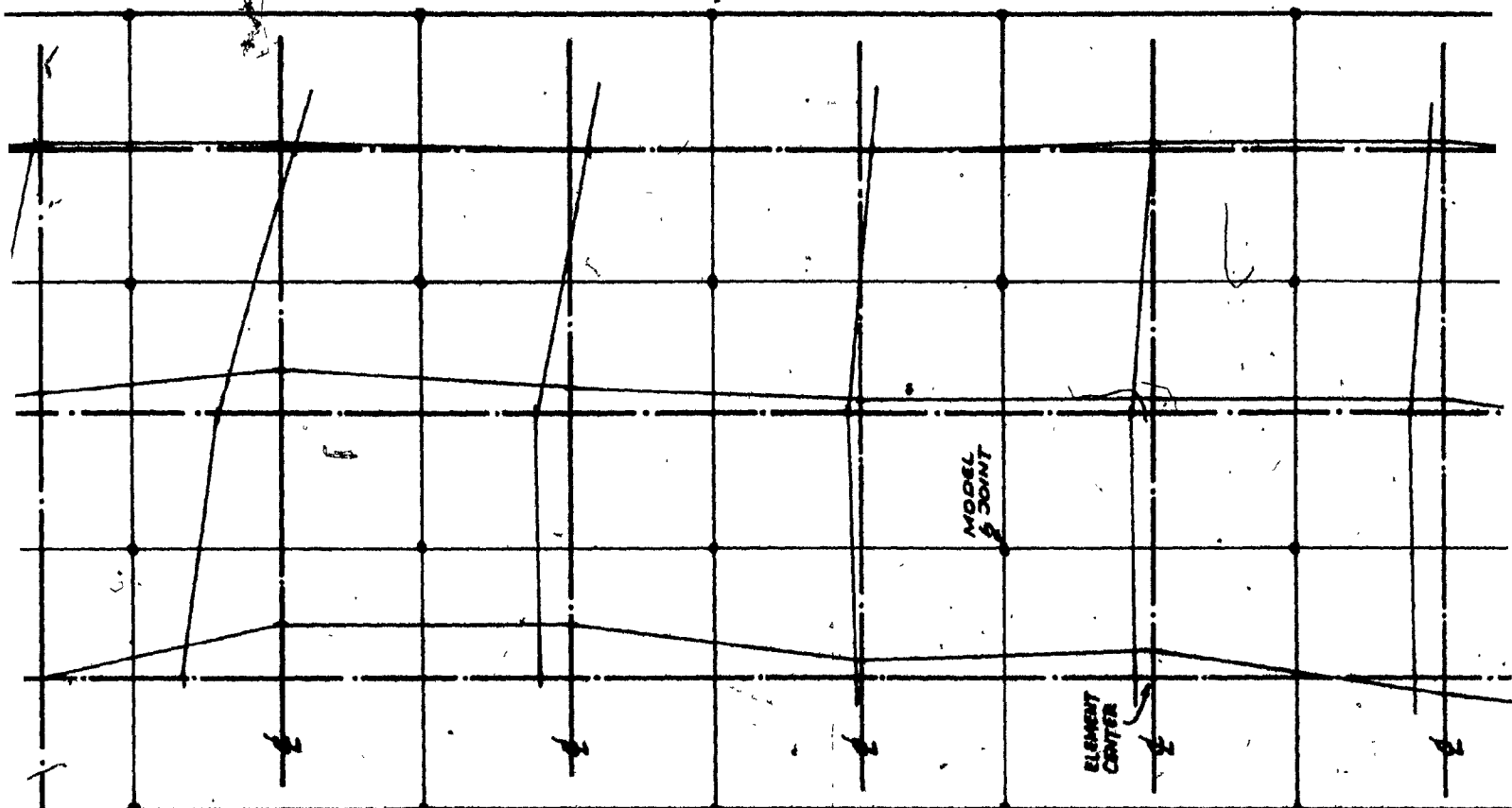
ELEMENT CENTER OF GRAVITY (CENTROID)

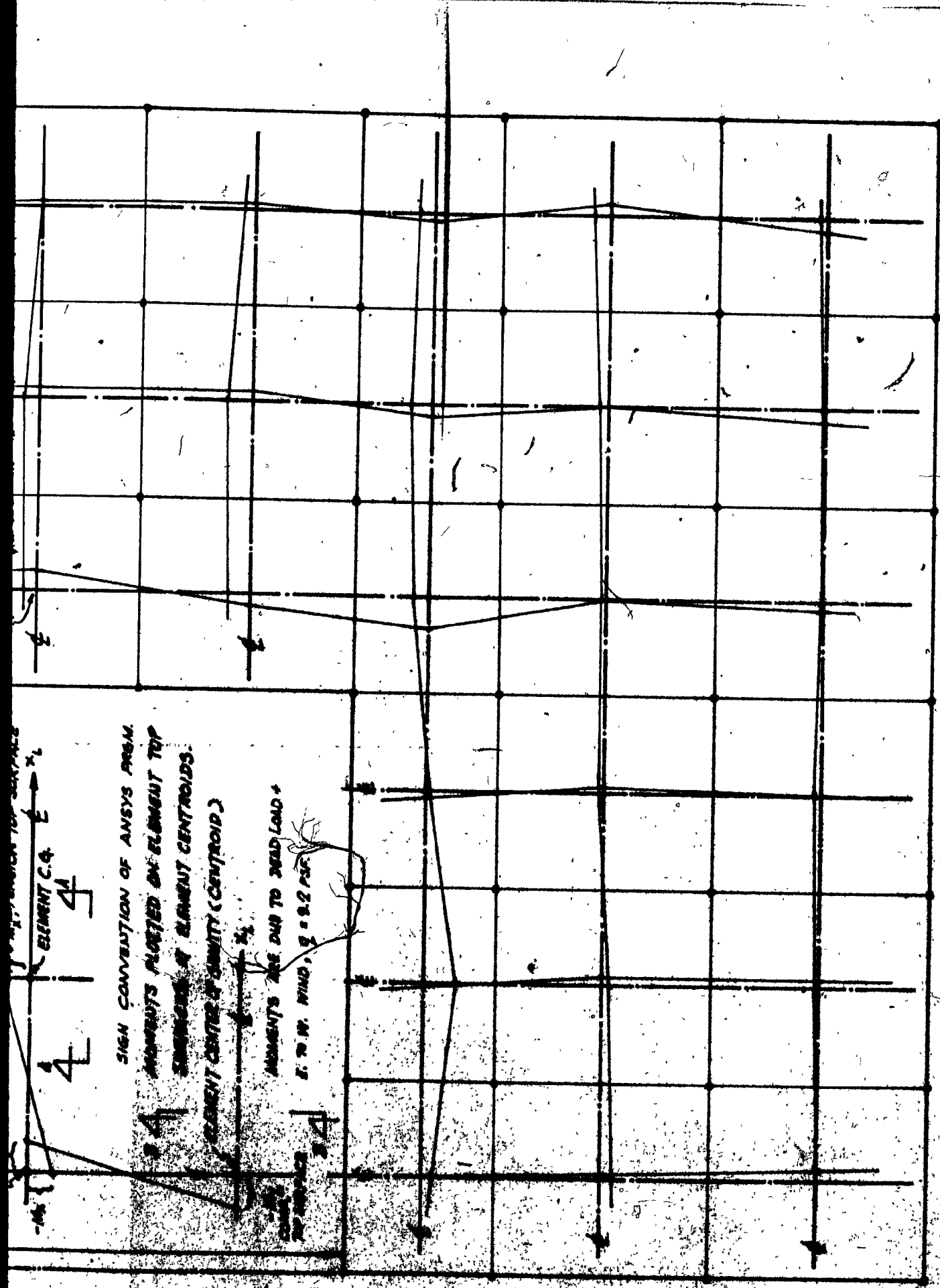
MOMENTS ARE DUE TO DEAD LOAD +

E. TO W. WIND. 0.0007 PER

$-M_x, -M_y$  TOP SURFACE

2.0



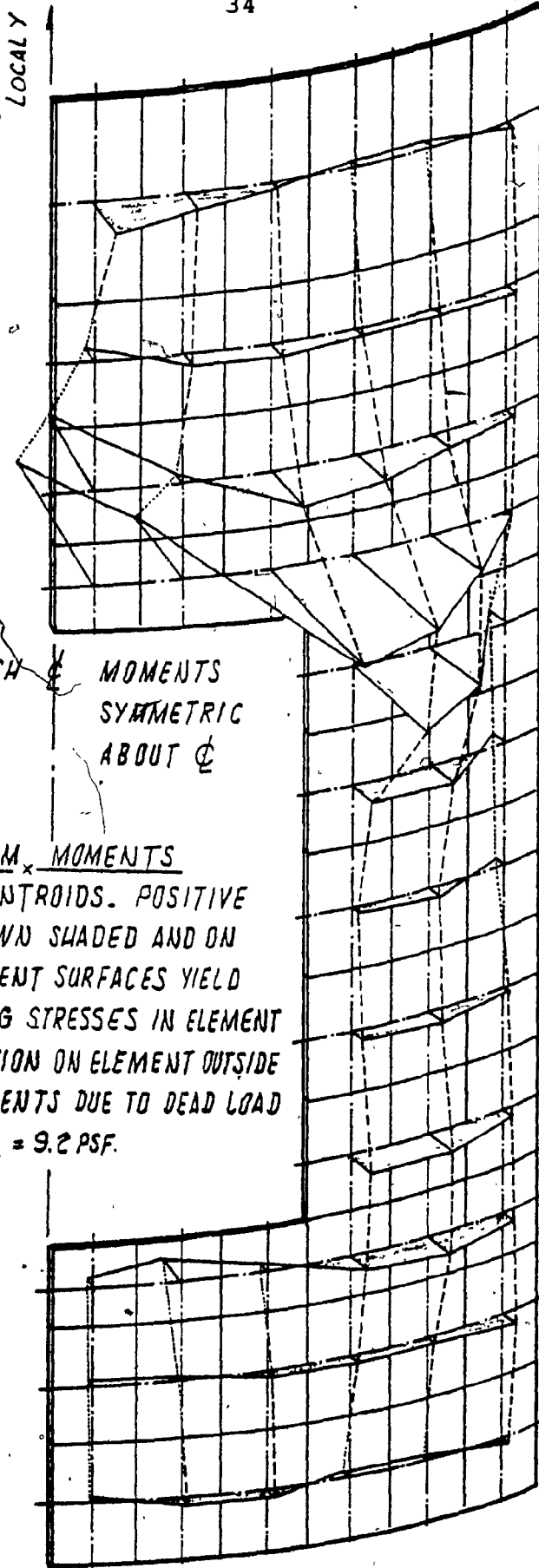


SIGN CONVENTION OF ANSYS PRGM.  
 MOMENTS PLOTTED ON ELEMENT TOP  
 SURFACES AT ELEMENT CENTROIDS.  
 ELEMENT CENTER OF GRAVITY (CENTROID)

MOMENTS ARE DUE TO DEAD LOAD +  
 E. TO W. WIND,  $q = 9.2 \text{ psf.}$

3 of 3

FIG. 2.6 MODEL II MOMENTS AT ELEMENT CENTERS

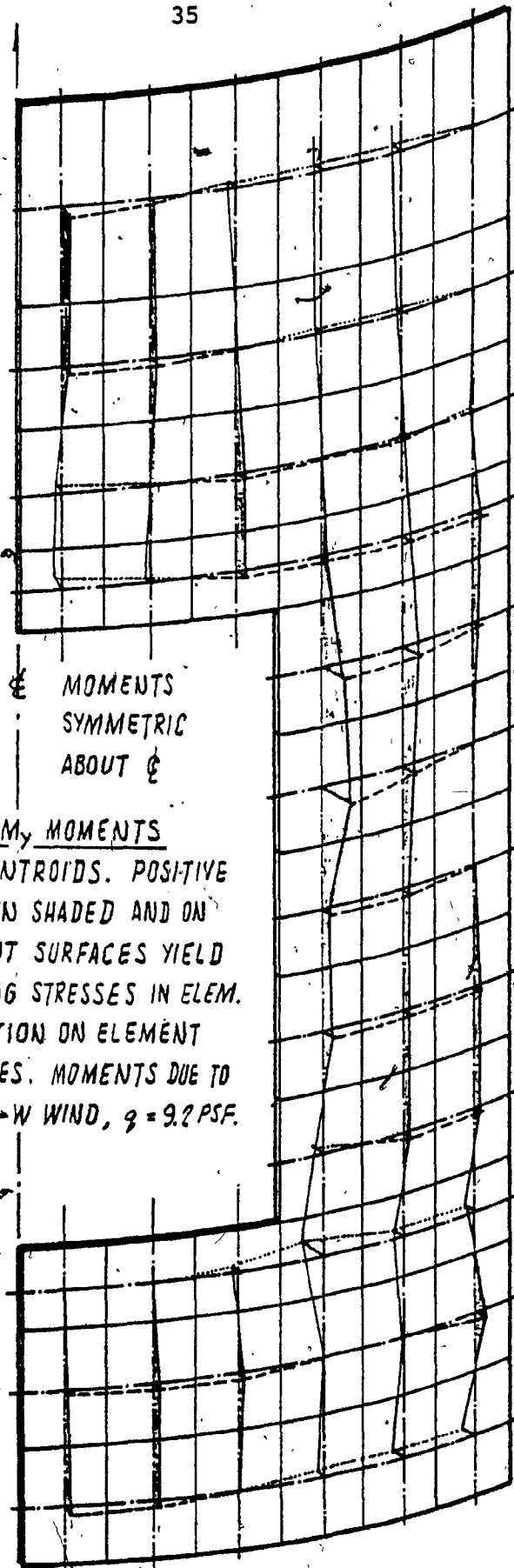


BREACH & MOMENTS  
 SYMMETRIC  
 ABOUT C

FIG. 2.7  
 MODEL II  
ISOMETRIC OF  $M_x$  MOMENTS  
 AT ELEMENT CENTROIDS. POSITIVE  
 MOMENTS SHOWN SHADED AND ON  
 OUTSIDE ELEMENT SURFACES YIELD  
 TENSILE BENDING STRESSES IN ELEMENT  
 LOCAL X DIRECTION ON ELEMENT OUTSIDE  
 SURFACES. MOMENTS DUE TO DEAD LOAD  
 + E → W WIND,  $q = 9.2$  PSF.

LOCAL X

LOCAL Y



BREECH & MOMENTS  
 SYMMETRIC  
 ABOUT &

FIG. 208  
 MODEL II

ISOMETRIC OF  $M_y$  MOMENTS

AT ELEMENT CENTROIDS. POSITIVE  
 MOMENTS SHOWN SHADED AND ON  
 OUTSIDE ELEMENT SURFACES YIELD  
 TENSILE BENDING STRESSES IN ELEM.  
 LOCAL Y DIRECTION ON ELEMENT  
 OUTSIDE SURFACES. MOMENTS DUE TO  
 DEAD LOAD + E → W WIND,  $q = 9.2$  PSF.

LOCAL X

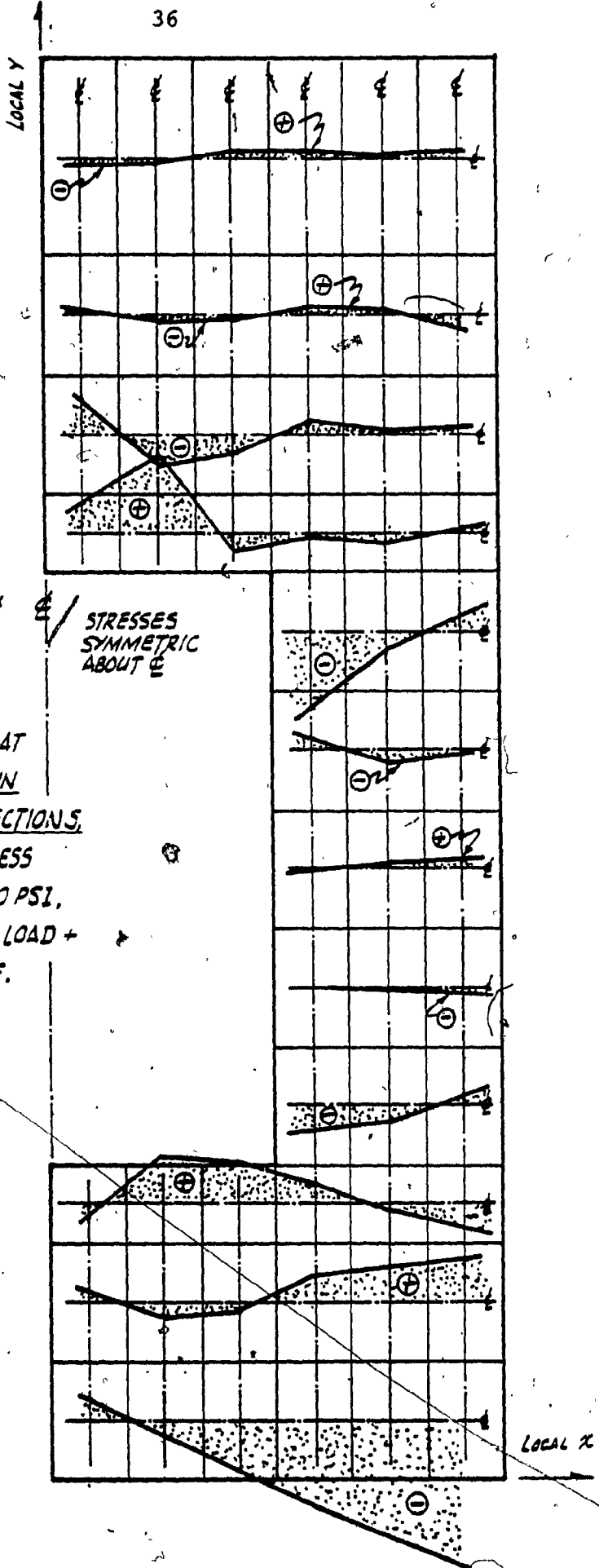


FIG. 2.9  
MODEL II  
MEMBRANE STRESSES AT  
CENTER OF ELEMENTS IN  
LOCAL ELEMENT X DIRECTIONS.  
 ⊕ IS TENSION, AND STRESS  
 PLOT SCALE IS 1" = 2000 PSI,  
 STRESSES DUE TO DEAD LOAD +  
 E → W WIND,  $q = 9.2$  PSF.

BREECH E / STRESSES SYMMETRIC ABOUT E

LOCAL X

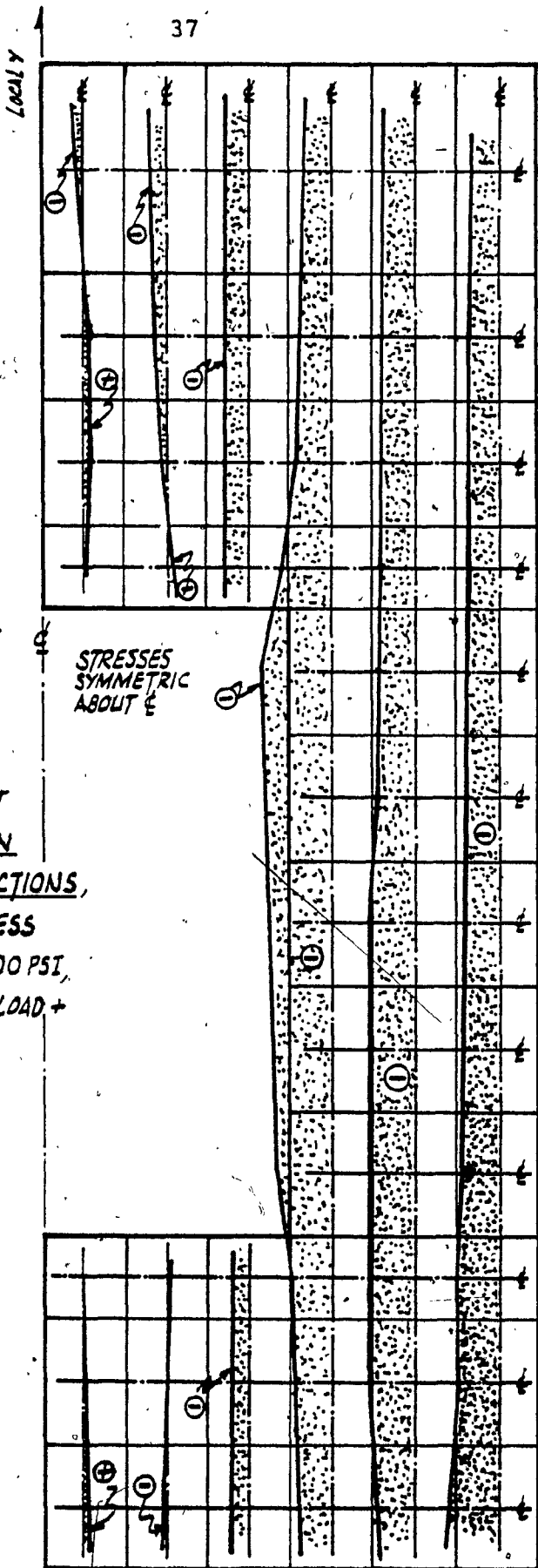


FIG. 2.10  
MODEL II  
MEMBRANE STRESSES AT  
CENTER OF ELEMENTS IN  
LOCAL ELEMENT Y DIRECTIONS,  
 ⊕ IS TENSION, AND STRESS  
 PLOT SCALE IS 1" = 5000 PSI,  
 STRESSES DUE TO DEAD LOAD +  
 E → W WIND,  $q = 9.2$  PSF.

BREECH

STRESSES  
SYMMETRIC  
ABOUT E

LOCAL X

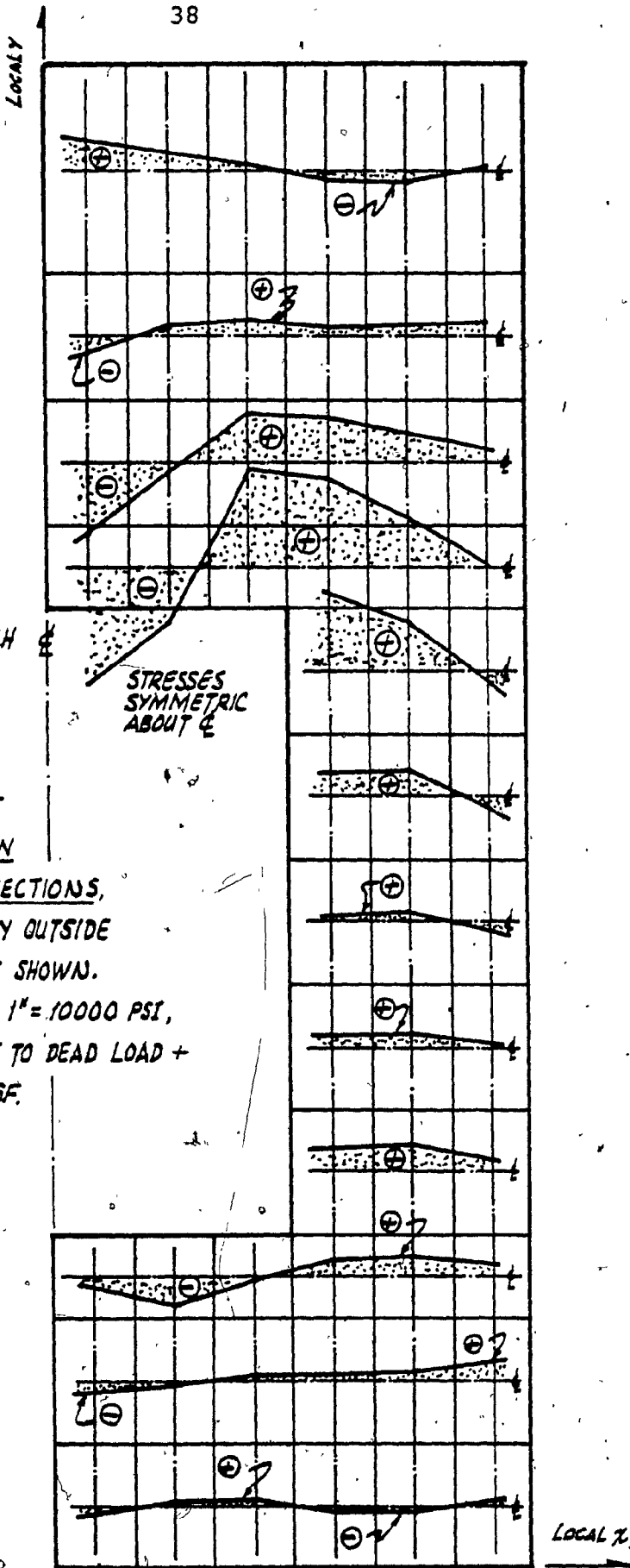


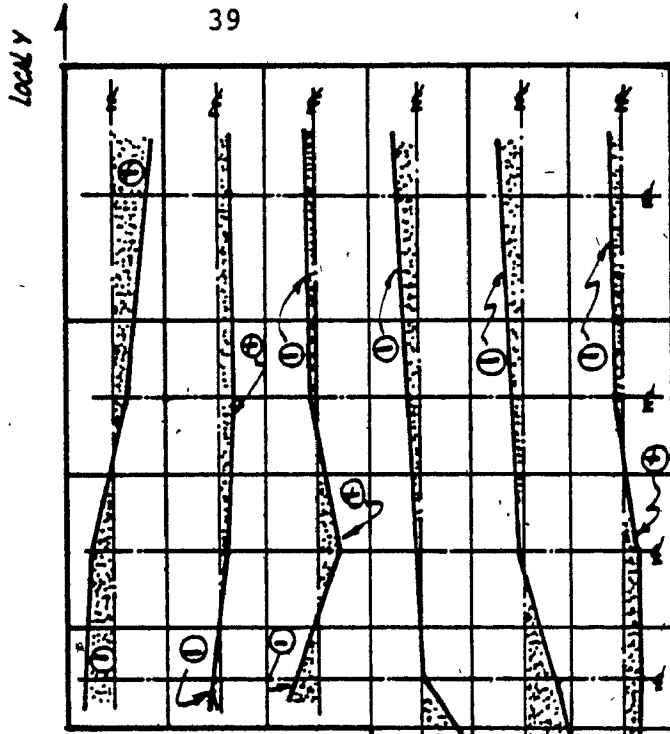
FIG. 2.11

MODEL II

BENDING STRESSES AT  
CENTER OF ELEMENTS IN  
LOCAL ELEMENT X DIRECTIONS,

⊕ IS TENSION, AND ONLY OUTSIDE  
 SURFACE STRESSES ARE SHOWN.

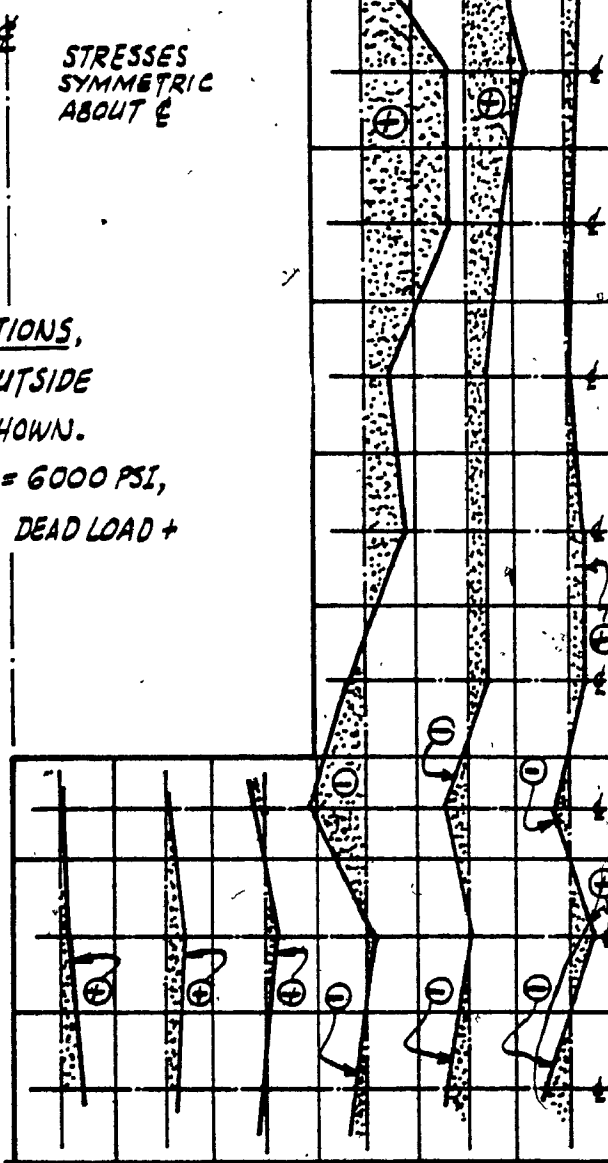
STRESS PLOT SCALE IS  $1'' = 10000 \text{ PSI}$ ,  
 AND STRESSES ARE DUE TO DEAD LOAD +  
 E → W WIND,  $q = 9.2 \text{ PSF}$ .



BREECH

STRESSES  
SYMMETRIC  
ABOUT E

FIG. 2.12  
**MODEL II**  
BENDING STRESSES AT  
CENTER OF ELEMENTS IN  
LOCAL ELEMENT Y DIRECTIONS,  
 ⊕ IS TENSION, AND ONLY OUTSIDE  
 SURFACE STRESSES ARE SHOWN.  
 STRESS PLOT SCALE IS 1" = 6000 PSI,  
 AND STRESSES ARE DUE TO DEAD LOAD +  
 E → W WIND,  $q = 9.2$  PSF.



LOCAL X



SECTION III

CONCLUSIONS AND RECOMMENDATIONS

The objective of this section is threefold. Firstly, it presents conclusions drawn from the analysis reflecting the behavior of the stack about the breech region due to the applied governing load combination of dead load plus East to West Wind; secondly, it presents the modelling recommendations; and thirdly, it presents recommendations for stiffening the breech opening to counter the resulting deflections and stresses about the breech region.

### 3.1: ANALYSIS CONCLUSIONS

Two models were employed to test the applicability of the short Model I and the assumptions with which Model I was developed. Fig. 2.1 summarizes the representative behavior of Models I and II by comparing the cross sectional deformations of both models at the 50 Ft stack elevation due to the applied governing load case. Although the cross sectional deformations are similar and expected in shape, due to the presence of lateral (tangential) shear, the magnitude of Model I displacements are unrealistically excessive. The primary causes of the Model I behavior may be found in its development assumptions. Specifically, Model I lacked the true stiffness of the stack at the 50 Ft elevation, and furthermore, the cross sectional tangential shear was unrealistically concentrated totally at the Model I top edge. To negate these factors Model II elevated the top free edge to the 200 Ft elevation. This in turn increased the stiffness of the stack model by providing material above the 50 Ft

elevation of Model I, stiffening the cross section appropriately. In addition, having greater height, Model II was loaded with more loads of lower magnitude at multiple elevations, reducing the localized concentrations of loads and therefore the associated edge distortions.

Close investigation of the Model II cross sectional displacements at 50, 200 and 100 Ft elevations, Figs. 2.2 and 2.3 respectively, reveals an egg or pear shaping of the cross section. This is anticipated behavior for a circular cross section undergoing lateral shear forces. Furthermore, it should be pointed out that Model II reflects more or less the same cross sectional distortion at all three elevations reviewed, which indicates consistency of cross sectional displacements, especially since the 100 Ft elevation was not a loaded elevation.

Accepting Model II results as satisfactory and reflective of total stack behavior when loaded with the governing load combination of self weight dead load plus East to West Wind, with  $q = 9.2$  PSF, the following conclusions based on the deflections, moments and stresses about the breech may be sighted.

By reviewing Figs. 2.1, 2.2, 2.4 and 2.5, it is obvious that the effects of the breech opening are localized within different areas about the breech. Below the breech, between elevation zero and eight feet, there are small deflections and forces in the longitudinal and tangential stack directions.

This is acceptable since this region is not directly loaded and is only reflecting the influence of the adjacent areas. To the sides of the breach, there are vertical and tangential deflections and forces reflecting localized behavior very similar to a column with end constraints loaded bi-axially or in two directions. It should be noted that this behavior seems to diminish very quickly and yields significantly smaller deflections and forces in the longitudinal direction within an angular arc of 45 degrees to the left and right of the breach center line. Furthermore, Figs. 2.9 to 2.12 reveal that the upper corners of the regions to the breach sides sustain significantly high values of membrane and bending stress, appearing simultaneously in both the longitudinal and tangential stack directions. Although these stresses are not necessarily the highest values present within their categories, their simultaneous existence at significant intensities makes the upper corners along the breach sides a critical location of stress concentration. This is also expected because the opening has compelled the forces to seek an alternative path to ground, and subsequently the upper corner is bottlenecking the loads before distribution further down from this region. Above the breach opening center line, the figures of deflections and stresses indicate very large bending in the cross section and vertical translations but very small values of longitudinal membrane stress. This is due to large tangential compressive membrane plate forces and arch action on axial loads downwardly approaching

the opening. Just above the breech opening, the vertical deflections from the breech center to the corners indicate the behavior of a beam element fixed to the side regions which were already sighted with behavior similar to that of a column. Discounting the cross sectional deflections above and to the sides of the breech, and considering only Fig. 2.4, the overall deflection pattern is indicative of a rigid frame being loaded vertically and horizontally by loads applied symmetrically about the frame center line, as illustrated in Fig. 3.1.1. With this prevailing behavior, the (following) stiffening recommendations in 3.3 were developed.

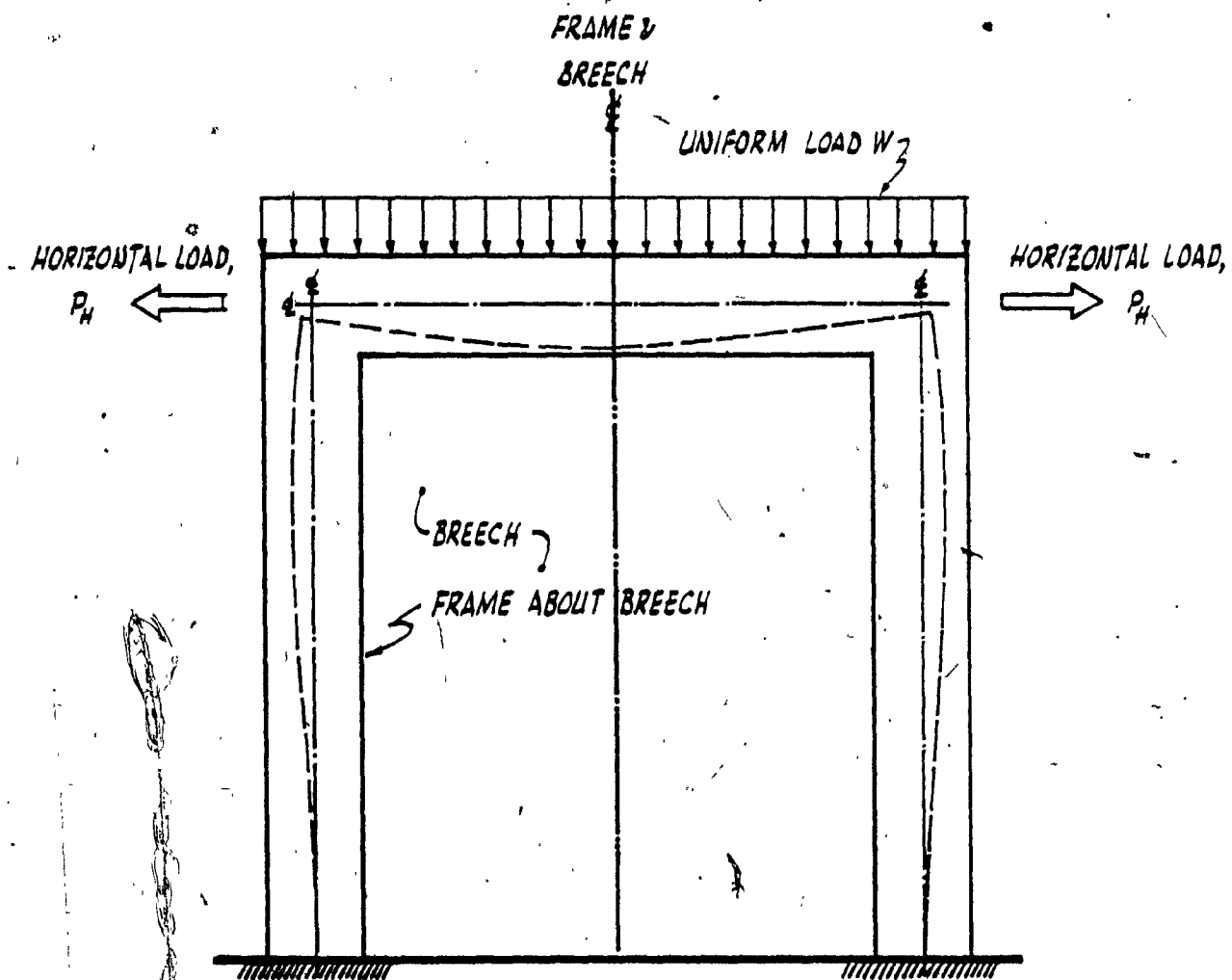


FIG. 3.1.1 ANALOGOUS FRAME ACTION ABOUT BREECH OPENING

### 3.2: MODELLING RECOMMENDATIONS

The program used is not of great importance in the finite element analysis of a simple structure such as a breeched steel stack. Attention should be given to the type of element to be used, the mesh layout, and the overall assumptions governing the model boundary conditions, size, and loading schemes. In this analysis two separate models were employed using the ANSYS program. The models basically differed from each other in size. By elevating the top free edge from 50 to 200 Ft, Model II verified the inadequate representation of the cross sectional stiffness at the 50 Ft elevation in Model I. In addition, it could also be sighted that Model II allowed for a better loading scheme by providing more elevations at which loads could be applied, possibly discounting the local effects at the 50 Ft elevation at which the total load was directly applied on Model I. In view of these two points, partial models of similar steel stacks should be modelled with not less than 40% of the total structure height above the breach. This will provide sufficient height over which the effects of the breach will subside, and more elevations at which a greater number of lower magnitude loads may be applied. If greater insight into the stress distributions is required, especially in the breach corners, a finer mesh in the regions of interest becomes necessary. However, for all intents and purposes, the mesh size implemented in this analysis seemed to be generally adequate. Similar element proportioning should yield acceptable results in other similar models.

The primary modelling goal should be to make the model as representative as possible without excessive expense. It is entirely possible to improve results by a very small percentage at a multiple increase in cost. On the other hand, it is also possible to develop a model that is relatively inexpensive, and still capable of yielding satisfactory results. The cost of the two models utilized in this analysis was, in total, under \$650.00, with approximately an equal division per model due to the fact that Model II had only one applied load case, but was much larger in the number of joints and elements used. Note that had Model II been used initially with all loads, as applied to Model I, the total cost would have been significantly reduced.



### 3.3: STIFFENING RECOMMENDATIONS

The following recommendations are based on the resulting deflected shape of the breech region. Since absolute deflection and stress values are a function of stack, material sizes and applied loads, it is obvious that the absolute magnitude of deflections and stresses found in this analysis are of little importance in comparison to general deflection and stress patterns developed. Primarily it is felt that if deflections can be reduced with stiffeners, bending stresses can be reduced if not removed from the stack plate, permitting the stack plate to be designed for an increased membrane stress with the stiffeners to sustain the bending stresses.

In addition to longitudinal stiffeners, it is felt that cross sectionally stiffening the stack above the opening at a few elevations will provide better redistribution of downward forces by maintaining the circular cross section under load, and also by reducing the bending and membrane stresses in the stack plate in the tangential direction. Note that these tangential stresses were highest just above the breech opening where the longitudinal membrane stresses were minimal. Furthermore, since the sides of the breech opening reflect column action with high longitudinal membrane and bending stresses, it is recommended that the side regions contain longitudinal stiffener reinforcement with increased spacing at further distances from the breech. Thirdly, since the highest tangential membrane stress in compression appears to be at the

base, a base cross section stiffener is also recommended.

However, this will more than likely already exist to facilitate anchoring. Fig. 3.3.1 reflects the stiffener cage recommendations shown in perspective with the stack plate removed.

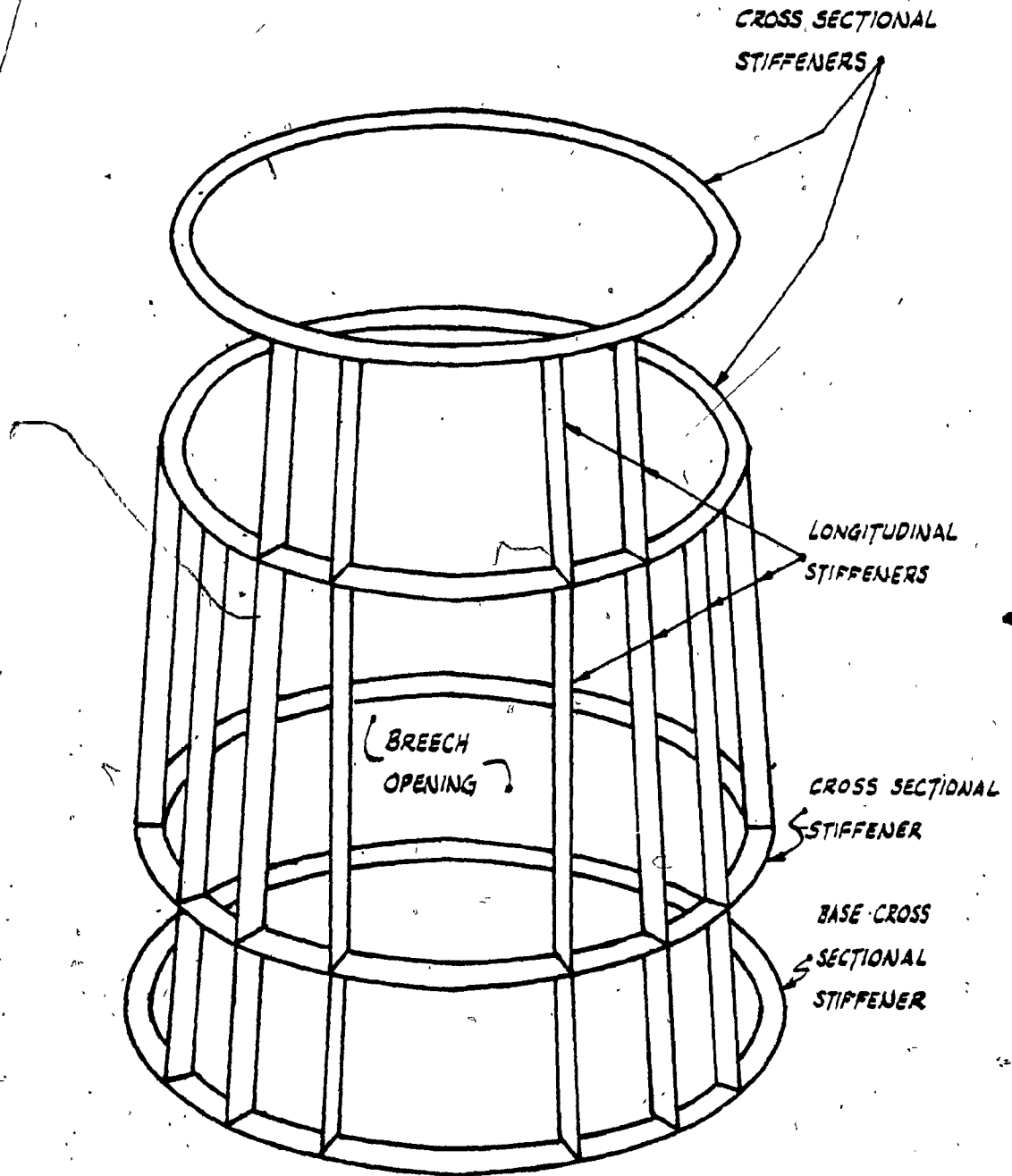


FIG. 3.3.1 STIFFENER RECOMMENDATIONS

## BIBLIOGRAPHY

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5. Roard, R.J., and Young, W.C., Formulas for Stress and Strain, McGraw-Hill, Inc., 1975
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APPENDIX A

A.1: EXACT CALCULATION OF STACK SELF WEIGHT DEAD LOAD,  
MODELS I AND II

Since neither Model I nor II modelled the entire stack in height, and because the model weights were determined by the ANSYS program, only the dead load of the stack above the top of each model was found as follows.

$$DL = Vol \times \rho_s \quad (A.1.1)$$

where Vol = the volume of the stack above the top of model elevation, In<sup>3</sup>

-  $A_m d$

$A_m$  = the average mean area of the stack cross section at the half height of stack above the top of model elevation, In<sup>2</sup>

-  $\pi D_m t$

$D_m$  = the average mean diameter at the same elevation of  $A_m$ , In

$$= 12 \left[ D_B - \left\{ \frac{D_B - D_T}{H} \times h \right\} \right] - t \quad (A.1.2)$$

where  $D_B$  = stack base diameter, 30 Ft

$D_T$  = stack top diameter, 15 Ft

$H$  = stack height, 500 Ft

$h$  = the elevation of  $A_m$  and  $D_m$ , Ft

$d$  = the height of stack from the top of model to the top of stack, Ft,  $d = 24H - 24h$ , In

$t$  = the stack plate thickness, In

$\rho_s$  = the plate density, 0.284 PCI

See Fig. A.1.1

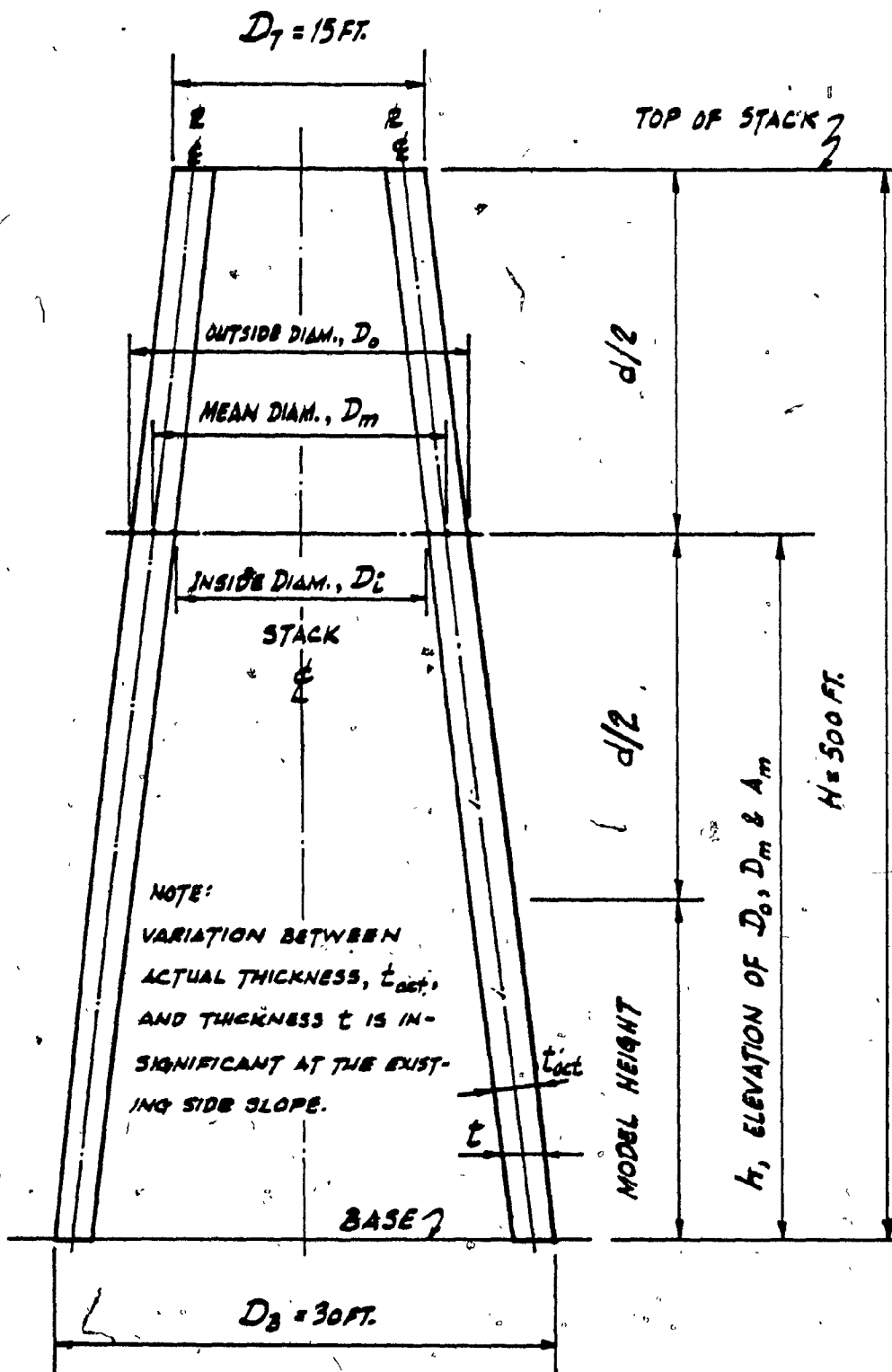


FIG. A.1.1 STACK GEOMETRY USED TO DETERMINE SELF WEIGHT WITH RESPECT TO PLATE THICKNESS AND MODEL HEIGHT

Therefore,

$$DL = \pi D_m t d \rho_s = 0.8922 d D_m t = 0.8922 (24) (H-h) D_m t$$

$$= 21.41 D_m t (H-h) \quad (A.1.3)$$

Equations A.1.2 and A.1.3 yield the following Tables for dead load determination.

TABLE A.1.1  
HEIGHTS AND OUTSIDE DIAMETERS

Item	Stack	Model I	Model II
Top Elevation (Ft)	500	50	200
d (Ft)	0	450	300
h (Ft)	250	275	350
Do (In)	270	261	234
H-h (Ft)	250	225	150
21.41 (H-h) (In)	5352.5	4817.25	3211.5

TABLE A.1.2  
MEAN DIAMETERS AND SELF WEIGHTS FOR  
PLATE THICKNESS OF 1, 1½ AND 1¾ INCHES

Plate t In	Stack		Model I		Model II	
	Dm In	DL #	Dm In	DL #	Dm In	DL #
1	269	1439823	260	1252485	233	748280
1½	268½	1798105	259½	1564101	232½	934346
1¾	268¾	2155719	259¾	1875115	232¾	1120011

Since preliminary dead load calculations indicated a required plate thickness of 1½" with dead loads of 1,572,000# and 934,480# for Model I and Model II, respectively; and because these values varied from the respective Table A.1.2 values by only 0.5% and 0.01%, no effort was made to change the joint loads determined by the preliminary dead load calculations for the Model I and Model II computer input data.



To determine a model plate thickness, the dead load stress, maximum at the base, was found as follows.

$$\sigma_{DL} = DL/A_{mb} \quad (A.1.4)$$

where  $\sigma_{DL}$  = the dead load stress, PSI

DL = the dead load stress, #

$A_{mb}$  = the mean area of the cross section at the base of the stack

$$= \pi D_{mb} t$$

$D_{mb} = 1.2D_B - t$  = (stack mean base diameter, In

Therefore,

$$\sigma_{DL} = DL/\pi D_{mb} t$$

Table A.1.3 verifies the observation of load and area being very linearly proportioned to plate thickness, which can be made by inspecting equations A.1.1 to A.1.3.

TABLE A.1.3  
BASE DEAD LOAD STRESSES FOR  $t=1, 1\frac{1}{2}$  AND  $1\frac{3}{4}$  INCHES

Plate t	1"	$1\frac{1}{2}$ "	$1\frac{3}{4}$ "
DL, #	1440000	1800000	2160000
$D_{mb}$ , In	359	358 $\frac{1}{2}$	358 $\frac{1}{2}$
$\pi D_{mb} t$ (In <sup>2</sup> )	1128	1409	1689
$\sigma_{DL}$ (PSI)	1277	1278	1279

Similarly, the dead load stresses at the top of the models were determined as follows.

TABLE A.1.4  
MODEL I AND MODEL II DEAD LOAD STRESSES

Model	Plate t	1	1½	1¾
I	DL, #	1252500	1564100	1875100
	Dm, In	260	259½	259½
	πDmt, In <sup>2</sup>	817	1020	1223
	σ <sub>DL</sub> , PSI	1533	1533	1533
II	DL, #	748300	934300	1120000
	Dm, In	233	232½	232½
	πDmt, In <sup>2</sup>	732	914	1096
	σ <sub>DL</sub> , PSI	1022	1022	1022

However, since Model I was modelled as a straight sided cylinder, the model mean diameter became the average of the top of model and stack base mean diameters.

i.e. 
$$D_{mI} = (D_{mb} + D_{mT})/2$$

For the 1½ inch plate, the  $D_{mI}$  for Model I was

$$(358½ + 340.75)/2 = 349.75"$$

The area,  $A_m$ , was

$$349.75(\pi)(1.25) = 1373 \text{ In}$$

Therefore, the Model I top edge stress, which was used to determine the Model I vertical top edge load joint forces, was

$$1564000/1373 = 1139 \text{ PSI}$$

#### A.2: STACK LATERAL WIND LOAD

Preliminary calculations indicated that wind governed. The lateral wind to which the stack was subjected was determined in accordance to the National Building Code of Canada as follows.

General Equation  $p = qC_e C_g C_p C_n$  (A.2.1)

where  $C_e$  (exposure factor) varied as follows

TABLE A.2.1

Height (Ft)	40	60	90	130	190	270	420	740
$C_e$	1.0	1.1	1.2	1.3	1.4	1.5	1.6	1.8

where  $C_g$  (gust factor) was taken as 2.0,

$C_p$  (external pressure coefficient) was taken as

$$1.0 - (-0.6) = 1.6,$$

$C_n$  (shape factor) was taken as 0.7.

For  $C_n$   $D_{AVG} \approx (30 + 15)/2 = 22.5$

$$H/D_{AVG} = 500/22.5 = 22.2$$

yielding  $C_n = 0.7$

with  $\text{Area} = 500 \times 22.2 \approx 11,110 \text{ Ft}^2$

Therefore, the general equation varied with respect to height as follows.

TABLE A.2.2

Height (Ft)	40	60	90	130	190	270	420	720
$p$	2.24q	2.46q	2.69q	2.91q	3.14q	3.36q	3.58q	4.03q

From these values the overturning wind moments and lateral shears were determined for Model I and Model II.

### A.3: MODEL I WIND LOADS

The wind moments and shears for Model I were determined tabularly in accordance to Fig. 1.2.1 and may be seen in Table A.3.1.

TABLE A.3.1  
LATERAL WIND SHEARS AND OVERTURNING MOMENTS FOR MODEL I BASE AND TOP

Stack Elevation Ft	Elevation Difference Ft	Average		Lateral		Moment Arm To		Moment About	
		Diameter Ft	Area Ft <sup>2</sup>	Wind qPSF	Shear q#	Base Ft	EL-50' Ft	Base q#-FT	EL-50' q#-Ft
0	40	29.4	1176	2.24	2634	20	0	52680	0
40	20	28.5	570	2.46	<del>701</del> 701	50	5	70100	3500
60	30	27.8	833	2.69	2239	75	25	167925	55975
90	40	26.7	1068	2.91	3108	110	60	341880	186480
130	60	25.2	1512	3.14	4748	160	110	759680	522280
190	80	23.1	1848	3.36	6209	230	180	1428070	1117620
270	150	19.7	2948	3.58	10552	345	295	3640440	3112840
420	80	16.2	1296	4.03	5223	460	410	2402580	2141430
500									
Total at Base					36115			8863355	
Total at EL-50'					32780				7140125

Note: The values shown are for  $q = 1.0$  PSF. For actual values, they must be multiplied by chosen value of  $q$ . For Model I, the analysis stresses due to the unit wind were multiplied by  $q = 9.2$ , the 1/100 wind for Montreal.

## A.4: MINIMUM REQUIRED STACK PLATE THICKNESS

The minimum required plate thickness in the stack was found as follows.

$$\sigma_{\max} = \sigma_a + \sigma_b = \frac{0.047tE}{D[1+0.004E/Y]} \leq 15000 \text{ PSI} \quad (\text{A.4.1})^1$$

where  $\sigma_a$  = the maximum axial stress (DL), PSI

$\sigma_b$  = the maximum flexural stress (Wind), PSI

t = the minimum plate thickness, In

E = Young's modulus, 29,000,000 PSI

D = mean stack diameter, at the base ~ 29.9'

T = material yield stress, 40,000 PSI

Therefore,

$$\sigma_{\max} = \frac{0.047 \times 29000000}{29.9 \times 12 \left[ 1 + 0.004 \times \frac{29000000}{40000} \right]} \times t = 11689t$$

Having previously determined the dead load self weight stresses for Model I, the flexural wind stresses were found at the stack base, where they were maximum as follows.

$$\sigma_b = M/S$$

where  $\sigma_b$  = the bending stress, PSI

M = the bending moment at base

= 12 x 9.2 x 8863355 #-In

S = the section modulus of the stack base

<sup>1</sup>Humphreys & Glasgow Ltd., Design of Self-Supporting Steel Stacks, No. M-19-1300-010/1, Issue 1, Sept. 1963, p. 2.

TABLE A.4.1  
SUMMARY OF MAXIMUM AXIAL AND FLEXURAL  
STRESSES IN STACK PLATE AT BASE ELEVATION

Plate t (In)	D <sub>out</sub> In	D <sub>out</sub> <sup>4</sup> 10 <sup>10</sup> In <sup>4</sup>	D <sub>in</sub> In	D <sub>in</sub> <sup>4</sup> 10 <sup>10</sup> In <sup>4</sup>	S In <sup>3</sup>	M 10 <sup>8</sup> #-In	σ <sub>b</sub> PSI	σ <sub>a</sub> PSI	σ <sub>a</sub> +σ <sub>b</sub> PSI	Allow PSI
1	360	1.68	358	1.64	100943	9.785	9694	1278	10972	11689
1½	"	"	357.5	1.63	125915	"	7771	"	9049	14611
1½	"	"	357	1.63	150784	"	6490	"	7768	15000

Although all plate thicknesses considered were below respective allowable stresses, the 1½ inch plate was chosen as the model material thickness for both Models I and II to realistically accommodate such things as corrosion. Note that the actual plate thickness was not of great importance since this was not a design problem, but an analysis, from which relative displacement shapes and stress distributions were being sought. In reality any reasonable thickness would have sufficed, however, since a fix on stresses to be expected in regions far removed from the breach discontinuity was required, the exercise of determining an acceptable plate thickness, and the respective stress, was carried out. By comparing the analysis stresses in regions far removed from the discontinuity to those found for a plate thickness of 1½ inch, a measure of the models' performance was determined.

#### A.5: MODEL II LATERAL WIND

Wind forces for Model II, determined with respect to the wind distribution shown in Fig. A.5.1 with  $q = 9.2$ PSF, were as shown in Table A.5.1.

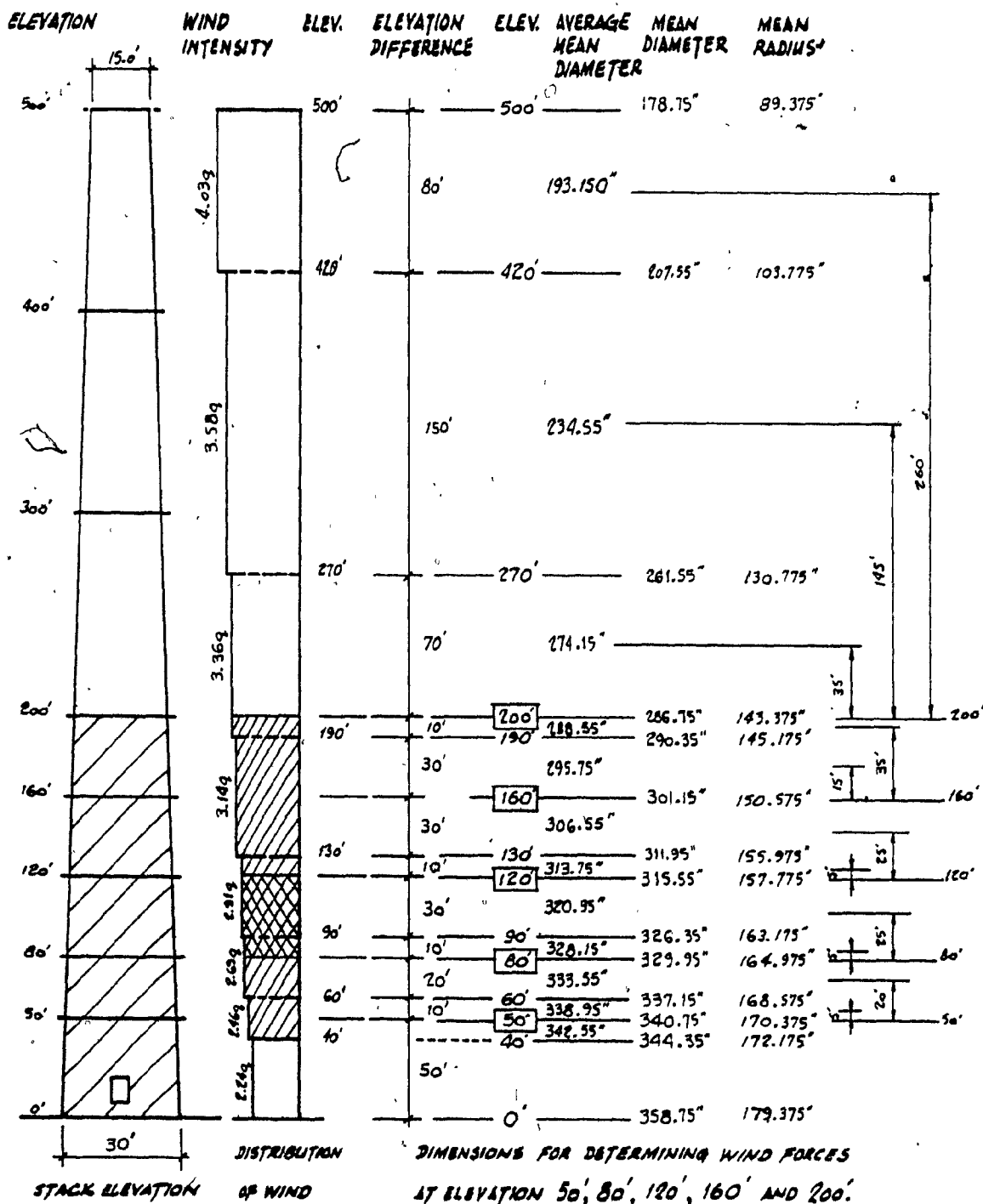


FIG. A.5.1 MODEL II DIMENSIONS REQUIRED TO DETERMINE WIND FORCES AT MODEL ELEVATIONS SHOWN BOXED IN

TABLE A.5.1  
WIND FORCES ON MODEL II FOR  $q=9.2$ PSF AT ELEVATIONS 50', 80', 120', 160' AND 200'

Stack Elevation Ft	Wind Intensity qPSF	Elevation Difference Ft	Average Diameter Ft	Shear q#	Moment Arm Ft	Moment q#-Ft	Shears and Moments for $q=9.2$ PSF
200	4.03	80	16.096	5189	260	1349231	
200	3.58	150	19.546	40496	145	1521936	
200	3.36	70	22.846	5373	35	188067	EL = 200'
$\Sigma$				21059		3059220	V=21059q=193743# M=3059220q=28144824#-Ft
160	3.36	10	24.046	808	35	28278	
160	3.14	30	24.646	2322	15	34825	EL = 160'
$\Sigma$				3130		63102	V=3130q=28796# M=63102q=580538#-Ft
120	3.14	30	25.546	2406	25	60160	
120	2.91	10	26.146	761	5	3804	EL = 120'
$\Sigma$				3167		63965	V=3167q=29136# M=63965q=588478#-Ft
80	2.91	30	26.746	2335	25	58373	
80	2.69	10	27.346	736	5	3678	EL = 80'
$\Sigma$				3071		62051	V=3071q=28253# M=62051q=570869#-Ft
50	2.69	20	27.796	1495	20	29908	
50	2.46	10	28.246	695	5	3474	EL = 50'
$\Sigma$				2190		33383	V=2190q=20148# M=33383q=307124#-Ft



## A.6.1: LOAD SUMMARY FOR MODELS I AND II

A summary of the loads applied to each model at respective elevations may be seen tabularized in Table A.6.1.1. Joint loads at specific elevations for each model were resolved from these loads.

TABLE A.6.1.1  
MODELS I AND II LOADS AT RESPECTIVE MODEL ELEVATIONS

Model	q	E Ft	L O A D				Approximate DL applied by ANSYS #
			Type	Self Weight DL	C A S E		
			P, # V, # M, #-Ft		Wind		
					E+W	N+S	
I	1.0 PSF	50	P	1564000	-	-	235000*
			V	-	32780	32780	
			M	-	7140000	7140000	
II	9.2 PSF	50	P	-	-	-	
			V	-	20148	-	
			M	-	307124	-	
"	"	80	P	-	-	-	
			V	-	28253	-	
			M	-	570869	-	
"	"	120	P	-	-	-	
			V	-	29136	-	
			M	-	588478	-	
"	"	160	P	-	-	-	
			V	-	28796	-	
			M	-	580538	-	
"	"	200	P	934480	-	-	864000*
			V	-	193743	-	
			M	-	28144824	-	

\* Estimated by  $Vol \times \rho_s$ , since the models are faceted and not truly cylindrical there will be slight variation between these values and what ANSYS determined.

E - Elevation, P - Axial, V - Shear, M - Moment

Note: The load cases of DL and E+W Wind (q=9.2PSF) were combined into a single load combination for Model II.

Table A.6.1.2 summarizes the stack dimensions which were required to determine the joint forces using the elevations and loads prescribed in Table A.6.1.1.

TABLE A.6.1.2

RADIAL DIMENSIONS AND SECTION MODULII FOR MODELS I AND II AT THE ELEVATIONS WHERE JOINT LOADS WERE APPLIED

Model	h Ft	t In	D <sub>B</sub> Ft	D <sub>T</sub> Ft	H Ft	Ro In	Ri In	Rm In	S In <sup>3</sup>
I	25*	1½	30	15	500	175.5	174.25	174.875	119666
II	50	"	"	"	"	171.0	169.75	170.375	113576
	80	"	"	"	"	165.6	164.35	164.975	106478
	120	"	"	"	"	158.4	157.15	157.775	97370
	160	"	"	"	"	151.2	149.95	150.575	88669
	200	"	"	"	"	144.0	142.75	143.375	80376

\* Because Model I was cylindrical, the diameter at the top edge where joint forces were applied was the average of stack diameters at the base and elevation 50 Ft.. The average of these two diameters coincides, of course, with a stack height of 25 Ft., which was utilized in Equation A.1.2

#### A.6.2: MODEL I JOINT LOADS

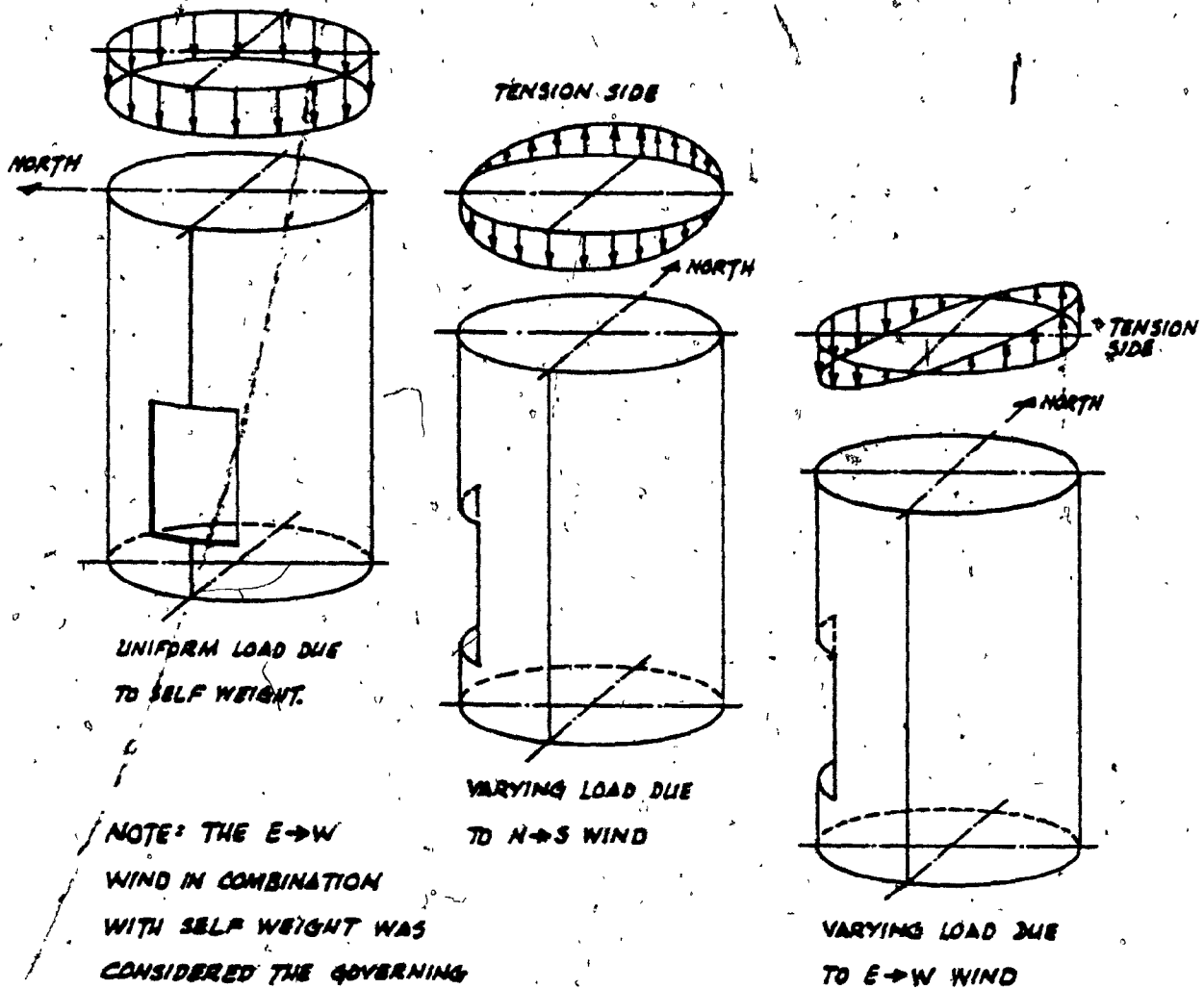
The Model I top edge vertical uniform and varying line loads from which top edge joint loads were determined are shown in Fig. A.6.2.1.

The lateral shear load due to wind was respectively applied tangentially to the top edge with a sinusoidal distribution to correspond to the appropriate wind direction.

Fig. A.6.2.2 reflects the lateral shear applied along the top edge.

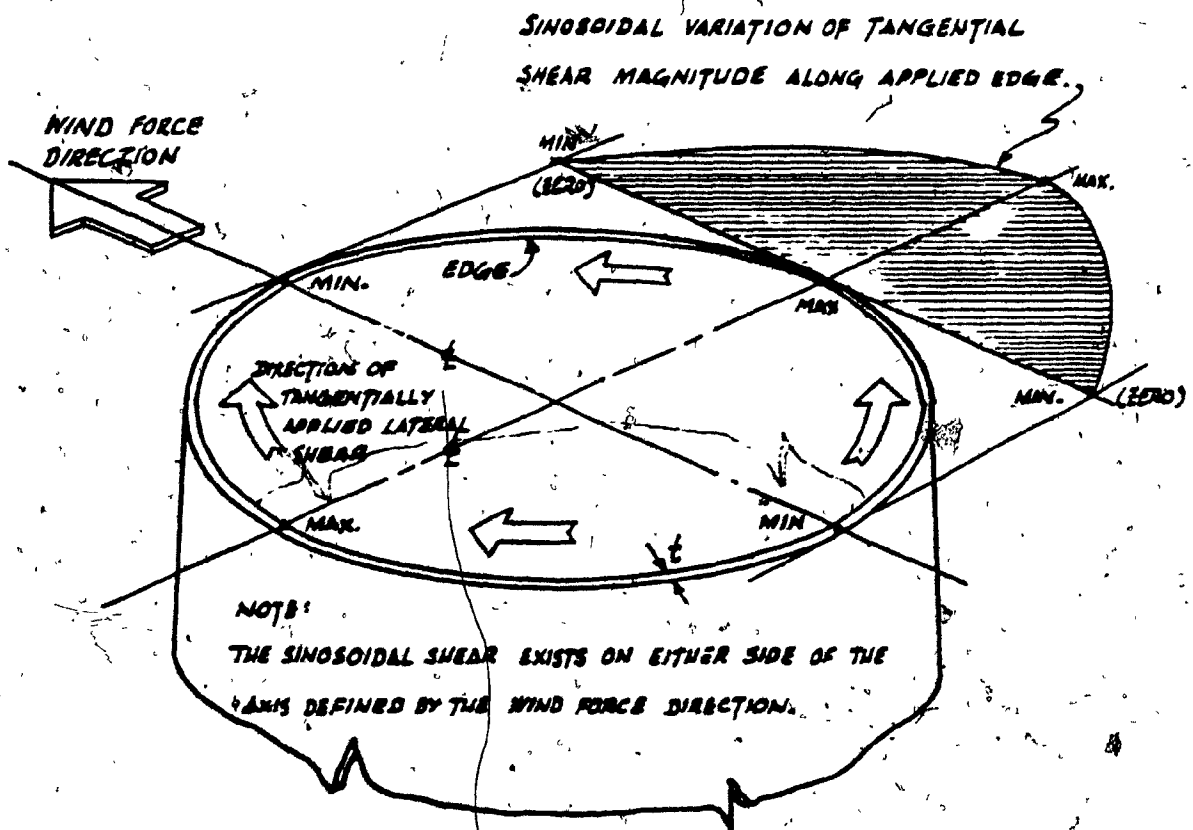
#### A.6.2.1: CALCULATION OF LOAD CASE 1 (DL) JOINT FORCES

The vertical joint loads applied to top edge joints were determined by the following equation.



**NOTE: THE E-W WIND IN COMBINATION WITH SELF WEIGHT WAS CONSIDERED THE GOVERNING LOAD COMBINATION FOR THE BREACH REGION.**

**FIG. A.6.2.1 MODEL I TOP EDGE VERTICAL LINE LOADS DUE TO LATERAL WIND AND SELF WEIGHT DEAD LOAD**



**FIG. A.6.2.2 LATERAL SHEAR ALONG CYLINDRICAL CROSS SECTION**

$$P_J = ARCL \times t \times \sigma_{DL} = ARCL \times t \times P_{DL}/A_{TE} \quad (A.6.2.1.1)$$

where  $P_J$  - joint load, #

ARCL - arc length supported by joint

e.g. Joint 5,  $ARCL' = 22\frac{1}{2}^\circ = 0.3927$  Radians,

and Joint 105,  $ARCL' = 7\frac{1}{2}^\circ = 0.1309$  Radians,

therefore  $ARCL_5 = ARCL'(R) = 0.3927 \times 174.875''$

$= 68.67''$

and  $ARCL_{105} = 0.1309 \times 174.875 = 22.89''$

$t$  - plate thickness,  $1\frac{1}{4}''$

$\sigma_{DL}$  - uniform DL stress -  $P_{DL}/A_{TE} \approx 1139$  PSI\*

$P_{DL}$  - total top edge dead load - 1,564,000#

$A_{TE}$  - top edge area -  $2\pi Rt = 2\pi(174.875)(1.25) = 1373$  In<sup>2</sup>

A summary of the top edge joint loads is in Table A.6.2.1.1.

TABLE A.6.2.1.1

MODEL I TOP EDGE SELF WEIGHT JOINT LOADS FOR LOAD CASE I

Joint Nos.	ARCL' RADS	ARCL IN	tARCL IN <sup>2</sup>	$P_J$ #	Factor F	$FP_J$ ( $F_2$ )
5, 10, 15, 20, 25, 223, 228, 233, 238	0.3927	68.67	85.84	97750	9	879750
34, 218	0.3709	64.86	81.08	92330	2	184660
49, 209	0.2662	46.55	58.19	66200	2	132520
64, 194	0.1614	28.22	35.28	40170	2	80340
79, 179	0.1396	24.41	30.52	34750	2	69500
94, 164	0.1309	22.89	28.61	32580	2	65160
105, 116, 127, 138, 149	0.1222	21.37	26.71	30420	5	152100 $\Sigma=1564030.$

A.6.2.2: VERTICAL JOINT LOADS DUE TO WIND MOMENT, TANGENTIAL  
HORIZONTAL JOINT LOADS DUE TO WIND SHEAR

With respect to Fig. A.6.2.2.1, the following equations were used to determine the vertical flexural and horizontal tangential shear loads at the joints along the Model I top edge for East to West (Load Case II) and North to South (Load Case III) Winds.

For moment,

$$P_b = M \times (\phi_2 - \phi_1) \times t \times (R_m \cos \theta) / S \quad (\text{A.6.2.2.1})$$

where  $M$  - the wind moment = 7,140,000#-Ft = 85,680,000#-In

$S$  - the section modulus of the cylinder

$$= \pi [(R_o^4 - R_i^4) / R_o] / 4$$

$$R_o = R_m + t/2$$

$$R_i = R_m - t/2$$

For shear,

$$\sigma_t = (V \sin \theta) / (\pi R_m t) \text{ and } \sigma_x = \sigma_t \sin \theta = (V \sin^2 \theta) / (\pi R_m t)$$

therefore  $dF_x = [(V \sin^2 \theta) / (\pi R_m t)] dA$  and  $dA = t R_m d\theta$

$$dF_x = [(V \sin^2 \theta) / (\pi R_m t)] (t R_m d\theta)$$

$$= [(V \sin^2 \theta) / \pi] d\theta$$

therefore

$$F_x = \frac{V}{\pi} \int_{\phi_1}^{\phi_2} \sin^2 \theta d\theta = \frac{V}{\pi} \left[ \theta - \sin \theta \cos \theta \right]_{\phi_1}^{\phi_2}$$

therefore per side,

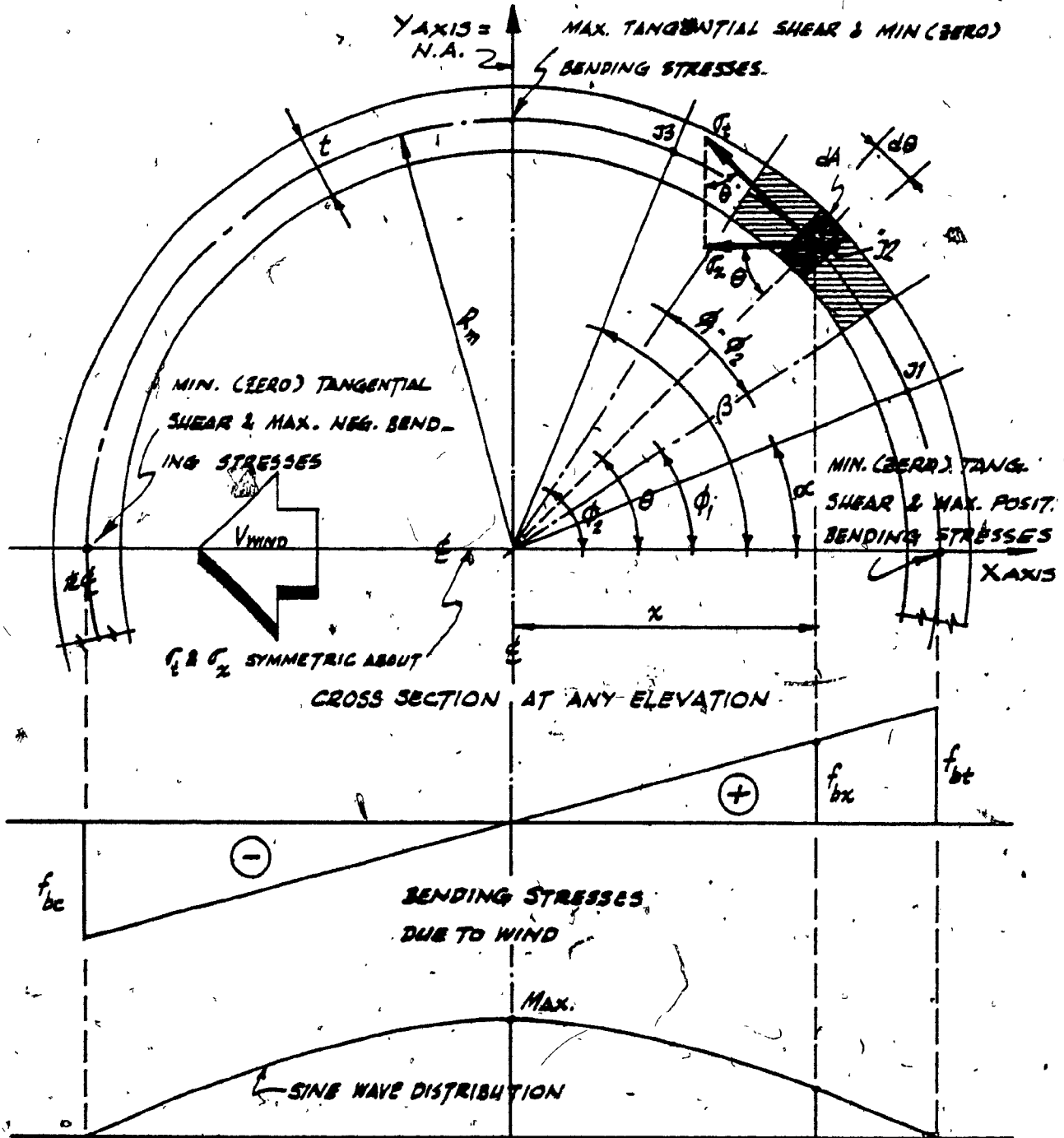
$$F_x = \frac{V}{2\pi} \left[ (\phi_2 - \sin \phi_2 \cos \phi_2) - (\phi_1 - \sin \phi_1 \cos \phi_1) \right]$$

and

$$F_t = F_x / \sin((\phi_1 + \phi_2) / 2),$$

where

$$0^\circ < (\phi_1 + \phi_2) / 2 < \pm 180^\circ$$



TANGENTIAL SHEAR STRESSES DUE TO WIND  
 FIG. A. 6.2.2.1 DESCRIPTION OF VARIABLES USED IN EQUATIONS FOR VERTICAL FLEXURAL AND HORIZONTAL TANGENTIAL SHEAR JOINT FORCES

The nomenclature for Fig. A.6.2.2.1 variables is as follows.

J1, J2, J3 - typical joints on cross section

$\alpha$  - angular location of J1

$\beta$  - same as  $\alpha$ , but for J3

$\theta$  - same as  $\alpha$ , but for J2

$\phi_1$  -  $(\alpha + \theta)/2$

$\phi_2$  -  $(\theta + \beta)/2$

$(\phi_1 - \phi_2)$  - the arc over which  $\int dA$  is taken

$R_m$  - the radius to the center line of plate of thickness  $t$

$(\phi_1 - \phi_2)R_m$  - the arc length supported by J2

$\sigma_t$  - tangential horizontal wind shear

$f_{bc}$  -  $-f_{bt}$  - compression and tension maximum bending stresses

$f_{bx}$  - bending stress at J2, and considered average bending stress over arc length supported by J2

$x$  - perpendicular distance of J2 from  $y$  or neutral axis

$= R_m \cos \theta$

The following Tables reflect the variables utilized in the solution of joint forces with an SR52 programmable calculator.



TABLE A.6.2.2.1  
SR52 INPUT DATA (See Fig. A.6.2.2.2)

Model-I East to West Wind

h Ft	M #-Ft	V #	t In	Rm In	Ro In	Ri In	S In <sup>3</sup>
50	7,140,000	32780	1½	174.875	175.5	174.25	119666

JN	θ	φ <sub>2</sub>	φ <sub>1</sub>	h
5	0	11½	- 11½	50
10	22½	33½	11½	"
15	45	56½	33½	"
20	67½	78½	56½	"
25	90	101½	78½	"
34	112½	122½	101½	"
49	132½	137½	122½	"
64	143	147	137½	"
79	151	155	147	"
94	159	162½	155	"
105	166	169½	162½	"
116	173	176½	169½	"
127	180	183½	176½	"
138	187	190½	183½	"
149	194	197½	190½	"
164	201	205	197½	"
179	209	213	205	"
194	217	222½	213	"
209	227½	237½	222½	"
218	247½	258½	237½	"
223	270	281½	258½	"
228	292½	303½	281½	"
233	315	326½	303½	"
238	337½	348½	326½	"

Table A.6.2.2.2 contains similar information for North to South Wind.

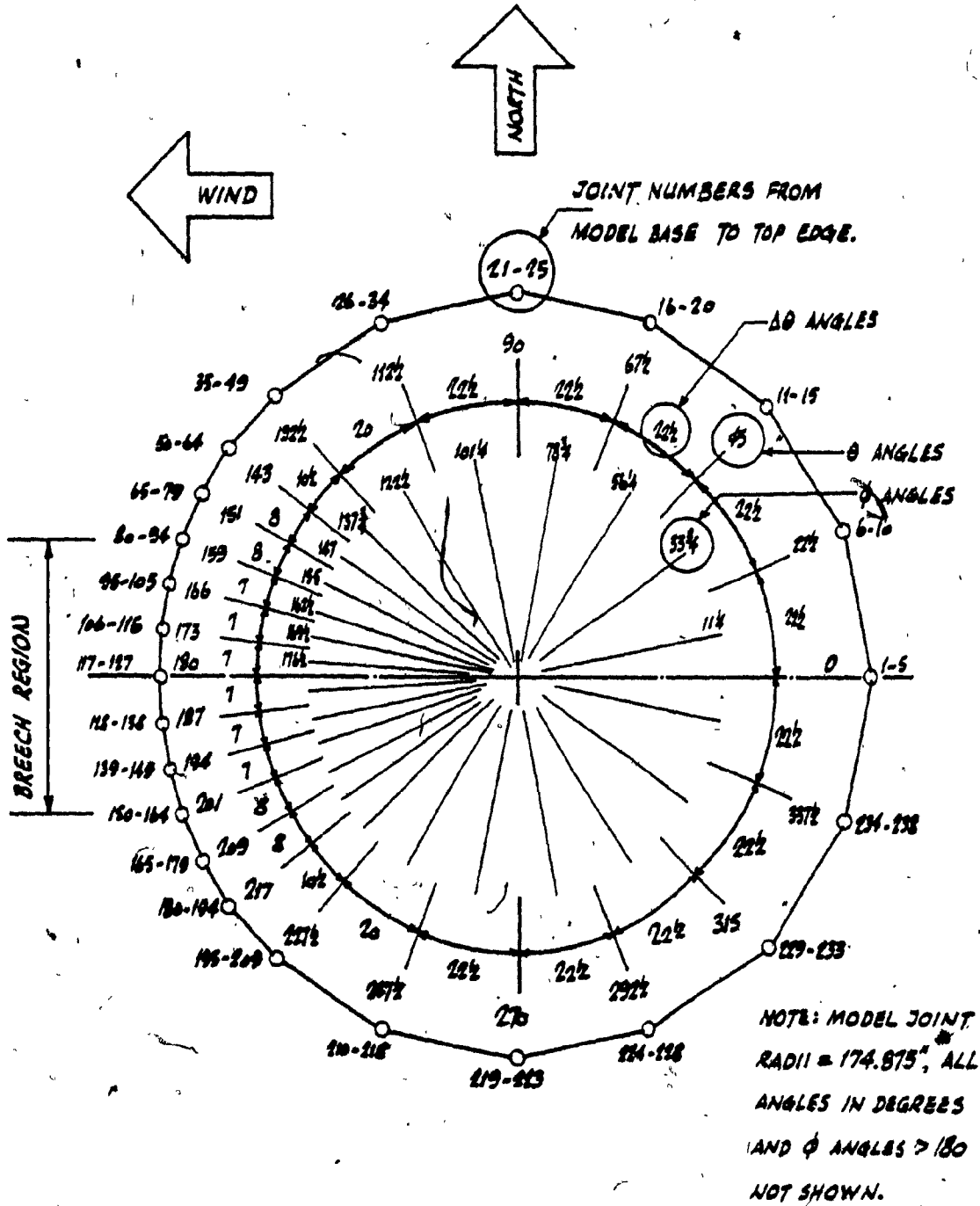


FIG. A.6.2.2.2 MODEL I CROSS SECTIONAL JOINT ANGLES FOR EAST TO WEST WIND

TABLE A.6.2.2.2

SR52 INPUT DATA (See Fig. A.6.2.2.3)

Model I North to South Wind

h Ft	M #-Ft	V #	t In	Rm In	Ro In	Ri In	S In <sup>3</sup>
50	7,140,000	32780	1½	174.875	175.5	174.25	119666

Jn	$\theta$	$\phi_2$	$\phi_1$	h
25	0	11½	- 11½	50
34	22½	32½	11½	"
49	42½	47½	32½	"
64	53	57	47½	"
79	61	65	57	"
94	69	72½	65	"
105	76	79½	72½	"
116	83	86½	79½	"
127	90	93½	86½	"
138	97	100½	93½	"
149	104	107½	100½	"
164	111	115	107½	"
179	119	123	115	"
194	127	132½	123	"
209	137½	147½	132½	"
218	157½	168½	147½	"
223	180	292½	168½	"
228	202½	213½	292½	"
233	225	236½	213½	"
238	247½	258½	236½	"
5	270	281½	258½	"
10	292½	303½	281½	"
15	315	326½	303½	"
20	337½	348½	326½	"

The sign convention of the tangential shear equation A.6.2.2.2 yields componential shears positive in the direction of the applied shear for integrations from 0 to 180° and 0 to -180°. Realizing the integration across the limits yield an undefined result, and realizing the results should be zero at the points on a cylinder which intersect a diametric axis.

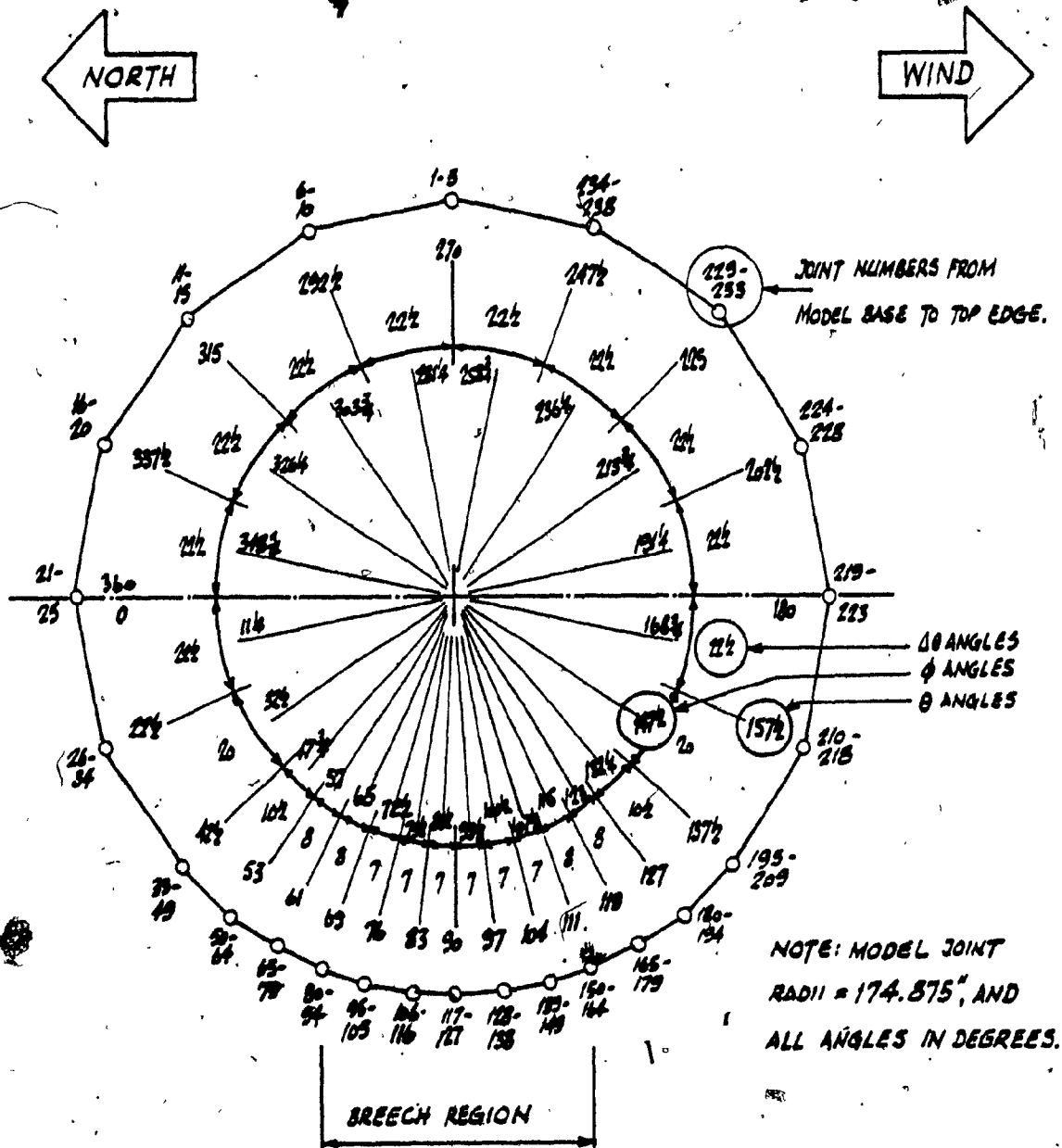


FIG. A.6.2.2.3 MODEL I CROSS SECTIONAL JOINT ANGLES FOR NORTH TO SOUTH WIND

parallel to the direction of applied wind, it was an easy task to program the SR52 to handle this discontinuity. The effect of having positive  $\phi_2$  angles larger than positive  $\phi_1$  angles in the third and fourth quadrants of a circle is the same as reversing the limits of the integral, which yields negative results. However, since the tangential shear forces were applied using the ANSYS cylindrical coordinate sign convention, where counter clockwise rotation is positive, the net results of reversing the sign of the tangential force (to negative) after  $(\phi_1 + \phi_2)/2 > 180^\circ$  were beneficial because the calculator's output matched the input sign convention required by the ANSYS program.

## A.6.3: MODEL II JOINT LOADS

Model II was a tapered cylinder of 200 Ft elevation, with only one load combination applied - Dead Load + E+W Wind. Because a single load combination was applied, the wind was taken with  $q = 9.2\text{PSF}$  for determining the appropriate moments and horizontal shears at the elevations at which joints were loaded. This simplified output interpretation, and the chosen load case, was that which governed in the Model I analysis.

The methods of determining the joint forces at the chosen elevations were identical to those used for Model I, utilizing the Model II summaries of Applied Loads - Table A.6.1.1, Section Properties - Table A.6.1.2, and the Procedures and Equations outlined in Section A.6.2.

Table A.6.3.1 summarizes the Model II data which was required to resolve joint force due to wind moment and shear at elevations 50, 80, 120, 160 and 200 Ft respectively.

The vertical joint forces due to the dead load of the stack above elevation 200 Ft was determined as follows.

## A.6.3.1: MODEL II TOP EDGE DEAD LOAD JOINT FORCES

The top edge, at elevation 200 Ft, was loaded with 934,480#. The configuration of the Model II top edge was 16 joints, each  $22\frac{1}{2}^{\circ}$  apart from adjacent joints. Therefore, the dead load joint forces per joint were

$$\frac{\text{EDL}}{16} = \frac{934,480}{16} = 58405\#/\text{Joint}$$

A.6.5.4 MODEL I: JOINT FORCES DUE TO WIND MOMENT AND SHEAR EAST TO WEST WIND

JOINT NUMBER	149.	PRT
ANGLES $\theta$	194.	PRT
$\phi_2$	197.5	PRT
$\phi_1$	190.5	PRT
ELEVATION, Ft	50.	PRT
MOMENT, #-Ft	7140000.	PRT
SHEAR, #	32780.	PRT
PLATE THICKNESS, In	1.25	PRT
MODEL RADII, MEAN, In	174.875	PRT
OUTSIDE, In	175.5	PRT
INSIDE, In	174.25	PRT
SECTION MODULUS, In <sup>3</sup>	119666.	PRT
VERTICAL JNT FORCE DUE TO MOM., #	-19121.49444	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, #	0.	PRT
COMPONENT IN DIRECTION OF WIND, #	0.	PRT
SUM OF COMPONENTS IN WIND DIRECTION, #	16363.08145	PRT
$\phi_{2r} - \sin\phi_2 \cos\phi_2$ FACTOR	3.141440801	PRT
$\phi_{1r} - \sin\phi_1 \cos\phi_1$ FACTOR	3.137517057	PRT

JOINT NUMBER	164.	PRT
ANGLES $\theta$	201.	PRT
$\phi_2$	205.	PRT
$\phi_1$	197.5	PRT
ELEVATION, Ft	50.	PRT
MOMENT, #-Ft	7140000.	PRT
SHEAR, #	32780.	PRT
PLATE THICKNESS, In	1.25	PRT
MODEL RADII, MEAN, In	174.875	PRT
OUTSIDE, In	175.5	PRT
INSIDE, In	174.25	PRT
SECTION MODULUS, In <sup>3</sup>	119666.	PRT
VERTICAL JNT FORCE DUE TO MOM., #	-19126.55672	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, #	-498.993818	PRT
COMPONENT IN DIRECTION OF WIND, #	180.8543405	PRT
SUM OF COMPONENTS IN WIND DIRECTION, #	16640.41325	PRT
$\phi_{2r} - \sin\phi_2 \cos\phi_2$ FACTOR	3.194902745	PRT
$\phi_{1r} - \sin\phi_1 \cos\phi_1$ FACTOR	3.160237055	PRT

A.6.5.5 MODEL I: JOINT FORCES DUE TO WIND MOMENT AND SHEAR EAST TO WEST WIND

JOINT NUMBER	179	PRT	179	PRT
ANGLES $\theta$	209	PRT	209	PRT
$\phi_2$	213	PRT	213	PRT
$\phi_1$	205	PRT	205	PRT
ELEVATION, Ft	50	PRT	50	PRT
MOMENT, #-Ft	7140000	PRT	7140000	PRT
SHEAR, #	32780	PRT	32780	PRT
PLATE THICKNESS, In	1.25	PRT	1.25	PRT
MODEL RADII, MEAN, In	174.875	PRT	174.875	PRT
OUTSIDE, In	175.5	PRT	175.5	PRT
INSIDE, In	174.25	PRT	174.25	PRT
SECTION MODULUS, In <sup>3</sup>	119666	PRT	119666	PRT
VERTICAL JNT FORCE DUE TO MOM., #	-19113.18385	PRT	-19113.18385	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, #	-708.898359	PRT	-708.898359	PRT
COMPONENT IN DIRECTION OF WIND, #	343.6807442	PRT	343.6807442	PRT
SUM OF COMPONENTS IN WIND DIRECTION, #	16984.094	PRT	16984.094	PRT
$\phi_{2r} - \sin\phi_2 \cos\phi_2$ FACTOR	3.260778578	PRT	3.260778578	PRT
$\phi_{1r} - \sin\phi_1 \cos\phi_1$ FACTOR	3.194902745	PRT	3.194902745	PRT

JOINT NUMBER	194	PRT	194	PRT
ANGLES $\theta$	217	PRT	217	PRT
$\phi_2$	222.25	PRT	222.25	PRT
$\phi_1$	213	PRT	213	PRT
ELEVATION, Ft	50	PRT	50	PRT
MOMENT, #-Ft	7140000	PRT	7140000	PRT
SHEAR, #	32780	PRT	32780	PRT
PLATE THICKNESS, In	1.25	PRT	1.25	PRT
MODEL RADII, MEAN, In	174.875	PRT	174.875	PRT
OUTSIDE, In	175.5	PRT	175.5	PRT
INSIDE, In	174.25	PRT	174.25	PRT
SECTION MODULUS, In <sup>3</sup>	119666	PRT	119666	PRT
VERTICAL JNT FORCE DUE TO MOM., #	-20179.67376	PRT	-20179.67376	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, #	-1029.912606	PRT	-1029.912606	PRT
COMPONENT IN DIRECTION OF WIND, #	628.7521786	PRT	628.7521786	PRT
SUM OF COMPONENTS IN WIND DIRECTION, #	17612.84618	PRT	17612.84618	PRT
$\phi_{2r} - \sin\phi_2 \cos\phi_2$ FACTOR	3.381296163	PRT	3.381296163	PRT
$\phi_{1r} - \sin\phi_1 \cos\phi_1$ FACTOR	3.260778578	PRT	3.260778578	PRT



A.6.5.6 MODEL I: JOINT FORCES DUE TO WIND MOMENT AND SHEAR EAST TO WEST WIND

JOINT NUMBER	223.	PRT	PRT
ANGLES $\theta$	270.	PRT	PRT
$\phi_2$	281.25	PRT	PRT
$\phi_1$	258.75	PRT	PRT
ELEVATION, Ft	50.	PRT	PRT
MOMENT, #-Ft	7140000.	PRT	PRT
SHEAR, #	32780.	PRT	PRT
PLATE THICKNESS, In	1.25	PRT	PRT
MODEL RADII, MEAN, In	174.875	PRT	PRT
OUTSIDE, In	175.5	PRT	PRT
INSIDE, In	174.25	PRT	PRT
SECTION MODULUS, In <sup>3</sup>	119666.	PRT	PRT
VERTICAL JNT FORCE DUE TO MOM., #	0.	PRT	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, #	-4045.247366	PRT	PRT
COMPONENT IN DIRECTION OF WIND, #	4045.247366	PRT	PRT
SUM OF COMPONENTS IN WIND DIRECTION, #	26579.91291	PRT	PRT
$\phi_{2r} - \sin\phi_2 \cos\phi_2$ FACTOR	5.100080237	PRT	PRT
$\phi_{1r} - \sin\phi_1 \cos\phi_1$ FACTOR	4.324697723	PRT	PRT

JOINT NUMBER	228.	PRT	PRT
ANGLES $\theta$	292.5	PRT	PRT
$\phi_2$	303.75	PRT	PRT
$\phi_1$	281.25	PRT	PRT
ELEVATION, Ft	50.	PRT	PRT
MOMENT, #-Ft	7140000.	PRT	PRT
SHEAR, #	32780.	PRT	PRT
PLATE THICKNESS, In	1.25	PRT	PRT
MODEL RADII, MEAN, In	174.875	PRT	PRT
OUTSIDE, In	175.5	PRT	PRT
INSIDE, In	174.25	PRT	PRT
SECTION MODULUS, In <sup>3</sup>	119666.	PRT	PRT
VERTICAL JNT FORCE DUE TO MOM., #	23520.45861	PRT	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, #	-3745.603949	PRT	PRT
COMPONENT IN DIRECTION OF WIND, #	3460.486826	PRT	PRT
SUM OF COMPONENTS IN WIND DIRECTION #	30040.39974	PRT	PRT
$\phi_{2r} - \sin\phi_2 \cos\phi_2$ FACTOR	5.763377369	PRT	PRT
$\phi_{1r} - \sin\phi_1 \cos\phi_1$ FACTOR	5.100080237	PRT	PRT

A.6.6: INPUT DATA CHECK OF MODEL I. TOP EDGE JOINT FORCES  
DUE TO NORTH+SOUTH WIND (LOAD CASE III)

As in Section A.6.5, the following data check was  
generated for Load Case II on Model I.

A.6.6.1 MODEL I: JOINT FORCES DUE TO WIND MOMENT AND SHEAR NORTH TO SOUTH WIND.

JOINT NUMBER	5.	PRT
ANGLES $\theta$	270.	PRT
$\phi_2$	281.25	PRT
$\phi_1$	258.75	PRT
ELEVATION, Ft	50.	PRT
MOMENT, #-Ft	7140000.	PRT
SHEAR, #	32780.	PRT
PLATE THICKNESS, In	1.25	PRT
MODEL RADII, MEAN, In	174.875	PRT
OUTSIDE, In	175.5	PRT
INSIDE, In	174.25	PRT
SECTION MODULUS, In <sup>3</sup>	119666.	PRT
VERTICAL JNT FORCE DUE TO MOM., #	0.	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, #	-4045.247366	PRT
COMPONENT IN DIRECTION OF WIND, #	4045.247366	PRT
SUM OF COMPONENTS IN WIND DIRECTION, #	26529.24473	PRT
$\phi_{2r} - \sin\phi_2 \cos\phi_2$ FACTOR	5.100080237	PRT
$\phi_{1r} - \sin\phi_1 \cos\phi_1$ FACTOR	4.324697723	PRT

JOINT NUMBER	10.	PRT
ANGLES $\theta$	292.5	PRT
$\phi_2$	303.75	PRT
$\phi_1$	281.25	PRT
ELEVATION, Ft	50.	PRT
MOMENT, #-Ft	7140000.	PRT
SHEAR, #	32780.	PRT
PLATE THICKNESS, In	1.25	PRT
MODEL RADII, MEAN, In	174.875	PRT
OUTSIDE, In	175.5	PRT
INSIDE, In	174.25	PRT
SECTION MODULUS, In <sup>3</sup>	119666.	PRT
VERTICAL JNT FORCE DUE TO MOM., #	29520.46861	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, #	-3745.603949	PRT
COMPONENT IN DIRECTION OF WIND, #	3460.486826	PRT
SUM OF COMPONENTS IN WIND DIRECTION, #	29989.73156	PRT
$\phi_{2r} - \sin\phi_2 \cos\phi_2$ FACTOR	5.763377369	PRT
$\phi_{1r} - \sin\phi_1 \cos\phi_1$ FACTOR	5.100080237	PRT

A.6.6.2 MODEL I: JOINT FORCES DUE TO WIND MOMENT AND SHEAR NORTH TO SOUTH WIND

JOINT NUMBER ..... 49.  
 ANGLES  $\theta$  ..... 42.5  
            $\phi_2$  ..... 47.75  
            $\phi_1$  ..... 32.5  
 ELEVATION, Ft ..... 50.  
 MOMENT, #-Ft ..... 714000.  
 SHEAR, # ..... 32780.  
 PLATE THICKNESS, In ..... 1.25  
 MODEL RADII, MEAN, In ..... 174.875  
                     OUTSIDE, In ..... 175.5  
                     INSIDE, In ..... 174.25  
 SECTION MODULUS, In<sup>3</sup> ..... 119666.  
 VERTICAL JNT FORCE DUE TO MOM., # ..... 61461.94642  
 TANGENTIAL JNT FORCE DUE TO SHEAR, # ..... 0.  
 COMPONENT IN DIRECTION OF WIND, # ..... 0.  
 SUM OF COMPONENTS IN WIND DIRECTION, # ..... 0.  
 $\phi_{2r}$  -  $\sin\phi_2 \cos\phi_2$  FACTOR ..... 0.  
 $\phi_{1r}$  -  $\sin\phi_1 \cos\phi_1$  FACTOR ..... 0.

PRT ..... 49.  
 PRT ..... 42.5  
 PRT ..... 47.75  
 PRT ..... 32.5  
 PRT ..... 50.  
 PRT ..... 714000.  
 PRT ..... 32780.  
 PRT ..... 1.25  
 PRT ..... 174.875  
 PRT ..... 175.5  
 PRT ..... 174.25  
 PRT ..... 119666.  
 PRT ..... 30713.16123  
 PRT ..... 1794.07641  
 PRT ..... 1156.20569  
 PRT ..... 1725.236188  
 PRT ..... 3356966186  
 PRT ..... 1140781134

JOINT NUMBER ..... 64.  
 ANGLES  $\theta$  ..... 53.  
            $\phi_1$  ..... 57.  
            $\phi_2$  ..... 47.75  
 ELEVATION, Ft ..... 50.  
 MOMENT, #-Ft ..... 714000.  
 SHEAR, # ..... 32780.  
 PLATE THICKNESS, In ..... 1.25  
 MODEL RADII, MEAN, In ..... 174.875  
                     OUTSIDE, In ..... 175.5  
                     INSIDE, In ..... 174.25  
 SECTION MODULUS, In<sup>3</sup> ..... 119666.  
 VERTICAL JNT FORCE DUE TO MOM., # ..... 53628.79908  
 TANGENTIAL JNT FORCE DUE TO SHEAR, # ..... 1527.258751  
 COMPONENT IN DIRECTION OF WIND, # ..... 569.0304979  
 SUM OF COMPONENTS IN WIND DIRECTION, # ..... 569.0304979  
 $\phi_{2r}$  -  $\sin\phi_2 \cos\phi_2$  FACTOR ..... 1140781134  
 $\phi_{1r}$  -  $\sin\phi_1 \cos\phi_1$  FACTOR ..... 0050078247

PRT ..... 64.  
 PRT ..... 53.  
 PRT ..... 57.  
 PRT ..... 47.75  
 PRT ..... 50.  
 PRT ..... 714000.  
 PRT ..... 32780.  
 PRT ..... 1.25  
 PRT ..... 174.875  
 PRT ..... 175.5  
 PRT ..... 174.25  
 PRT ..... 119666.  
 PRT ..... 15206.47489  
 PRT ..... 1333.010712  
 PRT ..... 1055.775599  
 PRT ..... 2781.011787  
 PRT ..... 5380649448  
 PRT ..... 3356966186

A.6.6.3 MODEL I: JOINT FORCES DUE TO WIND MOMENT AND SHEAR NORTH TO SOUTH WIND

JOINT NUMBER	79.	PRT
ANGLES $\theta$	61.	PRT
$\phi_2$	68.	PRT
$\phi_1$	57.	PRT
ELEVATION, Ft	50.	PRT
MOMENT, #-Ft	7140000.	PRT
SHEAR, #	32780.	PRT
PLATE THICKNESS, In	1.25	PRT
MODEL RADII, MEAN, In	174.875	PRT
OUTSIDE, In	175.5	PRT
INSIDE, In	174.25	PRT
SECTION MODULUS, In <sup>3</sup>	119666.	PRT
VERTICAL JNT FORCE DUE TO MOM., #	10594.61081	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, #	1272.791061	PRT
COMPONENT IN DIRECTION OF WIND, #	1113.208145	PRT
SUM OF COMPONENTS IN WIND DIRECTION, #	3894.219932	PRT
$\phi_{2r} - \sin\phi_2 \cos\phi_2$ FACTOR	7514417922	PRT
$\phi_{1r} - \sin\phi_1 \cos\phi_1$ FACTOR	5380649448	PRT

JOINT NUMBER	94.	PRT
ANGLES $\theta$	69.	PRT
$\phi_1$	72.5	PRT
$\phi_2$	65.	PRT
ELEVATION, Ft	50.	PRT
MOMENT, #-Ft	7140000.	PRT
SHEAR, #	32780.	PRT
PLATE THICKNESS, In	1.25	PRT
MODEL RADII, MEAN, In	174.875	PRT
OUTSIDE, In	175.5	PRT
INSIDE, In	174.25	PRT
SECTION MODULUS, In <sup>3</sup>	119666.	PRT
VERTICAL JNT FORCE DUE TO MOM., #	7341.997237	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, #	1271.425952	PRT
COMPONENT IN DIRECTION OF WIND, #	1184.978993	PRT
SUM OF COMPONENTS IN WIND DIRECTION, #	5079.198924	PRT
$\phi_{2r} - \sin\phi_2 \cos\phi_2$ FACTOR	9785754895	PRT
$\phi_{1r} - \sin\phi_1 \cos\phi_1$ FACTOR	7514417922	PRT

A.6.6.4 MODEL I: JOINT FORCES DUE TO WIND MOMENT AND SHEAR NORTH TO SOUTH WIND

JOINT NUMBER ..... 127. PRT  
 ANGLES  $\theta$  ..... 90. PRT  
 $\phi_2$  ..... 93.5 PRT  
 $\phi_1$  ..... 86.5 PRT  
 ELEVATION, Ft ..... 50. PRT  
 MOMENT, #-Ft ..... 7140000. PRT  
 SHEAR, # ..... 32780. PRT  
 PLATE THICKNESS, In ..... 1.25 PRT  
 MODEL RADII, MEAN, In ..... 174.875 PRT  
     OUTSIDE, In ..... 175.5 PRT  
     INSIDE, In ..... 174.25 PRT  
 SECTION MODULUS, In<sup>3</sup> ..... 119666. PRT  
 VERTICAL JNT FORCE DUE TO MOM., # ..... 0. PRT  
 TANGENTIAL JNT FORCE DUE TO SHEAR, # ..... 1273.193323 PRT  
 COMPONENT IN DIRECTION OF WIND, # ..... 1273.193323 PRT  
 SUM OF COMPONENTS IN WIND DIRECTION, # ..... 8805.470344 PRT  
 $\phi_{2r}$  - Sin $\phi_2$  Cos $\phi_2$  FACTOR ..... 1.692817522 PRT  
 $\phi_{1r}$  - Sin $\phi_1$  Cos $\phi_1$  FACTOR ..... 1.448775131 PRT

149. PRT  
 104. PRT  
 107.5 PRT  
 100.5 PRT  
 50. PRT  
 7140000. PRT  
 32780. PRT  
 1.25 PRT  
 174.875 PRT  
 175.5 PRT  
 174.25 PRT  
 119666. PRT  
 -4625.908182 PRT  
 1235.469611 PRT  
 1198.770883 PRT  
 11258.54844 PRT  
 2.163017164 PRT  
 1.933239873 PRT

JOINT NUMBER ..... 164. PRT  
 ANGLES  $\theta$  ..... 111. PRT  
 $\phi_2$  ..... 115. PRT  
 $\phi_1$  ..... 107.5 PRT  
 ELEVATION, Ft ..... 50. PRT  
 MOMENT, #-Ft ..... 7140000. PRT  
 SHEAR, # ..... 32780. PRT  
 PLATE THICKNESS, In ..... 1.25 PRT  
 MODEL RADII, MEAN, In ..... 174.875 PRT  
     OUTSIDE, In ..... 175.5 PRT  
     INSIDE, In ..... 174.25 PRT  
 SECTION MODULUS, In<sup>3</sup> ..... 119666. PRT  
 VERTICAL JNT FORCE DUE TO MOM., # ..... -2330.323972 PRT  
 TANGENTIAL JNT FORCE DUE TO SHEAR, # ..... 1263.726842 PRT  
 COMPONENT IN DIRECTION OF WIND, # ..... 1254.307214 PRT  
 SUM OF COMPONENTS IN WIND DIRECTION, # ..... 10059.7756 PRT  
 $\phi_{2r}$  - Sin $\phi_2$  Cos $\phi_2$  FACTOR ..... 1.933239873 PRT  
 $\phi_{1r}$  - Sin $\phi_1$  Cos $\phi_1$  FACTOR ..... 1.692817522 PRT

164. PRT  
 111. PRT  
 115. PRT  
 107.5 PRT  
 50. PRT  
 7140000. PRT  
 32780. PRT  
 1.25 PRT  
 174.875 PRT  
 175.5 PRT  
 174.25 PRT  
 119666. PRT  
 -2330.323972 PRT  
 1263.726842 PRT  
 1254.307214 PRT  
 10059.7756 PRT  
 1.933239873 PRT  
 1.692817522 PRT

A.6.6.5 MODEL I: JOINT FORCES DUE TO WIND MOMENT AND SHEAR NORTH TO SOUTH WIND

JOINT NUMBER ..... 179. PRT  
 ANGLES  $\theta$  ..... 119. PRT  
 $\phi_2$  ..... 123. PRT  
 $\phi_1$  ..... 115. PRT  
 ELEVATION, Ft ..... 50. PRT  
 MOMENT, #-Ft ..... 7140000. PRT  
 SHEAR, # ..... 32780. PRT  
 PLATE THICKNESS, In ..... 1.25 PRT  
 MODEL RADII, MEAN, In ..... 174.875 PRT  
 OUTSIDE, In ..... 175.5 PRT  
 INSIDE, In ..... 174.25 PRT  
 SECTION MODULUS, In<sup>3</sup> ..... 119666. PRT  
 VERTICAL JNT FORCE DUE TO MOM., # ..... -10594.61081 PRT  
 TANGENTIAL JNT FORCE DUE TO SHEAR, # ..... 1272.791061 PRT  
 COMPONENT IN DIRECTION OF WIND, # ..... 1113.208145 PRT  
 SUM OF COMPONENTS IN WIND DIRECTION, # ..... 13556.73558 PRT  
 $\phi_{2r} - \sin\phi_2 \cos\phi_2$  FACTOR ..... 2.603527709 PRT  
 $\phi_{1r} - \sin\phi_1 \cos\phi_1$  FACTOR ..... 2.390150861 PRT

JOINT NUMBER ..... 218. PRT  
 ANGLES  $\theta$  ..... 157.5 PRT  
 $\phi_2$  ..... 168.75 PRT  
 $\phi_1$  ..... 147.5 PRT  
 ELEVATION, Ft ..... 50. PRT  
 MOMENT, #-Ft ..... 7140000. PRT  
 SHEAR, # ..... 32780. PRT  
 PLATE THICKNESS, In ..... 1.25 PRT  
 MODEL RADII, MEAN, In ..... 174.875 PRT  
 OUTSIDE, In ..... 175.5 PRT  
 INSIDE, In ..... 174.25 PRT  
 SECTION MODULUS, In<sup>3</sup> ..... 119666. PRT  
 VERTICAL JNT FORCE DUE TO MOM., # ..... -53628.79908 PRT  
 TANGENTIAL JNT FORCE DUE TO SHEAR, # ..... 1527.258751 PRT  
 COMPONENT IN DIRECTION OF WIND, # ..... 569.0304979 PRT  
 SUM OF COMPONENTS IN WIND DIRECTION, # ..... 16337.74737 PRT  
 $\phi_{2r} - \sin\phi_2 \cos\phi_2$  FACTOR ..... 3.136584829 PRT  
 $\phi_{1r} - \sin\phi_1 \cos\phi_1$  FACTOR ..... 3.02751454 PRT

A.6.6.6 MODEL I: JOINT FORCES DUE TO WIND MOMENT AND SHEAR NORTH TO SOUTH WIND

JOINT NUMBER	233.	PRT	233.	PRT
ANGLES $\theta$	180.	PRT	180.	PRT
$\phi_2$	191.25	PRT	191.25	PRT
$\phi_1$	168.75	PRT	168.75	PRT
ELEVATION, Ft	50.	PRT	50.	PRT
MOMENT, #-ft	7140000.	PRT	7140000.	PRT
SHEAR, #	32780.	PRT	32780.	PRT
PLATE THICKNESS, In	1.25	PRT	1.25	PRT
MODEL RADII, MEAN, In	174.875	PRT	174.875	PRT
OUTSIDE, In	175.5	PRT	175.5	PRT
INSIDE, In	174.25	PRT	174.25	PRT
SECTION MODULUS, In <sup>3</sup>	119666.	PRT	119666.	PRT
VERTICAL JNT FORCE DUE TO MOM., #	-61461.94642	PRT	-43460.1591	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, #	0.	PRT	-2897.370036	PRT
COMPONENT IN DIRECTION OF WIND, #	0.	PRT	2048.75	PRT
SUM OF COMPONENTS IN WIND DIRECTION, #	16337.74737	PRT	19023.51054	PRT
$\phi_{2r} - \text{Sin}\phi_2 \text{Cos}\phi_2 \text{ FACTOR}$	3.136584829	PRT	3.661400592	PRT
$\phi_{1r} - \text{Sin}\phi_1 \text{Cos}\phi_1 \text{ FACTOR}$	3.02751454	PRT	3.26870151	PRT

JOINT NUMBER	238.	PRT	238.	PRT
ANGLES $\theta$	207.5	PRT	247.5	PRT
$\phi_2$	213.75	PRT	258.75	PRT
$\phi_1$	191.25	PRT	236.25	PRT
ELEVATION, Ft	50.	PRT	50.	PRT
MOMENT, #-ft	7140000.	PRT	7140000.	PRT
SHEAR, #	32780.	PRT	32780.	PRT
PLATE THICKNESS, In	1.25	PRT	1.25	PRT
MODEL RADII, MEAN, In	174.875	PRT	174.875	PRT
OUTSIDE, In	175.5	PRT	175.5	PRT
INSIDE, In	174.25	PRT	174.25	PRT
SECTION MODULUS, In <sup>3</sup>	119666.	PRT	119666.	PRT
VERTICAL JNT FORCE DUE TO MOM., #	-56789.43482	PRT	-23520.46881	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, #	-1664.595643	PRT	-3745.603947	PRT
COMPONENT IN DIRECTION OF WIND, #	637.0131742	PRT	3460.486826	PRT
SUM OF COMPONENTS IN WIND DIRECTION, #	16974.76054	PRT	22483.99737	PRT
$\phi_{2r} - \text{Sin}\phi_2 \text{Cos}\phi_2 \text{ FACTOR}$	3.26870151	PRT	4.324697723	PRT
$\phi_{1r} - \text{Sin}\phi_1 \text{Cos}\phi_1 \text{ FACTOR}$	3.146600478	PRT	3.661400592	PRT



A.6.7: INPUT DATA CHECK OF MODEL II JOINT FORCES DUE TO  
EAST-WEST WIND (LOAD CASE 3)

Again with the procedures of Section A.6.2.2, the SR52 was utilized to generate the joint forces applied to Model II to represent the applied wind load. Note that a wind pressure of  $q = 9.2\text{PSF}$  was utilized in forming the joint forces in the Model II load combination.

A.6.7.1 MODEL II: JOINT FORCES DUE TO WIND MOMENT AND SHEAR EAST TO WEST WIND

JOINT NUMBER	213.	PRT	215.	PRT
ANGLES $\theta$	0.	PRT	45.	PRT
$\phi_2$	11.25	PRT	56.25	PRT
$\phi_1$	-11.25	PRT	33.75	PRT
ELEVATION, Ft	50.	PRT	50.	PRT
MOMENT, #-Ft	307124.	PRT	307124.	PRT
SHEAR, #	20148.	PRT	20148.	PRT
PLATE THICKNESS, In	1.25	PRT	1.25	PRT
MODEL RADII, MEAN, In	170.375	PRT	170.375	PRT
OUTSIDE, In	171.	PRT	171.	PRT
INSIDE, In	169.75	PRT	169.75	PRT
SECTION MODULUS, In <sup>3</sup>	113576.	PRT	113576.	PRT
VERTICAL JNT FORCE DUE TO MOM., #	2713.839752	PRT	1918.974492	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, #	0.	PRT	1780.848428	PRT
COMPONENT IN DIRECTION OF WIND, #	0.	PRT	1259.25	PRT
SUM OF COMPONENTS IN WIND DIRECTION, #	0.	PRT	1650.785736	PRT
$\phi_{2r} - \text{Sin}\phi_2 \text{Cos}\phi_2$ FACTOR	0.	PRT	0.519807938	PRT
$\phi_{1r} - \text{Sin}\phi_1 \text{Cos}\phi_1$ FACTOR	0.	PRT	.1271088563	PRT

JOINT NUMBER	214.	PRT	216.	PRT
ANGLES $\theta$	22.5	PRT	67.5	PRT
$\phi_2$	33.75	PRT	78.75	PRT
$\phi_1$	11.25	PRT	56.25	PRT
ELEVATION, Ft	50.	PRT	50.	PRT
MOMENT, #-Ft	307124.	PRT	307124.	PRT
SHEAR, #	20148.	PRT	20148.	PRT
PLATE THICKNESS, In	1.25	PRT	1.25	PRT
MODEL RADII, MEAN, In	170.375	PRT	170.375	PRT
OUTSIDE, In	171.	PRT	171.	PRT
INSIDE, In	169.75	PRT	169.75	PRT
SECTION MODULUS, In <sup>3</sup>	113576.	PRT	113576.	PRT
VERTICAL JNT FORCE DUE TO MOM., #	2507.261002	PRT	1038.541511	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, #	1023.132185	PRT	2302.209529	PRT
COMPONENT IN DIRECTION OF WIND, #	391.5357362	PRT	2126.964264	PRT
SUM OF COMPONENTS IN WIND DIRECTION, #	391.5357362	PRT	3777.75	PRT
$\phi_{2r} - \text{Sin}\phi_2 \text{Cos}\phi_2$ FACTOR	.1271088563	PRT	1.18310507	PRT
$\phi_{1r} - \text{Sin}\phi_1 \text{Cos}\phi_1$ FACTOR	.0050078247	PRT	0.519807938	PRT

A.6.7.2 MODEL II: JOINT FORCES DUE TO WIND MOMENT AND SHEAR EAST TO WEST WIND

JOINT NUMBER	217.	PRT	219.	PRT
ANGLES $\theta$	90.	PRT	135.	PRT
$\phi_2$	101.25	PRT	138.75	PRT
$\phi_1$	78.75	PRT	123.75	PRT
ELEVATION, Ft	50.	PRT	50.	PRT
MOMENT, #-Ft	307124.	PRT	307124.	PRT
SHEAR, #	20148.	PRT	20148.	PRT
PLATE THICKNESS, In	1.25	PRT	1.25	PRT
MODEL RADII, MEAN, In	170.375	PRT	170.375	PRT
OUTSIDE, In	171.	PRT	171.	PRT
INSIDE, In	169.75	PRT	169.75	PRT
SECTION MODULUS, In <sup>3</sup>	113576.	PRT	113576.	PRT
VERTICAL JNT FORCE DUE TO MOM., #	0.	PRT	-1279.316328	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, #	2486.38328	PRT	-1260.679868	PRT
COMPONENT IN DIRECTION OF WIND, #	2486.38328	PRT	947.829309	PRT
SUM OF COMPONENTS IN WIND DIRECTION, #	6264.13328	PRT	9338.926853	PRT
$\phi_2r - \sin\phi_2 \cos\phi_2$ FACTOR	1.958487584	PRT	2.917366768	PRT
$\phi_1r - \sin\phi_1 \cos\phi_1$ FACTOR	1.18310507	PRT	2.621784716	PRT

JOINT NUMBER	218.	PRT	220.	PRT
ANGLES $\theta$	112.5	PRT	142.5	PRT
$\phi_2$	123.75	PRT	146.25	PRT
$\phi_1$	101.25	PRT	138.75	PRT
ELEVATION, Ft	50.	PRT	50.	PRT
MOMENT, #-Ft	307124.	PRT	307124.	PRT
SHEAR, #	20148.	PRT	20148.	PRT
PLATE THICKNESS, In	1.25	PRT	1.25	PRT
MODEL RADII, MEAN, In	170.375	PRT	170.375	PRT
OUTSIDE, In	171.	PRT	171.	PRT
INSIDE, In	169.75	PRT	169.75	PRT
SECTION MODULUS, In <sup>3</sup>	113576.	PRT	113576.	PRT
VERTICAL JNT FORCE DUE TO MOM., #	-1038.541511	PRT	-717.6179441	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, #	2302.209529	PRT	511.5644259	PRT
COMPONENT IN DIRECTION OF WIND, #	2126.964264	PRT	311.4206909	PRT
SUM OF COMPONENTS IN WIND DIRECTION, #	8391.097544	PRT	9650.347544	PRT
$\phi_2r - \sin\phi_2 \cos\phi_2$ FACTOR	2.621784716	PRT	3.014483797	PRT
$\phi_1r - \sin\phi_1 \cos\phi_1$ FACTOR	1.958487584	PRT	2.917366768	PRT

A.6.7.3 MODEL II: JOINT FORCES DUE TO WIND MOMENT AND SHEAR EAST TO WEST WIND

JOINT NUMBER ..... 221. PRT  
 ANGLES  $\theta$  ..... 150. PRT  
 $\phi_2$  ..... 153.75 PRT  
 $\phi_1$  ..... 146.25 PRT  
 ELEVATION, Ft ..... 50. PRT  
 MOMENT, #-Ft ..... 307124. PRT  
 SHEAR, # ..... 20148. PRT  
 PLATE THICKNESS, In ..... 1.25 PRT  
 MODEL RADII, MEAN, In ..... 170.375 PRT  
 OUTSIDE, In ..... 171. PRT  
 INSIDE, In ..... 169.75 PRT  
 SECTION MODULUS, In<sup>3</sup> ..... 113576. PRT  
 VERTICAL JNT FORCE DUE TO MOM., # ..... -783.4180557 PRT  
 TANGENTIAL JNT FORCE DUE TO SHEAR, # ..... 420.9476905 PRT  
 COMPONENT IN DIRECTION OF WIND, # ..... 210.4738453 PRT  
 SUM OF COMPONENTS IN WIND DIRECTION, # ..... 9860.821389 PRT  
 $\phi_{2r} - \sin\phi_2 \cos\phi_2$  FACTOR ..... 3.080120395 PRT  
 $\phi_{1r} - \sin\phi_1 \cos\phi_1$  FACTOR ..... 3.014483797 PRT

JOINT NUMBER ..... 224. PRT  
 ANGLES  $\theta$  ..... 157.5 PRT  
 $\phi_2$  ..... 161.25 PRT  
 $\phi_1$  ..... 153.75 PRT  
 ELEVATION, Ft ..... 50. PRT  
 MOMENT, #-Ft ..... 307124. PRT  
 SHEAR, # ..... 20148. PRT  
 PLATE THICKNESS, In ..... 1.25 PRT  
 MODEL RADII, MEAN, In ..... 170.375 PRT  
 OUTSIDE, In ..... 171. PRT  
 INSIDE, In ..... 169.75 PRT  
 SECTION MODULUS, In<sup>3</sup> ..... 113576. PRT  
 VERTICAL JNT FORCE DUE TO MOM., # ..... -835.7536672 PRT  
 TANGENTIAL JNT FORCE DUE TO SHEAR, # ..... 323.4757851 PRT  
 COMPONENT IN DIRECTION OF WIND, # ..... 123.7888237 PRT  
 SUM OF COMPONENTS IN WIND DIRECTION # ..... 9984.610213 PRT  
 $\phi_{2r} - \sin\phi_2 \cos\phi_2$  FACTOR ..... 3.118724133 PRT  
 $\phi_{1r} - \sin\phi_1 \cos\phi_1$  FACTOR ..... 3.080120395 PRT

A.6.7.4 MODEL II: JOINT FORCES DUE TO WIND MOMENT AND SHEAR EAST TO WEST WIND

JOINT NUMBER	225.	PRT	227.	PRT
ANGLES $\theta$	180.	PRT	195.	PRT
$\phi_2$	183.75	PRT	198.75	PRT
$\phi_1$	176.25	PRT	191.25	PRT
ELEVATION, Ft	50.	PRT	50.	PRT
MOMENT, #-Ft	307124.	PRT	307124.	PRT
SHEAR, #	20148.	PRT	20148.	PRT
PLATE THICKNESS, In	1.25	PRT	1.25	PRT
MODEL RADII, MEAN, In	170.375	PRT	170.375	PRT
OUTSIDE, In	171.	PRT	171.	PRT
INSIDE, In	169.75	PRT	169.75	PRT
SECTION MODULUS, In <sup>3</sup>	113576.	PRT	113576.	PRT
VERTICAL JNT FORCE DUE TO MOM., #	-904.6132507	PRT	-873.7893017	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, #	0.	PRT	-221.286139	PRT
COMPONENT IN DIRECTION OF WIND, #	0.	PRT	57.2730672	PRT
SUM OF COMPONENTS IN WIND DIRECTION, #	10057.34279	PRT	10130.07538	PRT
$\phi_{2r}$ - $\sin\phi_2 \cos\phi_2$ FACTOR	3.141405903	PRT	3.164461174	PRT
$\phi_{1r}$ - $\sin\phi_1 \cos\phi_1$ FACTOR	3.136584829	PRT	3.146600478	PRT

JOINT NUMBER	226.	PRT	228.	PRT
ANGLES $\theta$	187.5	PRT	202.5	PRT
$\phi_2$	191.25	PRT	206.25	PRT
$\phi_1$	183.75	PRT	198.75	PRT
ELEVATION, Ft	50.	PRT	50.	PRT
MOMENT, #-Ft	307124.	PRT	307124.	PRT
SHEAR, #	20148.	PRT	20148.	PRT
PLATE THICKNESS, In	1.25	PRT	1.25	PRT
MODEL RADII, MEAN, In	170.375	PRT	170.375	PRT
OUTSIDE, In	171.	PRT	171.	PRT
INSIDE, In	169.75	PRT	169.75	PRT
SECTION MODULUS, In <sup>3</sup>	113576.	PRT	113576.	PRT
VERTICAL JNT FORCE DUE TO MOM., #	-896.874159	PRT	-835.7536672	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, #	-118.4399423	PRT	-323.475785	PRT
COMPONENT IN DIRECTION OF WIND, #	15.45951468	PRT	123.7888237	PRT
SUM OF COMPONENTS IN WIND DIRECTION #	10072.80231	PRT	10253.8642	PRT
$\phi_{2r}$ - $\sin\phi_2 \cos\phi_2$ FACTOR	3.146600478	PRT	3.203064912	PRT
$\phi_{1r}$ - $\sin\phi_1 \cos\phi_1$ FACTOR	3.141779404	PRT	3.164461174	PRT

A.6.7.5 MODEL II: JOINT FORCES DUE TO WIND MOMENT AND SHEAR EAST TO WEST WIND

JOINT NUMBER .....	229.	PRT	231.	PRT
ANGLES $\theta$ .....	210.	PRT	225.	PRT
$\phi_2$ .....	213.75	PRT	236.25	PRT
$\phi_1$ .....	206.25	PRT	221.25	PRT
ELEVATION, Ft .....	50.	PRT	50.	PRT
MOMENT, #-Ft .....	307124.	PRT	307124.	PRT
SHEAR, # .....	20148.	PRT	20148.	PRT
PLATE THICKNESS, In .....	1.25	PRT	1.25	PRT
MODEL RADII, MEAN, In .....	170.375	PRT	170.375	PRT
OUTSIDE, In .....	171.	PRT	171.	PRT
INSIDE, In .....	169.75	PRT	169.75	PRT
SECTION MODULUS, In <sup>3</sup> .....	113576.	PRT	113576.	PRT
VERTICAL JNT FORCE DUE TO MOM., # .....	-783.4180557	PRT	-1277.316328	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, # .....	-420.9476905	PRT	-1260.679868	PRT
COMPONENT IN DIRECTION OF WIND, # .....	210.4738453	PRT	947.829309	PRT
SUM OF COMPONENTS IN WIND DIRECTION, # .....	10464.33805	PRT	11723.58805	PRT
$\phi_{2r}$ - $\sin\phi_2 \cos\phi_2$ FACTOR .....	3.26870151	PRT	3.661400592	PRT
$\phi_{1r}$ - $\sin\phi_1 \cos\phi_1$ FACTOR .....	3.203064912	PRT	3.365818539	PRT

JOINT NUMBER .....	230.	PRT	232.	PRT
ANGLES $\theta$ .....	217.5	PRT	247.5	PRT
$\phi_2$ .....	221.25	PRT	258.75	PRT
$\phi_1$ .....	213.75	PRT	236.25	PRT
ELEVATION, Ft .....	50.	PRT	50.	PRT
MOMENT, #-Ft .....	307124.	PRT	307124.	PRT
SHEAR, # .....	20148.	PRT	20148.	PRT
PLATE THICKNESS, In .....	1.25	PRT	1.25	PRT
MODEL RADII, MEAN, In .....	170.375	PRT	170.375	PRT
OUTSIDE, In .....	171.	PRT	171.	PRT
INSIDE, In .....	169.75	PRT	169.75	PRT
SECTION MODULUS, In <sup>3</sup> .....	113576.	PRT	113576.	PRT
VERTICAL JNT FORCE DUE TO MOM., # .....	-717.6779441	PRT	-1038.541511	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, # .....	-511.5644259	PRT	-2302.209529	PRT
COMPONENT IN DIRECTION OF WIND, # .....	311.4206909	PRT	2126.964264	PRT
SUM OF COMPONENTS IN WIND DIRECTION, # .....	10775.75874	PRT	13850.55231	PRT
$\phi_{2r}$ - $\sin\phi_2 \cos\phi_2$ FACTOR .....	3.365818539	PRT	4.324697723	PRT
$\phi_{1r}$ - $\sin\phi_1 \cos\phi_1$ FACTOR .....	3.26870151	PRT	3.661400592	PRT

A.6.7.6 MODEL II: JOINT FORCES DUE TO WIND MOMENT AND SHEAR EAST TO WEST WIND

JOINT NUMBER	233.	PRT
ANGLES $\theta$	270.	PRT
$\phi_2$	281.25	PRT
$\phi_1$	258.75	PRT
ELEVATION, Ft	50.	PRT
MOMENT, #-Ft	307124.	PRT
SHEAR, #	20148.	PRT
PLATE THICKNESS, In	1.25	PRT
MODEL RADII, MEAN, In	170.375	PRT
OUTSIDE, In	171.	PRT
INSIDE, In	169.75	PRT
SECTION MODULUS, In <sup>3</sup>	113576.	PRT
VERTICAL JNT FORCE DUE TO MOM., #	0.	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, #	-2486.38328	PRT
COMPONENT IN DIRECTION OF WIND, #	2486.38328	PRT
SUM OF COMPONENTS IN WIND DIRECTION, #	16336.93559	PRT
$\phi_{2r} - \sin\phi_2 \cos\phi_2$ FACTOR	5.100080237	PRT
$\phi_{1r} - \sin\phi_1 \cos\phi_1$ FACTOR	4.324697723	PRT

JOINT NUMBER	236.	PRT
ANGLES $\theta$	337.5	PRT
$\phi_2$	348.75	PRT
$\phi_1$	326.25	PRT
ELEVATION, Ft	50.	PRT
MOMENT, #-Ft	307124.	PRT
SHEAR, #	20148.	PRT
PLATE THICKNESS, In	1.25	PRT
MODEL RADII, MEAN, In	170.375	PRT
OUTSIDE, In	171.	PRT
INSIDE, In	169.75	PRT
SECTION MODULUS, In <sup>3</sup>	113576.	PRT
VERTICAL JNT FORCE DUE TO MOM., #	1038.541511	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, #	-2302.209529	PRT
COMPONENT IN DIRECTION OF WIND, #	2126.964264	PRT
SUM OF COMPONENTS IN WIND DIRECTION, #	18463.89985	PRT
$\phi_{2r} - \sin\phi_2 \cos\phi_2$ FACTOR	5.768377369	PRT
$\phi_{1r} - \sin\phi_1 \cos\phi_1$ FACTOR	5.100080237	PRT

235.	PRT
315.	PRT
326.25	PRT
303.75	PRT
50.	PRT
307124.	PRT
20148.	PRT
1.25	PRT
170.375	PRT
171.	PRT
169.75	PRT
113576.	PRT
1918.974492	PRT
-1780.848428	PRT
1259.25	PRT
49723.14985	PRT
6.156076451	PRT
5.768377369	PRT

236.	PRT
337.5	PRT
348.75	PRT
326.25	PRT
50.	PRT
307124.	PRT
20148.	PRT
1.25	PRT
170.375	PRT
171.	PRT
169.75	PRT
113576.	PRT
2507.261002	PRT
-1023.132185	PRT
391.5357362	PRT
20114.68559	PRT
6.278177483	PRT
6.156076451	PRT

A.6.7.7 MODEL II: JOINT FORCES DUE TO WIND MOMENT AND SHEAR EAST TO WEST WIND

JOINT NUMBER	287.	PRT	289.	PRT
ANGLES $\theta$	0.	PRT	45.	PRT
$\phi_2$	11.25	PRT	56.25	PRT
$\phi_1$	11.25	PRT	33.75	PRT
ELEVATION, Ft	80.	PRT	80.	PRT
MOMENT, #-ft	570869.	PRT	570869.	PRT
SHEAR, #	28253.	PRT	28253.	PRT
PLATE THICKNESS, In	1.25	PRT	1.25	PRT
MODEL RADII, MEAN, In	164.975	PRT	164.975	PRT
OUTSIDE, In	165.6	PRT	165.6	PRT
INSIDE, In	164.35	PRT	164.35	PRT
SECTION MODULUS, In <sup>3</sup>	106478.	PRT	106478.	PRT
VERTICAL JNT FORCE DUE TO MOM., #	5210.697574	PRT	3684.095325	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, #	0.	PRT	2497.235986	PRT
COMPONENT IN DIRECTION OF WIND, #	0.	PRT	1765.8125	PRT
SUM OF COMPONENTS IN WIND DIRECTION, #	0.	PRT	2314.852561	PRT
$\phi_{2r} - \text{Sin}\phi_2 \text{Cos}\phi_2$ FACTOR	0.	PRT	0.519807938	PRT
$\phi_{1r} - \text{Sin}\phi_1 \text{Cos}\phi_1$ FACTOR	0.	PRT	.1271088563	PRT

JOINT NUMBER	288.	PRT	290.	PRT
ANGLES $\theta$	22.5	PRT	67.5	PRT
$\phi_2$	33.75	PRT	78.75	PRT
$\phi_1$	11.25	PRT	56.25	PRT
ELEVATION, Ft	80.	PRT	80.	PRT
MOMENT, #-ft	570869.	PRT	570869.	PRT
SHEAR, #	28253.	PRT	28253.	PRT
PLATE THICKNESS, In	1.25	PRT	1.25	PRT
MODEL RADII, MEAN, In	164.975	PRT	164.975	PRT
OUTSIDE, In	165.6	PRT	165.6	PRT
INSIDE, In	164.35	PRT	164.35	PRT
SECTION MODULUS, In <sup>3</sup>	106478.	PRT	106478.	PRT
VERTICAL JNT FORCE DUE TO MOM., #	4813.502511	PRT	1993.818023	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, #	1434.710824	PRT	3228.326674	PRT
COMPONENT IN DIRECTION OF WIND, #	549.0400613	PRT	2982.584999	PRT
SUM OF COMPONENTS IN WIND DIRECTION, #	549.0400613	PRT	5297.4375	PRT
$\phi_{2r} - \text{Sin}\phi_2 \text{Cos}\phi_2$ FACTOR	.1271088563	PRT	1.18310507	PRT
$\phi_{1r} - \text{Sin}\phi_1 \text{Cos}\phi_1$ FACTOR	.0050078247	PRT	0.519807938	PRT



A.6.7.8 MODEL II: JOINT FORCES DUE TO WIND MOMENT AND SHEAR EAST TO WEST WIND

JOINT NUMBER ..... 291. PRT  
 ANGLES  $\theta$  ..... 90. PRT  
 $\phi_2$  ..... 101.25 PRT  
 $\phi_1$  ..... 78.75 PRT  
 ELEVATION, Ft ..... 80. PRT  
 MOMENT, #-Ft ..... 570869. PRT  
 SHEAR, # ..... 28253. PRT  
 PLATE THICKNESS, In ..... 1.25 PRT  
 MODEL RADII, MEAN: In ..... 164.975 PRT  
     OUTSIDE, In ..... 165.6 PRT  
     INSIDE, In ..... 164.35 PRT  
 SECTION MODULUS, In<sup>3</sup> ..... 106478. PRT  
 VERTICAL JNT FORCE DUE TO MOM., # ..... 0. PRT  
 TANGENTIAL JNT FORCE DUE TO SHEAR, # ..... 3486.588585 PRT  
 COMPONENT IN DIRECTION OF WIND, # ..... 3486.588585 PRT  
 SUM OF COMPONENTS IN WIND DIRECTION, # ..... 8784.026085 PRT  
 $\phi_{2r}$  -  $\sin\phi_2 \cos\phi_2$  FACTOR ..... 1.958487584 PRT  
 $\phi_{1r}$  -  $\sin\phi_1 \cos\phi_1$  FACTOR ..... 1.18310507 PRT

JOINT NUMBER ..... 294. PRT  
 ANGLES  $\theta$  ..... 112.5 PRT  
 $\phi_2$  ..... 123.75 PRT  
 $\phi_1$  ..... 101.25 PRT  
 ELEVATION, Ft ..... 80. PRT  
 MOMENT, #-Ft ..... 570869. PRT  
 SHEAR, # ..... 28253. PRT  
 PLATE THICKNESS, In ..... 1.25 PRT  
 MODEL RADII, MEAN, In ..... 164.975 PRT  
     OUTSIDE, In ..... 165.6 PRT  
     INSIDE, In ..... 164.35 PRT  
 SECTION MODULUS, In<sup>3</sup> ..... 106478. PRT  
 VERTICAL JNT FORCE DUE TO MOM., # ..... -1993.818023 PRT  
 TANGENTIAL JNT FORCE DUE TO SHEAR, # ..... 3228.326674 PRT  
 COMPONENT IN DIRECTION OF WIND, # ..... 2982.584939 PRT  
 SUM OF COMPONENTS IN WIND DIRECTION # ..... 11766.61102 PRT  
 $\phi_{2r}$  -  $\sin\phi_2 \cos\phi_2$  FACTOR ..... 2.621784716 PRT  
 $\phi_{1r}$  -  $\sin\phi_1 \cos\phi_1$  FACTOR ..... 1.958487584 PRT

A.6.7.9 MODEL II: JOINT FORCES DUE TO WIND MOMENT AND SHEAR EAST TO WEST WIND

JOINT NUMBER	295.	PRT
ANGLES $\theta$	150.	PRT
$\phi_2$	153.75	PRT
$\phi_1$	146.25	PRT
ELEVATION, Ft	80.	PRT
MOMENT, #-Ft	570869.	PRT
SHEAR, #	28253.	PRT
PLATE THICKNESS, In	1.25	PRT
MODEL RADII, MEAN, In	164.975	PRT
OUTSIDE, In	165.6	PRT
INSIDE, In	164.35	PRT
SECTION MODULUS, In <sup>3</sup>	106478.	PRT
VERTICAL JNT FORCE DUE TO MOM., #	-1504.025618	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, #	590.283656	PRT
COMPONENT IN DIRECTION OF WIND, #	295.141828	PRT
SUM OF COMPONENTS IN WIND DIRECTION, #	13827.56535	PRT
$\phi_{2r} - \text{Sin}\phi_2 \text{Cos}\phi_2$ FACTOR	3.080120395	PRT
$\phi_{1r} - \text{Sin}\phi_1 \text{Cos}\phi_1$ FACTOR	3.014483797	PRT

JOINT NUMBER	296.	PRT
ANGLES $\theta$	157.5	PRT
$\phi_2$	161.25	PRT
$\phi_1$	153.75	PRT
ELEVATION, Ft	80.	PRT
MOMENT, #-Ft	570869.	PRT
SHEAR, #	28253.	PRT
PLATE THICKNESS, In	1.25	PRT
MODEL RADII, MEAN, In	164.975	PRT
OUTSIDE, In	165.6	PRT
INSIDE, In	164.35	PRT
SECTION MODULUS, In <sup>3</sup>	106478.	PRT
VERTICAL JNT FORCE DUE TO MOM., #	-1604.500837	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, #	453.6014173	PRT
COMPONENT IN DIRECTION OF WIND, #	173.5857473	PRT
SUM OF COMPONENTS IN WIND DIRECTION, #	14001.1511	PRT
$\phi_{2r} - \text{Sin}\phi_2 \text{Cos}\phi_2$ FACTOR	3.118724133	PRT
$\phi_{1r} - \text{Sin}\phi_1 \text{Cos}\phi_1$ FACTOR	3.080120395	PRT

A.6.7.10 MODEL II: JOINT FORCES DUE TO WIND MOMENT AND SHEAR EAST TO WEST WIND

JOINT NUMBER	299.	PRT	301.	PRT
ANGLES $\theta$	180.	PRT	195.	PRT
$\phi_2$	183.75	PRT	198.75	PRT
$\phi_1$	176.25	PRT	191.25	PRT
ELEVATION, Ft	80.	PRT	80.	PRT
MOMENT, #-FT	570869.	PRT	570869.	PRT
SHEAR, #	28253.	PRT	28253.	PRT
PLATE THICKNESS, In	1.25	PRT	1.25	PRT
MODEL RADII, MEAN, In	164.975	PRT	164.975	PRT
OUTSIDE, In	165.6	PRT	165.6	PRT
INSIDE, In	164.35	PRT	164.35	PRT
SECTION MODULUS, In <sup>3</sup>	106478.	PRT	106478.	PRT
VERTICAL JNT FORCE DUE TO MOM., #	-1736.699191	PRT	-1677.522601	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, #	0.	PRT	-310.3036175	PRT
COMPONENT IN DIRECTION OF WIND, #	0.	PRT	80.31248599	PRT
SUM OF COMPONENTS IN WIND DIRECTION, #	14103.14205	PRT	14205.133	PRT
$\phi_{2r} - \sin\phi_2 \cos\phi_2$ FACTOR	3.141405903	PRT	3.164461174	PRT
$\phi_{1r} - \sin\phi_1 \cos\phi_1$ FACTOR	3.136584829	PRT	3.146600478	PRT

JOINT NUMBER	300.	PRT	302.	PRT
ANGLES $\theta$	187.5	PRT	202.5	PRT
$\phi_2$	191.25	PRT	206.25	PRT
$\phi_1$	183.75	PRT	198.75	PRT
ELEVATION, Ft	80.	PRT	80.	PRT
MOMENT, #-FT	570869.	PRT	570869.	PRT
SHEAR, #	28253.	PRT	28253.	PRT
PLATE THICKNESS, In	1.25	PRT	1.25	PRT
MODEL RADII, MEAN, In	164.975	PRT	164.975	PRT
OUTSIDE, In	165.6	PRT	165.6	PRT
INSIDE, In	164.35	PRT	164.35	PRT
SECTION MODULUS, In <sup>3</sup>	106478.	PRT	106478.	PRT
VERTICAL JNT FORCE DUE TO MOM., #	-1721.841489	PRT	-1604.500897	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, #	-166.0851544	PRT	-453.6014172	PRT
COMPONENT IN DIRECTION OF WIND, #	21.67846279	PRT	173.5857473	PRT
SUM OF COMPONENTS IN WIND DIRECTION, #	14124.82051	PRT	14378.71874	PRT
$\phi_{2r} - \sin\phi_2 \cos\phi_2$ FACTOR	3.146600478	PRT	3.203064912	PRT
$\phi_{1r} - \sin\phi_1 \cos\phi_1$ FACTOR	3.141779404	PRT	3.164461174	PRT

A.6.7.11 MODEL II: JOINT FORCES DUE TO WIND MOMENT AND SHEAR EAST TO WEST WIND-

JOINT NUMBER ..... 303. PRT  
 ANGLES  $\theta$  ..... 210. PRT  
 $\phi_2$  ..... 213.75 PRT  
 $\phi_1$  ..... 206.25 PRT  
 ELEVATION, Ft ..... 80. PRT  
 MOMENT, #-Ft ..... 570869. PRT  
 SHEAR, # ..... 28253. PRT  
 PLATE THICKNESS, In ..... 1.25 PRT  
 MODEL RADII, MEAN, In ..... 164.975 PRT  
   OUTSIDE, In ..... 165.6 PRT  
   INSIDE, In ..... 164.35 PRT  
 SECTION MODULUS, In<sup>3</sup> ..... 106478. PRT  
 VERTICAL JNT FORCE DUE TO MOM., # ..... -1504.025618 PRT  
 TANGENTIAL JNT FORCE DUE TO SHEAR, # ..... -590.283656 PRT  
 COMPONENT IN DIRECTION OF WIND, # ..... 295.141828 PRT  
 SUM OF COMPONENTS IN WIND DIRECTION, # ..... 14673.86057 PRT  
 $\phi_{2r}$  - Sin $\phi_2$  Cos $\phi_2$  FACTOR ..... 3.26870151 PRT  
 $\phi_{1r}$  - Sin $\phi_1$  Cos $\phi_1$  FACTOR ..... 3.203064912 PRT

305. PRT  
 225. PRT  
 236.25 PRT  
 221.25 PRT  
 80. PRT  
 570869. PRT  
 28253. PRT  
 1.25 PRT  
 164.975 PRT  
 165.6 PRT  
 164.35 PRT  
 106478. PRT  
 -2456.06355 PRT  
 -1767.817565 PRT  
 1329.115618 PRT  
 16439.67307 PRT  
 3.661400592 PRT  
 3.365818553 PRT

JOINT NUMBER ..... 304. PRT  
 ANGLES  $\theta$  ..... 217.5 PRT  
 $\phi_2$  ..... 221.25 PRT  
 $\phi_1$  ..... 213.75 PRT  
 ELEVATION, Ft ..... 80. PRT  
 MOMENT, #-Ft ..... 570869. PRT  
 SHEAR, # ..... 28253. PRT  
 PLATE THICKNESS, In ..... 1.25 PRT  
 MODEL RADII, MEAN, In ..... 164.975 PRT  
   OUTSIDE, In ..... 165.6 PRT  
   INSIDE, In ..... 164.35 PRT  
 SECTION MODULUS, In<sup>3</sup> ..... 106478. PRT  
 VERTICAL JNT FORCE DUE TO MOM., # ..... -1377.816104 PRT  
 TANGENTIAL JNT FORCE DUE TO SHEAR, # ..... 717.5530735 PRT  
 COMPONENT IN DIRECTION OF WIND, # ..... 435.8968821 PRT  
 SUM OF COMPONENTS IN WIND DIRECTION, # ..... 15110.55745 PRT  
 $\phi_{2r}$  - Sin $\phi_2$  Cos $\phi_2$  FACTOR ..... 3.365818539 PRT  
 $\phi_{1r}$  - Sin $\phi_1$  Cos $\phi_1$  FACTOR ..... 3.26870151 PRT

306. PRT  
 247.5 PRT  
 258.75 PRT  
 236.25 PRT  
 80. PRT  
 570869. PRT  
 28253. PRT  
 1.25 PRT  
 164.975 PRT  
 165.6 PRT  
 164.35 PRT  
 106478. PRT  
 -1993.818023 PRT  
 -3228.326674 PRT  
 2982.584939 PRT  
 19422.25801 PRT  
 4.324697723 PRT  
 3.661400592 PRT

A.6.7.12 MODEL II: JOINT FORCES DUE TO WIND MOMENT AND SHEAR EAST TO WEST WIND

JOINT NUMBER	307.	*	PRT	309.	PRT
ANGLES $\theta$	270.		PRT	315.	PRT
$\phi_2$	281.25		PRT	326.25	PRT
$\phi_1$	258.75		PRT	303.75	PRT
ELEVATION, Ft	80.		PRT	80.	PRT
MOMENT, #-Ft	570869.		PRT	570869.	PRT
SHEAR, #	28253.		PRT	28253.	PRT
PLATE THICKNESS, In	1.25		PRT	1.25	PRT
MODEL RADII, MEAN, In	164.975		PRT	164.975	PRT
OUTSIDE, In	165.6		PRT	165.6	PRT
INSIDE, In	164.35		PRT	164.35	PRT
SECTION MODULUS, In <sup>3</sup>	106478.		PRT	106478.	PRT
VERTICAL JNT FORCE DUE TO MOM., #	0.		PRT	0.	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, #	-3486.588585		PRT	3484.095325	PRT
COMPONENT IN DIRECTION OF WIND, #	3486.588585		PRT	2497.235986	PRT
SUM OF COMPONENTS IN WIND DIRECTION, #	22908.8466		PRT	1765.8125	PRT
$\phi_{2r} - \sin\phi_2 \cos\phi_2$ FACTOR	5.100080237		PRT	27657.24403	PRT
$\phi_{1r} - \sin\phi_1 \cos\phi_1$ FACTOR	4.324697723		PRT	6.156076451	PRT
			PRT	5.763377369	PRT
JOINT NUMBER	308.		PRT	310.	PRT
ANGLES $\theta$	292.5		PRT	337.5	PRT
$\phi_2$	303.75		PRT	348.75	PRT
$\phi_1$	281.25		PRT	326.25	PRT
ELEVATION, Ft	80.		PRT	80.	PRT
MOMENT, #-Ft	570869.		PRT	570869.	PRT
SHEAR, #	28253.		PRT	28253.	PRT
PLATE THICKNESS, In	1.25		PRT	1.25	PRT
MODEL RADII, MEAN, In	164.975		PRT	164.975	PRT
OUTSIDE, In	165.6		PRT	165.6	PRT
INSIDE, In	164.35		PRT	164.35	PRT
SECTION MODULUS, In <sup>3</sup>	106478.		PRT	106478.	PRT
VERTICAL JNT FORCE DUE TO MOM., #	1993.818023		PRT	4813.502511	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, #	-3228.326674		PRT	-1434.710821	PRT
COMPONENT IN DIRECTION OF WIND, #	2982.584939		PRT	549.0400613	PRT
SUM OF COMPONENTS IN WIND DIRECTION, #	25891.43153		PRT	28206.2841	PRT
$\phi_{2r} - \sin\phi_2 \cos\phi_2$ FACTOR	5.763377369		PRT	6.278177483	PRT
$\phi_{1r} - \sin\phi_1 \cos\phi_1$ FACTOR	5.100080237		PRT	6.156076451	PRT

A.6.7.13 MODEL II: JOINT FORCES DUE TO WIND MOMENT AND SHEAR EAST TO WEST WIND

JOINT NUMBER	334.	PRT	336.	PRT
ANGLES $\theta$	0.	PRT	45.	PRT
$\phi_2$	11.25	PRT	56.25	PRT
$\phi_1$	-11.25	PRT	33.75	PRT
ELEVATION, Ft	120.	PRT	120.	PRT
MOMENT, #-Ft	588478.	PRT	588478.	PRT
SHEAR, #	29136.	PRT	29136.	PRT
PLATE THICKNESS, In	1.25	PRT	1.25	PRT
MODEL RADII, MEAN, In	157.775	PRT	157.775	PRT
OUTSIDE, In	158.4	PRT	158.4	PRT
INSIDE, In	157.15	PRT	157.15	PRT
SECTION MODULUS, In <sup>3</sup>	97370.	PRT	97370.	PRT
VERTICAL JNT FORCE DUE TO MOM., #	5616.870294	PRT	3971.727074	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, #	0.	PRT	2575.282897	PRT
COMPONENT IN DIRECTION OF WIND, #	0.	PRT	1821.	PRT
SUM OF COMPONENTS IN WIND DIRECTION, #	0.	PRT	2387.199385	PRT
$\phi_{2r} - \text{Sin}\phi_2 \text{Cos}\phi_2$ FACTOR	0.	PRT	0.519807938	PRT
$\phi_{1r} - \text{Sin}\phi_1 \text{Cos}\phi_1$ FACTOR	0.	PRT	.1271088563	PRT

JOINT NUMBER	335.	PRT	337.	PRT
ANGLES $\theta$	22.5	PRT	67.5	PRT
$\phi_2$	33.75	PRT	78.75	PRT
$\phi_1$	11.25	PRT	56.25	PRT
ELEVATION, Ft	120.	PRT	120.	PRT
MOMENT, #-Ft	588478.	PRT	588478.	PRT
SHEAR, #	29136.	PRT	29136.	PRT
PLATE THICKNESS, In	1.25	PRT	1.25	PRT
MODEL RADII, MEAN, In	157.775	PRT	157.775	PRT
OUTSIDE, In	158.4	PRT	158.4	PRT
INSIDE, In	157.15	PRT	157.15	PRT
SECTION MODULUS, In <sup>3</sup>	97370.	PRT	97370.	PRT
VERTICAL JNT FORCE DUE TO MOM., #	5189.311501	PRT	3149.483283	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, #	1479.550295	PRT	3329.222595	PRT
COMPONENT IN DIRECTION OF WIND, #	566.199385	PRT	3075.800615	PRT
SUM OF COMPONENTS IN WIND DIRECTION, #	566.199385	PRT	5463.	PRT
$\phi_{2r} - \text{Sin}\phi_2 \text{Cos}\phi_2$ FACTOR	.1271088563	PRT	1.18310507	PRT
$\phi_{1r} - \text{Sin}\phi_1 \text{Cos}\phi_1$ FACTOR	.005007824	PRT	0.519807938	PRT

A.6.7.14 MODEL II: JOINT FORCES DUE TO WIND MOMENT AND SHEAR EAST TO WEST WIND

JOINT NUMBER	338.	PRT	340.	PRT
ANGLES $\theta$	90.	PRT	135.	PRT
$\phi_2$	101.25	PRT	146.25	PRT
$\phi_1$	78.75	PRT	123.75	PRT
ELEVATION, Ft	120.	PRT	120.	PRT
MOMENT, #-Ft	588478.	PRT	588478.	PRT
SHEAR, #	29136.	PRT	29136.	PRT
PLATE THICKNESS, In	1.25	PRT	1.25	PRT
MODEL RADII, MEAN, In	157.775	PRT	157.775	PRT
OUTSIDE, In	158.4	PRT	158.4	PRT
INSIDE, In	157.15	PRT	157.15	PRT
SECTION MODULUS, In <sup>3</sup>	97370.	PRT	97370.	PRT
VERTICAL JNT FORCE DUE TO MOM., #	0.	PRT	-3971.727074	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, #	3595.556048	PRT	3575.282897	PRT
COMPONENT IN DIRECTION OF WIND, #	3595.556048	PRT	1821.	PRT
SUM OF COMPONENTS IN WIND DIRECTION, #	9058.556048	PRT	13955.356666	PRT
$\phi_{2r} - \text{Sin}\phi_2 \text{Cos}\phi_2$ FACTOR	1.958487584	PRT	3.014483797	PRT
$\phi_{1r} - \text{Sin}\phi_1 \text{Cos}\phi_1$ FACTOR	1.18310507	PRT	2.621784716	PRT

JOINT NUMBER	339.	PRT	341.	PRT
ANGLES $\theta$	112.5	PRT	157.5	PRT
$\phi_2$	123.75	PRT	168.75	PRT
$\phi_1$	101.25	PRT	146.25	PRT
ELEVATION, Ft	120.	PRT	120.	PRT
MOMENT, #-Ft	588478.	PRT	588478.	PRT
SHEAR, #	29136.	PRT	29136.	PRT
PLATE THICKNESS, In	1.25	PRT	1.25	PRT
MODEL RADII, MEAN, In	157.775	PRT	157.775	PRT
OUTSIDE, In	158.4	PRT	158.4	PRT
INSIDE, In	157.15	PRT	157.15	PRT
SECTION MODULUS, In <sup>3</sup>	97370.	PRT	97370.	PRT
VERTICAL JNT FORCE DUE TO MOM., #	-2149.483203	PRT	-5189.311501	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, #	3329.222595	PRT	1479.550299	PRT
COMPONENT IN DIRECTION OF WIND, #	3075.800615	PRT	566.1993857	PRT
SUM OF COMPONENTS IN WIND DIRECTION #	12134.35666	PRT	14521.55605	PRT
$\phi_{2r} - \text{Sin}\phi_2 \text{Cos}\phi_2$ FACTOR	2.621784716	PRT	3.136584829	PRT
$\phi_{1r} - \text{Sin}\phi_1 \text{Cos}\phi_1$ FACTOR	1.958487584	PRT	3.014483797	PRT

A.6.7.15 MODEL II: JOINT FORCES DUE TO WIND MOMENT AND SHEAR EAST TO WEST WIND

JOINT NUMBER	342.	PRT	344.	PRT
ANGLES $\theta$	180.	PRT	225.	PRT
$\phi_2$	191.25	PRT	236.25	PRT
$\phi_1$	168.75	PRT	213.75	PRT
ELEVATION, Ft	120.	PRT	120.	PRT
MOMENT, #-Ft	588478.	PRT	588478.	PRT
SHEAR, #	29136.	PRT	29136.	PRT
PLATE THICKNESS, In	1.25	PRT	1.25	PRT
MODEL RADII, MEAN, In	157.775	PRT	157.775	PRT
OUTSIDE, In	158.4	PRT	158.4	PRT
INSIDE, In	157.15	PRT	157.15	PRT
SECTION MODULUS, In <sup>3</sup>	97370.	PRT	97370.	PRT
VERTICAL JNT FORCE DUE TO MOM., #	-5616.870294	PRT	-3971.727074	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, #	0.	PRT	-2575.282897	PRT
COMPONENT IN DIRECTION OF WIND, #	0.	PRT	1821.	PRT
SUM OF COMPONENTS IN WIND DIRECTION, #	14521.55605	PRT	16908.75543	PRT
$\phi_{2r} - \sin\phi_2 \cos\phi_2$ FACTOR	3.136584829	PRT	3.661400592	PRT
$\phi_{1r} - \sin\phi_1 \cos\phi_1$ FACTOR	3.014483797	PRT	3.26870151	PRT

JOINT NUMBER	343.	PRT	345.	PRT
ANGLES $\theta$	202.5	PRT	247.5	PRT
$\phi_2$	213.75	PRT	258.75	PRT
$\phi_1$	191.25	PRT	236.25	PRT
ELEVATION, Ft	120.	PRT	120.	PRT
MOMENT, #-Ft	588478.	PRT	588478.	PRT
SHEAR, #	29136.	PRT	29136.	PRT
PLATE THICKNESS, In	1.25	PRT	1.25	PRT
MODEL RADII, MEAN, In	157.775	PRT	157.775	PRT
OUTSIDE, In	158.4	PRT	158.4	PRT
INSIDE, In	157.15	PRT	157.15	PRT
SECTION MODULUS, In <sup>3</sup>	97370.	PRT	97370.	PRT
VERTICAL JNT FORCE DUE TO MOM., #	-5189.311501	PRT	-2149.483203	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, #	-1479.550295	PRT	-3329.222595	PRT
COMPONENT IN DIRECTION OF WIND, #	566.1993851	PRT	3075.800615	PRT
SUM OF COMPONENTS IN WIND DIRECTION, #	15087.75543	PRT	19484.55605	PRT
$\phi_{2r} - \sin\phi_2 \cos\phi_2$ FACTOR	3.26870151	PRT	4.324697223	PRT
$\phi_{1r} - \sin\phi_1 \cos\phi_1$ FACTOR	3.146600478	PRT	3.661400592	PRT



A.6.7.16 MODEL II: JOINT FORCES DUE TO WIND MOMENT AND SHEAR EAST TO WEST WIND

JOINT NUMBER	346.	PRT	PRT
ANGLES $\theta$	270.	PRT	PRT
$\phi_2$	381.25	PRT	PRT
$\phi_1$	258.75	PRT	PRT
ELEVATION, Ft	120.	PRT	PRT
MOMENT, #-Ft	588478.	PRT	PRT
SHEAR, #	29136.	PRT	PRT
PLATE THICKNESS, In	1.25	PRT	PRT
MODEL RADII, MEAN, In	157.775	PRT	PRT
OUTSIDE, In	158.4	PRT	PRT
INSIDE, In	157.15	PRT	PRT
SECTION MODULUS, In <sup>3</sup>	97370.	PRT	PRT
VERTICAL JNT FORCE DUE TO MOM., #	0.	PRT	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, #	-3595.556048	PRT	PRT
COMPONENT IN DIRECTION OF WIND, #	3595.556048	PRT	PRT
SUM OF COMPONENTS IN WIND DIRECTION, #	23580.11209	PRT	PRT
$\phi_{2r} - \sin\phi_2 \cos\phi_2$ FACTOR	5.100080237	PRT	PRT
$\phi_{1r} - \sin\phi_1 \cos\phi_1$ FACTOR	4.324697723	PRT	PRT

JOINT NUMBER	347.	PRT	PRT
ANGLES $\theta$	292.5	PRT	PRT
$\phi_2$	303.75	PRT	PRT
$\phi_1$	281.25	PRT	PRT
ELEVATION, Ft	120.	PRT	PRT
MOMENT, #-Ft	588478.	PRT	PRT
SHEAR, #	29136.	PRT	PRT
PLATE THICKNESS, In	1.25	PRT	PRT
MODEL RADII, MEAN, In	157.775	PRT	PRT
OUTSIDE, In	158.4	PRT	PRT
INSIDE, In	157.15	PRT	PRT
SECTION MODULUS, In <sup>3</sup>	97370.	PRT	PRT
VERTICAL JNT FORCE DUE TO MOM., #	3149.483203	PRT	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, #	-3329.222595	PRT	PRT
COMPONENT IN DIRECTION OF WIND, #	3075.800615	PRT	PRT
SUM OF COMPONENTS IN WIND DIRECTION, #	26655.91277	PRT	PRT
$\phi_{2r} - \sin\phi_2 \cos\phi_2$ FACTOR	5.763377369	PRT	PRT
$\phi_{1r} - \sin\phi_1 \cos\phi_1$ FACTOR	5.100080237	PRT	PRT

A.6.7.17 MODEL II: JOINT FORCES DUE TO WIND MOMENT AND SHEAR EAST TO WEST WIND

JOINT NUMBER ..... 366. PRT  
 ANGLES  $\theta$  ..... 0. PRT  
 $\phi_2$  ..... 11.25 PRT  
 $\phi_1$  ..... -11.25 PRT  
 ELEVATION, Ft ..... 160. PRT  
 MOMENT, #-Ft ..... 580538. PRT  
 SHEAR, # ..... 28796. PRT  
 PLATE THICKNESS, In ..... 1.25 PRT  
 MODEL RADII, MEAN, In ..... 150.575 PRT  
                   OUTSIDE, In ..... 151.2 PRT  
                   INSIDE, In ..... 149.95 PRT  
 SECTION MODULUS, In<sup>3</sup> ..... 88669. PRT  
 VERTICAL JNT FORCE DUE TO MOM., # ..... 5807.147518 PRT  
 TANGENTIAL JNT FORCE DUE TO SHEAR, # ..... 0. PRT  
 COMPONENT IN DIRECTION OF WIND, # ..... 0. PRT  
 SUM OF COMPONENTS IN WIND DIRECTION, # ..... 0. PRT  
 $\phi_{2r} - \sin\phi_2 \cos\phi_2$  FACTOR ..... 0. PRT  
 $\phi_{1r} - \sin\phi_1 \cos\phi_1$  FACTOR ..... 0. PRT

368. PRT  
 45. PRT  
 56.25 PRT  
 33.75 PRT  
 160. PRT  
 580538. PRT  
 28796. PRT  
 1.25 PRT  
 150.575 PRT  
 151.2 PRT  
 149.95 PRT  
 88669. PRT  
 4106.273389 PRT  
 2545.230859 PRT  
 1799.75 PRT  
 2359.342171 PRT  
 0.519807938 PRT  
 .1271088563 PRT

JOINT NUMBER ..... 369. PRT  
 ANGLES  $\theta$  ..... 22.5 PRT  
 $\phi_2$  ..... 33.75 PRT  
 $\phi_1$  ..... 11.25 PRT  
 ELEVATION, Ft ..... 160. PRT  
 MOMENT, #-Ft ..... 580538. PRT  
 SHEAR, # ..... 28796. PRT  
 PLATE THICKNESS, In ..... 1.25 PRT  
 MODEL RADII, MEAN, In ..... 150.575 PRT  
                   OUTSIDE, In ..... 151.2 PRT  
                   INSIDE, In ..... 149.95 PRT  
 SECTION MODULUS, In<sup>3</sup> ..... 88669. PRT  
 VERTICAL JNT FORCE DUE TO MOM., # ..... 5365.104734 PRT  
 TANGENTIAL JNT FORCE DUE TO SHEAR, # ..... 1462.284812 PRT  
 COMPONENT IN DIRECTION OF WIND, # ..... 559.592171 PRT  
 SUM OF COMPONENTS IN WIND DIRECTION, # ..... 559.592171 PRT  
 $\phi_{2r} - \sin\phi_2 \cos\phi_2$  FACTOR ..... .1271088563 PRT  
 $\phi_{1r} - \sin\phi_1 \cos\phi_1$  FACTOR ..... .005007824 PRT

369. PRT  
 67.5 PRT  
 78.75 PRT  
 56.25 PRT  
 160. PRT  
 580538. PRT  
 28796. PRT  
 1.25 PRT  
 150.575 PRT  
 151.2 PRT  
 149.95 PRT  
 88669. PRT  
 2222.299144 PRT  
 3290.372524 PRT  
 3039.907823 PRT  
 5399.25 PRT  
 1.18310507 PRT  
 .0519807938 PRT

A.6.7.18 MODEL II: JOINT FORCES DUE TO WIND MOMENT AND SHEAR EAST TO WEST WIND

JOINT NUMBER	370.	PRT	372.	PRT
ANGLES $\theta$	90.	PRT	135.	PRT
$\phi_2$	101.25	PRT	146.25	PRT
$\phi_1$	78.75	PRT	123.75	PRT
ELEVATION, Ft	160.	PRT	160.	PRT
MOMENT, #-Ft	580538.	PRT	580538.	PRT
SHEAR, #	28796.	PRT	28796.	PRT
PLATE THICKNESS, In	1.25	PRT	1.25	PRT
MODEL RADII, MEAN, In	150.575	PRT	150.575	PRT
OUTSIDE, In	151.2	PRT	151.2	PRT
INSIDE, In	149.95	PRT	149.95	PRT
SECTION MODULUS, In <sup>3</sup>	88669.	PRT	88669.	PRT
VERTICAL JNT FORCE DUE TO MOM., #	0.	PRT	-4106.273389	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, #	3553.598021	PRT	2545.230854	PRT
COMPONENT IN DIRECTION OF WIND, #	3553.598021	PRT	1799.75	PRT
SUM OF COMPONENTS IN WIND DIRECTION, #	8952.848021	PRT	13792.50585	PRT
$\phi_{2r} - \sin\phi_2 \cos\phi_2$ FACTOR	1.958487584	PRT	3.014483797	PRT
$\phi_{1r} - \sin\phi_1 \cos\phi_1$ FACTOR	1.18310507	PRT	2.621784716	PRT

JOINT NUMBER	371.	PRT	373.	PRT
ANGLES $\theta$	112.5	PRT	157.5	PRT
$\phi_2$	123.75	PRT	168.75	PRT
$\phi_1$	101.25	PRT	146.25	PRT
ELEVATION, Ft	160.	PRT	160.	PRT
MOMENT, #-Ft	580538.	PRT	580538.	PRT
SHEAR, #	28796.	PRT	28796.	PRT
PLATE THICKNESS, In	1.25	PRT	1.25	PRT
MODEL RADII, MEAN, In	150.575	PRT	150.575	PRT
OUTSIDE, In	151.2	PRT	151.2	PRT
INSIDE, In	149.95	PRT	149.95	PRT
SECTION MODULUS, In <sup>3</sup>	88669.	PRT	88669.	PRT
VERTICAL JNT FORCE DUE TO MOM., #	-2223.20944	PRT	-5365.104734	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, #	3290.372324	PRT	1462.284812	PRT
COMPONENT IN DIRECTION OF WIND, #	3039.907829	PRT	559.592171	PRT
SUM OF COMPONENTS IN WIND DIRECTION, #	11992.75585	PRT	14352.09802	PRT
$\phi_{2r} - \sin\phi_2 \cos\phi_2$ FACTOR	2.621784716	PRT	3.136584829	PRT
$\phi_{1r} - \sin\phi_1 \cos\phi_1$ FACTOR	1.958487584	PRT	3.014483797	PRT

A.6.7.19 MODEL II: JOINT FORCES DUE TO WIND MOMENT AND SHEAR EAST TO WEST WIND

JOINT NUMBER	374.	PRT	376.	PRT
ANGLES $\theta$	180.	PRT	225.	PRT
$\phi_2$	191.25	PRT	236.25	PRT
$\phi_1$	168.75	PRT	213.75	PRT
ELEVATION, Ft	160.	PRT	160.	PRT
MOMENT, #-Ft	580538.	PRT	580538.	PRT
SHEAR, #	28796.	PRT	28796.	PRT
PLATE THICKNESS, In	1.25	PRT	1.25	PRT
MODEL RADII, MEAN, In	150.575	PRT	150.575	PRT
OUTSIDE, In	151.2	PRT	151.2	PRT
INSIDE, In	149.95	PRT	149.95	PRT
SECTION MODULUS, In <sup>3</sup>	88669.	PRT	88669.	PRT
VERTICAL JNT FORCE DUE TO MOM., #	-5807.147518	PRT	-4106.273389	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, #	0.	PRT	-2545.130859	PRT
COMPONENT IN DIRECTION OF WIND, #	0.	PRT	1799.75	PRT
SUM OF COMPONENTS IN WIND DIRECTION, #	14352.09802	PRT	16711.44019	PRT
$\phi_{2r} - \sin\phi_2 \cos\phi_2$ FACTOR	3.136584829	PRT	3.661400592	PRT
$\phi_{1r} - \sin\phi_1 \cos\phi_1$ FACTOR	3.014483797	PRT	3.26870151	PRT

JOINT NUMBER	375.	PRT	377.	PRT
ANGLES $\theta$	202.5	PRT	247.5	PRT
$\phi_2$	213.75	PRT	258.75	PRT
$\phi_1$	191.25	PRT	236.25	PRT
ELEVATION, Ft	160.	PRT	160.	PRT
MOMENT, #-Ft	580538.	PRT	580538.	PRT
SHEAR, #	28796.	PRT	28796.	PRT
PLATE THICKNESS, In	1.25	PRT	1.25	PRT
MODEL RADII, MEAN, In	150.575	PRT	150.575	PRT
OUTSIDE, In	151.2	PRT	151.2	PRT
INSIDE, In	149.95	PRT	149.95	PRT
SECTION MODULUS, In <sup>3</sup>	88669.	PRT	88669.	PRT
VERTICAL JNT FORCE DUE TO MOM., #	-5365.104731	PRT	-2222.299144	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, #	-1462.284812	PRT	-3290.372524	PRT
COMPONENT IN DIRECTION OF WIND, #	559.5921704	PRT	3039.907829	PRT
SUM OF COMPONENTS IN WIND DIRECTION, #	14911.69019	PRT	19751.34802	PRT
$\phi_{2r} - \sin\phi_2 \cos\phi_2$ FACTOR	3.26870151	PRT	4.324697723	PRT
$\phi_{1r} - \sin\phi_1 \cos\phi_1$ FACTOR	3.146600478	PRT	3.661400592	PRT

A.6.7.20 MODEL II: JOINT FORCES DUE TO WIND MOMENT AND SHEAR EAST TO WEST WIND

JOINT NUMBER	378.	PRT	380.	PRT
ANGLES $\theta$	270.	PRT	315.	PRT
$\phi_2$	281.25	PRT	326.25	PRT
$\phi_1$	258.75	PRT	303.75	PRT
ELEVATION, Ft	160.	PRT	160.	PRT
MOMENT, #-Ft	580538.	PRT	580538.	PRT
SHEAR, #	28796.	PRT	28796.	PRT
PLATE THICKNESS, In	1.25	PRT	1.25	PRT
MODEL RADII, MEAN, In	150.575	PRT	150.575	PRT
OUTSIDE, In	151.2	PRT	151.2	PRT
INSIDE, In	149.95	PRT	149.95	PRT
SECTION MODULUS, In <sup>3</sup>	88669.	PRT	88669.	PRT
VERTICAL JNT FORCE DUE TO MOM., #	0.	PRT	4106.273389	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, #	-3553.598021	PRT	-2545.230859	PRT
COMPONENT IN DIRECTION OF WIND, #	3553.598021	PRT	1799.75	PRT
SUM OF COMPONENTS IN WIND DIRECTION, #	23304.94604	PRT	28144.60387	PRT
$\phi_{2r} - \sin\phi_2 \cos\phi_2$ FACTOR	5.100080237	PRT	6.156076451	PRT
$\phi_{1r} - \sin\phi_1 \cos\phi_1$ FACTOR	4.324697723	PRT	5.763377369	PRT

JOINT NUMBER	379.	PRT	381.	PRT
ANGLES $\theta$	292.5	PRT	337.5	PRT
$\phi_2$	303.75	PRT	348.75	PRT
$\phi_1$	281.25	PRT	326.25	PRT
ELEVATION, Ft	160.	PRT	160.	PRT
MOMENT, #-Ft	580538	PRT	580538.	PRT
SHEAR, #	28796.	PRT	28796.	PRT
PLATE THICKNESS, In	1.25	PRT	1.25	PRT
MODEL RADII, MEAN, In	150.575	PRT	150.575	PRT
OUTSIDE, In	151.2	PRT	151.2	PRT
INSIDE, In	149.95	PRT	149.95	PRT
SECTION MODULUS, In <sup>3</sup>	88669.	PRT	88669.	PRT
VERTICAL JNT FORCE DUE TO MOM., #	2222.299144	PRT	5365.104734	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, #	-3290.372524	PRT	-1462.284812	PRT
COMPONENT IN DIRECTION OF WIND, #	3039.907829	PRT	559.592171	PRT
SUM OF COMPONENTS IN WIND DIRECTION #	26344.85387	PRT	28704.19604	PRT
$\phi_{2r} - \sin\phi_2 \cos\phi_2$ FACTOR	5.763377369	PRT	6.278172483	PRT
$\phi_{1r} - \sin\phi_1 \cos\phi_1$ FACTOR	5.100080237	PRT	6.156076451	PRT

A.6.7.21 MODEL II: JOINT FORCES DUE TO WIND MOMENT AND SHEAR EAST TO WEST WIND

JOINT NUMBER	398.	PRT	400.	PRT
ANGLES $\theta$	0.	PRT	45.	PRT
$\phi_2$	11.25	PRT	56.25	PRT
$\phi_1$	-11.25	PRT	33.75	PRT
ELEVATION, Ft	200.	PRT	200.	PRT
MOMENT, #-Ft	28144824.	PRT	28144824.	PRT
SHEAR, #	193745.	PRT	193745.	PRT
PLATE THICKNESS, in	1.25	PRT	1.25	PRT
MODEL RADII, MEAN, In	143.375	PRT	143.375	PRT
OUTSIDE, In	144.	PRT	144.	PRT
INSIDE, In	142.75	PRT	142.75	PRT
SECTION MODULUS, in <sup>3</sup>	80376.	PRT	80376.	PRT
VERTICAL JNT FORCE DUE TO MOM., #	295730.9127	PRT	209113.3338	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, #	0.	PRT	17124.80041	PRT
COMPONENT IN DIRECTION OF WIND, #	0.	PRT	12109.0625	PRT
SUM OF COMPONENTS IN WIND DIRECTION, #	0.	PRT	15874.10574	PRT
$\phi_{2r} - \sin\phi_2 \cos\phi_2$ FACTOR	0.	PRT	0.519807938	PRT
$\phi_{1r} - \sin\phi_1 \cos\phi_1$ FACTOR	0.	PRT	.1271088563	PRT

JOINT NUMBER	399.	PRT	401.	PRT
ANGLES $\theta$	22.5	PRT	67.5	PRT
$\phi_2$	33.75	PRT	78.75	PRT
$\phi_1$	11.25	PRT	56.25	PRT
ELEVATION, Ft	200.	PRT	200.	PRT
MOMENT, #-Ft	28144824.	PRT	28144824.	PRT
SHEAR, #	193745.	PRT	193745.	PRT
PLATE THICKNESS, in	1.25	PRT	1.25	PRT
MODEL RADII, MEAN, In	143.375	PRT	143.375	PRT
OUTSIDE, In	144.	PRT	144.	PRT
INSIDE, In	142.75	PRT	142.75	PRT
SECTION MODULUS, in <sup>3</sup>	80376.	PRT	80376.	PRT
VERTICAL JNT FORCE DUE TO MOM., #	273219.7374	PRT	113171.3207	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, #	9838.582119	PRT	22138.25617	PRT
COMPONENT IN DIRECTION OF WIND, #	3765.043241	PRT	20453.08176	PRT
SUM OF COMPONENTS IN WIND DIRECTION #	3765.043241	PRT	36327.1875	PRT
$\phi_{2r} - \sin\phi_2 \cos\phi_2$ FACTOR	.1271088563	PRT	1.18310507	PRT
$\phi_{1r} - \sin\phi_1 \cos\phi_1$ FACTOR	.0050078247	PRT	0.519807938	PRT

A.6.7.22 MODEL II: JOINT FORCES DUE TO WIND MOMENT AND SHEAR EAST TO WEST WIND

JOINT NUMBER	402.	PRT	404.	PRT
ANGLES $\theta$	90.	PRT	135.	PRT
$\phi_2$	101.25	PRT	146.25	PRT
$\phi_1$	78.75	PRT	123.75	PRT
ELEVATION, Ft	200.	PRT	200.	PRT
MOMENT, #-Ft	28144824.	PRT	28144824.	PRT
SHEAR, #	193745.	PRT	193745.	PRT
PLATE THICKNESS, In	1.25	PRT	1.25	PRT
MODEL RADII, MEAN, In	143.375	PRT	143.375	PRT
OUTSIDE, In	144.	PRT	144.	PRT
INSIDE, In	142.75	PRT	142.75	PRT
SECTION MODULUS, In <sup>3</sup>	80376.	PRT	80376.	PRT
VERTICAL JNT FORCE DUE TO MOM., #	0.	PRT	-209113.3338	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, #	23909.2877	PRT	17124.80041	PRT
COMPONENT IN DIRECTION OF WIND, #	23909.2877	PRT	12109.0625	PRT
SUM OF COMPONENTS IN WIND DIRECTION, #	60236.4752	PRT	92798.61946	PRT
$\phi_{2r} - \sin\phi_2 \cos\phi_2$ FACTOR	1.9558487584	PRT	3.014483797	PRT
$\phi_{1r} - \sin\phi_1 \cos\phi_1$ FACTOR	1.18310507	PRT	2.621784716	PRT

JOINT NUMBER	403.	PRT	405.	PRT
ANGLES $\theta$	112.5	PRT	157.5	PRT
$\phi_2$	123.75	PRT	168.75	PRT
$\phi_1$	101.25	PRT	146.25	PRT
ELEVATION, Ft	200.	PRT	200.	PRT
MOMENT, #-Ft	28144824.	PRT	28144824.	PRT
SHEAR, #	193745.	PRT	193745.	PRT
PLATE THICKNESS, In	1.25	PRT	1.25	PRT
MODEL RADII, MEAN, In	143.375	PRT	143.375	PRT
OUTSIDE, In	144.	PRT	144.	PRT
INSIDE, In	142.75	PRT	142.75	PRT
SECTION MODULUS, In <sup>3</sup>	80376.	PRT	80376.	PRT
VERTICAL JNT FORCE DUE TO MOM., #	-113171.3203	PRT	-273219.7374	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, #	22138.25617	PRT	9838.532119	PRT
COMPONENT IN DIRECTION OF WIND, #	20453.08176	PRT	3765.043241	PRT
SUM OF COMPONENTS IN WIND DIRECTION #	80689.55696	PRT	96563.6627	PRT
$\phi_{2r} - \sin\phi_2 \cos\phi_2$ FACTOR	2.621784716	PRT	3.136584829	PRT
$\phi_{1r} - \sin\phi_1 \cos\phi_1$ FACTOR	1.9558487584	PRT	3.014483797	PRT

A. 6.7-23 MODEL II: JOINT FORCES DUE TO WIND MOMENT AND SHEAR EAST TO WEST WIND

JOINT NUMBER	406.	PRT	408.	PRT
ANGLES $\theta$	180.	PRT	225.	PRT
$\phi_2$	191.25	PRT	236.25	PRT
$\phi_1$	168.75	PRT	213.75	PRT
ELEVATION, Ft	200.	PRT	200.	PRT
MOMENT, #-Ft	28144824.	PRT	28144824.	PRT
SHEAR, #	193745.	PRT	193745.	PRT
PLATE THICKNESS, In	1.25	PRT	1.25	PRT
MODEL RADII, MEAN, In	143.375	PRT	143.375	PRT
OUTSIDE, In	144.	PRT	144.	PRT
INSIDE, In	142.75	PRT	142.75	PRT
SECTION MODULUS, In <sup>3</sup>	80376.	PRT	80376.	PRT
VERTICAL JNT FORCE DUE TO MOM., #	-295730.9127	PRT	-209113.3338	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, #	0.	PRT	-17124.80041	PRT
COMPONENT IN DIRECTION OF WIND, #	0.	PRT	12109.0625	PRT
SUM OF COMPONENTS IN WIND DIRECTION, #	96563.6627	PRT	112437.7684	PRT
$\phi_{2r} - \sin\phi_2 \cos\phi_2$ FACTOR	3.136584829	PRT	3.661400592	PRT
$\phi_{1r} - \sin\phi_1 \cos\phi_1$ FACTOR	3.014483797	PRT	3.26870151	PRT

JOINT NUMBER	407.	PRT	409.	PRT
ANGLES $\theta$	202.5	PRT	247.5	PRT
$\phi_2$	213.75	PRT	258.75	PRT
$\phi_1$	191.25	PRT	236.25	PRT
ELEVATION, Ft	200.	PRT	200.	PRT
MOMENT, #-Ft	28144824.	PRT	28144824.	PRT
SHEAR, #	193745.	PRT	193745.	PRT
PLATE THICKNESS, In	1.25	PRT	1.25	PRT
MODEL RADII, MEAN, In	143.375	PRT	143.375	PRT
OUTSIDE, In	144.	PRT	144.	PRT
INSIDE, In	142.75	PRT	142.75	PRT
SECTION MODULUS, In <sup>3</sup>	80376.	PRT	80376.	PRT
VERTICAL JNT FORCE DUE TO MOM., #	-273219.7374	PRT	-113171.3207	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, #	-9838.532119	PRT	-2138.25617	PRT
COMPONENT IN DIRECTION OF WIND, #	3765.043241	PRT	20453.08176	PRT
SUM OF COMPONENTS IN WIND DIRECTION #	100328.7059	PRT	132890.8502	PRT
$\phi_{2r} - \sin\phi_2 \cos\phi_2$ FACTOR	3.26870151	PRT	4.324697723	PRT
$\phi_{1r} - \sin\phi_1 \cos\phi_1$ FACTOR	3.146600478	PRT	3.661400592	PRT



A.6.7.24 MODEL II: JOINT FORCES DUE TO WIND MOMENT AND SHEAR EAST TO WEST WIND

JOINT NUMBER	410.	PRT	PRT
ANGLES $\theta$	270.	PRT	PRT
$\phi_2$	281.25	PRT	PRT
$\phi_1$	258.75	PRT	PRT
ELEVATION, Ft	200.	PRT	PRT
MOMENT, #-Ft	28144824.	PRT	PRT
SHEAR, #	193745.	PRT	PRT
PLATE THICKNESS, In	1.25	PRT	PRT
MODEL RADII, MEAN, In	143.375	PRT	PRT
OUTSIDE, In	144.	PRT	PRT
INSIDE, In	142.75	PRT	PRT
SECTION MODULUS, in <sup>3</sup>	80376.	PRT	PRT
VERTICAL JNT FORCE DUE TO MOM., #	0.	PRT	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, #	-23909.2877	PRT	PRT
COMPONENT IN DIRECTION OF WIND, #	23909.2877	PRT	PRT
SUM OF COMPONENTS IN WIND DIRECTION, #	156800.1379	PRT	PRT
$\phi_{2r} - \sin\phi_2 \cos\phi_2$ FACTOR	5.100080237	PRT	PRT
$\phi_{1r} - \sin\phi_1 \cos\phi_1$ FACTOR	4.324697723	PRT	PRT

JOINT NUMBER	411.	PRT	PRT
ANGLES $\theta$	292.5	PRT	PRT
$\phi_2$	303.75	PRT	PRT
$\phi_1$	281.25	PRT	PRT
ELEVATION, Ft	200.	PRT	PRT
MOMENT, #-Ft	28144824.	PRT	PRT
SHEAR, #	193745.	PRT	PRT
PLATE THICKNESS, In	1.25	PRT	PRT
MODEL RADII, MEAN, In	143.375	PRT	PRT
OUTSIDE, In	144.	PRT	PRT
INSIDE, In	142.75	PRT	PRT
SECTION MODULUS, in <sup>3</sup>	80376.	PRT	PRT
VERTICAL JNT FORCE DUE TO MOM., #	113171.3207	PRT	PRT
TANGENTIAL JNT FORCE DUE TO SHEAR, #	-22138.2561	PRT	PRT
COMPONENT IN DIRECTION OF WIND, #	20453.08176	PRT	PRT
SUM OF COMPONENTS IN WIND DIRECTION, #	177253.2197	PRT	PRT
$\phi_{2r} - \sin\phi_2 \cos\phi_2$ FACTOR	5.769277369	PRT	PRT
$\phi_{1r} - \sin\phi_1 \cos\phi_1$ FACTOR	5.100080237	PRT	PRT

APPENDIX B

B.1: MODEL I ELEMENT STRESSES FOR LOAD CASE 1,2,3,1+2,  
AND 1+3 FOR ELEMENTS ABOUT AND ADJACENT TO THE BREACH.

The following Tables reflect the analysis stresses due  
to the three applied load cases as well as two load combina-  
tions with Wind  $q = 9.2$  PSF.

B.1.1

50 FT TALL BREACHED STEEL STACK MODEL CENTER OF ELEMENT STRESSES

ELEMENT NO	STRESS DIRECTION	39.1	39.2	40.1	40.2	41.1	41.2
TOP SURFACE STRESSES	LC1	-295.	-2556.	243.	-2020.	92.	-2108.
	LC2	7179.	-1505.	148.	-1116.	36.	-1216.
	LC3	79.	799.	-110.	722.	-125.	735.
	LC4	-1941.8	-16402.	1604.6	-12287.2	423.2	-13295.2
	LC5	431.8	4794.8	-769.	4622.4	-1058.	4654.
BOTTOM SURFACE STRESSES	LC1	-346.	-2319.	-92.	-2200.	-178.	-2091.
	LC2	-205.	-1346.	-49.	-1217.	-107.	-1201.
	LC3	106.	727.	67.	776.	148.	738.
	LC4	-2332.	-14702.2	-542.8	-13396.4	-1162.4	-13140.2
	LC5	629.2	4369.4	524.4	4939.2	1183.6	4698.6
MEMBRANE STRESSES	LC1	-320.5	-2437.5	75.9	-2110.	-43.	-2099.5
	LC4	-2086.9	-15552.1	530.9	-12841.8	-269.6	-13217.7
	LC5	530.5	4582.1	-122.3	4780.8	62.8	4676.3
	LC1	25.5	-118.5	167.5	90.	135.	-8.5
	LC4	145.1	-849.9	1073.7	554.6	792.8	-77.5
TOP BENDING STRESSES	LC5	-98.7	212.7	-646.7	-158.4	-1120.8	-22.3
	LC1	-25.5	118.5	-167.5	-90.	-135.	8.5
	LC4	-145.1	849.9	-1073.7	-554.6	-792.8	77.5
	LC5	98.7	-212.7	646.7	158.4	1120.8	-22.3

Note: 1 = Local element x stresses, 2 = Local element y stresses  
 LC = Load case, LC1 = Dead load, LC2 = E-W Wind, q=1.0PSF,  
 LC3 = N+S Wind, q=1.0PSF, LC4 = LC1 + 9.2LC2, LC5 = LC1 + 9.2LC3  
 Top Surface = Stack outside surface, Bottom Surface = Stack inside surface  
 All stresses are in PSI

## B.1.2

## 50 FT TALL BREACHED STEEL STACK MODEL CENTER OF ELEMENT STRESSES

ELEMENT NO	STRESS DIRECTION	42.1	42.2	43.1	43.2	44.1	44.2
TOP SURFACE STRESSES	LC1	82.	-1782.	-55.	41700.	-166.	-1539.
	LC2	38.	-948.	-61.	-921.	-119.	-791.
	LC3	-136.	698.	-80.	644.	35.	612.
	LC4	431.6	-10503.6	-616.2	-10173.2	-1260.8	-8816.2
	LC5	-1169.2	4639.6	-791.	4224.8	156.	4091.4
BOTTOM SURFACE STRESSES	LC1	-69.	-1877.	44.	-1770.	171.	-1552.
	LC2	-12.	-995.	45.	-959.	134.	-789.
	LC3	170.	738.	85.	657.	-25.	585.
	LC4	-179.4	-11031.	458.	-10592.8	1403.8	-8810.8
	LC5	1495.	4912.6	826.	4274.4	-59.	3830.
MEMBRANE STRESSES	LC1	6.5	-1829.5	-5.5	-1735.	2.5	-1545.5
	LC4	126.1	-10767.3	-79.1	-10383.	71.5	-8813.5
	LC5	162.9	4776.1	17.5	4249.6	48.5	3960.7
TOP BENDING STRESSES	LC1	75.5	47.5	-49.5	35.	-168.5	6.5
	LC4	305.5	263.7	-537.1	209.8	-1332.3	-2.7
	LC5	-1332.1	-136.5	-808.5	-24.8	107.5	130.7
BOTTOM BENDING STRESSES	LC1	-75.5	-47.5	49.5	-35.	168.5	-6.5
	LC4	-305.5	-263.7	537.1	-209.8	1332.3	2.7
	LC5	1332.1	136.5	808.5	24.8	-107.5	-130.

Note: 1 = Local element x stresses, 2 = Local element y stresses  
 LC = Load case, LC1 = Dead load, LC2 = E+W Wind, q=1.0PSF,  
 LC3 = N+S Wind, q=1.0PSF, LC4 = LC1 + 9.2LC2, LC5 = LC1 + 9.2LC3  
 Top Surface = Stack outside surface, Bottom Surface = Stack inside surface  
 All stresses are in PSI.

B.1.3

50 FT TALL BREACHED STEEL STACK MODEL CENTER OF ELEMENT STRESSES

ELEMENT NO	STRESS DIRECTION	45.1	45.2	46.1	46.2	47.1	47.2
TOP SURFACE STRESSES	LC1	-222.	-1365.	-220.	1281.	-150.	-1270.
	LC2	-168.	-734.	-159.	-634.	-132.	-661.
	LC3	121.	513.	191.	517.	153.	439.
	LC4	-1767.6	-8117.8	-1682.8	-7113.8	-1364.4	-7351.2
	LC5	891.2	3354.6	1537.2	3475.4	1349.6	2768.8
BOTTOM SURFACE STRESSES	LC1	249.	-1407.	270.	-1333.	190.	-1318.
	LC2	172.	-743.	201.	-646.	146.	-668.
	LC3	-145.	457.	-212.	431.	-209.	370.
	LC4	1831.4	-8242.6	2119.2	-7276.2	1533.2	-7463.6
	LC5	-1085.	2797.4	-1680.4	2632.2	-1732.8	2086.
MEMBRANE STRESSES	LC1	13.5	-1386.	25.	-1307.	20.	-1294.
	LC4	31.9	-8180.2	218.2	-7195.	84.4	-7407.4
	LC5	-96.9	3076.	-71.6	3053.8	-191.6	2427.4
TOP BENDING STRESSES	LC1	-235.5	21.	-245.	26.	-170.	24.
	LC4	-1799.5	62.4	-1901.	81.2	-1448.8	56.2
	LC5	988.1	278.6	1608.8	421.6	1541.2	341.4
BOTTOM BENDING STRESSES	LC1	235.5	-21.	245.	-26.	170.	-24.
	LC4	1799.5	-62.4	1901.	-81.2	1448.8	-56.2
	LC5	-988.1	-278.6	-1608.8	-421.6	-1541.2	-341.4

Note: 1 = Local element x stresses, 2 = Local element y stresses  
 LC = Load case, LC1 = Dead load, LC2 = E-W Wind, q=1.0PSF,  
 LC3 = N-S Wind, q=1.0PSF, LC4 = LC1 + 9.2LC2, LC5 = LC1 + 9.2LC3  
 Top Surface = Stack outside surface, Bottom Surface = Stack inside surface  
 All stresses are in PSI

B.1.1.4

50 FT TALL BREECHEED STEEL STACK MODEL CENTER OF ELEMENT STRESSES

ELEMENT NO	STRESS DIRECTION	48.1	48.2	49.1	49.2	50.1	50.2
TOP SURFACE STRESSES	LC1	-1347.	-152.	-1477.	-366.	-1434.	
	LC2	-90.	-154.	-820.	-286.	-725.	
	LC3	101.	-12.	404.	-78.	459.	
	LC4	-924.	-1568.8	-9021.	-2997.2	-8104.	
	LC5	836.2	-262.4	2239.8	-1083.6	2788.8	
BOTTOM SURFACE STRESSES	LC1	69.	139.	-1308.	368.	-1305.	
	LC2	88.	128.	-670.	300.	-605.	
	LC3	-98.	434.	430.	85.	503.	
	LC4	878.6	1316.6	-7472.	3128.	-6871.	
	LC5	-832.6	65.4	2648.	1450.	3222.6	
MEMBRANE STRESSES	LC1	-12.	-6.5	-1392.5	1.	-1369.5	
	LC4	-21.2	-126.1	-8246.5	65.4	-7487.5	
	LC5	1.8	-98.5	2443.9	33.2	3055.7	
TOP BENDING STRESSES	LC1	-81.	19.	-84.5	-367.	-64.5	
	LC4	-899.8	14.4	-774.5	-3062.6	-616.5	
	LC5	834.4	184.6	-204.1	-1116.8	-262.9	
BOTTOM BENDING STRESSES	LC1	81.	-19.	84.5	367.	64.5	
	LC4	899.8	-14.4	774.5	3062.6	616.5	
	LC5	-834.4	-184.6	204.1	1116.8	266.9	

Note: 1 = Local element x stresses, 2 = Local element y stresses  
 LC = Load case, LC1 = Dead load, LC2 = E+W Wind, q=1.0PSF,  
 LC3 = N+S Wind, q=1.0PSF, LC4 = LC1 + 9.2LC2, LC5 = LC1 + 9.2LC3  
 Top Surface = Stack outside surface, Bottom Surface = Stack inside surface  
 All stresses are in PSI

B.I.5  
50 FT TALL BREECHEED STEEL STACK MODEL CENTER OF ELEMENT STRESSES

ELEMENT NO	STRESS DIRECTION	53.1	53.2	54.1	54.2	55.1	55.2
TOP SURFACE STRESSES	LC1	-303.	-2581.	212.	-2578.	270.	-2718.
	LC2	-189.	-1636.	140.	1631.	186.	-1671.
	LC3	51.	415.	-39.	432.	-71.	605.
	LC4	-2041.8	-17632.2	1500.	-17583.2	1981.2	-18091.2
	LC5	166.2	1237.	-146.8	1396.4	2883.2	2848.
BOTTOM SURFACE STRESSES	LC1	-269.	-2305.	-64.	-2592.	-327.	-2507.
	LC2	-170.	-1442.	-50.	-1627.	-218.	-1553.
	LC3	37.	388.	12.	464.	94.	456.
	LC4	-1833.	-15571.4	-524.	-17560.4	-2332.6	-16794.6
	LC5	71.4	1264.6	46.4	1676.8	537.8	1688.2
MEMBRANE STRESSES	LC1	-286.	-2443.	74.	-2585.	-38.5	-2612.5
	LC4	-1937.4	-16601.8	488.	-17571.8	-175.	-17442.9
	LC5	118.8	1250.8	-50.2	1536.6	77.3	2268.1
TOP BENDING STRESSES	LC1	-17.	-138.	138.	7.	298.5	-105.5
	LC4	-104.4	-1030.4	1012.	-11.4	2156.9	-648.3
	LC5	47.4	-13.8	-96.6	-140.2	-460.5	579.9
BOTTOM BENDING STRESSES	LC1	17.	138.	-138.	7.	-298.5	105.5
	LC4	104.4	1030.4	-1012.	11.4	-2156.9	648.3
	LC5	-47.4	13.8	96.6	140.2	460.5	-579.9

Note: 1 = Local element x stresses, 2 = Local element y stresses  
 LC = Load case, LC1 = Dead load, LC2 = E+W Wind, q=1.0PSF,  
 LC3 = N+S Wind, q=1.0PSF, LC4 = LC1 + 9.2LC2, LC5 = LC1 + 9.2LC3  
 Top Surface = Stack outside surface, Bottom Surface = Stack inside surface  
 All stresses are in PSI



## B.1.6

## 50 FT TALL BREACHED STEEL STACK MODEL CENTER OF ELEMENT STRESSES

ELEMENT NO	STRESS DIRECTION	56.1	56.2	57.1	57.2	58.1	58.2
TOP SURFACE STRESSES	LC1	343.	-2409.	213.	-2328.	169.	-2218.
	LC2	233.	-1471.	134.	-1421.	98.	-1365.
	LC3	-90.	668.	-56.	535.	-19.	417.
	LC4	2394.6	-15942.2	1445.8	-15401.2	1070.6	-14776.
	LC5	-485.	3736.6	-302.2	3594.	-5.8	1448.4
BOTTOM SURFACE STRESSES	LC1	-425.	-2635.	-249.	-2622.	-160.	-2547.
	LC2	-279.	-1640.	-155.	-1615.	-97.	-1562.
	LC3	119.	504.	55.	473.	16.	439.
	LC4	-2991.8	-17723.	-1675.	-17480.	-1052.4	-16981.8
MEMBRANE STRESSES	LC5	669.8	2001.8	257.	1729.6	-12.8	1399.8
	LC1	-41.	-2522.	-18.	-2475.	4.5	-2382.5
	LC4	-298.6	-16832.6	-114.6	-16440.6	9.1	-15878.9
	LC5	92.4	2869.2	-22.6	2161.8	-9.3	1509.1
	LC1	384.	113.	231.	147.	164.5	184.5
TOP BENDING STRESSES	LC4	2693.2	890.4	1560.4	1039.4	1061.5	1102.9
	LC5	-577.4	867.4	-279.6	432.2	3.5	109.9
	LC1	-384.	-113.	-231.	-147.	-164.5	-164.5
	LC4	-2693.2	-890.4	-1560.4	-1039.4	-1061.5	-1102.9
	LC5	577.4	-867.4	279.6	-432.2	-3.5	-109.9

Note: .1 = Local element x stresses, 2 = Local element y stresses

LC = Load case, LC1 = Dead load, LC2 = E+W Wind, q=1.0PSF,

LC3 = N+S Wind, q=1.0PSF, LC4 = LC1 + 9.2LC2, LC5 = LC1 + 9.2LC3

Top Surface = Stack outside surface, Bottom Surface = Stack inside surface

All stresses are in PSI

B.1.7

50 FT TALL BREACHED STEEL STACK MODEL CENTER OF ELEMENT STRESSES

ELEMENT NO	STRESS DIRECTION	59.1	59.2	60.1	60.2	61.1	61.2
TOP SURFACE STRESSES	LC1	299.	-1960.	607.	-1522.	655.	-1846.
	LC2	179.	-1186.	357.	-927.	378.	-1118.
	LC3	-5.	308.	-61.	106.	-127.	175.
	LC4	1945.8	-12871.2	3891.4	-10050.4	4132.6	-12131.6
	LC5	253.	873.6	45.8	-546.8	-513.4	-236.
BOTTOM SURFACE STRESSES	LC1	-375.	-2549.	-671.	-2388.	-727.	-2377.
	LC2	-225.	-1546.	-402.	-1445.	-416.	-1411.
	LC3	24.	412.	42.	364.	119.	420.
	LC4	-2445.	-16772.2	-4369.4	-15682.	-4554.2	-15358.2
	LC5	-154.2	1241.4	-284.6	960.8	367.8	1487.
MEMBRANE STRESSES	LC1	-38.	-2254.5	-32.	-1955.	-36.	-2111.5
	LC4	-249.6	-14821.7	-239.	-12866.2	-210.8	-13744.9
	LC5	49.4	1057.5	-119.4	207.	-72.8	625.5
TOP BENDING STRESSES	LC1	337.	294.5	639.	433.	691.	265.5
	LC4	2195.4	1950.5	4130.4	2815.8	4343.4	1613.3
	LC5	203.6	-183.9	165.2	-753.8	-440.6	-861.5
BOTTOM BENDING STRESSES	LC1	-337.	-294.5	-639.	-433.	-691.	-265.5
	LC4	-2195.4	-1950.5	-4130.4	-2815.8	-4343.4	-1613.3
	LC5	-203.6	183.9	-165.2	753.8	440.6	861.5

Note: 1 = Local element x stresses, 2 = Local element y stresses  
 LC = Load case, LC1 = Dead load, LC2 = E+W Wind, q=1.0PSF,  
 LC3 = N+S Wind, q=1.0PSF, LC4 = LC1 + 9.2LC2, LC5 = LC1 + 9.2LC3  
 Top Surface = Stack outside surface, Bottom Surface = Stack inside surface  
 All stresses are in PSI

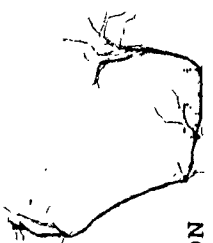
B.1.8  
50 FT TALL BREACHED STEEL STACK MODEL CENTER OF ELEMENT STRESSES

ELEMENT NO	STRESS DIRECTION	62.1	62.2	63.1	63.2	64.1	64.2
TOP SURFACE STRESSES	LC1	413.	-2024.	31.	-2018.	-343.	-1809.
	LC2	216.	-1228.	-24.	-1196.	-269.	-1077.
	LC3	-205.	271.	-243.	299.	-253.	202.
	LC4	2400.2	-13321.6	-189.8	-13021.2	-2817.8	-11717.4
	LC5	-1473.	469.2	-2204.6	732.8	-2670.6	969.4
BOTTOM SURFACE STRESSES	LC1	-431.	-2129.	-36.	-1834.	361.	-1582.
	LC2	-228.	-1259.	28.	-1046.	281.	-899.
	LC3	203.	407.	257.	466.	256.	454.
	LC4	-2528.6	-13711.8	221.6	-11457.2	2946.2	-9797.6
	LC5	1436.6	1615.4	2328.4	2453.2	2716.2	2594.8
MEMBRANE STRESSES	LC1	-9.	-2076.5	-2.5	-1926.	9.	-1695.5
	LC4	-64.2	-13516.7	15.9	-12239.2	64.2	-10757.5
	LC5	-18.2	1042.3	61.9	1593.	22.8	1782.1
TOP BENDING STRESSES	LC1	422.	52.5	33.5	-92.	-352.	-113.5
	LC4	2464.4	195.1	-205.7	-782.	-2882.4	-959.9
	LC5	-1454.8	-573.1	-2346.5	-860.2	-2693.4	-812.7
BOTTOM BENDING STRESSES	LC1	-422.	-52.5	-33.5	92.	352.	113.5
	LC4	-2464.4	-195.1	205.7	782.	2882.4	959.9
	LC5	1454.8	573.1	2346.5	860.2	2693.4	812.7

Note: 1 = Local element x stresses, 2 = Local element y stresses  
 LC = Load case, LC1 = Dead load, LC2 = E+W Wind, q=1.0PSF,  
 LC3 = N+S Wind, q=1.0PSF, LC4 = LC1 + 9.2LC2, LC5 = LC1 + 9.2LC3  
 Top Surface = Stack outside surface, Bottom Surface = Stack inside surface  
 All stresses are in PSI

B.1.9

50 FT TALL BREACHED STEEL STACK MODEL CENTER OF ELEMENT STRESSES



ELEMENT NO	STRESS DIRECTION	67.2	68.1	68.2	69.1	69.2
TOP SURFACE STRESSES	LC1	-256.	152.	-2079.	-2795.	271.
	LC2	-168.	119.	-1383.	-1845.	214.
	LC3	5.	40.	-10.	186.	110.
	LC4	-1801.6	1246.8	-14802.6	-19769.	2239.8
	LC5	-210.	520.	-2171.	10831.3	1283.
BOTTOM SURFACE STRESSES	LC1	-184.	-24.	-2098.	-2007.	-197.
	LC2	-125.	-30.	-1381.	-1358.	-172.
	LC3	-1.	0.	94.	-203.	-129.
	LC4	-1334.	-300.	-14803.2	-14500.6	-1779.4
	LC5	-193.2	-24.	-1233.2	-3874.6	-1383.8
MEMBRANE STRESSES	LC1	-220.	64.	-2088.5	-2401.	37.
	LC4	-1567.8	473.4	-14802.9	-17134.8	230.3
	LC5	-201.6	248.	-1702.1	-2479.2	-50.4
TOP BENDING STRESSES	LC1	-36.	88.	9.5	-394.	234.
	LC4	-233.8	773.4	0.3	-2634.2	2009.6
	LC5	-8.4	272.	-468.9	1395.4	1393.4
BOTTOM BENDING STRESSES	LC1	36.	-88.	-9.5	394.	-234.
	LC4	233.8	-773.4	-0.3	2634.2	-2009.6
	LC5	8.4	-272.	468.9	-1395.4	-1393.4

Note: 1 = Local element x stresses, 2 = Local element y stresses  
 LC = Load case, LC1 = Dead load, LC2 = E+W Wind, q=1.0PSF,  
 LC3 = N+S Wind, q=1.0PSF, LC4 = LC1 + 9.2LC2, LC5 = LC1 + 9.2LC3  
 Top Surface = Stack outside surface, Bottom Surface = Stack inside surface  
 All stresses are in PSI

B.1.1.0  
50 FT TALL BREACHED STEEL STACK MODEL CENTER OF ELEMENT STRESSES

ELEMENT NO	STRESS DIRECTION	70.1	70.2	71.1	71.2	72.1	72.2
TOP SURFACE STRESSES	LC1	259.	-3437.	190.	-3255.	59.	-3574.
	LC2	192.	-2272.	127.	2140.	36.	-2322.
	LC3	44.	151.	-10.	121.	-7.	274.
	LC4	2025.4	-24339.4	1358.4	-22943.	390.2	-24936.4
	LC5	663.8	-2047.8	198.	-2141.8	-5.4	-1053.2
BOTTOM SURFACE STRESSES	LC1	-391.	-3201.	-189.	-3822.	-66.	-3992.
	LC2	-296.	-2180.	-125.	-2527.	-40.	-2582.
	LC3	-146.	-447.	26.	7.	6.	296.
	LC4	-3114.2	-23257.	-1339.	-27020.4	-434.	-27746.4
	LC5	-1734.2	-7313.4	50.2	-3757.6	-10.8	-1268.8
MEMBRANE STRESSES	LC1	-66.	-3319.	0.5	-3538.5	-3.5	-3783.
	LC4	-544.4	-23798.2	9.7	-25006.7	-21.9	-26341.4
	LC5	-535.2	-4680.6	74.1	-2949.7	-8.1	-1161.
TOP BENDING STRESSES	LC1	325.	-118.	189.5	283.5	62.5	209.
	LC4	2569.8	-541.2	1348.7	2063.7	412.1	1405.
	LC5	1199.	2632.8	23.9	807.9	3.7	107.8
BOTTOM BENDING STRESSES	LC1	-325.	118.	-189.5	-283.5	-62.5	-209.
	LC4	-2569.8	541.2	-1348.7	-2063.7	-412.1	-1405.
	LC5	-1199.	-2632.8	-23.9	-807.9	-3.7	-107.8

Note: 1 = Local element x stresses, 2 = Local element y stresses  
 LC = Load case, LC1 = Dead load, LC2 = E-W Wind, q=1.0PSF,  
 LC3 = N+S Wind, q=1.0PSF, LC4 = LC1 + 9.2LC2, LC5 = LC1 + 9.2LC3  
 Top Surface = Stack outside surface, Bottom Surface = Stack inside surface  
 All stresses are in PSI

B.1.11  
50 FT TALL BREACHED STEEL STACK MODEL CENTER OF ELEMENT STRESSES

ELEMENT NO	STRESS DIRECTION	73.1	73.2	74.1	74.2	75.1	75.2
TOP SURFACE STRESSES	LC1	330.	-3449.	258.	-3803.	1282.	-2882.
	LC2	203.	-2222.	525.	-2420.	780.	-1822.
	LC3	43.	374.	-266.	371.	-533.	213.
	LC4	2197.6	-23891.4	5688.	-26067.	8458.	-19644.4
	LC5	-65.6	-8.2	-1589.2	-389.8	-3621.6	-922.4
BOTTOM SURFACE STRESSES	LC1	-293.	-4647.	+1296.	-4913.	-1238.	-2999.
	LC2	-180.	-2958.	-797.	-3082.	-814.	-1862.
	LC3	30.	636.	392.	1094.	563.	748.
	LC4	-1949.	-31860.6	-8628.4	-39267.4	-8826.8	-20129.4
	LC5	-17.	1204.2	2310.4	5151.8	3841.6	3882.6
MEMBRANE STRESSES	LC1	18.5	-4048.	-219.	-4358.	-28.	-2940.5
	LC4	124.3	-27876.	-1470.2	-29667.2	-184.4	-19886.9
	LC5	-41.3	598.	360.6	2381.	110.	1480.1
TOP BENDING STRESSES	LC1	311.5	599.	1077.	555.	1310.	58.5
	LC4	2073.3	3984.6	7158.2	3600.2	8642.4	242.5
	LC5	-24.3	-606.2	-1949.8	-2770.8	-3731.6	-2402.5
BOTTOM BENDING STRESSES	LC1	-311.5	-599.	-1077.	-555.	-1310.	-58.5
	LC4	-2073.3	-3984.6	-7158.2	-3600.2	-8642.4	-242.5
	LC5	24.3	606.2	1949.8	2770.8	3731.6	2402.5

Note: 1 = Local element x stresses, 2 = Local element y stresses  
 LC = Load case, LC1 = Dead load, LC2 = E+W Wind, q=1.0PSF,  
 LC3 = N+S Wind, q=1.0PSF, LC4 = LC1 + 9.2LC2, LC5 = LC1 + 9.2LC3  
 Top Surface = Stack, outside surface, Bottom Surface = Stack inside surface  
 All stresses are in PSI

B.1.12

50 FT TALL BREACHED STEEL STACK MODEL CENTER OF ELEMENT STRESSES

ELEMENT NO	STRESS DIRECTION	76.1	76.2	77.1	77.2	78.2	78.1
TOP SURFACE STRESSES	LC1	709.4	-2363.	155.	-2039.	-1774.	-136.
	LC2	420.	4475.	75.	-1262.	-1088.	-99.
	LC3	-497.	291.	-423.	224.	247.	-340.
	LC4	4573.	-15933.	845.	-13649.4	-11783.6	-1046.8
	LC5	-3863.4	314.2	-3736.6	21.8	498.4	-3264.
BOTTOM SURFACE STRESSES	LC1	-648.	-2374.	-135.	-1939.	-1577.	165.
	LC2	-377.	-1456.	-59.	-11181.	-946.	117.
	LC3	481.	498.	436.	489.	442.	350.
	LC4	-4116.4	-15769.2	-677.8	-12804.2	-10280.2	1241.4
	LC5	3777.2	2207.6	3876.2	2559.8	2489.4	3385.
MEMBRANE STRESSES	LC1	30.5	-2368.5	10.	-1989.	-1675.5	14.5
	LC4	228.3	-15851.1	83.6	-13226.8	-11031.9	97.3
	LC5	-43.1	1260.9	69.8	1290.8	1493.9	60.5
TOP BENDING STRESSES	LC1	678.5	5.5	145.	-50.	-98.5	150.5
	LC4	4344.7	-81.9	761.4	-422.6	-751.7	1144.1
	LC5	-3820.9	-946.7	-3806.4	-1269.	-995.5	-3324.5
BOTTOM BENDING STRESSES	LC1	-678.5	-5.5	-145.	50.	98.5	150.5
	LC4	-4344.7	81.9	-761.4	422.6	751.7	1144.1
	LC5	3820.9	946.7	3806.4	1269.	995.5	3324.5

Note: 1 = Local element x stresses, 2 = Local element y stresses  
 LC = Load case, LC1 = Dead load, LC2 = E+W Wind, q=1.0PSF,  
 LC3 = N+S Wind, q=1.0PSF, LC4 = LC1 + 9.2LC2, LC5 = LC1 + 9.2LC3  
 Top Surface = Stack outside surface, Bottom Surface = Stack inside surface  
 All stresses are in PSI

B.1.13  
 50 FT TALL BREACHED STEEL STACK MODEL CENTER OF ELEMENT STRESSES

ELEMENT NO	STRESS DIRECTION	81.1	81.2	82.1	82.2	83.1	83.2
TOP SURFACE STRESSES	LC1	-108.	-1062.	45.	-938.	159.	-929.
	LC2	-75.	-740.	52.	-654.	156.	-646.
	LC3	-31.	-239.	132.	-177.	238.	-187.
	LC4	-798.	-7870.	523.4	-6954.8	1594.2	-6866.2
	LC5	-393.2	-3260.8	1259.4	-2566.4	2348.6	-2643.4
BOTTOM SURFACE STRESSES	LC1	-154.	-1037.	-101.	-1094.	43.	-792.
	LC2	-105.	-704.	-84.	-757.	-44.	-564.
	LC3	-23.	-132.	-72.	-189.	-390.	-241.
	LC4	-1120.	-7513.8	-873.8	-8058.4	-361.8	-5980.8
MEMBRANE STRESSES	LC5	-365.6	-2251.4	-763.4	-2832.8	-3545.	-3009.2
	LC1	-131.	-1049.5	-28.	-1016.	101.	-857.5
	LC4	-959.	-7691.9	-175.2	-7506.6	616.2	-6423.5
	LC5	-379.4	-2756.1	248.	-2699.6	-598.2	-2826.3
	LC1	23.	-12.5	73.	78.	58.	-65.5
BENDING STRESSES	LC4	161.	-178.1	698.6	551.8	978.	-442.7
	LC5	-13.8	-504.7	1011.4	133.2	2946.8	182.9
	LC1	-23.	12.5	-73.	-78.	-58.	6.5
BENDING STRESSES	LC4	-161.	178.1	-698.6	-551.8	-978.	442.7
	LC5	13.8	504.7	-1011.4	-133.2	-2946.8	-182.9

Note: 1 = Local element x stresses, 2 = Local element y stresses  
 LC = Load case, LC1 = Dead load, LC2 = E+W Wind, q=1.0PSF,  
 LC3 = N+S Wind, q=1.0PSF, LC4 = LC1 + 9.2LC2, LC5 = LC1 + 9.2LC3  
 Top Surface = Stack outside surface, Bottom Surface = Stack inside surface  
 All stresses are in PSI



B.1.14

50 FT TALL BREACHED STEEL STACK MODEL CENTER OF ELEMENT STRESSES

ELEMENT NO	STRESS DIRECTION	84.1	84.2	85.1	85.2	86.1	86.2
TOP SURFACE STRESSES	LC1	1498.	-1404.	674.	-1473.	-1449.	327.
	LC2	936.	-883.	433.	-926.	-901.	226.
	LC3	-972.	185.	-709.	112.	187.	-500.
	LC4	10104.2	-9527.6	4657.6	-9992.2	-9738.2	2406.2
	LC5	-7449.4	398.	-5848.8	-442.6	271.4	-4273.
BOTTOM SURFACE STRESSES	LC1	-1653.	-1695.	-764.	-1751.	-1556.	-333.
	LC2	-1033.	-1063.	-490.	-1098.	-979.	-230.
	LC3	1120.	510.	727.	554.	439.	522.
	LC4	-11156.6	-11474.6	-5272.	-11852.6	-10562.8	-2449.
	LC5	8651.	2997.	5924.4	3345.8	2482.8	4469.4
MEMBRANE STRESSES	LC1	-80.	-1549.5	-45.	-1612.	-1502.5	-3.
	LC4	-526.2	-10501.1	-307.2	-10922.4	-10150.9	-21.4
	LC5	600.8	1647.5	37.8	1451.6	1377.1	98.2
TOP BENDING STRESSES	LC1	1573.	145.5	719.	139.	53.5	330.
	LC4	10630.4	973.5	4964.8	930.2	412.3	2427.6
	LC5	-8050.2	-1349.5	-5886.6	-1894.2	-1105.7	-4371.2
BOTTOM BENDING STRESSES	LC1	-1573.	-145.5	-719.	-139.	-53.5	-330.
	LC4	-10630.4	-973.5	-4964.8	-930.2	-412.3	-2427.6
	LC5	8050.2	1349.5	5886.6	1894.2	1105.7	4371.2

Note: 1 = Local element x stresses, 2 = Local element y stresses  
 LC = Load case, LC1 = Dead load, LC2 = E+W Wind, LC3 = 1.0PSF,  
 LC3 = N+S Wind, q=1.0PSF, LC4 = LC1 + 9.2LC2, LC5 = LC1 + 9.2LC3  
 Top Surface = Stack outside surface, Bottom Surface = Stack inside surface  
 All stresses are in PSI

B.1.15

## 50 FT TALL BREACHED STEEL STACK MODEL CENTER OF ELEMENT STRESSES

ELEMENT NO	STRESS DIRECTION	87.1	87.2	90.1	90.2	91.1	91.2
TOP SURFACE STRESSES	LC1	253.	-1320.	-17.	-171.	-93.	-16.
	LC2	190.	-816.	-14.	-134.	-85.	-24.
	LC3	364.	187.	-8.	-259.	89.	-121.
	LC4	2001.	-8827.2	-145.8	-1403.8	-691.	-236.8
	LC5	-3095.8	400.4	-90.6	-2553.8	725.8	-1129.2
BOTTOM SURFACE STRESSES	LC1	-240.	-1360.	-68.	-299.	1.	-208.
	LC2	-184.	-851.	-47.	-209.	-1.	-150.
	LC3	378.	401.	-51.	-169.	-109.	-168.
	LC4	-1932.8	-9189.2	-500.4	-2221.8	-8.2	-1588.
	LC5	3237.6	2329.2	-537.2	-1853.8	-1001.8	-1753.6
MEMBRANE STRESSES	LC1	6.5	-1340.	-42.5	-235.	-46.	-112.
	LC4	34.1	-9008.2	-323.1	-1812.8	-349.6	-912.4
	LC5	70.9	1364.8	-313.9	-2203.8	-138.	-1441.4
TOP BENDING STRESSES	LC1	246.5	20.	25.5	64.	-47.	96.
	LC4	1966.9	181.	177.3	409.	-341.4	675.6
	LC5	-3166.7	-964.4	223.3	-350.	863.8	312.2
BOTTOM BENDING STRESSES	LC1	-246.5	-20.	-25.5	-64.	47.	-96.
	LC4	-1966.9	-181.	-177.3	-409.	341.4	-675.6
	LC5	3166.7	964.4	-223.3	350.	-863.8	-312.2

Note: 1 = Local element x stresses, 2 = Local element y stresses

LC = Load case, LC1 = Dead load, LC2 = E+W Wind, q=1.0PSF,

LC3 = N+S Wind, q=1.0PSF, LC4 = LC1 + 9.2LC2, LC5 = LC1 + 9.2LC3

Top Surface = Stack outside surface, Bottom Surface = Stack inside surface

All stresses are in PSI

B.1.1.16

50 FT TALL BREACHED STEEL STACK MODEL CENTER OF ELEMENT STRESSES

ELEMENT NO	STRESS DIRECTION	92.1	92.2	93.1	93.2	94.1	94.2
TOP SURFACE STRESSES	LC1	-289.	93.	-439.	94.	-69.	-397.
	LC2	-199.	61.	-231.	65.	8.	-240.
	LC3	122.	11.	-610.	-120.	-543.	-42.
	LC4	-2119.8	654.2	-2567.2	692.	4.6	-2605.
	LC5	833.4	102.2	-6051.	1010.	-5064.6	-783.4
BOTTOM SURFACE STRESSES	LC1	523.	106.	915.	109.	-116.	-478.
	LC2	365.	74.	530.	59.	-129.	-319.
	LC3	-87.	-24.	551.	110.	638.	336.
	LC4	3881.	786.8	5791.	651.8	-1302.8	-3412.8
	LC5	-277.4	-114.8	5984.2	1121.	5753.6	2613.2
MEMBRANE STRESSES	LC1	117.	99.5	238.	101.5	-92.5	-437.5
	LC4	880.6	720.5	1613.4	671.9	-649.1	-3008.9
	LC5	278.	-6.3	-33.4	55.5	344.5	914.9
TOP BENDING STRESSES	LC1	-406.	-6.5	-677.	-7.5	23.5	40.5
	LC4	-3000.4	-66.3	-4177.6	20.1	653.7	403.9
	LC5	555.4	108.5	-6017.6	-1065.5	-5409.1	-1698.3
BOTTOM BENDING STRESSES	LC1	406.	6.5	677.	7.5	-23.5	-40.5
	LC4	3000.4	66.3	4177.6	-20.1	-653.7	-403.9
	LC5	-555.4	-108.5	6017.6	1065.5	5409.1	1698.3

Note: 1 = Local element x stresses, 2 = Local element y stresses  
 LC = Load case, LC1 = Dead load, LC2 = E+W Wind, q=1.0PSF,  
 LC3 = N+S Wind, q=1.0PSF, LC4 = LC1 + 9.2LC2, LC5 = LC1 + 9.2LC3  
 Top Surface = Stack outside surface, Bottom Surface = Stack inside surface  
 All stresses are in PSI

B.1.17

50 FT TALL BREACHED STEEL STACK MODEL CENTER OF ELEMENT STRESSES

ELEMENT NO	STRESS DIRECTION	95.1	95.2	96.1	96.2	99.1	99.2
TOP SURFACE STRESSES	LC1	394.	-480.	660.	-675.	-11.	401.
	LC2	309.	-289.	483.	-413.	-13.	269.
	LC3	-450.	83.	-286.	128.	-5.	210.
	LC4	3236.8	-3138.8	5103.6	-4474.6	-130.6	2875.8
	LC5	-3746.	283.6	-1971.2	502.6	-57.	-611.
BOTTOM SURFACE STRESSES	LC1	-413.	-932.	-681.	-1015.	76.	185.
	LC2	-324.	-612.	-493.	-659.	54.	127.
	LC3	473.	325.	305.	288.	-30.	-74.
	LC4	-999.0	-6562.4	-5262.6	-7077.8	572.0	1853.4
	LC5	9938.6	2058.	2125.	1634.6	-200.	-495.8
MEMBRANE STRESSES	LC1	-9.5	-706.	-10.5	-845.	32.5	293.
	LC4	-78.5	-4850.6	-79.5	-5776.2	221.7	2114.6
	LC5	96.3	1170.3	76.9	1068.6	-128.5	-553.4
TOP BENDING STRESSES	LC1	403.5	226.	670.5	170.	-43.5	108.
	LC4	9315.1	1711.8	5183.1	1301.6	-351.7	761.2
	LC5	-3842.1	-887.2	-2048.1	-566.	71.5	-57.6
BOTTOM BENDING STRESSES	LC1	-403.5	-226.	-670.5	-170.	43.5	-108.
	LC4	-9315.1	-1711.8	-5183.1	-1301.6	351.7	-761.2
	LC5	3842.1	887.2	2048.1	566.	-71.5	57.6

Note: 1 = Local element x stresses, 2 = Local element y stresses  
 LC = Load case, LC1 = Dead load, LC2 = E+W Wind, q=1.0PSF,  
 LC3 = N+S Wind, q=1.0PSF, LC4 = LC1 + 9.2LC2, LC5 = LC1 + 9.2LC3  
 Top Surface = Stack outside surface, Bottom Surface = Stack inside surface  
 All stresses are in PSI

B.1.18

50 FT TALL BREACHED STEEL STACK MODEL CENTER OF ELEMENT/STRESSES

ELEMENT NO	STRESS DIRECTION	100.1	100.2	101.2	101.1	102.2	102.1
TOP SURFACE STRESSES	LC1	-186.	257.	37.	-290.	-132.	-1499.
	LC2	-148.	174.	20.	-240.	173.	-869.
	LC3	31.	-24.	-10.	17.	-92.	-190.
	LC4	-1547.6	1857.8	221.	-2493.	-803.6	-9493.8
	LC5	99.2	36.2	-95.	-133.6	-426.4	-9247.
BOTTOM SURFACE STRESSES	LC1	227.	191.	62.	256.	239.	1787.
	LC2	177.	135.	51.	230.	135.	1049.
	LC3	-33.	-51.	15.	8.	52.	163.
	LC4	1855.4	1442.2	591.2	2672.	1481.	11437.8
	LC5	576.6	-278.2	200.	329.6	717.4	3286.8
MEMBRANE STRESSES	LC1	20.5	224.	49.5	-17.	53.5	144.
	LC4	153.9	1650.	376.1	-63.	338.7	972.
	LC5	11.3	-121.	72.5	98.	145.5	19.8
TOP BENDING STRESSES	LC1	-206.5	33.	-12.5	-273.	-185.5	-1643.
	LC4	-1701.5	207.8	-155.	-2435.	-1142.3	-10465.8
	LC5	87.9	157.2	-127.5	-2317.6	-571.9	-3266.8
BOTTOM BENDING STRESSES	LC1	286.5	-33.	12.5	273.	185.5	1643.
	LC4	1701.5	-207.8	155.1	2435.	1142.3	10465.8
	LC5	-87.9	-157.2	127.5	231.6	571.9	3266.8

Note: 1 = Local element x stresses, 2 = Local element y stresses  
 LC = Load case, LC1 = Dead load; LC2 = E+W Wind, q=1.0PSF,  
 LC3 = N+S Wind, q=1.0PSF, LC4 = LC1 + 9.2LC2, LC5 = LC1 + 9.2LC3  
 Top Surface = Stack outside surface, Bottom Surface = Stack inside surface  
 All stresses are in PSI

B.1.1.19

50 FT TALL BREACHED STEEL STACK MODEL CENTER OF ELEMENT STRESSES

ELEMENT NO	STRESS DIRECTION	103.1	103.2	104.1	104.2	105.1	105.2
TOP SURFACE STRESSES	LC1	-712.	11.	229.	237.	919.	-186.
	LC2	-370.	23.	230.	167.	866.	-106.
	LC3	-203.	-57.	-178.	18.	-112.	45.
	LC4	-4116.	222.6	2345.	1773.4	7046.2	-1161.2
	LC5	-2579.6	-513.4	-1408.6	402.6	-111.4	228.
BOTTOM SURFACE STRESSES	LC1	877.	297.	-243.	-272.	-967.	-748.
	LC2	467.	157.	-242.	-272.	-700.	-504.
	LC3	227.	94.	199.	119.	120.	106.
	LC4	5173.4	1741.4	-2469.4	-2875.4	-7407.	-5384.0
	LC5	2965.4	1161.8	1587.8	721.8	137.	227.2
MEMBRANE STRESSES	LC1	82.5	154.	-7.	-68.	-24.	-467.
	LC4	528.	982.	-62.2	-551.	120.4	-3273.
	LC5	192.9	324.2	89.6	562.2	12.8	227.6
TOP BENDING STRESSES	LC1	-794.5	-143.	236.	305.	943.	281.
	LC4	-4644.	-759.4	2407.2	2324.4	7226.6	2111.8
	LC5	-2772.5	-837.6	-1498.2	-159.6	-124.2	0.4
BOTTOM BENDING STRESSES	LC1	794.5	143.	-236.	-305.	-943.	-281.
	LC4	4644.	759.4	-2407.2	-2324.4	-7226.6	-2111.8
	LC5	2772.5	837.6	1498.2	159.6	124.2	-0.4

Note: 1 = Local element x stresses, 2 = Local element y stresses  
 LC = Load case, LC1 = Dead load, LC2 = E+W Wind, q=1.0PSF,  
 LC3 = N+S Wind, q=1.0PSF, LC4 = LC1 + 9.2LC2, LC5 = LC1 + 9.2LC3  
 Top Surface = Stack outside surface, Bottom Surface = Stack inside surface  
 All stresses are in PSI

B.1.1.20  
50 FT TALL BREACHED STEEL STACK MODEL CENTER OF ELEMENT STRESSES

ELEMENT NO	STRESS DIRECTION	108.1	108.2	109.1	109.2	110.1	110.2
TOP SURFACE STRESSES	LC1	-11.	401.	-186.	257.	-290.	37.
	LC2	-13.	269.	-148.	174.	-240.	20.
	LC3	5.	110.	-31.	24.	-17.	10.
	LC4	-130.6	2875.8	-1547.6	1857.8	-2498.	221.
	LC5	35.	1413.	-471.2	477.8	-446.4	129.
BOTTOM SURFACE STRESSES	LC1	76.	185.	227.	191.	256.	62.
	LC2	54.	127.	177.	136.	230.	51.
	LC3	30.	74.	33.	51.	-8.	15.
	LC4	572.8	1353.4	1855.4	1442.2	2372.	531.2
	LC5	352.	865.8	530.6	660.2	182.4	200.
MEMBRANE STRESSES	LC1	32.5	293.	20.5	224.	-17.	49.5
	LC4	221.1	2114.6	153.9	1650.	-63.	376.1
	LC5	193.5	1139.4	29.7	569.	-132.	164.5
TOP BENDING STRESSES	LC1	-43.5	108.	-206.5	33.	-273.	-12.5
	LC4	-351.7	761.2	-1701.5	207.8	-2435.	-155.1
	LC5	-158.5	273.6	-500.9	-91.2	-314.4	-35.5
BOTTOM BENDING STRESSES	LC1	43.5	-108.	206.5	-33.	273.	12.5
	LC4	351.7	-761.2	1701.5	-207.8	2435.	155.1
	LC5	158.5	-273.6	500.9	91.2	-314.4	35.5

Note: 1 = Local element x stresses, 2 = Local element y stresses  
 LC = Load case, LC1 = Dead load, LC2 = E+W Wind, q=1.0PSF,  
 LC3 = N+S Wind, q=1.0PSF, LC4 = LC1 + 9.2LC2, LC5 = LC1 + 9.2LC3  
 Top Surface = Stack outside surface, Bottom Surface = Stack inside surface  
 All stresses are in PSI

B.1.21  
50 FT TALL BREACHED STEEL STACK MODEL CENTER OF ELEMENT STRESSES

ELEMENT NO	STRESS DIRECTION	111.1	111.2	112.1	112.2	113.1	113.2
TOP SURFACE STRESSES	LC1	-1499.	-132.	-712.	11.	229.	237.
	LC2	-869.	-73.	-370.	23.	230.	167.
	LC3	190.	32.	203.	57.	178.	-18.
	LC4	-9493.8	-803.6	-4116.	222.6	2345.	1773.4
	LC5	249.	162.4	1155.6	535.4	1866.6	71.4
BOTTOM SURFACE STRESSES	LC1	1787.	239.	877.	297.	-243.	-373.
	LC2	1049.	135.	467.	157.	-242.	-272.
	LC3	-163.	-52.	-227.	-94.	-199.	-119.
	LC4	11437.8	1481.	5173.4	1741.4	-2469.4	-2875.4
	LC5	287.4	-239.4	-1211.4	-567.8	-2073.8	-1467.8
MEMBRANE STRESSES	LC1	144.	53.5	82.5	154.	-7.	-69.
	LC4	972.	338.7	528.7	982.	-62.2	-551.
	LC5	268.2	-38.5	-27.9	-16.2	-103.6	-698.2
TOP BENDING STRESSES	LC1	-1643.	-185.5	-794.5	-143.	236.	305.
	LC4	-10465.8	-1142.3	-4644.2	-759.4	2407.2	2324.4
	LC5	-19.2	200.9	1183.5	551.6	1970.2	769.6
BOTTOM BENDING STRESSES	LC1	1643.	185.5	794.5	143.	-236.	-305.
	LC4	10465.8	1142.3	4644.2	759.4	-2407.2	-2324.4
	LC5	19.2	-200.9	-1183.5	-551.6	-1970.2	-769.6

Note: 1 = Local element x stresses, 2 = Local element y stresses  
 LC = Load case, LC1 = Dead load, LC2 = E+W Wind, q=1.0PSF,  
 LC3 = N+S Wind, q=1.0PSF, LC4 = LC1 + 9.2LC2, LC5 = LC1 + 9.2LC3  
 Top Surface = Stack outside surface, Bottom Surface = Stack inside surface  
 All stresses are in PSI



B.1.22  
50 FT TALL BREECHEED STEEL STACK MODEL CENTER OF ELEMENT STRESSES

ELEMENT NO	STRESS DIRECTION	114.1	114.2	117.1	117.2	118.1	118.2
TOP SURFACE STRESSES	LC1	919.	-186.	-17.	-171.	-93.	-16.
	LC2	666.	-106.	-14.	-134.	-65.	-24.
	LC3	112.	-45.	8.	259.	-89.	121.
	LC4	7046.2	-1161.2	-145.8	-1403.8	-691.	-236.8
	LC5	1949.4	-600.	56.6	2211.8	-911.8	1097.2
BOTTOM SURFACE STRESSES	LC1	-967.	-743.	-68.	-299.	1.	-208.
	LC2	-700.	-504.	-47.	-209.	-1.	-150.
	LC3	-120.	-106.	51.	169.	109.	168.
	LC4	-7407.	-5384.8	-500.4	-2221.8	-8.2	-1588.
	LC5	-2071.	-1723.2	401.2	1255.8	1003.8	1337.6
MEMBRANE STRESSES	LC1	-24.	-467.	-42.5	-235.	-46.	-112.
	LC4	-180.4	-3273.	-223.1	-1812.8	-349.6	-912.4
	LC5	-60.8	-1161.6	228.9	1733.8	46.	1217.4
TOP BENDING STRESSES	LC1	943.	281.	25.5	64.	-47.	96.
	LC4	7226.6	2111.8	177.3	403.	-341.4	675.9
	LC5	2010.2	561.6	-172.3	478.	-957.8	-120.2
BOTTOM BENDING STRESSES	LC1	-943.	-281.	-25.5	-64.	47.	-96.
	LC4	-7226.6	-2111.8	-177.3	-409.	341.4	-675.6
	LC5	-2010.2	-561.6	172.3	-478.	-957.8	120.2

Note: 1 = Local element x stresses, 2 = Local element y stresses  
 LC = Load case, LC1 = Dead load, LC2 = E+W Wind, q=1.0PSF,  
 LC3 = N+S Wind, q=1.0PSF, LC4 = LC1 + 9.2LC2, LC5 = LC1 + 9.2LC3  
 Top Surface = Stack outside surface, Bottom Surface = Stack inside surface  
 All stresses are in PSI

B.1.1.23  
50 FT TALL BREACHED STEEL STACK MODEL CENTER OF ELEMENT STRESSES

ELEMENT NO	STRESS DIRECTION	119.1	119.2	120.1	120.2	121.1	121.2
TOP SURFACE STRESSES	LC1	-289.	93.	-439.	94.	-69.	-397.
	LC2	-199.	61.	-231.	65.	8.	-240.
	LC3	-122.	-1.	610.	120.	542.	42.
	LC4	-2119.8	654.2	-2564.2	692.	4.6	-2605.
	LC5	-1411.4	83.8	5173.	1198.	4926.6	-10.6
BOTTOM SURFACE STRESSES	LC1	523.	106.	915.	109.	-116.	-478.
	LC2	365.	74.	530.	57.	-129.	-319.
	LC3	87.	24.	-551.	-110.	-638.	-336.
	LC4	3881.	786.8	5791.	651.8	-1302.8	-3412.8
	LC5	1323.4	326.8	-4154.2	-903.	-5985.6	-3569.8
MEMBRANE STRESSES	LC1	117.	99.5	233.	101.5	-92.5	-437.5
	LC4	880.6	720.5	1613.4	671.9	-649.1	-3008.9
	LC5	-44.	205.3	509.4	147.5	-529.5	-1789.9
TOP BENDING STRESSES	LC1	-405.	-6.5	-677.	-7.5	23.5	40.5
	LC4	-3000.4	-66.3	-4177.6	20.1	653.7	403.9
	LC5	-1367.4	-121.5	4663.6	1050.5	5456.1	1779.3
BOTTOM BENDING STRESSES	LC1	406.	6.5	677.	7.5	-23.5	-40.5
	LC4	3000.4	66.3	4177.6	-20.1	-653.7	-403.9
	LC5	1367.4	121.5	-4663.6	-1050.5	-5456.1	-1779.3

Note: 1 = Local element x stresses, 2 = Local element y stresses  
 LC = Load case, LC1 = Dead load, LC2 = E+W Wind, q=1.0PSF,  
 LC3 = N+S Wind, q=1.0PSF, LC4 = LC1 + 9.2LC2, LC5 = LC1 + 9.2LC3  
 Top Surface = Stack outside surface, Bottom Surface = Stack inside surface  
 All stresses are in PSI

B.1.24  
50 FT TALL BREACHED STEEL STACK MODEL CENTER OF ELEMENT STRESSES

ELEMENT NO	STRESS DIRECTION	122.2	123.1	123.2	126.1	126.2
TOP SURFACE STRESSES	LC1	394.	660.	-675.	-108.	-1062.
	LC2	309.	483.	-413.	-75.	-740.
	LC3	450.	286.	-128.	31.	239.
	LC4	3236.8	5103.6	-4474.4	-798.	-7870.
	LC5	4534.	3291.2	-1852.6	177.2	1136.8
BOTTOM SURFACE STRESSES	LC1	-413.	-681.	-1015.	-154.	-1037.
	LC2	-324.	-498.	-659.	-105.	-704.
	LC3	-473.	-305.	-288.	23.	132.
	LC4	-3393.8	-5262.6	-7077.8	-1120.	-7513.8
	LC5	-4764.6	-3487.	-3664.6	57.6	177.4
MEMBRANE STRESSES	LC1	-9.5	-10.5	-845.	-131.	-1049.5
	LC4	-78.5	-79.5	-5776.2	-953.	-7691.9
	LC5	-115.3	-97.3	-2758.6	117.4	657.1
TOP BENDING STRESSES	LC1	403.5	670.5	170.	23.	-12.5
	LC4	3315.3	5183.1	1301.6	161.	-178.1
	LC5	4649.3	3389.1	906.	59.8	479.7
BOTTOM BENDING STRESSES	LC1	-403.5	-670.5	-170.	-23.	13.5
	LC4	-3315.3	-5183.1	-1301.6	-161.	178.1
	LC5	-4649.3	-3399.1	-906.	-59.8	-479.7

Note: 1 = Local element x stresses, 2 = Local element y stresses  
 LC = Load case, LC1 = Dead load, LC2 = E+W Wind, q=1.0PSF,  
 LC3 = N+S Wind, q=1.0PSF, LC4 = LC1 + 9.2LC2, LC5 = LC1 + 9.2LC3  
 Top Surface = Stack outside surface, Bottom Surface = Stack inside surface  
 All stresses are in PSI

B.1.25  
50 FT TALL BREECHEED STEEL STACK MODEL CENTER OF ELEMENT STRESSES

ELEMENT NO	STRESS DIRECTION	127.1	127.2	128.1	128.2	129.1	129.2
TOP SURFACE STRESSES	LC1	45.	-938.	159.	-923.	1493.	-1404.
	LC2	52.	-654.	156.	-646.	936.	-883.
	LC3	-132.	177.	-238.	187.	972.	-185.
	LC4	523.4	-6954.8	1594.2	-6866.2	10104.2	-9527.6
	LC5	-1169.4	690.4	-2030.6	797.4	10435.4	-3106.
BOTTOM SURFACE STRESSES	LC1	-101.	-1094.	43.	-792.	-1653.	-1695.
	LC2	-84.	-757.	-44.	-564.	-1033.	-1063.
	LC3	72.	189.	390.	241.	-1120.	-510.
	LC4	-873.8	-8058.4	-361.8	-5980.8	-11156.6	-11474.6
	LC5	561.4	644.8	3631.	1425.2	-11957.	-6387.
MEMBRANE STRESSES	LC1	-28.	-1016.	101.	-857.5	-80.	-1549.5
	LC4	-175.2	-7506.6	616.2	-6223.5	-526.2	-10501.1
	LC5	-304.	667.6	800.2	1111.9	-760.8	-4746.5
TOP BENDING STRESSES	LC1	73.	78.	58.	-65.5	1573.	145.5
	LC4	698.6	551.8	978.	-442.7	10630.4	973.5
	LC5	-865.4	22.8	-2830.8	-313.9	11196.2	1640.5
BOTTOM BENDING STRESSES	LC1	-73.	-78.	-58.	65.5	-1573.	-145.5
	LC4	-698.6	-551.8	-978.	442.7	-10630.4	-973.5
	LC5	865.4	-22.8	2830.8	313.9	-11196.2	-1640.5

Note: 1 = Local element x stresses, 2 = Local element y stresses  
 LC = Load case, LC1 = Dead load, LC2 = E+W Wind, q=1.0PSF,  
 LC3 = N+S Wind, q=1.0PSF, LC4 = LC1 + 9.2LC2, LC5 = LC1 + 9.2LC3  
 Top Surface = Stack outside surface, Bottom Surface = Stack inside surface  
 All stresses are in PSI.

B.1.26  
50 FT TALL BREACHED STEEL STACK MODEL CENTER OF ELEMENT STRESSES

ELEMENT NO	STRESS DIRECTION	130.1	130.2	131.1	131.2	132.1	132.2
TOP SURFACE STRESSES	LC1	674.	-1473.	927.	-1449.	253.	-1320.
	LC2	433.	-926.	226.	-901.	190.	-816.
	LC3	709.	-112.	500.	-187.	364.	-187.
	LC4	4657.6	-9992.2	2406.2	-9798.2	2001.8	-8827.2
	LC5	7196.8	-2503.4	4927.	-3169.4	3601.8	-3040.4
BOTTOM SURFACE STRESSES	LC1	-764.	-1751.	-333.	-1556.	-240.	-1360.
	LC2	-490.	-1098.	-230.	-979.	-184.	-851.
	LC3	-727.	-554.	-522.	-439.	-378.	-401.
	LC4	-5272.	-11852.6	-2449.	-10562.8	-1932.8	-9189.2
	LC5	-7452.4	-6847.8	-5135.4	-5594.8	-3717.6	-5049.2
MEMBRANE STRESSES	LC1	-45.	-1612.	-3.	-1502.5	6.5	-1340.
	LC4	-607.2	-10922.4	-21.4	-10150.5	34.1	-9008.2
	LC5	-127.8	-4675.6	-104.2	-4382.1	-57.9	-4044.8
TOP BENDING STRESSES	LC1	719.	139.	-3.	53.5	246.5	20.
	LC4	4964.8	930.2	2427.6	412.3	1966.9	181.
	LC5	7324.6	2172.2	5031.2	1212.7	3659.7	1004.4
BOTTOM BENDING STRESSES	LC1	-719.	-139.	-330.	-53.5	-246.5	-20.
	LC4	-4964.8	-930.2	-2427.6	-412.3	-1966.9	-181.
	LC5	-7324.6	-2172.2	-5031.2	-1212.7	-3659.7	-1004.4

Note: 1 = Local element x stresses, 2 = Local element y stresses  
 LC = Load case, LC1 = Dead load, LC2 = E+W Wind, q=1.0PSF,  
 LC3 = N+S Wind, q=1.0PSF, LC4 = LC1 + 9.2LC2, LC5 = LC1 + 9.2LC3  
 Top Surface = Stack outside surface, Bottom Surface = Stack inside surface  
 All stresses are in PSI

B.1.1.27

50 FT TALL BREACHED STEEL STACK MODEL CENTER OF ELEMENT STRESSES

ELEMENT NO.	STRESS DIRECTION	135.1	135.2	136.1	136.2	137.1	137.2
TOP SURFACE STRESSES	LC1	-256.	-1998.	152.	-2079.	271.	-2795.
	LC2	-168.	-1332.	119.	-1383.	214.	-1845.
	LC3	-5.	-26.	-40.	10.	-110.	-186.
	LC4	-1801.6	-14252.4	1246.8	-14802.6	2299.8	-19769.
	LC5	-302.	-2237.2	-216.	-1987.	741.	-4506.2
BOTTOM SURFACE STRESSES	LC1	-184.	-1807.	-24.	-2098.	-197.	-2007.
	LC2	-125.	-1188.	-30.	-1381.	-172.	-1358.
	LC3	1.	-61.	0.	-94.	129.	203.
	LC4	-1334.	-12736.6	-300.	-14803.2	-1779.4	-14500.6
	LC5	-174.8	-2368.2	-24.	-2962.8	989.8	-139.4
MEMBRANE STRESSES	LC1	-220.	-1902.5	64.	-2088.5	37.	-2401.
	LC4	-1567.8	-13494.5	473.4	-14802.9	230.2	-17134.8
	LC5	-238.4	-2302.7	-120.	-2474.9	124.4	-2322.8
TOP BENDING STRESSES	LC1	-36.	-95.5	88.	9.5	234.	-394.
	LC4	-233.8	-757.9	773.4	0.3	2009.6	-2634.2
	LC5	-63.6	65.5	-96.	487.9	-865.4	-2183.4
BOTTOM BENDING STRESSES	LC1	36.	95.5	-88.	-9.5	-234.	394.
	LC4	233.8	757.9	-773.4	-0.3	-2009.6	2634.2
	LC5	63.6	-65.5	96.	-487.9	865.4	2183.4

Note: 1 = Local element x stresses, 2 = Local element y stresses  
 LC = Load case, LC1 = Dead load, LC2 = E-W Wind, q=1.0PSF,  
 LC3 = N-S Wind, q=1.0PSF, LC4 = LC1 + 9.2LC2, LC5 = LC1 + 9.2LC3  
 Top Surface = Stack outside surface, Bottom Surface = Stack inside surface  
 All stresses are in PSI

B.1.28

50 FT TALL BREECHEED STEEL STACK MODEL CENTER OF ELEMENT STRESSES

ELEMENT NO	STRESS DIRECTION	138.1	138.2	139.1	139.2	140.1	140.2
TOP SURFACE STRESSES	LC1	259.	-3437.	190.	-3255.	59.	-3574.
	LC2	192.	-2272.	127.	-2140.	36.	-2322.
	LC3	-44.	-151.	10.	-121.	1.	-274.
	LC4	2025.4	-24339.4	1358.4	-22943.	390.2	-24936.4
	LC5	-145.8	-4826.2	282.	-4368.2	123.4	-6094.8
BOTTOM SURFACE STRESSES	LC1	-391.	-3201.	-189.	-3822.	-66.	-3992.
	LC2	-296.	-2180.	-125.	-2327.	-40.	-2582.
	LC3	145.	447.	-26.	17.	-6.	-296.
	LC4	-3114.2	-23257.	-1339.	-27070.4	-434.	-27746.4
	LC5	952.2	911.4	-428.2	-3886.4	-121.2	-6715.2
MEMBRANE STRESSES	LC1	65.	-3319.	0.5	-3538.5	-3.5	-3783.
	LC4	-544.4	-23798.2	4.7	-25006.7	-21.9	-26341.4
	LC5	403.2	-1957.4	73.1	-4127.3	1.1	-6405.
TOP BENDING STRESSES	LC1	325.	-118.	189.5	283.5	62.5	203.
	LC4	2569.8	-541.2	1348.7	2063.7	412.1	1405.
	LC5	-549.	-2668.8	355.1	-240.9	122.3	310.2
BOTTOM BENDING STRESSES	LC1	-325.	118.	-189.5	-283.5	-62.5	-209.
	LC4	-2569.8	541.2	-1348.7	-2063.7	-412.1	-1405.
	LC5	549.	-2668.8	-355.1	240.9	-122.3	-310.2

Note: 1 = Local element x stresses, 2 = Local element y stresses  
 LC = Load case, LC1 = Dead load, LC2 = E+W Wind, q=1.0PSF,  
 LC3 = N+S Wind, q=1.0PSF, LC4 = LC1 + 9.2LC2, LC5 = LC1 + 9.2LC3  
 Top Surface = Stack outside surface, Bottom Surface = Stack inside surface  
 All stresses are in PSI

B.I.29  
50 FT TALL BREECHEED STEEL STACK MODEL CENTER OF ELEMENT STRESSES.

ELEMENT NO	STRESS DIRECTION	141.1	141.2	142.1	142.2	143.1	143.2
TOP SURFACE STRESSES	LC1	330.	-3449.	859.	-3803.	1282.	-2882.
	LC2	203.	-2222.	525.	-2420.	780.	-1822.
	LC3	43.	-374.	266.	-371.	533.	-213.
	LC4	2197.6	-23891.4	5688.	-26067.	8458.	-19644.4
	LC5	725.6	-6889.8	3305.2	-7216.2	6185.6	-4841.6
BOTTOM SURFACE STRESSES	LC1	-293.	-4647.	-1296.	-4913.	-1338.	-2999.
	LC2	-180.	-2958.	-797.	-3082.	-814.	-1862.
	LC3	-30.	-636.	-392.	-1094.	-563.	-748.
	LC4	-1949.	-31860.6	-8628.4	-33267.4	-8826.8	-20129.4
	LC5	-569.	-10498.2	-4902.4	-14977.8	46517.6	-9880.8
MEMBRANE STRESSES	LC1	18.5	-4048.	-219.	-4358.	-28.	-2940.5
	LC4	124.3	-27876.	-1470.2	-29667.2	-184.4	-19886.9
	LC5	78.3	-8694.	-798.6	-11097.	-166.	-7361.1
	LC1	311.5	599.	1077.	555.	1310.	58.5
	LC4	2073.3	3984.6	7158.2	3600.2	8642.4	422.5
BENDING STRESSES	LC5	647.3	1804.2	4103.8	3880.8	6251.6	2519.5
	LC1	-311.5	-599.	-1077.	-555.	-1310.	-58.5
	LC4	-2073.3	-3984.6	-7158.2	-3600.2	-8642.4	-422.5
	LC5	-647.3	-1804.2	-4103.8	-3880.8	-6251.6	-2519.5

Note: 1 = Local element x stresses, 2 = Local element y stresses  
 LC = Load case, LC1 = Dead load, LC2 = E+W Wind, q=1.0PSF,  
 LC3 = N+S Wind, q=1.0PSF, LC4 = LC1 + 9.2LC2, LC5 = LC1 + 9.2LC3  
 Top Surface = Stack outside surface, Bottom Surface = Stack inside surface  
 All stresses are in PSI



B.1.30  
50 FT TALL BREACHED STEEL STACK MODEL CENTER OF ELEMENT STRESSES

ELEMENT NO	STRESS DIRECTION	144.1	144.2	145.1	145.2	146.1	146.2
TOP SURFACE STRESSES	LC1	709.	-2363.	155.	-2039.	-136.	-1774.
	LC2	420.	-1475.	75.	-1262.	-99.	-1088.
	LC3	497.	-291.	423.	-224.	340.	-247.
	LC4	4573.	-15933.	845.	-13649.4	-1046.8	-11783.6
	LC5	5281.4	-5040.2	4046.6	-4099.8	2992.	-4046.4
BOTTOM SURFACE STRESSES	LC1	-648.	-2374.	-135.	-1939.	165.	-1577.
	LC2	-377.	-1456.	-59.	-1181.	117.	-946.
	LC3	-481.	-498.	7436.	-489.	-350.	-442.
	LC4	-4116.4	-15769.2	-677.8	-12804.2	1241.4	-10288.2
	LC5	-5073.2	-6955.6	-4146.2	-6497.8	-3055.	-5643.4
MEMBRANE STRESSES	LC1	30.5	-2368.5	10.	-1989.	14.5	-1675.5
	LC4	228.3	-15851.1	83.6	-13226.8	97.3	-11031.9
	LC5	104.1	-5997.9	-49.8	-5268.8	-31.5	-4844.9
TOP BENDING STRESSES	LC1	678.5	5.5	145.	-50.	-150.5	-98.5
	LC4	4344.	-81.9	761.4	-4226.6	-1144.1	-751.7
	LC5	5177.3	957.7	4096.4	1169.	3023.5	798.5
BOTTOM BENDING STRESSES	LC1	-678.5	-5.5	-145.	50.	150.5	98.5
	LC4	-4344.	81.9	-761.4	422.6	1144.1	751.7
	LC5	-5177.3	-957.7	-4096.4	-1169.	-3023.5	-798.5

Note: 1 = Local element x stresses, 2 = Local element y stresses  
 LC = Load case, LC1 = Dead load, LC2 = E+W Wind, q=1.0PSF,  
 LC3 = N+S Wind, q=1.0PSF, LC4 = LC1 + 9.2LC2, LC5 = LC1 + 9.2LC3  
 Top Surface = Stack outside surface, Bottom Surface = Stack inside surface  
 All stresses are in PSI

B.1.1.31  
50 FT TALL BREACHED STEEL STACK MODEL CENTER OF ELEMENT STRESSES

ELEMENT NO	STRESS DIRECTION	149.1	149.2	150.1	150.2	151.1	151.2
TOP SURFACE STRESSES	LC1	-303.	-2581.	212.	-2578.	270.	-2718.
	LC2	-189.	-1636.	140.	-1631.	186.	-1671.
	LC3	-51.	-415.	39.	-432.	71.	-605.
	LC4	-2041.8	-17632.2	1500.	-17583.2	1981.2	-18091.2
	LC5	-772.2	-6399.	570.8	-6552.4	923.2	-8284.
BOTTOM SURFACE STRESSES	LC1	-269.	-2305.	-64.	-2592.	-327.	-2507.
	LC2	-170.	-1442.	-50.	-1627.	-218.	-1553.
	LC3	-37.	-388.	-12.	-464.	-94.	-456.
	LC4	-1833.	-15571.4	-524.	-17560.4	-2332.6	-16794.6
	LC5	-609.4	-5874.6	-174.4	-6860.8	-1191.8	-6702.2
MEMBRANE STRESSES	LC1	-286.	-2443.	74.	-2585.	-28.5	-2612.5
	LC4	-1937.4	-16601.8	488.	-17571.8	-175.7	-17442.9
	LC5	-690.8	-6136.8	198.2	-6706.6	-134.3	-7493.1
TOP BENDING STRESSES	LC1	-17.	-138.	138.	7.	298.5	-105.5
	LC4	-104.4	-1030.4	1012.	-11.4	2156.9	-648.3
	LC5	-81.4	-262.2	372.6	154.2	1057.5	-790.9
BOTTOM BENDING STRESSES	LC1	17.	138.	-138.	-7.	-298.5	105.5
	LC4	104.4	1030.4	-1012.	11.4	-2156.9	648.3
	LC5	81.4	262.2	-372.6	-154.2	-1057.5	790.9

Note: 1 = Local element x stresses, 2 = Local element y stresses  
 LC = Load case, LC1 = Dead load, LC2 = E+W Wind, q=1.0PSF,  
 LC3 = N+S Wind, q=1.0PSF, LC4 = LC1 + 9.2LC2, LC5 = LC1 + 9.2LC3  
 Top Surface = Stack outside surface, Bottom Surface = Stack inside surface  
 All stresses are in PSI

B.1.32  
50 FT TALL BREACHED STEEL STACK MODEL CENTER OF ELEMENT STRESSES

ELEMENT NO	STRESS DIRECTION	152.1	152.2	153.1	153.2	154.1	154.2
TOP SURFACE STRESSES	LC1	343.	-2409.	213.	-2328.	169.	22218.
	LC2	223.	-1471.	134.	-1421.	98.	-1365.
	LC3	90.	-668.	56.	-535.	19.	-417.
	LC4	2394.6	-15942.2	1445.8	-15401.2	1070.6	-14776.
	LC5	1171.	-8554.6	728.2	-7250.	343.8	-6054.4
BOTTOM SURFACE STRESSES	LC1	-425.	-2635.	-249.	-2622.	-160.	-2547.
	LC2	-279.	-1640.	-155.	-1615.	-97.	-1569.
	LC3	-119.	-504.	-55.	-473.	-16.	-429.
	LC4	-2991.8	-17723.	-1675.	-17480.	-1052.4	-16981.8
	LC5	-1519.8	-7271.8	-755.	-6973.6	-307.2	-6493.8
MEMBRANE STRESSES	LC1	-41.	-2522.	-18.	-2475.	4.5	-2382.5
	LC4	-298.6	-16832.6	-114.6	-16440.6	9.1	-15878.9
	LC5	-174.4	-7913.2	-13.4	-7111.8	18.3	-6274.1
TOP BENDING STRESSES	LC1	384.	113.	231.	147.	164.5	164.5
	LC4	2693.2	890.4	1560.4	1039.4	1061.5	1102.9
	LC5	1345.4	-641.4	741.6	-138.2	325.5	219.7
BOTTOM BENDING STRESSES	LC1	-384.	-113.	-231.	-147.	-164.5	-164.5
	LC4	-2693.2	-890.4	-1560.4	-1039.4	-1061.5	-1102.9
	LC5	-1345.4	641.4	-741.6	138.2	-325.5	-219.7

Note: 1 = Local element x stresses, 2 = Local element y stresses  
 LC = Load case, LC1 = Dead load, LC2 = E+W Wind, q=1.0PSF,  
 LC3 = N+S Wind, q=1.0PSF, LC4 = LC1 + 9.2LC2, LC5 = LC1 + 9.2LC3  
 Top Surface = Stack outside surface, Bottom Surface = Stack inside surface  
 All stresses are in PSI

B.1.33

50 FT TALL BREACHED STEEL STACK MODEL CENTER OF ELEMENT STRESSES

ELEMENT NO	STRESS DIRECTION	155.1	155.2	156.1	156.2	157.1	157.2
TOP SURFACE STRESSES	LC1	299.	-1960.	607.	-1522.	655.	-1846.
	LC2	179.	-1186.	357.	-927.	378.	-1118.
	LC3	5.	-308.	61.	-106.	127.	-175.
	LC4	1945.8	-12871.2	3891.4	-10050.4	4132.6	-12131.6
	LC5	345.	-4793.6	1168.2	-2497.2	1823.4	-3456.
BOTTOM SURFACE STRESSES	LC1	-375.	-2549.	-671.	-2388.	-727.	-2377.
	LC2	-225.	-1546.	-402.	-1445.	-416.	-1411.
	LC3	-24.	-412.	-42.	-364.	-119.	-420.
	LC4	-2445.	-16772.2	-4369.4	-15682.	-4554.2	-15358.2
	LC5	-595.8	-6339.4	-1057.4	-5736.8	-1821.8	-6241.
MEMBRANE STRESSES	LC1	-38.	-2254.5	-32.	-1955.	-36.	-2111.5
	LC4	-249.6	-14821.7	-239.	-12866.2	-210.8	-13744.9
	LC5	-125.4	-5566.5	55.4	-4117.	0.8	-4848.5
TOP BENDING STRESSES	LC1	337.	294.5	639.	433.	691.	265.5
	LC4	2195.4	1950.5	-4130.4	2815.8	4343.4	1613.3
	LC5	470.4	772.9	1112.8	1619.8	1822.6	1392.5
BOTTOM BENDING STRESSES	LC1	-337.	-294.5	-639.	-433.	-691.	-265.5
	LC4	-2195.4	-1950.5	-4130.4	-2815.8	-4343.4	-1613.3
	LC5	-470.4	-772.9	-1112.8	-1619.8	-1822.6	-1392.5

Note: 1 = Local element x stresses, 2 = Local element y stresses  
 LC = Load case, LC1 = Dead load, LC2 = E+W Wind, q=1.0PSF,  
 LC3 = N+S Wind, q=1.0PSF, LC4 = LC1 + 9.2LC2, LC5 = LC1 + 9.2LC3  
 Top Surface = Stack outside surface, Bottom Surface = Stack inside surface  
 All stresses are in PSI

B.1.34  
50 FT TALL BREECHEDED STEEL STACK MODEL CENTER OF ELEMENT STRESSES

ELEMENT NO	STRESS DIRECTION	158.1	158.2	159.1	159.2	160.1	160.2
TOP SURFACE STRESSES	LC1	413.	-2024.	31.	-2018.	-343.	-1809.
	LC2	215.	-1228.	-24.	-1196.	-269.	-1077.
	LC3	205.	-271.	243.	-293.	253.	-302.
	LC4	2400.2	-13321.6	-189.8	-13021.2	-2817.8	-11717.4
	LC5	2299.	-4517.2	2266.6	-4768.8	1984.6	-4587.4
BOTTOM SURFACE STRESSES	LC1	-431.	-2129.	-36.	-1834.	361.	-1582.
	LC2	-228.	-1259.	28.	-1046.	281.	-893.
	LC3	-203.	-407.	-257.	-468.	-256.	-454.
	LC4	-2528.6	-13711.8	221.6	-11457.2	2946.2	-9797.6
	LC5	-2298.6	-5873.4	-2400.4	-6121.2	-1994.2	-5758.8
MEMBRANE STRESSES	LC1	-9.	-2076.5	2.5	-1926.	9.	-1695.5
	LC4	-64.2	-13516.7	15.9	-12239.2	64.2	-10757.5
	LC5	0.2	-5195.3	-66.9	-5445.	-4.8	-5172.1
TOP BENDING STRESSES	LC1	422.	52.5	33.5	-92.	352.	-113.5
	LC4	2464.4	195.1	-205.7	-782.	-2882.	-959.9
	LC5	2298.8	678.1	2333.5	676.2	1989.4	585.7
BOTTOM BENDING STRESSES	LC1	-422.	-52.5	-33.5	92.	-352.	113.5
	LC4	-2464.4	-195.1	205.7	782.	2882.	959.9
	LC5	-2298.8	-678.1	-2333.5	-676.2	-1989.4	-585.7

Note: 1 = Local element x stresses, 2 = Local element y stresses  
 LC = Load case, LC1 = Dead load, LC2 = E+W Wind, q=1.0PSF,  
 LC3 = N+S Wind, q=1.0PSF, LC4 = LC1 + 9.2LC2, LC5 = LC1 + 9.2LC3  
 Top Surface = Stack outside surface, Bottom Surface = Stack inside surface  
 All stresses are in PSI

B.1.35  
50 FT TALL BREACHED STEEL STACK MODEL CENTER OF ELEMENT STRESSES

ELEMENT NO	STRESS DIRECTION	163.1	163.2	164.1	164.2	165.1	165.2
TOP SURFACE STRESSES	LC1	-295.	-2556.	243.	-2020.	92.	-2108.
	LC2	-179.	-1505.	148.	-1116.	36.	-1216.
	LC3	-79.	-799.	110.	-722.	125.	-735.
	LC4	-1941.8	-16402.	1604.6	-12287.3	423.2	-13295.2
	LC5	-1021.8	-9906.8	1255.	-8662.4	1242.	-8870.
BOTTOM SURFACE STRESSES	LC1	-846.	-2319.	-92.	-2200.	-178.	-2091.
	LC2	-205.	-1346.	-49.	-1217.	-107.	-1201.
	LC3	-106.	-727.	-67.	-776.	-148.	-738.
	LC4	-2232.	-14702.2	-542.8	-13396.4	-1162.4	-13140.2
	LC5	-1321.2	-9007.4	-708.4	-9339.2	-1599.6	-8880.6
MEMBRANE STRESSES	LC1	-320.5	-2437.5	75.5	-2110.	-43.	-2099.5
	LC4	-2086.9	-15552.1	530.9	-12841.8	-369.6	-13217.7
	LC5	-1171.5	-8757.1	273.9	-9000.8	-148.8	-8875.3
TOP BENDING STRESSES	LC1	25.5	-118.5	167.5	90.	135.	-8.5
	LC4	145.1	-849.9	1073.7	554.6	792.8	-77.5
	LC5	149.7	-449.7	981.7	338.4	1390.8	5.3
BOTTOM BENDING STRESSES	LC1	-25.5	118.5	-167.5	-90.	-135.	8.5
	LC4	-145.1	849.9	-1073.7	-554.6	-792.8	77.5
	LC5	-149.7	-449.7	-981.7	-338.4	-1390.8	-5.3

Note: 1 = Local element x stresses, 2 = Local element y stresses  
 LC = Load case, LC1 = Dead load, LC2 = E+W Wind, q=1.0PSF,  
 LC3 = N+S Wind, q=1.0PSF, LC4 = LC1 + 9.2LC2, LC5 = LC1 + 9.2LC3  
 Top Surface = Stack outside surface, Bottom Surface = Stack inside surface  
 All stresses are in PSI

B.1.36  
50 FT TALL BREACHED STEEL STACK MODEL CENTER OF ELEMENT STRESSES

ELEMENT NO	STRESS DIRECTION	166.1	166.2	167.1	167.2	168.1	168.2
TOP SURFACE STRESSES	LC1	82.	-1782.	-55.	-1700.	-166.	-1539.
	LC2	38.	-948.	-61.	-921.	-119.	-791.
	LC3	136.	-698.	80.	-644.	-35.	-612.
	LC4	431.6	-10503.6	-616.2	-10173.2	-1260.8	-8816.2
	LC5	1333.2	-8203.6	681.	-7624.8	-488.	-7169.4
BOTTOM SURFACE STRESSES	LC1	-69.	-1877.	44.	-1770.	171.	-1552.
	LC2	-12.	-995.	45.	-959.	134.	-789.
	LC3	-170.	-738.	-85.	-657.	25.	-585.
	LC4	-179.4	-11031.	458.	-10592.8	1403.8	-8810.8
	LC5	-1633.	-8666.6	-738.	-7814.4	401.	-6934.
MEMBRANE STRESSES	LC1	6.5	-1829.5	-5.5	-1735.	2.5	-1545.5
	LC4	126.1	-10767.3	-79.1	-91383.	71.5	-8813.5
	LC5	-149.9	-8435.1	-28.5	-7719.6	-43.5	-7051.5
TOP BENDING STRESSES	LC1	75.5	47.5	149.5	35.	-168.5	6.5
	LC4	305.5	263.7	-537.1	209.8	-1332.3	-2.7
	LC5	1483.1	234.5	709.5	94.8	-444.5	-117.7
BOTTOM BENDING STRESSES	LC1	-75.5	-47.5	49.5	-35.	168.5	-6.5
	LC4	-305.5	-263.7	-537.1	-209.8	1332.3	2.7
	LC5	-1483.1	-231.5	-709.5	-94.8	444.5	117.7

Note: 1 = Local element x stresses, 2 = Local element y stresses  
 LC = Load case, LC1 = Dead load, LC2 = E-W Wind, q=1.0PSF,  
 LC3 = N+S Wind, q=1.0PSF, LC4 = LC1 + 9.2LC2, LC5 = LC1 + 9.2LC3  
 Top Surface = Stack outside surface, Bottom Surface = Stack inside surface  
 All stresses are in PSI

B.1.37  
50 FT TALL BREACHED STEEL STACK MODEL CENTER OF ELEMENT STRESSES

ELEMENT NO	STRESS DIRECTION	169.1	169.2	170.1	170.2	171.1	171.2
TOP SURFACE STRESSES	LC1	-222.	-1365.	-220.	-1281.	-150.	-1270.
	LC2	-168.	-734.	-159.	-634.	-132.	-661.
	LC3	-121.	-513.	-191.	-517.	-163.	-439.
	LC4	-1767.6	-8117.8	-1682.8	-7113.8	-1364.4	-7351.2
	LC5	-1335.2	-6084.6	-1977.2	-6037.4	-1649.6	-5308.3
BOTTOM SURFACE STRESSES	LC1	249.	-1407.2	270.	-1333.	190.	-1319.
	LC2	172.	-743.	201.	-646.	146.	-669.
	LC3	145.	-457.	212.	-431.	203.	-370.
	LC4	1831.4	-8242.6	2119.2	-7276.2	1533.2	-7463.6
	LC5	1583.	-5611.4	2220.4	-5298.2	2112.8	-4722.
MEMBRANE STRESSES	LC1	13.5	-1386.	35.	-1307.	20.	-1294.
	LC4	31.9	-8180.2	218.2	-7195.	84.4	-7407.4
	LC5	123.9	-5848.	121.6	-5667.8	231.6	-5015.4
TOP BENDING STRESSES	LC1	-235.5	21.	-245.	25.	-177.	24.
	LC4	-1799.5	62.4	-1901.	81.2	-1448.8	56.2
	LC5	-1459.1	-236.6	-2098.8	-369.6	-1881.2	-293.4
BOTTOM BENDING STRESSES	LC1	235.5	-21.	245.	-26.	177.	-24.
	LC4	1799.5	-62.4	1901.	-81.2	1448.8	-56.2
	LC5	1459.1	236.6	2098.8	369.6	1881.2	293.4

Note: 1 = Local element x stresses, 2 = Local element y stresses  
 LC = Load case, LC1 = Dead load, LC2 = E+W Wind, q=1.0PSF,  
 LC3 = N+S Wind, q=1.0PSF, LC4 = LC1 + 9.2LC2, LC5 = LC1 + 9.2LC3  
 Top Surface = Stack outside surface, Bottom Surface = Stack inside surface  
 All stresses are in PSI



B.1.38  
50 FT TALL BREACHED STEEL STACK MODEL CENTER OF ELEMENT STRESSES

ELEMENT NO	STRESS DIRECTION	172.1	172.2	173.1	173.2	174.1	174.2
TOP SURFACE STRESSES	LC1	-93.	-1347.	-152.	-1477.	-366.	-1434.
	LC2	-90.	-680.	-154.	-820.	-286.	-725.
	LC3	-101.	-470.	12.	-404.	78.	-453.
	LC4	-921.	-7603.	-1568.8	-9021.	-2997.2	-8104.
	LC5	-1022.2	-5671.	-41.6	-5193.8	351.6	-5656.8
BOTTOM SURFACE STRESSES	LC1	89.	-1385.	139.	-1309.	363.	-1305.
	LC2	83.	-673.	123.	-670.	300.	-605.
	LC3	93.	-434.	3.	-430.	85.	-503.
	LC4	878.6	-7631.8	1316.6	-7472.	3123.	-6871.
	LC5	970.6	-5377.8	212.6	-5264.	-414.	-5932.8
MEMBRANE STRESSES	LC1	-12.	-1363.	-6.5	-1392.5	1.	-1369.5
	LC4	-21.2	-7617.4	-126.1	-8246.5	65.4	-7487.5
	LC5	-25.8	-5524.4	85.5	-5228.9	-31.2	-5794.5
TOP BENDING STRESSES	LC1	81.	13.	-145.5	-84.5	-367.	-64.5
	LC4	-899.8	14.4	-1442.7	-774.5	-3062.6	-616.5
	LC5	-996.4	-146.6	-127.1	95.1	382.8	137.9
BOTTOM BENDING STRESSES	LC1	81.	-13.	145.5	84.5	367.	64.5
	LC4	899.8	-14.4	1442.7	774.5	3062.6	616.5
	LC5	996.4	146.6	127.1	-95.1	-382.8	-137.9

Note: 1 = Local element x stresses, 2 = Local element y stresses  
 LC = Load case, LC1 = Dead load, LC2 = E+W Wind, q=1.0PSF,  
 LC3 = N+S Wind, q=1.0PSF, LC4 = LC1 + 9.2LC2, LC5 = LC1 + 9.2LC3  
 Top Surface = Stack outside surface, Bottom Surface = Stack inside surface  
 All stresses are in PSI

B.2: MODEL II ELEMENT STRESSES FOR LOAD CASE 1+2 FOR  
ELEMENTS ABOUT AND ADJACENT TO THE BREACH

The following tables reflect the analysis stresses due  
to dead load self weight and East+West Wind at  $q = 9.2$  PSF.

B.2.1 200 FT TALL BREECHEED STEEL STACK  
MODEL CENTER OF ELEMENT STRESSES

ELEMENT NUMBER .....	13.	PRT
TOP SURF STRESS IN X .....	24.	PRT
TOP SURF STRESS IN Y .....	2110.	PRT
BOT SURF STRESS IN X .....	398.	PRT
BOT SURF STRESS IN Y .....	1107.	PRT
MEMBRANE STRESS IN X .....	211.	PRT
TOP SURF BENDING STRESS IN X .....	-187.	PRT
BOT SURF BENDING STRESS IN X .....	187.	PRT
MEMBRANE STRESS IN Y .....	1608.5	PRT
TOP SURF BENDING STRESS IN Y .....	501.5	PRT
BOT SURF BENDING STRESS IN Y .....	-501.5	PRT
ELEMENT NUMBER .....	14.	PRT
TOP SURF STRESS IN X .....	-61.	PRT
TOP SURF STRESS IN Y .....	-751.	PRT
BOT SURF STRESS IN X .....	-352.	PRT
BOT SURF STRESS IN Y .....	-1335.	PRT
MEMBRANE STRESS IN X .....	-206.5	PRT
TOP SURF BENDING STRESS IN X .....	145.5	PRT
BOT SURF BENDING STRESS IN X .....	-145.5	PRT
MEMBRANE STRESS IN Y .....	-1043.	PRT
TOP SURF BENDING STRESS IN Y .....	292.	PRT
BOT SURF BENDING STRESS IN Y .....	-292.	PRT
ELEMENT NUMBER .....	15.	PRT
TOP SURF STRESS IN X .....	-574.	PRT
TOP SURF STRESS IN Y .....	-5437.	PRT
BOT SURF STRESS IN X .....	-807.	PRT
BOT SURF STRESS IN Y .....	-5331.	PRT
MEMBRANE STRESS IN X .....	-690.5	PRT
TOP SURF BENDING STRESS IN X .....	116.5	PRT
BOT SURF BENDING STRESS IN X .....	-116.5	PRT
MEMBRANE STRESS IN Y .....	-5384.	PRT
TOP SURF BENDING STRESS IN Y .....	-53.	PRT
BOT SURF BENDING STRESS IN Y .....	53.	PRT

B.2.2            200 FT TALL BREECED STEEL STACK  
 MODEL CENTER OF ELEMENT STRESSES

ELEMENT NUMBER .....	16.	PRT
TOP SURF STRESS IN X .....	-1382.	PRT
TOP SURF STRESS IN Y .....	-10387.	PRT
BOT SURF STRESS IN X .....	-943.	PRT
BOT SURF STRESS IN Y .....	-9446.	PRT

MEMBRANE STRESS IN X .....	-1162.5	PRT
TOP SURF BENDING STRESS IN X .....	-219.5	PRT
BOT SURF BENDING STRESS IN X .....	219.5	PRT

MEMBRANE STRESS IN Y .....	-9906.5	PRT
TOP SURF BENDING STRESS IN Y .....	-460.5	PRT
BOT SURF BENDING STRESS IN Y .....	460.5	PRT

ELEMENT NUMBER .....	17.	PRT
TOP SURF STRESS IN X .....	-1721.	PRT
TOP SURF STRESS IN Y .....	-13682.	PRT
BOT SURF STRESS IN X .....	-1309.	PRT
BOT SURF STRESS IN Y .....	-12253.	PRT

MEMBRANE STRESS IN X .....	-1515.	PRT
TOP SURF BENDING STRESS IN X .....	-206.	PRT
BOT SURF BENDING STRESS IN X .....	206.	PRT

MEMBRANE STRESS IN Y .....	-12967.5	PRT
TOP SURF BENDING STRESS IN Y .....	-714.5	PRT
BOT SURF BENDING STRESS IN Y .....	714.5	PRT

ELEMENT NUMBER .....	18.	PRT
TOP SURF STRESS IN X .....	-1747.	PRT
TOP SURF STRESS IN Y .....	-15157.	PRT
BOT SURF STRESS IN X .....	-2051.	PRT
BOT SURF STRESS IN Y .....	-13595.	PRT

MEMBRANE STRESS IN X .....	-1899.	PRT
TOP SURF BENDING STRESS IN X .....	152.	PRT
BOT SURF BENDING STRESS IN X .....	-152.	PRT

MEMBRANE STRESS IN Y .....	-14376.	PRT
TOP SURF BENDING STRESS IN Y .....	-781.	PRT
BOT SURF BENDING STRESS IN Y .....	781.	PRT

B.2.3 200 FT TALL BREACHED STEEL STACK  
MODEL CENTER OF ELEMENT STRESSES

ELEMENT NUMBER .....	32.	PRT
TOP SURF STRESS IN X .....	-639.	PRT
TOP SURF STRESS IN Y .....	1224.	PRT
BOT SURF STRESS IN X .....	801.	PRT
BOT SURF STRESS IN Y .....	933.	PRT
MEMBRANE STRESS IN X .....	81.	PRT
TOP SURF BENDING STRESS IN X .....	-720.	PRT
BOT SURF BENDING STRESS IN X .....	720.	PRT
MEMBRANE STRESS IN Y .....	1078.5	PRT
TOP SURF BENDING STRESS IN Y .....	145.5	PRT
BOT SURF BENDING STRESS IN Y .....	-145.5	PRT
ELEMENT NUMBER .....	33.	PRT
TOP SURF STRESS IN X .....	-420.	PRT
TOP SURF STRESS IN Y .....	151.	PRT
BOT SURF STRESS IN X .....	17.	PRT
BOT SURF STRESS IN Y .....	-858.	PRT
MEMBRANE STRESS IN X .....	-201.5	PRT
TOP SURF BENDING STRESS IN X .....	-218.5	PRT
BOT SURF BENDING STRESS IN X .....	218.5	PRT
MEMBRANE STRESS IN Y .....	-353.5	PRT
TOP SURF BENDING STRESS IN Y .....	504.5	PRT
BOT SURF BENDING STRESS IN Y .....	-504.5	PRT
ELEMENT NUMBER .....	34.	PRT
TOP SURF STRESS IN X .....	165.	PRT
TOP SURF STRESS IN Y .....	-4693.	PRT
BOT SURF STRESS IN X .....	-396.	PRT
BOT SURF STRESS IN Y .....	-5517.	PRT
MEMBRANE STRESS IN X .....	-115.5	PRT
TOP SURF BENDING STRESS IN X .....	280.5	PRT
BOT SURF BENDING STRESS IN X .....	-280.5	PRT
MEMBRANE STRESS IN Y .....	-5105.	PRT
TOP SURF BENDING STRESS IN Y .....	412.	PRT
BOT SURF BENDING STRESS IN Y .....	-412.	PRT

B.2.4 200 FT TALL BREECHEDED STEEL STACK  
MODEL CENTER OF ELEMENT STRESSES

ELEMENT NUMBER .....	35.	PRT
TOP SURF STRESS IN X .....	604.	PRT
TOP SURF STRESS IN Y .....	-10848.	PRT
BOT SURF STRESS IN X .....	69.	PRT
BOT SURF STRESS IN Y .....	-10949.	PRT
MEMBRANE STRESS IN X .....	336.5	PRT
TOP SURF BENDING STRESS IN X .....	267.5	PRT
BOT SURF BENDING STRESS IN X .....	-267.5	PRT
MEMBRANE STRESS IN Y .....	-10898.5	PRT
TOP SURF BENDING STRESS IN Y .....	50.5	PRT
BOT SURF BENDING STRESS IN Y .....	-50.5	PRT
ELEMENT NUMBER .....	36.	PRT
TOP SURF STRESS IN X .....	860.	PRT
TOP SURF STRESS IN Y .....	-13872.	PRT
BOT SURF STRESS IN X .....	-31.	PRT
BOT SURF STRESS IN Y .....	-13891.	PRT
MEMBRANE STRESS IN X .....	414.5	PRT
TOP SURF BENDING STRESS IN X .....	445.5	PRT
BOT SURF BENDING STRESS IN X .....	-445.5	PRT
MEMBRANE STRESS IN Y .....	-13881.5	PRT
TOP SURF BENDING STRESS IN Y .....	9.5	PRT
BOT SURF BENDING STRESS IN Y .....	-9.5	PRT
ELEMENT NUMBER .....	37.	PRT
TOP SURF STRESS IN X .....	1602.	PRT
TOP SURF STRESS IN Y .....	-11737.	PRT
BOT SURF STRESS IN X .....	-575.	PRT
BOT SURF STRESS IN Y .....	-13089.	PRT
MEMBRANE STRESS IN X .....	513.5	PRT
TOP SURF BENDING STRESS IN X .....	1088.5	PRT
BOT SURF BENDING STRESS IN X .....	-1088.5	PRT
MEMBRANE STRESS IN Y .....	-12413.	PRT
TOP SURF BENDING STRESS IN Y .....	676.	PRT
BOT SURF BENDING STRESS IN Y .....	-676.	PRT

B.2.5 200 FT TALL BREACHED STEEL STACK  
MODEL CENTER OF ELEMENT STRESSES

ELEMENT NUMBER .....	47.	PRT
TOP SURF STRESS IN X .....	-779.	PRT
TOP SURF STRESS IN Y .....	201.	PRT
BOT SURF STRESS IN X .....	434.	PRT
BOT SURF STRESS IN Y .....	187.	PRT

MEMBRANE STRESS IN X .....	-172.5	PRT
TOP SURF BENDING STRESS IN X .....	-606.5	PRT
BOT SURF BENDING STRESS IN X .....	606.5	PRT

MEMBRANE STRESS IN Y .....	194.	PRT
TOP SURF BENDING STRESS IN Y .....	7.	PRT
BOT SURF BENDING STRESS IN Y .....	-7.	PRT

ELEMENT NUMBER .....	48.	PRT
TOP SURF STRESS IN X .....	-1390.	PRT
TOP SURF STRESS IN Y .....	629.	PRT
BOT SURF STRESS IN X .....	2485.	PRT
BOT SURF STRESS IN Y .....	578.	PRT

MEMBRANE STRESS IN X .....	547.5	PRT
TOP SURF BENDING STRESS IN X .....	-1937.5	PRT
BOT SURF BENDING STRESS IN X .....	1937.5	PRT

MEMBRANE STRESS IN Y .....	603.5	PRT
TOP SURF BENDING STRESS IN Y .....	25.5	PRT
BOT SURF BENDING STRESS IN Y .....	-25.5	PRT

ELEMENT NUMBER .....	49.	PRT
TOP SURF STRESS IN X .....	391.	PRT
TOP SURF STRESS IN Y .....	-4567.	PRT
BOT SURF STRESS IN X .....	588.	PRT
BOT SURF STRESS IN Y .....	-3993.	PRT

MEMBRANE STRESS IN X .....	489.5	PRT
TOP SURF BENDING STRESS IN X .....	-98.5	PRT
BOT SURF BENDING STRESS IN X .....	98.5	PRT

MEMBRANE STRESS IN Y .....	-4280.	PRT
TOP SURF BENDING STRESS IN Y .....	-287.	PRT
BOT SURF BENDING STRESS IN Y .....	287.	PRT

B.2.6      200° FT TALL BREECHE<sup>d</sup> STEEL STACK  
 MODEL CENTER OF ELEMENT STRESSES

ELEMENT NUMBER .....	50.	PRT
TOP SURF STRESS IN X .....	1153.	PRT
TOP SURF STRESS IN Y .....	-14400.	PRT
BOT SURF STRESS IN X .....	-721.	PRT
BOT SURF STRESS IN Y .....	-10614.	PRT

MEMBRANE STRESS IN X .....	216.	PRT
TOP SURF BENDING STRESS IN X .....	937.	PRT
BOT SURF BENDING STRESS IN X .....	-937.	PRT

MEMBRANE STRESS IN Y .....	-12507.	PRT
TOP SURF BENDING STRESS IN Y .....	-1893.	PRT
BOT SURF BENDING STRESS IN Y .....	1893.	PRT

ELEMENT NUMBER .....	51.	PRT
TOP SURF STRESS IN X .....	1130.	PRT
TOP SURF STRESS IN Y .....	-14689.	PRT
BOT SURF STRESS IN X .....	-1338.	PRT
BOT SURF STRESS IN Y .....	-13392.	PRT

MEMBRANE STRESS IN X .....	-104.	PRT
TOP SURF BENDING STRESS IN X .....	1234.	PRT
BOT SURF BENDING STRESS IN X .....	-1234.	PRT

MEMBRANE STRESS IN Y .....	-14040.5	PRT
TOP SURF BENDING STRESS IN Y .....	-648.5	PRT
BOT SURF BENDING STRESS IN Y .....	648.5	PRT

ELEMENT NUMBER .....	52.	PRT
TOP SURF STRESS IN X .....	432.	PRT
TOP SURF STRESS IN Y .....	-13617.	PRT
BOT SURF STRESS IN X .....	-1148.	PRT
BOT SURF STRESS IN Y .....	-12943.	PRT

MEMBRANE STRESS IN X .....	-358.	PRT
TOP SURF BENDING STRESS IN X .....	790.	PRT
BOT SURF BENDING STRESS IN X .....	-790.	PRT

MEMBRANE STRESS IN Y .....	-13280.	PRT
TOP SURF BENDING STRESS IN Y .....	337.	PRT
BOT SURF BENDING STRESS IN Y .....	337.	PRT



B.2.7            200 FT TALL BREECED STEEL STACK  
 MODEL CENTER OF ELEMENT STRESSES

ELEMENT NUMBER .....	59.	PRT
TOP SURF STRESS IN X .....	1103.	PRT
TOP SURF STRESS IN Y .....	-17878.	PRT
BOT SURF STRESS IN X .....	-1854.	PRT
BOT SURF STRESS IN Y .....	-16963.	PRT
MEMBRANE STRESS IN X .....	-375.5	PRT
TOP SURF BENDING STRESS IN X .....	1478.5	PRT
BOT SURF BENDING STRESS IN X .....	-1478.5	PRT
MEMBRANE STRESS IN Y .....	-17420.5	PRT
TOP SURF BENDING STRESS IN Y .....	-457.5	PRT
BOT SURF BENDING STRESS IN Y .....	457.5	PRT
ELEMENT NUMBER .....	60.	PRT
TOP SURF STRESS IN X .....	1461.	PRT
TOP SURF STRESS IN Y .....	-13436.	PRT
BOT SURF STRESS IN X .....	-1940.	PRT
BOT SURF STRESS IN Y .....	-14459.	PRT
MEMBRANE STRESS IN X .....	-239.5	PRT
TOP SURF BENDING STRESS IN X .....	1700.5	PRT
BOT SURF BENDING STRESS IN X .....	-1700.5	PRT
MEMBRANE STRESS IN Y .....	-13947.5	PRT
TOP SURF BENDING STRESS IN Y .....	511.5	PRT
BOT SURF BENDING STRESS IN Y .....	-511.5	PRT
ELEMENT NUMBER .....	61.	PRT
TOP SURF STRESS IN X .....	706.	PRT
TOP SURF STRESS IN Y .....	-10917.	PRT
BOT SURF STRESS IN X .....	-515.	PRT
BOT SURF STRESS IN Y .....	-14472.	PRT
MEMBRANE STRESS IN X .....	95.5	PRT
TOP SURF BENDING STRESS IN X .....	610.5	PRT
BOT SURF BENDING STRESS IN X .....	-610.5	PRT
MEMBRANE STRESS IN Y .....	-11194.5	PRT
TOP SURF BENDING STRESS IN Y .....	277.5	PRT
BOT SURF BENDING STRESS IN Y .....	-277.5	PRT

B.2.8 200 FT TALL BREACHED STEEL STACK  
 MODEL CENTER OF ELEMENT STRESSES

ELEMENT NUMBER .....	72.	PRT
TOP SURF STRESS IN X .....	834.	PRT
TOP SURF STRESS IN Y .....	-17076.	PRT
BOT SURF STRESS IN X .....	-828.	PRT
BOT SURF STRESS IN Y .....	-19629.	PRT
MEMBRANE STRESS IN X .....	3.	PRT
TOP SURF BENDING STRESS IN X .....	831.	PRT
BOT SURF BENDING STRESS IN X .....	-831.	PRT
MEMBRANE STRESS IN Y .....	-18352.5	PRT
TOP SURF BENDING STRESS IN Y .....	1276.5	PRT
BOT SURF BENDING STRESS IN Y .....	-1276.5	PRT
ELEMENT NUMBER .....	73.	PRT
TOP SURF STRESS IN X .....	859.	PRT
TOP SURF STRESS IN Y .....	-13327.	PRT
BOT SURF STRESS IN X .....	-977.	PRT
BOT SURF STRESS IN Y .....	-14530.	PRT
MEMBRANE STRESS IN X .....	-59.	PRT
TOP SURF BENDING STRESS IN X .....	918.	PRT
BOT SURF BENDING STRESS IN X .....	-918.	PRT
MEMBRANE STRESS IN Y .....	-13928.5	PRT
TOP SURF BENDING STRESS IN Y .....	601.5	PRT
BOT SURF BENDING STRESS IN Y .....	-601.5	PRT
ELEMENT NUMBER .....	74.	PRT
TOP SURF STRESS IN X .....	-50.	PRT
TOP SURF STRESS IN Y .....	-10867.	PRT
BOT SURF STRESS IN X .....	-94.	PRT
BOT SURF STRESS IN Y .....	-11309.	PRT
MEMBRANE STRESS IN X .....	-72.	PRT
TOP SURF BENDING STRESS IN X .....	22.	PRT
BOT SURF BENDING STRESS IN X .....	-22.	PRT
MEMBRANE STRESS IN Y .....	-11088.	PRT
TOP SURF BENDING STRESS IN Y .....	221.	PRT
BOT SURF BENDING STRESS IN Y .....	-221.	PRT

B.2.9 200 FT TALL BREECHEd STEEL STACK  
MODEL CENTER OF ELEMENT STRESSES

ELEMENT NUMBER .....	85.	PRT
TOP SURF STRESS IN X .....	177.	PRT
TOP SURF STRESS IN Y .....	-18591.	PRT
BOT SURF STRESS IN X .....	-201.	PRT
BOT SURF STRESS IN Y .....	-20121.	PRT
MEMBRANE STRESS IN X .....	-12.	PRT
TOP SURF BENDING STRESS IN X .....	189.	PRT
BOT SURF BENDING STRESS IN X .....	-189.	PRT
MEMBRANE STRESS IN Y .....	-19356.	PRT
TOP SURF BENDING STRESS IN Y .....	765.	PRT
BOT SURF BENDING STRESS IN Y .....	-765.	PRT
ELEMENT NUMBER .....	86.	PRT
TOP SURF STRESS IN X .....	594.	PRT
TOP SURF STRESS IN Y .....	-13197.	PRT
BOT SURF STRESS IN X .....	-576.	PRT
BOT SURF STRESS IN Y .....	-14435.	PRT
MEMBRANE STRESS IN X .....	9.	PRT
TOP SURF BENDING STRESS IN X .....	585.	PRT
BOT SURF BENDING STRESS IN X .....	-585.	PRT
MEMBRANE STRESS IN Y .....	-13816.	PRT
TOP SURF BENDING STRESS IN Y .....	619.	PRT
BOT SURF BENDING STRESS IN Y .....	-619.	PRT
ELEMENT NUMBER .....	87.	PRT
TOP SURF STRESS IN X .....	-469.	PRT
TOP SURF STRESS IN Y .....	-9972.	PRT
BOT SURF STRESS IN X .....	608.	PRT
BOT SURF STRESS IN Y .....	-10076.	PRT
MEMBRANE STRESS IN X .....	69.5	PRT
TOP SURF BENDING STRESS IN X .....	-538.5	PRT
BOT SURF BENDING STRESS IN X .....	538.5	PRT
MEMBRANE STRESS IN Y .....	-10024.	PRT
TOP SURF BENDING STRESS IN Y .....	52.	PRT
BOT SURF BENDING STRESS IN Y .....	-52.	PRT

B.2.10 200 FT TALL BREECHEED STEEL STACK  
MODEL CENTER OF ELEMENT STRESSES

ELEMENT NUMBER .....	94.	PRT
TOP SURF STRESS IN X .....	1378.	PRT
TOP SURF STRESS IN Y .....	-17872.	PRT
BOT SURF STRESS IN X .....	-1223.	PRT
BOT SURF STRESS IN Y .....	-23200.	PRT

MEMBRANE STRESS IN X .....	77.5	PRT
TOP SURF BENDING STRESS IN X .....	1300.5	PRT
BOT SURF BENDING STRESS IN X .....	-1300.5	PRT

MEMBRANE STRESS IN Y .....	-20536.	PRT
TOP SURF BENDING STRESS IN Y .....	2664.	PRT
BOT SURF BENDING STRESS IN Y .....	-2664.	PRT

ELEMENT NUMBER .....	95.	PRT
TOP SURF STRESS IN X .....	1301.	PRT
TOP SURF STRESS IN Y .....	-11932.	PRT
BOT SURF STRESS IN X .....	-1698.	PRT
BOT SURF STRESS IN Y .....	-14526.	PRT

MEMBRANE STRESS IN X .....	-198.5	PRT
TOP SURF BENDING STRESS IN X .....	1499.5	PRT
BOT SURF BENDING STRESS IN X .....	-1499.5	PRT

MEMBRANE STRESS IN Y .....	-13229.	PRT
TOP SURF BENDING STRESS IN Y .....	1297.	PRT
BOT SURF BENDING STRESS IN Y .....	-1297.	PRT

ELEMENT NUMBER .....	96.	PRT
TOP SURF STRESS IN X .....	-869.	PRT
TOP SURF STRESS IN Y .....	-9795.	PRT
BOT SURF STRESS IN X .....	804.	PRT
BOT SURF STRESS IN Y .....	-10037.	PRT

MEMBRANE STRESS IN X .....	-32.5	PRT
TOP SURF BENDING STRESS IN X .....	-836.5	PRT
BOT SURF BENDING STRESS IN X .....	836.5	PRT

MEMBRANE STRESS IN Y .....	-9916.	PRT
TOP SURF BENDING STRESS IN Y .....	121.	PRT
BOT SURF BENDING STRESS IN Y .....	-121.	PRT

B.2.11 200 FT TALL BREACHED STEEL STACK  
MODEL CENTER OF ELEMENT STRESSES

ELEMENT NUMBER .....	103.	PRT
TOP SURF STRESS IN X .....	3674.	PRT
TOP SURF STRESS IN Y .....	-19472.	PRT
BOT SURF STRESS IN X .....	-5703.	PRT
BOT SURF STRESS IN Y .....	-24487.	PRT
MEMBRANE STRESS IN X .....	-1014.5	PRT
TOP SURF BENDING STRESS IN X .....	4688.5	PRT
BOT SURF BENDING STRESS IN X .....	-4688.5	PRT
MEMBRANE STRESS IN Y .....	-21979.5	PRT
TOP SURF BENDING STRESS IN Y .....	2507.5	PRT
BOT SURF BENDING STRESS IN Y .....	-2507.5	PRT
ELEMENT NUMBER .....	104.	PRT
TOP SURF STRESS IN X .....	2701.	PRT
TOP SURF STRESS IN Y .....	-10011.	PRT
BOT SURF STRESS IN X .....	-3117.	PRT
BOT SURF STRESS IN Y .....	-13908.	PRT
MEMBRANE STRESS IN X .....	-208.	PRT
TOP SURF BENDING STRESS IN X .....	2909.	PRT
BOT SURF BENDING STRESS IN X .....	-2909.	PRT
MEMBRANE STRESS IN Y .....	-11959.5	PRT
TOP SURF BENDING STRESS IN Y .....	1948.5	PRT
BOT SURF BENDING STRESS IN Y .....	-1948.5	PRT
ELEMENT NUMBER .....	105.	PRT
TOP SURF STRESS IN X .....	-509.	PRT
TOP SURF STRESS IN Y .....	-8732.	PRT
BOT SURF STRESS IN X .....	857.	PRT
BOT SURF STRESS IN Y .....	-9207.	PRT
MEMBRANE STRESS IN X .....	174.	PRT
TOP SURF BENDING STRESS IN X .....	-683.	PRT
BOT SURF BENDING STRESS IN X .....	683.	PRT
MEMBRANE STRESS IN Y .....	-8969.5	PRT
TOP SURF BENDING STRESS IN Y .....	237.5	PRT
BOT SURF BENDING STRESS IN Y .....	-237.5	PRT

B.2.12 200 FT TALL BREECED STEEL STACK  
 MODEL CENTER OF ELEMENT STRESSES

ELEMENT NUMBER .....	119.	PRT
TOP SURF STRESS IN X .....	-6627.	PRT
TOP SURF STRESS IN Y .....	-711.	PRT
BOT SURF STRESS IN X .....	7394.	PRT
BOT SURF STRESS IN Y .....	1150.	PRT

MEMBRANE STRESS IN X .....	383.5	PRT
TOP SURF BENDING STRESS IN X .....	-7010.5	PRT
BOT SURF BENDING STRESS IN X .....	7010.5	PRT

MEMBRANE STRESS IN Y .....	219.5	PRT
TOP SURF BENDING STRESS IN Y .....	-930.5	PRT
BOT SURF BENDING STRESS IN Y .....	930.5	PRT

ELEMENT NUMBER .....	120.	PRT
TOP SURF STRESS IN X .....	-2531.	PRT
TOP SURF STRESS IN Y .....	614.	PRT
BOT SURF STRESS IN X .....	4522.	PRT
BOT SURF STRESS IN Y .....	671.	PRT

MEMBRANE STRESS IN X .....	995.5	PRT
TOP SURF BENDING STRESS IN X .....	-3526.5	PRT
BOT SURF BENDING STRESS IN X .....	3526.5	PRT

MEMBRANE STRESS IN Y .....	642.5	PRT
TOP SURF BENDING STRESS IN Y .....	-28.5	PRT
BOT SURF BENDING STRESS IN Y .....	28.5	PRT

ELEMENT NUMBER .....	121.	PRT
TOP SURF STRESS IN X .....	5759.	PRT
TOP SURF STRESS IN Y .....	-6478.	PRT
BOT SURF STRESS IN X .....	-6258.	PRT
BOT SURF STRESS IN Y .....	-7597.	PRT

MEMBRANE STRESS IN X .....	-249.5	PRT
TOP SURF BENDING STRESS IN X .....	6008.5	PRT
BOT SURF BENDING STRESS IN X .....	-6008.5	PRT

MEMBRANE STRESS IN Y .....	-7037.5	PRT
TOP SURF BENDING STRESS IN Y .....	559.5	PRT
BOT SURF BENDING STRESS IN Y .....	-559.5	PRT

B.2.13 200 FT TALL BREECHEED STEEL STACK  
MODEL CENTER OF ELEMENT STRESSES

ELEMENT NUMBER .....	122.	PRT
TOP SURF STRESS IN X .....	5521.	PRT
TOP SURF STRESS IN Y .....	-14824.	PRT
BOT SURF STRESS IN X .....	-5618.	PRT
BOT SURF STRESS IN Y .....	-15033.	PRT

MEMBRANE STRESS IN X .....	-48.5	PRT
TOP SURF BENDING STRESS IN X .....	5569.5	PRT
BOT SURF BENDING STRESS IN X .....	-5569.5	PRT

MEMBRANE STRESS IN Y .....	-14928.5	PRT
TOP SURF BENDING STRESS IN Y .....	104.5	PRT
BOT SURF BENDING STRESS IN Y .....	-104.5	PRT

ELEMENT NUMBER .....	123.	PRT
TOP SURF STRESS IN X .....	2851.	PRT
TOP SURF STRESS IN Y .....	-11460.	PRT
BOT SURF STRESS IN X .....	-3090.	PRT
BOT SURF STRESS IN Y .....	-13496.	PRT

MEMBRANE STRESS IN X .....	-119.5	PRT
TOP SURF BENDING STRESS IN X .....	2970.5	PRT
BOT SURF BENDING STRESS IN X .....	-2970.5	PRT

MEMBRANE STRESS IN Y .....	-12478.	PRT
TOP SURF BENDING STRESS IN Y .....	1018.	PRT
BOT SURF BENDING STRESS IN Y .....	-1018.	PRT

ELEMENT NUMBER .....	124.	PRT
TOP SURF STRESS IN X .....	39.	PRT
TOP SURF STRESS IN Y .....	-9047.	PRT
BOT SURF STRESS IN X .....	-31.	PRT
BOT SURF STRESS IN Y .....	-9610.	PRT

MEMBRANE STRESS IN X .....	4.	PRT
TOP SURF BENDING STRESS IN X .....	35.	PRT
BOT SURF BENDING STRESS IN X .....	35.	PRT

MEMBRANE STRESS IN Y .....	-9328.5	PRT
TOP SURF BENDING STRESS IN Y .....	281.5	PRT
BOT SURF BENDING STRESS IN Y .....	-281.5	PRT

B.2.14      200 FT TALL BREECHEDED STEEL STACK  
 MODEL CENTER OF ELEMENT STRESSES

ELEMENT NUMBER .....	138.	PRT
TOP SURF STRESS IN X .....	-4057.	PRT
TOP SURF STRESS IN Y .....	118.	PRT
BOT SURF STRESS IN X .....	4806.	PRT
BOT SURF STRESS IN Y .....	1674.	PRT
MEMBRANE STRESS IN X .....	374.5	PRT
TOP SURF BENDING STRESS IN X .....	-4431.5	PRT
BOT SURF BENDING STRESS IN X .....	4431.5	PRT
MEMBRANE STRESS IN Y .....	896.	PRT
TOP SURF BENDING STRESS IN Y .....	-778.	PRT
BOT SURF BENDING STRESS IN Y .....	778.	PRT
ELEMENT NUMBER .....	139.	PRT
TOP SURF STRESS IN X .....	-1057.	PRT
TOP SURF STRESS IN Y .....	-1475.	PRT
BOT SURF STRESS IN X .....	247.	PRT
BOT SURF STRESS IN Y .....	-1753.	PRT
MEMBRANE STRESS IN X .....	-405.	PRT
TOP SURF BENDING STRESS IN X .....	-652.	PRT
BOT SURF BENDING STRESS IN X .....	652.	PRT
MEMBRANE STRESS IN Y .....	-1614.	PRT
TOP SURF BENDING STRESS IN Y .....	139.	PRT
BOT SURF BENDING STRESS IN Y .....	-139.	PRT
ELEMENT NUMBER .....	140.	PRT
TOP SURF STRESS IN X .....	2575.	PRT
TOP SURF STRESS IN Y .....	-6945.	PRT
BOT SURF STRESS IN X .....	-3061.	PRT
BOT SURF STRESS IN Y .....	-8231.	PRT
MEMBRANE STRESS IN X .....	-243.	PRT
TOP SURF BENDING STRESS IN X .....	2818.	PRT
BOT SURF BENDING STRESS IN X .....	-2818.	PRT
MEMBRANE STRESS IN Y .....	-7588.	PRT
TOP SURF BENDING STRESS IN Y .....	643.	PRT
BOT SURF BENDING STRESS IN Y .....	-643.	PRT



## C.1.1 MODEL I JOINT DEFLECTIONS

Joint No.	117.	119.	121.	123.
Ux-LC1	0.	-0.01909	-0.06724	0.06459
Ux-LC2	0.	-0.01327	-0.02626	0.06293
Ux-LC3	0.	0.	0.	0.
Ux-LC4	0.	-0.141174	-0.308832	0.642546
Ux-LC5	0.	-0.01909	-0.06724	0.06459
Uy-LC1	0.	0.	0.	0.
Uy-LC2	0.	0.	0.	0.
Uy-LC3	0.	-0.0007	0.05631	0.05668
Uy-LC4	0.	0.	0.	0.
Uy-LC5	0.	-0.00644	0.518052	0.521456
Uz-LC1	0.	0.00083	-0.05977	-0.05948
Uz-LC2	0.	0.00057	-0.03992	-0.03975
Uz-LC3	0.	0.	0.	0.
Uz-LC4	0.	0.006074	-0.427034	-0.42518
Uz-LC5	0.	0.00083	-0.05977	-0.05948

Joint No.	118.	120.	122.	124.
Ux-LC1	-0.00587	-0.02985	-0.01196	0.12746
Ux-LC2	-0.00398	-0.02096	0.01094	0.10632
Ux-LC3	0.	0.	0.	0.
Ux-LC4	-0.042486	-0.222682	0.082688	1.105604
Ux-LC5	-0.00587	-0.02985	-0.01196	0.12746
Uy-LC1	0.	0.	0.	0.
Uy-LC2	0.	0.	0.	0.
Uy-LC3	-0.00018	-0.00109	0.05643	0.05722
Uy-LC4	0.	0.	0.	0.
Uy-LC5	-0.001656	-0.010028	0.519156	0.526424
Uz-LC1	0.00053	0.0009	-0.0597	-0.05931
Uz-LC2	0.00037	0.00063	-0.03988	-0.03965
Uz-LC3	0.	0.	0.	0.
Uz-LC4	0.003934	0.006696	-0.426596	-0.42409
Uz-LC5	0.00053	0.0009	-0.0597	-0.05931

Note: LC1 - Dead load, LC2 - Unit Wind, E+W,  
 LC3 - Unit Wind, N+S, LC4 - LC1 + 9.2 LC2,  
 LC5 - LC1 + 9.2 LC3, Ux - Global radial translation, In  
 Uy - Global tangential translation, In  
 Uz - Global vertical translation, In  
 Global coordinate system is the same as Nodal cylindrical  
 coordinate system.

## C.1.2 MODEL I JOINT DEFLECTIONS

Joint No.	125.	127.	129.	131.
Ux-LC1	0.21168	0.40544	-0.00355	-0.02725
Ux-LC2	0.16599	0.30888	-0.00228	-0.01838
Ux-LC3	0.	0.	-0.00198	-0.0062
Ux-LC4	1.738788	3.247136	-0.024526	-0.196346
Ux-LC5	0.21168	0.40544	-0.021766	-0.08429
Uy-LC1	0.	0.	0.00049	0.00343
Uy-LC2	0.	0.	0.00033	0.00237
Uy-LC3	0.05891	0.06764	-0.00007	-0.00073
Uy-LC4	0.	0.	0.003526	0.025234
Uy-LC5	0.541972	0.622288	-0.000154	-0.003286
Uz-LC1	-0.06002	-0.06558	0.00017	0.00043
Uz-LC2	-0.04012	-0.04367	0.00011	0.00028
Uz-LC3	0.	0.	0.00021	0.00033
Uz-LC4	-0.429124	-0.467344	0.001182	0.003006
Uz-LC5	-0.06002	-0.06558	0.002102	0.003466

Joint No.	126.	128.	130.	132.
Ux-LC1	0.30998	0.	-0.01537	-0.0421
Ux-LC2	-0.23784	0.	-0.01025	-0.01146
Ux-LC3	0.	0.	-0.00464	0.04421
Ux-LC4	2.498108	0.	-0.10967	-0.147532
Ux-LC5	0.30998	0.	-0.058058	0.364632
Uy-LC1	0.	0.	0.00212	0.00682
Uy-LC2	0.	0.	0.00145	0.0024
Uy-LC3	0.06267	0.	-0.00041	0.05364
Uy-LC4	0.	0.	0.01546	0.0289
Uy-LC5	0.576564	0.	-0.001652	0.500308
Uz-LC1	-0.06243	0.	0.00042	-0.0568
Uz-LC2	-0.04167	0.	0.00028	-0.03791
Uz-LC3	0.	0.	0.00031	-0.00012
Uz-LC4	-0.445794	0.	0.002996	-0.405572
Uz-LC5	-0.06243	0.	0.003272	-0.057904

Note: LC1 - Dead load, LC2 - Unit Wind, E+W,  
 LC3 - Unit Wind, N+S, LC4 - LC1 + 9.2 LC2,  
 LC5 - LC1 + 9.2 LC3, Ux - Global radial translation, In  
 Uy - Global tangential translation, In  
 Uz - Global vertical translation, In  
 Global coordinate system is the same as Nodal cylindrical  
 coordinate system.

## C.1.3

## MODEL I JOINT DEFLECTIONS

Joint No.	133.	135.	137.	139.
Ux-LC1	0.00551	0.11991	0.28491	0.
Ux-LC2	0.02085	0.10032	0.2206	0.
Ux-LC3	0.04185	0.03232	0.01843	0.
Ux-LC4	0.19733	1.042854	2.31443	0.
Ux-LC5	0.39053	0.417254	0.454466	0.
Uy-LC1	0.00045	-0.01511	-0.03621	0.
Uy-LC2	-0.00191	-0.01262	-0.02793	0.
Uy-LC3	0.05386	0.05526	0.06156	0.
Uy-LC4	-0.017122	-0.131214	-0.293166	0.
Uy-LC5	0.495962	0.493282	0.530142	0.
Uz-LC1	-0.05686	-0.05708	-0.06092	0.
Uz-LC2	-0.03795	-0.03811	-0.04057	0.
Uz-LC3	-0.00015	-0.00035	-0.00101	0.
Uz-LC4	-0.406	-0.407692	-0.434164	0.
Uz-LC5	-0.05824	-0.0603	-0.070212	0.

Joint No.	134.	136.	138.	140.
Ux-LC1	0.06827	0.19429	0.37355	0.00062
Ux-LC2	0.06406	0.15371	0.28722	0.00074
Ux-LC3	0.03746	0.02545	0.01237	-0.00245
Ux-LC4	0.657622	1.608422	3.015974	0.007428
Ux-LC5	0.412902	0.42843	0.487354	-0.02192
Uy-LC1	-0.00813	-0.02468	-0.04741	0.00071
Uy-LC2	-0.00777	-0.01946	-0.03631	0.00045
Uy-LC3	0.05439	0.05737	0.06689	0.00015
Uy-LC4	-0.079614	-0.203712	-0.381462	0.00485
Uy-LC5	0.492258	0.503124	0.567978	0.00209
Uz-LC1	-0.05675	-0.05828	-0.0641	-0.00073
Uz-LC2	-0.03789	-0.03889	-0.04257	-0.00051
Uz-LC3	-0.00018	-0.00066	-0.00129	0.00003
Uz-LC4	-0.485338	-0.416068	-0.455744	-0.005422
Uz-LC5	0.058406	-0.064352	-0.075968	0.00203

Note: LC1 - Dead load, LC2 - Unit Wind, E+W,  
 LC3 - Unit Wind, N+S, LC4 - LC1 + 9.2 LC2,  
 LC5 - LC1 + 9.2 LC3, Ux - Global radial translation, In  
 Uy - Global tangential translation, In  
 Uz - Global vertical translation, In  
 Global coordinate system is the same as Nodal cylindrical  
 coordinate system.

## C.1.4 MODEL I JOINT DEFLECTIONS

Joint No.	141.	143.	145.	147.
Ux-LC1	-0.00493	0.0246	0.07105	0.14765
Ux-LC2	-0.00218	0.02781	0.0626	0.12059
Ux-LC3	-0.0077	0.07982	0.06498	0.04712
Ux-LC4	-0.024986	0.280452	0.64697	1.257078
Ux-LC5	-0.07577	0.758944	0.668866	0.581154
Uy-LC1	0.0033	0.00829	-0.0165	-0.04538
Uy-LC2	0.00217	0.00165	-0.01542	-0.0361
Uy-LC3	0.00035	0.04614	0.04816	0.05297
Uy-LC4	0.023264	0.02347	-0.158364	-0.3775
Uy-LC5	0.00652	0.432778	0.426572	0.441944
Uz-LC1	-0.00121	-0.0487	-0.04971	-0.05341
Uz-LC2	-0.00087	-0.03242	-0.03306	-0.03539
Uz-LC3	0.00056	-0.00043	-0.0007	-0.00158
Uz-LC4	-0.009214	-0.346964	-0.353862	-0.378998
Uz-LC5	0.003942	-0.052656	-0.05615	-0.067946

Joint No.	142.	144.	146.	148.
Ux-LC1	-0.01334	0.04444	0.09824	0.21638
Ux-LC2	-0.0071	0.04246	0.08323	0.17329
Ux-LC3	-0.01204	0.0736	0.05739	0.03592
Ux-LC4	-0.07866	0.435072	0.863956	1.810648
Ux-LC5	-0.124108	0.72156	0.626228	0.546844
Uy-LC1	0.0061	-0.0027	-0.0283	-0.06666
Uy-LC2	0.00406	-0.00585	-0.02376	-0.05189
Uy-LC3	0.00035	0.04676	0.04981	0.05827
Uy-LC4	0.043452	-0.05652	-0.248892	-0.544048
Uy-LC5	0.00932	0.427492	0.429952	0.469424
Uz-LC1	-0.00112	-0.04859	-0.05113	-0.05661
Uz-LC2	-0.00081	-0.03236	-0.03396	-0.03738
Uz-LC3	0.00056	-0.00041	-0.00104	-0.00223
Uz-LC4	-0.008572	-0.346302	-0.363562	-0.400506
Uz-LC5	0.004032	-0.052362	-0.060698	-0.077126

Note: LC1 = Dead load, LC2 = Unit Wind, E-W,  
 LC3 = Unit Wind, N-S, LC4 = LC1 + 9.2 LC2,  
 LC5 = LC1 + 9.2 LC3, Ux = Global radial translation, In  
 Uy = Global tangential translation, In  
 Uz = Global vertical translation, In  
 Global coordinate system is the same as Nodal cylindrical  
 coordinate system.

## C.1.5 MODEL I JOINT DEFLECTIONS

Joint No.	149.	150.	152.	154.
Ux-LC1	0.28522	0.	0.00594	0.06357
Ux-LC2	0.22705	0.	0.00565	0.04578
Ux-LC3	0.0261	0.	-0.00646	0.00334
Ux-LC4	2.37408	0.	0.05792	0.484746
Ux-LC5	0.52534	0.	-0.053492	0.094290
Uy-LC1	-0.08746	0.	0.00338	0.01354
Uy-LC2	-0.06762	0.	0.00206	0.00805
Uy-LC3	0.06458	0.	0.00113	0.00683
Uy-LC4	-0.709564	0.	0.022332	0.0876
Uy-LC5	0.506676	0.	0.013776	0.076376
Uz-LC1	-0.05987	0.	-0.0038	-0.01042
Uz-LC2	-0.03939	0.	-0.00259	-0.007
Uz-LC3	-0.00279	0.	0.00036	0.00162
Uz-LC4	-0.422258	0.	-0.027628	-0.07452
Uz-LC5	-0.085446	0.	-0.000488	0.004784

Joint No.	151.	153.	155.
Ux-LC1	0.00317	-0.01417	0.10952
Ux-LC2	0.00254	0.01256	0.07544
Ux-LC3	-0.00149	-0.01185	0.0318
Ux-LC4	0.026538	0.129722	0.803568
Ux-LC5	-0.010538	-0.09485	0.40208
Uy-LC1	0.00075	0.00644	0.01856
Uy-LC2	0.00044	0.00398	0.0104
Uy-LC3	0.00034	0.00131	0.01424
Uy-LC4	0.004798	0.043056	0.11424
Uy-LC5	0.003878	0.024012	0.149568
Uz-LC1	-0.00178	-0.00536	-0.01527
Uz-LC2	-0.00121	-0.00369	-0.01046
Uz-LC3	0.00015	0.00068	0.00192
Uz-LC4	-0.012912	-0.039308	-0.111502
Uz-LC5	-0.0004	0.000896	0.002394

Note: LC1 - Dead load, LC2 - Unit Wind, E+W,  
 LC3 - Unit Wind, N+S, LC4 - LC1 + 9.2 LC2,  
 LC5 - LC1 + 9.2 LC3, Ux - Global radial translation, In.  
 Uy - Global tangential translation, In  
 Uz - Global vertical translation, In  
 Global coordinate system is the same as Nodal cylindrical  
 coordinate system.

0:1.6

## MODEL I JOINT DEFLECTIONS

Joint No.	156.	158.	160.	162.
Ux-LC1	0.13599	0.08962	0.08259	0.08524
Ux-LC2	0.09231	0.06543	0.05294	0.07577
Ux-LC3	0.06206	0.09386	0.07776	0.06224
Ux-LC4	0.985242	0.691576	0.549638	0.782324
Ux-LC5	-0.706942	0.953132	0.777982	0.657848
Uy-LC1	0.01943	0.00158	-0.02432	-0.05933
Uy-LC2	0.00985	-0.00398	-0.02226	-0.04793
Uy-LC3	0.02248	0.03549	0.03949	0.04634
Uy-LC4	0.11005	-0.034116	-0.229112	-0.500286
Uy-LC5	0.226246	0.328088	0.338988	0.366998
Uz-LC1	-0.0211	-0.0353	-0.04078	-0.04636
Uz-LC2	-0.0143	-0.02337	-0.0268	-0.03026
Uz-LC3	0.00165	-0.00072	-0.00181	-0.00291
Uz-LC4	-0.15266	-0.250304	-0.28734	-0.324752
Uz-LC5	-0.00592	-0.041924	-0.057432	-0.073132

Joint No.	157.	159.	161.	163.
Ux-LC1	0.13073	0.06991	0.06654	0.12169
Ux-LC2	0.08964	0.05475	0.05843	0.10741
Ux-LC3	0.08635	0.08598	0.07146	0.05125
Ux-LC4	0.955418	0.57361	0.604096	1.109862
Ux-LC5	0.92515	0.860926	0.723978	0.59319
Uy-LC1	0.01423	-0.0094	-0.03807	-0.08709
Uy-LC2	0.00534	-0.01161	-0.03223	-0.06891
Uy-LC3	0.03011	0.03707	0.04199	0.05299
Uy-LC4	0.063358	-0.116212	-0.334586	-0.721062
Uy-LC5	0.291242	0.331644	0.348238	0.400418
Uz-LC1	-0.02757	-0.03794	-0.04431	-0.05014
Uz-LC2	-0.01846	-0.02502	-0.02824	-0.03255
Uz-LC3	0.00082	-0.00128	-0.00224	-0.00378
Uz-LC4	-0.197402	-0.268124	-0.302908	-0.3496
Uz-LC5	-0.020026	-0.049716	-0.063708	-0.084916

Note: LC1 - Dead load, LC2 - Unit Wind, E+W,  
 LC3 - Unit Wind, N+S, LC4 - LC1 + 9.2 LC2,  
 LC5 - LC1 + 9.2 LC3, Ux - Global radial translation, In  
 Uy - Global tangential translation, In  
 Uz - Global vertical translation, In  
 Global coordinate system is the same as Nodal cylindrical  
 coordinate system.

## C.1.7 MODEL I JOINT DEFLECTIONS

Joint No.	164.	166.	168.	170.
Ux-LC1	0.16038	0.00574	0.02965	0.08324
Ux-LC2	0.14135	0.00417	0.02186	0.05695
Ux-LC3	0.04149	-0.00001	-0.00131	0.02871
Ux-LC4	1.4608	0.044104	0.230762	0.60718
Ux-LC5	0.542088	0.00482	0.017598	0.347372
Uy-LC1	-0.11447	0.00072	0.00408	0.00601
Uy-LC2	-0.09	0.00036	0.00203	0.00175
Uy-LC3	0.0605	0.00047	0.00288	0.01006
Uy-LC4	-0.94247	0.004032	0.022756	0.02211
Uy-LC5	0.44213	0.005044	0.030576	0.098562
Uz-LC1	-0.05351	-0.00015	-0.00843	-0.0155
Uz-LC2	-0.03455	-0.00181	-0.00546	-0.00996
Uz-LC3	-0.00456	-0.00025	-0.00864	-0.00133
Uz-LC4	-0.37137	-0.016802	-0.058662	-0.107132
Uz-LC5	-0.095462	-0.00245	-0.087918	-0.033256

Joint No.	165.	167.	169.	171.
Ux-LC1	0.	0.01528	0.05737	0.1003
Ux-LC2	0.	0.01153	0.04073	0.06784
Ux-LC3	0.	-0.00149	0.00958	0.04969
Ux-LC4	0.	0.121356	0.432686	0.724428
Ux-LC5	0.	0.001572	0.146106	0.557448
Uy-LC1	0.	0.00247	0.0059	0.00389
Uy-LC2	0.	0.00124	0.00258	-0.00071
Uy-LC3	0.	0.00165	0.00589	0.01476
Uy-LC4	0.	0.013878	0.029636	-0.002642
Uy-LC5	0.	0.01765	0.060088	0.139682
Uz-LC1	0.	-0.006	-0.01186	-0.01905
Uz-LC2	0.	-0.00389	-0.00763	-0.01221
Uz-LC3	0.	-0.00056	-0.00148	-0.00237
Uz-LC4	0.	-0.041788	-0.082056	-0.131382
Uz-LC5	0.	-0.011152	-0.025476	-0.040854

Note: LC1 = Dead load, LC2 = Unit Wind, E+W,  
 LC3 = Unit Wind, N+S, LC4 = LC1 + 9.2 LC2,  
 LC5 = LC1 + 9.2 LC3, Ux = Global radial translation, In  
 Uy = Global tangential translation, In  
 Uz = Global vertical translation, In  
 Global coordinate system is the same as Nodal cylindrical  
 coordinate system.

## C.1.8 MODEL I JOINT DEFLECTIONS

Joint No.	172.	174.	176.	178.
Ux-LC1	0.10424	0.06617	0.02608	0.00847
Ux-LC2	0.07064	0.04836	0.02708	0.02744
Ux-LC3	0.06839	0.07856	0.07997	0.06425
Ux-LC4	0.754128	0.511082	0.275216	0.260918
Ux-LC5	0.733428	0.788922	0.706604	0.59957
Uy-LC1	-0.00108	-0.01821	-0.04407	-0.09586
Uy-LC2	-0.00516	-0.01838	-0.03792	-0.07814
Uy-LC3	0.01946	0.02569	0.0319	0.04499
Uy-LC4	-0.048552	-0.187306	-0.392934	-0.814748
Uy-LC5	0.177952	0.218138	0.24941	0.318048
Uz-LC1	-0.02265	-0.03779	-0.03343	-0.0413
Uz-LC2	-0.01449	-0.0177	-0.02117	-0.02578
Uz-LC3	-0.0028	-0.00325	-0.00416	-0.00594
Uz-LC4	-0.155958	-0.19063	-0.228244	-0.278476
Uz-LC5	-0.04841	-0.05769	-0.071752	-0.095948

Joint No.	173.	175.	177.	179.
Ux-LC1	0.08784	0.04112	0.01301	0.00683
Ux-LC2	0.06107	0.03441	0.02301	0.03436
Ux-LC3	0.07884	0.07625	0.07001	0.05935
Ux-LC4	0.649684	0.357692	0.224702	0.322942
Ux-LC5	0.813168	0.74262	0.657102	0.55235
Uy-LC1	-0.0101	-0.03103	-0.06581	-0.12586
Uy-LC2	-0.01226	-0.02803	-0.0546	-0.10212
Uy-LC3	0.0235	0.02881	0.03718	0.05353
Uy-LC4	-0.122892	-0.288906	-0.56813	-1.065364
Uy-LC5	0.2061	0.234022	0.276246	0.366616
Uz-LC1	-0.02557	-0.03085	-0.03717	-0.04476
Uz-LC2	-0.01632	-0.01957	-0.02338	-0.02771
Uz-LC3	-0.00302	-0.00371	-0.00491	-0.00698
Uz-LC4	-0.175714	-0.210894	-0.252266	-0.299692
Uz-LC5	-0.053354	-0.064982	-0.082342	-0.108976

Note: LC1 = Dead load, LC2 = Unit Wind, E+W,  
 LC3 = Unit Wind, N+S, LC4 = LC1 + 9.2 LC2,  
 LC5 = LC1 + 9.2 LC3, Ux = Global radial translation, In  
 Uy = Global tangential translation, In  
 Uz = Global vertical translation, In  
 Global coordinate system is the same as Nodal cylindrical  
 coordinate system.



## C.1.9 MODEL I JOINT DEFLECTIONS

Joint No.	180.	182.	184.	186.
Ux-LC1	0.	0.01809	0.0417	0.05949
Ux-LC2	0.	0.01247	0.02849	0.04019
Ux-LC3	0.	0.0039	0.01406	0.03622
Ux-LC4	0.	0.132814	0.303808	0.429238
Ux-LC5	0.	0.05397	0.171052	0.392714
Uy-LC1	0.	0.00081	-0.00046	-0.00663
Uy-LC2	0.	-0.06001	-0.00189	-0.0079
Uy-LC3	0.	0.0016	0.00438	0.00853
Uy-LC4	0.	0.000718	-0.017848	-0.07936
Uy-LC5	0.	0.01553	0.039836	0.071796
Uz-LC1	0.	-0.00633	-0.01101	-0.01587
Uz-LC2	0.	-0.00387	-0.00664	-0.00946
Uz-LC3	0.	-0.00153	-0.00292	-0.00434
Uz-LC4	0.	-0.041934	-0.072098	-0.102902
Uz-LC5	0.	-0.020406	-0.037874	-0.055798

Joint No.	181.	183.	185.
Ux-LC1	0.00838	0.02773	0.05274
Ux-LC2	0.00549	0.01903	0.03574
Ux-LC3	0.001758	0.00697	0.02435
Ux-LC4	0.058888	0.202358	0.381548
Ux-LC5	0.0245536	0.091854	0.27676
Uy-LC1	0.00049	0.00067	-0.00289
Uy-LC2	0.00016	-0.00047	-0.00436
Uy-LC3	0.00049	0.00261	0.00647
Uy-LC4	0.001962	-0.003654	-0.043002
Uy-LC5	0.004998	0.024682	0.056634
Uz-LC1	-0.00306	-0.00821	-0.0135
Uz-LC2	-0.00188	-0.00497	-0.00806
Uz-LC3	-0.00071	-0.00209	-0.00373
Uz-LC4	-0.020356	-0.053934	-0.087652
Uz-LC5	-0.009592	-0.027438	-0.047816

Note: LC1 - Dead load, LC2 - Unit Wind, E+W,  
 LC3 - Unit Wind, N+S, LC4 - LC1 + 9.2 LC2,  
 LC5 - LC1 + 9.2 LC3, Ux - Global radial translation, In  
 Uy - Global tangential translation, In  
 Uz - Global vertical translation, In  
 Global coordinate system is the same as Nodal cylindrical  
 coordinate system.

C.1.10

## MODEL I JOINT DEFLECTIONS

Joint No.	187.	189.	191.	193.
Ux-LC1	0.05906	0.03384	-0.01234	-0.02386
Ux-LC2	0.04006	0.02489	-0.00207	-0.03963
Ux-LC3	0.0472	0.05891	0.06511	0.07059
Ux-LC4	0.427612	0.262828	-0.031384	-0.448456
Ux-LC5	0.4933	0.575812	0.586672	0.565568
Uy-LC1	-0.01205	-0.02466	-0.04455	-0.09023
Uy-LC2	-0.01262	-0.02317	-0.03938	-0.0771
Uy-LC3	0.01144	0.01616	0.02227	0.03566
Uy-LC4	-0.128154	-0.237824	-0.406846	-0.79955
Uy-LC5	0.093198	0.124012	0.160334	0.237842
Uz-LC1	-0.01784	-0.021	-0.02524	-0.03358
Uz-LC2	-0.01056	-0.01236	-0.01475	-0.01879
Uz-LC3	-0.0048	-0.00542	-0.00631	-0.00835
Uz-LC4	-0.114992	-0.134712	-0.16094	-0.205448
Uz-LC5	-0.062	-0.070864	-0.083292	-0.1094

Joint No.	188.	190.	192.	194.
Ux-LC1	0.04743	0.0102	-0.04539	-0.12077
Ux-LC2	0.03303	0.01101	-0.02039	-0.05725
Ux-LC3	0.05519	0.06262	0.06795	0.07363
Ux-LC4	0.351306	0.111492	-0.232978	-0.64747
Ux-LC5	0.555178	0.586304	0.57975	0.556626
Uy-LC1	-0.01909	-0.03412	-0.06315	-0.11759
Uy-LC2	-0.01856	-0.03092	-0.05456	-0.10036
Uy-LC3	0.0142	0.01918	0.02763	0.04434
Uy-LC4	-0.189842	-0.318584	-0.565102	-1.040902
Uy-LC5	0.11155	0.142336	0.191046	0.290338
Uz-LC1	-0.01974	-0.02309	-0.02858	-0.03608
Uz-LC2	-0.01166	-0.01357	-0.01663	-0.02061
Uz-LC3	-0.00516	-0.00581	-0.00712	-0.00957
Uz-LC4	-0.127012	-0.147934	-0.181576	-0.235692
Uz-LC5	-0.067212	-0.076542	-0.094084	-0.124124

Note: LC1 - Dead load, LC2 - Unit Wind, E+W,  
 LC3 - Unit Wind, N+S, LC4 - LC1 + 9.2 LC2,  
 LC5 - LC1 + 9.2 LC3, Ux - Global radial translation, In  
 Uy - Global tangential translation, In  
 Uz - Global vertical translation, In  
 Global coordinate system is the same as Nodal cylindrical  
 coordinate system.

## C.1.11 MODEL I JOINT DEFLECTIONS

Joint No.	195.	197.	200.	202.
Ux-LC1	0.	0.01081	0.01007	-0.00251
Ux-LC2	0.	0.00664	0.0068	-0.00095
Ux-LC3	0.	0.00623	0.01604	0.02372
Ux-LC4	0.	0.071698	0.07263	-0.01125
Ux-LC5	0.	0.067926	0.157638	0.215714
Uy-LC1	0.	-0.00117	-0.00811	-0.01678
Uy-LC2	0.	-0.0014	-0.00797	-0.01597
Uy-LC3	0.	0.00092	0.00296	0.00509
Uy-LC4	0.	-0.01405	-0.081434	-0.163704
Uy-LC5	0.	0.007294	0.019122	0.030048
Uz-LC1	0.	-0.00478	-0.00991	-0.01286
Uz-LC2	0.	-0.00245	-0.005	-0.0063
Uz-LC3	0.	-0.00217	-0.00457	-0.00619
Uz-LC4	0.	-0.02732	-0.05591	-0.07082
Uz-LC5	0.	-0.024744	-0.051954	-0.069808

Joint No.	196.	198.	201.	203.
Ux-LC1	0.00738	0.0112	0.00539	-0.01259
Ux-LC2	0.0043	0.00691	0.00407	-0.00682
Ux-LC3	0.00346	0.00858	0.01948	0.02875
Ux-LC4	0.04694	0.074772	0.042834	-0.075334
Ux-LC5	0.039212	0.090136	0.184606	0.25191
Uy-LC1	-0.00005	-0.00227	-0.01216	-0.02184
Uy-LC2	-0.00022	-0.00249	-0.01172	-0.02075
Uy-LC3	0.00031	0.0014	0.00392	0.00666
Uy-LC4	-0.002074	-0.025178	-0.119984	-0.21274
Uy-LC5	0.002802	0.01061	0.023904	0.039432
Uz-LC1	-0.00274	-0.00635	-0.01138	-0.01422
Uz-LC2	-0.00152	-0.00335	-0.00557	-0.00681
Uz-LC3	-0.00111	-0.00282	-0.00544	-0.00701
Uz-LC4	-0.016724	-0.03717	-0.062624	-0.076872
Uz-LC5	-0.012952	-0.032294	-0.061428	-0.078712

Note: LC1 - Dead load, LC2 - Unit Wind, E+W,  
 LC3 - Unit Wind, N+S, LC4 - LC1 + 9.2 LC2,  
 LC5 - LC1 + 9.2 LC3, Ux - Global radial translation, In  
 Uy - Global tangential translation, In  
 Uz - Global vertical translation, In  
 Global coordinate system is the same as Nodal cylindrical  
 coordinate system.

## C.1.12 MODEL I JOINT DEFLECTIONS

Joint No.	204.	206.	208.
Ux-LC1	-0.02061	-0.05486	-0.15074
Ux-LC2	-0.01185	-0.03322	-0.09138
Ux-LC3	0.03286	0.04508	0.06722
Ux-LC4	-0.12963	-0.360484	-0.991436
Ux-LC5	0.281702	0.359876	0.467684
Uy-LC1	-0.02542	-0.0379	-0.06823
Uy-LC2	-0.02414	-0.03589	-0.06482
Uy-LC3	0.00788	0.0123	0.02314
Uy-LC4	-0.247508	-0.368088	-0.664574
Uy-LC5	0.047076	0.07536	0.144658
Uz-LC1	-0.01512	-0.01771	-0.02356
Uz-LC2	-0.00721	-0.00834	-0.01081
Uz-LC3	-0.00743	-0.00871	-0.01131
Uz-LC4	-0.081452	-0.094438	-0.123012
Uz-LC5	-0.083476	-0.097842	-0.127612

Joint No.	205.	207.	209.
Ux-LC1	-0.03588	-0.0929	-0.20964
Ux-LC2	-0.02128	-0.05596	-0.12594
Ux-LC3	0.0388	0.05434	0.0792
Ux-LC4	-0.231656	-0.607732	-1.368288
Ux-LC5	0.32108	0.407028	0.519
Uy-LC1	-0.03133	-0.05001	-0.08681
Uy-LC2	0.02976	-0.04733	-0.08333
Uy-LC3	0.01001	0.01655	0.03045
Uy-LC4	-0.305122	-0.485446	-0.853446
Uy-LC5	0.060762	0.10225	0.19333
Uz-LC1	-0.01641	-0.02004	-0.02697
Uz-LC2	-0.00769	-0.00922	-0.01243
Uz-LC3	-0.00817	-0.00989	-0.01277
Uz-LC4	-0.087158	-0.104864	-0.141326
Uz-LC5	-0.091574	-0.111028	-0.144454

Note: LC1 - Dead load, LC2 - Unit Wind, E+W,  
 LC3 - Unit Wind, N+S, LC4 - LC1 + 9.2 LC2,  
 LC5 - LC1 + 9.2 LC3, Ux - Global radial translation, In  
 Uy - Global tangential translation, In  
 Uz - Global vertical translation, In  
 Global coordinate system is the same as Nodal cylindrical  
 coordinate system.

APPENDIX D

D.1: MODEL II ANSYS ANALYSIS

This appendix contains the entire Model II ANSYS analysis output.



TX HAADUUU, 4986647E005, 1161.610-0032-26-78-00-005, MEMOTE 39, AYVAZYAN, 6582,  
HAADUUU DHCU 05/11/77 09.38.26 7600 DU





















UP181 STRESSES AND MODE MORPHS ADDED TO TAPE10 FOR STIF9  
 UP181 SASI ANNOUNCES THE ADDITION OF MIKE WHEELER TO STAFF  
 UP184 P8S111 ALLOWS USER SPECIFICATION OF CONTOUR LINES  
 UP184 POWER FUNCTION CREEP LAW ADDED W/ LINEAR INTERPOL  
 UP184 SPECIFIC HEAT STRESSES ADDED TO INT. POINTS, STIF55 AND 70  
 UP185 MORE STRAIN STRESSES ADDED TO TAPE10 FOR STIF20,60  
 UP185 60 SCALE FACTORS MAY BE USED IN S.E. STRESS PASS  
 UP185 TIME AND THRESHOLD DEPENDENT CREEP ADDED- STIF34  
 UP185 SPECIFIC HEAT COMPUTED AT EACH INT. PT.-STIF67,77  
 UP185 FORCES ON TAPE10 ADDED TO STIF62  
 UP185 PLASTICITY CAPABILITY ADDED TO STIF85  
 UP185 UPDATE PKG. NO.3 DOCUMENTING ABOVE FEATURES IS AVAIL  
 UP186 POST12 NOW ALLOWS SIX MATERIAL TABLES  
 UP186 RADIATION CREEP AND SWELLING ADDED FOR 20PL Ck 316SS  
 UP186 STRESSES ADDED TO STIF20 AND STIF60 PRINTOUT  
 UP186 ANSYS REV.3 USERS MAN. SUPPLEMENT AVAILABLE, 180 PG.  
 UP187 VIB MIXES STRESS PLANS. ADDED TO PUSTS  
 UP187 SINKING POWER FUNCTION CREEP LAW ADDED  
 UP187 NFA OPTIONS FROM FILM CUFF EVALUATION STIF55  
 UP187 SASI ANNOUNCES THE ADDITION OF BARRY COLTON TO STAFF  
 UP187 SASI ADDS FAX SERVICE 710-789-3883 ANSYS EXBH

ANSYS - ENGINEERING ANALYSIS SYSTEM UPI88 REV 2 U+W/600 JAN 1, 1972  
 SWANSON ANALYSIS SYSTEMS, INC. ELIZABETH, PENNSYLVANIA 15037 PHONE (412) 751-1940  
 BRECHRD STEEL STACK ANSYS ANALYSIS FOR DL + 9.2PSF E ID W LAT. MID CP = 9.8425 5/11/77  
 2.594 PP = 0.000

\*\*\*\*\* ANALYST = MEA

\*\*\*\*\* ANALYSIS OPTIONS (CARDS C1 AND C2) \*\*\*\*\*  
 VALUE VARIABLE NAME COLUMNS  
 NSTEPS 1-4  
 K20 5-7  
 KTB 11-12  
 K15 15-16  
 K17 18  
 TREF 1-12  
 IUNIT 13-24  
 ICORE 73-80 (CARD B)  
 500 1500 500 500 0 0 0 0  
 NUMBER OF LOAD STEPS . . . . . 1  
 ANALYSIS TYPE . . . . . 1  
 ELEMENT CONSTANT TABLE . . . . . 1  
 REACTION FORCE KEY . . . . . 2  
 BOUNDARY CONDITION KEY . . . . . 1  
 REFERENCE TEMPERATURE . . . . . 70.00  
 UNIT TEMP TEMPERATURE . . . . . 70.00  
 CORE SIZE PARAMETER . . . . . 00100000  
 NUENK SIZES 500  
 CORE SIZE REQUESTED (ACTUAL) . . . 00114000  
 LCM SIZE REQUESTED (ACTUAL) . . . 00100000

\*\*\*\*\* ELEMENT TYPES (CARD D) \*\*\*\*\*

TYPE	STIFF	DESCRIPTION	KEYSUR	OPTIONS	KC	INDTPK
1	0.65	UNID FLAT SHELL ELEM 8/01/73	1B 1A 1	2B, 2A 2	0	0

\*\*\*\*\* TABLE OF ELEMENT REAL CONSTANTS (CARD D2) \*\*\*\*\*

NO.	1	1.2500				
1	0	0	0	0	0	0

\*\*\*\*\* ELEMENT DEFINITIONS (CARD E) \*\*\*\*\*

ELEMENT	NODES	MAT	TYPE	ELEMENT REAL CONSTANTS
1	1 2 3 4 5 6 7 8 9 10 11 12 13 14	1	1	
2	67 68 69 70 71 72 73 74 75 76 77 78 79 80	1	1	
3	67 68 69 70 71 72 73 74 75 76 77 78 79 80	1	1	
4	67 68 69 70 71 72 73 74 75 76 77 78 79 80	1	1	
5	67 68 69 70 71 72 73 74 75 76 77 78 79 80	1	1	
6	67 68 69 70 71 72 73 74 75 76 77 78 79 80	1	1	
7	67 68 69 70 71 72 73 74 75 76 77 78 79 80	1	1	
8	67 68 69 70 71 72 73 74 75 76 77 78 79 80	1	1	
9	67 68 69 70 71 72 73 74 75 76 77 78 79 80	1	1	
10	67 68 69 70 71 72 73 74 75 76 77 78 79 80	1	1	
11	67 68 69 70 71 72 73 74 75 76 77 78 79 80	1	1	
12	67 68 69 70 71 72 73 74 75 76 77 78 79 80	1	1	
13	67 68 69 70 71 72 73 74 75 76 77 78 79 80	1	1	
14	67 68 69 70 71 72 73 74 75 76 77 78 79 80	1	1	

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134	135	136	137	138	139	140	141	142	143	144	145	146	147	148	149	150	151	152	153	154	155	156	157	158	159	160	
161	162	163	164	165	166	167	168	169	170	171	172	173	174	175	176	177	178	179	180	181	182	183	184	185	186	187	188
189	190	191	192	193	194	195	196	197	198	199	200	201	202	203	204	205	206	207	208	209	210	211	212	213	214	215	216
217	218	219	220	221	222	223	224	225	226	227	228	229	230	231	232	233	234	235	236	237	238	239	240	241	242	243	244
245	246	247	248	249	250	251	252	253	254	255	256	257	258	259	260	261	262	263	264	265	266	267	268	269	270	271	272
273	274	275	276	277	278	279	280	281	282	283	284	285	286	287	288	289	290	291	292	293	294	295	296	297	298	299	300
301	302	303	304	305	306	307	308	309	310	311	312	313	314	315	316	317	318	319	320	321	322	323	324	325	326	327	328
329	330	331	332	333	334	335	336	337	338	339	340	341	342	343	344	345	346	347	348	349	350	351	352	353	354	355	356
357	358	359	360	361	362	363	364	365	366	367	368	369	370	371	372	373	374	375	376	377	378	379	380	381	382	383	384
385	386	387	388	389	390	391	392	393	394	395	396	397	398	399	400	401	402	403	404	405	406	407	408	409	410	411	412
413	414	415	416	417	418	419	420	421	422	423	424	425	426	427	428	429	430	431	432	433	434	435	436	437	438	439	440
441	442	443	444	445	446	447	448	449	450	451	452	453	454	455	456	457	458	459	460	461	462	463	464	465	466	467	468
469	470	471	472	473	474	475	476	477	478	479	480	481	482	483	484	485	486	487	488	489	490	491	492	493	494	495	496
497	498	499	500	501	502	503	504	505	506	507	508	509	510	511	512	513	514	515	516	517	518	519	520	521	522	523	524
525	526	527	528	529	530	531	532	533	534	535	536	537	538	539	540	541	542	543	544	545	546	547	548	549	550	551	552
553	554	555	556	557	558	559	560	561	562	563	564	565	566	567	568	569	570	571	572	573	574	575	576	577	578	579	580
581	582	583	584	585	586	587	588	589	590	591	592	593	594	595	596	597	598	599	600	601	602	603	604	605	606	607	608
609	610	611	612	613	614	615	616	617	618	619	620	621	622	623	624	625	626	627	628	629	630	631	632	633	634	635	636
637	638	639	640	641	642	643	644	645	646	647	648	649	650	651	652	653	654	655	656	657	658	659	660	661	662	663	664
665	666	667	668	669	670	671	672	673	674	675	676	677	678	679	680	681	682	683	684	685	686	687	688	689	690	691	692
693	694	695	696	697	698	699	700	701	702	703	704	705	706	707	708	709	710	711	712	713	714	715	716	717	718	719	720
721	722	723	724	725	726	727	728	729	730	731	732	733	734	735	736	737	738	739	740	741	742	743	744	745	746	747	748
749	750	751	752	753	754	755	756	757	758	759	760	761	762	763	764	765	766	767	768	769	770	771	772	773	774	775	776
777	778	779	780	781	782	783	784	785	786	787	788	789	790	791	792	793	794	795	796	797	798	799	800	801	802	803	804
805	806	807	808	809	810	811	812	813	814	815	816	817	818	819	820	821	822	823	824	825	826	827	828	829	830	831	832
833	834	835	836	837	838	839	840	841	842	843	844	845	846	847	848	849	850	851	852	853	854	855	856	857	858	859	860
861	862	863	864	865	866	867	868	869	870	871	872	873	874	875	876	877	878	879	880	881	882	883	884	885	886	887	888
889	890	891	892	893	894	895	896	897	898	899	900	901	902	903	904	905	906	907	908	909	910	911	912	913	914	915	916
917	918	919	920	921	922	923	924	925	926	927	928	929	930	931	932	933	934	935	936	937	938	939	940	941	942	943	944
945	946	947	948	949	950	951	952	953	954	955	956	957	958	959	960	961	962	963	964	965	966	967	968	969	970	971	972
973	974	975	976	977	978	979	980	981	982	983	984	985	986	987	988	989	990	991	992	993	994	995	996	997	998	999	1000

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297 298 299 300 301 302 303 304 305 306 307 308 309 310 311 312 313 314 315 316 317 318 319 320 321 322 323 324 325 326 327 328 329 330 331 332 333 334 335 336 337 338 339 340 341 342 343 344 345 346

278 279 280 281 282 283 284 285 286 287 288 289 290 291 292 293 294 295 296 297 298 299 300 301 302 303 304 305 306 307 308 309 310 311 312 313 314 315 316 317 318 319 320 321 322 323 324 325 326 327 328 329 330 331

277 278 279 280 281 282 283 284 285 286 287 288 289 290 291 292 293 294 295 296 297 298 299 300 301 302 303 304 305 306 307 308 309 310 311 312 313 314 315 316 317 318 319 320 321 322 323 324 325 326 327 328 329 330 331

279 280 281 282 283 284 285 286 287 288 289 290 291 292 293 294 295 296 297 298 299 300 301 302 303 304 305 306 307 308 309 310 311 312 313 314 315 316 317 318 319 320 321 322 323 324 325 326 327 328 329 330 331

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410 597 582 598 413 1 1 1.25  
NUMBER OF ELEMENTS = 410 MAXIMUM NODE POINT USED = 413

\*\*\* ELEMENT STIFFNESS FORMULATION TIME ESTIMATE (LDC 7600) \*\*\*

TYPE	STIFF	NUMBER	TIME(EACH)	TIME(ALL)
1	55	410	.0346	14.170

TOTAL TIME = 14.170 SECONDS.

ANSYS - ENGINEERING ANALYSIS SYSTEM UPI08 REV 2 B\*W7000 JAN 1, 1972  
 SWANSON ANALYSIS SYSTEMS, INC. ELIZABETH, PENNSYLVANIA 15037 PHONE (412) 751-1940  
 BRETFLED STEEL STACK ANSYS ANALYSIS FOR DL + 9.2PSF E TO W LAT. WIND CP = 9.7850 5/11/77  
 PF = 0.000 3.078

\*\*\*\*\* NODE DEFINITIONS (CARD F) \*\*\*\*\*

NODE	X (OR R)	Y (OR META)	Z (OR PHI)	INXZ (OR RT)	ROTATION (DEGREES)		CYLINDRICAL
					(TZ OR TP)	(TX OR MP)	
1	179.38	0.	0.	0.	0.	0.	CYLINDRICAL
2	179.38	25.500	0.	0.	0.	0.	CYLINDRICAL
3	179.38	50.999	0.	0.	0.	0.	CYLINDRICAL
4	179.38	76.499	0.	0.	0.	0.	CYLINDRICAL
5	179.38	101.999	0.	0.	0.	0.	CYLINDRICAL
6	179.38	127.499	0.	0.	0.	0.	CYLINDRICAL
7	179.38	152.999	0.	0.	0.	0.	CYLINDRICAL
8	179.38	178.499	0.	0.	0.	0.	CYLINDRICAL
9	179.38	203.999	0.	0.	0.	0.	CYLINDRICAL
10	179.38	229.499	0.	0.	0.	0.	CYLINDRICAL
11	179.38	254.999	0.	0.	0.	0.	CYLINDRICAL
12	179.38	280.499	0.	0.	0.	0.	CYLINDRICAL
13	179.38	305.999	0.	0.	0.	0.	CYLINDRICAL
14	179.38	331.499	0.	0.	0.	0.	CYLINDRICAL
15	179.38	356.999	0.	0.	0.	0.	CYLINDRICAL
16	179.38	382.499	0.	0.	0.	0.	CYLINDRICAL
17	179.38	407.999	0.	0.	0.	0.	CYLINDRICAL
18	179.38	433.499	0.	0.	0.	0.	CYLINDRICAL
19	179.38	458.999	0.	0.	0.	0.	CYLINDRICAL
20	179.38	484.499	0.	0.	0.	0.	CYLINDRICAL
21	179.38	509.999	0.	0.	0.	0.	CYLINDRICAL
22	179.38	535.499	0.	0.	0.	0.	CYLINDRICAL
23	179.38	560.999	0.	0.	0.	0.	CYLINDRICAL
24	179.38	586.499	0.	0.	0.	0.	CYLINDRICAL
25	179.38	611.999	0.	0.	0.	0.	CYLINDRICAL

29	174.84	142.50	36.000	0.00	CYLINDRICAL
30	174.84	148.87	36.000	0.00	CYLINDRICAL
31	174.84	150.00	36.000	0.00	CYLINDRICAL
32	174.84	157.50	36.000	0.00	CYLINDRICAL
33	174.84	157.50	36.000	0.00	CYLINDRICAL
34	174.84	165.00	36.000	0.00	CYLINDRICAL
35	174.84	172.50	36.000	0.00	CYLINDRICAL
36	174.84	180.00	36.000	0.00	CYLINDRICAL
37	174.84	167.50	36.000	0.00	CYLINDRICAL
38	174.84	195.00	36.000	0.00	CYLINDRICAL
39	174.84	202.50	36.000	0.00	CYLINDRICAL
40	174.84	210.00	36.000	0.00	CYLINDRICAL
41	174.84	217.50	36.000	0.00	CYLINDRICAL
42	174.84	225.00	36.000	0.00	CYLINDRICAL
43	174.84	112.50	72.000	0.00	CYLINDRICAL
44	174.84	135.00	72.000	0.00	CYLINDRICAL
45	174.84	142.50	72.000	0.00	CYLINDRICAL
46	174.84	150.00	72.000	0.00	CYLINDRICAL
47	174.84	157.50	72.000	0.00	CYLINDRICAL
48	174.84	165.00	72.000	0.00	CYLINDRICAL
49	174.84	172.50	72.000	0.00	CYLINDRICAL
50	174.84	180.00	72.000	0.00	CYLINDRICAL
51	174.84	187.50	72.000	0.00	CYLINDRICAL
52	174.84	195.00	72.000	0.00	CYLINDRICAL
53	174.84	202.50	72.000	0.00	CYLINDRICAL
54	174.84	210.00	72.000	0.00	CYLINDRICAL
55	174.84	217.50	72.000	0.00	CYLINDRICAL
56	174.84	225.00	72.000	0.00	CYLINDRICAL
57	174.84	247.50	96.000	0.00	CYLINDRICAL
58	174.84	135.00	96.000	0.00	CYLINDRICAL
59	174.84	142.50	96.000	0.00	CYLINDRICAL
60	174.84	150.00	96.000	0.00	CYLINDRICAL
61	174.84	157.50	96.000	0.00	CYLINDRICAL
62	174.84	165.00	96.000	0.00	CYLINDRICAL
63	174.84	172.50	96.000	0.00	CYLINDRICAL

54	177.94	180.00	96.000	0.00	CYLINDRICAL
55	177.94	187.50	96.000	0.00	CYLINDRICAL
56	176.41	195.00	96.000	0.00	CYLINDRICAL
57	177.94	202.50	96.000	0.00	CYLINDRICAL
58	177.94	210.00	96.000	0.00	CYLINDRICAL
59	177.94	217.50	96.000	0.00	CYLINDRICAL
60	177.94	225.00	96.000	0.00	CYLINDRICAL
61	177.94	0.00	96.000	0.00	CYLINDRICAL
62	177.94	23.500	96.000	0.00	CYLINDRICAL
63	177.94	45.000	96.000	0.00	CYLINDRICAL
64	177.94	67.500	96.000	0.00	CYLINDRICAL
65	177.94	90.000	96.000	0.00	CYLINDRICAL
66	177.94	112.50	96.000	0.00	CYLINDRICAL
67	177.94	135.00	96.000	0.00	CYLINDRICAL
68	177.94	157.50	96.000	0.00	CYLINDRICAL
69	177.94	180.00	96.000	0.00	CYLINDRICAL
70	177.94	202.50	96.000	0.00	CYLINDRICAL
71	177.94	210.00	96.000	0.00	CYLINDRICAL
72	177.94	217.50	96.000	0.00	CYLINDRICAL
73	177.94	225.00	96.000	0.00	CYLINDRICAL
74	177.94	247.50	96.000	0.00	CYLINDRICAL
75	177.94	270.00	96.000	0.00	CYLINDRICAL
76	177.94	292.50	96.000	0.00	CYLINDRICAL
77	177.94	315.00	96.000	0.00	CYLINDRICAL
78	177.94	337.50	96.000	0.00	CYLINDRICAL
79	177.94	350.00	96.000	0.00	CYLINDRICAL
80	177.94	0.00	96.000	0.00	CYLINDRICAL
81	177.94	42.50	96.000	0.00	CYLINDRICAL
82	177.94	85.00	96.000	0.00	CYLINDRICAL
83	177.94	127.50	96.000	0.00	CYLINDRICAL
84	177.94	170.00	96.000	0.00	CYLINDRICAL
85	177.94	212.50	96.000	0.00	CYLINDRICAL
86	177.94	255.00	96.000	0.00	CYLINDRICAL
87	177.94	297.50	96.000	0.00	CYLINDRICAL
88	177.94	340.00	96.000	0.00	CYLINDRICAL
89	177.94	382.50	96.000	0.00	CYLINDRICAL
90	177.94	425.00	96.000	0.00	CYLINDRICAL
91	177.94	467.50	96.000	0.00	CYLINDRICAL
92	177.94	510.00	96.000	0.00	CYLINDRICAL
93	177.94	552.50	96.000	0.00	CYLINDRICAL
94	177.94	595.00	96.000	0.00	CYLINDRICAL
95	177.94	637.50	96.000	0.00	CYLINDRICAL
96	177.94	680.00	96.000	0.00	CYLINDRICAL
97	177.94	722.50	96.000	0.00	CYLINDRICAL
98	177.94	765.00	96.000	0.00	CYLINDRICAL
99	177.94	807.50	96.000	0.00	CYLINDRICAL
100	177.94	850.00	96.000	0.00	CYLINDRICAL





158	174.34	202.50	336.00	0.00	0.00	CYLINDRICAL
159	170.04	-61.15	336.00	0.00	0.00	CYLINDRICAL
160	150.94	-210.00	336.00	0.00	0.00	CYLINDRICAL
161	174.34	-217.50	336.00	0.00	0.00	CYLINDRICAL
162	174.34	-225.00	336.00	0.00	0.00	CYLINDRICAL
163	174.34	-247.50	336.00	0.00	0.00	CYLINDRICAL
164	173.09	-135.00	336.00	0.00	0.00	CYLINDRICAL
165	173.09	-142.50	336.00	0.00	0.00	CYLINDRICAL
166	173.09	-150.00	336.00	0.00	0.00	CYLINDRICAL
167	173.09	-157.50	336.00	0.00	0.00	CYLINDRICAL
168	173.09	-165.00	336.00	0.00	0.00	CYLINDRICAL
169	173.09	-172.50	336.00	0.00	0.00	CYLINDRICAL
170	173.09	-180.00	336.00	0.00	0.00	CYLINDRICAL
171	173.09	-187.50	336.00	0.00	0.00	CYLINDRICAL
172	173.09	-195.00	336.00	0.00	0.00	CYLINDRICAL
173	173.09	-202.50	336.00	0.00	0.00	CYLINDRICAL
174	173.09	-210.00	336.00	0.00	0.00	CYLINDRICAL
175	173.09	-217.50	336.00	0.00	0.00	CYLINDRICAL
176	173.09	-225.00	336.00	0.00	0.00	CYLINDRICAL
177	173.09	0.00	336.00	0.00	0.00	CYLINDRICAL
178	173.09	23.50	336.00	0.00	0.00	CYLINDRICAL
179	173.09	45.00	336.00	0.00	0.00	CYLINDRICAL
180	173.09	67.50	336.00	0.00	0.00	CYLINDRICAL
181	173.09	90.00	336.00	0.00	0.00	CYLINDRICAL
182	173.09	112.50	336.00	0.00	0.00	CYLINDRICAL
183	173.09	135.00	336.00	0.00	0.00	CYLINDRICAL
184	173.09	142.50	336.00	0.00	0.00	CYLINDRICAL
185	173.09	150.00	336.00	0.00	0.00	CYLINDRICAL
186	173.09	157.50	336.00	0.00	0.00	CYLINDRICAL
187	173.09	165.00	336.00	0.00	0.00	CYLINDRICAL
188	173.09	172.50	336.00	0.00	0.00	CYLINDRICAL
189	173.09	180.00	336.00	0.00	0.00	CYLINDRICAL
190	173.09	187.50	336.00	0.00	0.00	CYLINDRICAL
191	173.09	195.00	336.00	0.00	0.00	CYLINDRICAL

191	172.90	202.50	432.00	CYLINDRICAL	0.00
192	159.73	-66.164	432.00	CYLINDRICAL	0.00
193	172.90	-210.448	432.00	CYLINDRICAL	0.00
194	172.90	-217.50	432.00	CYLINDRICAL	0.00
195	172.90	-225.00	432.00	CYLINDRICAL	0.00
196	172.90	-247.50	432.00	CYLINDRICAL	0.00
197	264.58E-08	270.00	432.00	CYLINDRICAL	0.00
198	66.164	292.50	432.00	CYLINDRICAL	0.00
199	172.90	315.00	432.00	CYLINDRICAL	0.00
200	172.90	337.50	432.00	CYLINDRICAL	0.00
201	171.64	35.00	516.00	CYLINDRICAL	0.00
202	171.64	142.50	516.00	CYLINDRICAL	0.00
203	171.64	150.00	516.00	CYLINDRICAL	0.00
204	171.64	157.50	516.00	CYLINDRICAL	0.00
205	171.64	165.00	516.00	CYLINDRICAL	0.00
206	171.64	172.50	516.00	CYLINDRICAL	0.00
207	171.64	180.00	516.00	CYLINDRICAL	0.00
208	171.64	187.50	516.00	CYLINDRICAL	0.00
209	171.64	195.00	516.00	CYLINDRICAL	0.00
210	171.64	202.50	516.00	CYLINDRICAL	0.00
211	171.64	210.00	516.00	CYLINDRICAL	0.00
212	171.64	217.50	516.00	CYLINDRICAL	0.00
213	171.64	225.00	516.00	CYLINDRICAL	0.00
214	170.38	0.00	600.00	CYLINDRICAL	0.00
215	170.38	22.500	600.00	CYLINDRICAL	0.00
216	170.38	45.000	600.00	CYLINDRICAL	0.00
217	170.38	67.500	600.00	CYLINDRICAL	0.00
218	170.38	90.000	600.00	CYLINDRICAL	0.00
219	170.38	112.50	600.00	CYLINDRICAL	0.00
220	170.38	135.00	600.00	CYLINDRICAL	0.00
221	170.38	142.50	600.00	CYLINDRICAL	0.00
222	170.38	150.00	600.00	CYLINDRICAL	0.00
223	170.38	157.50	600.00	CYLINDRICAL	0.00
224	170.38	165.00	600.00	CYLINDRICAL	0.00
225	164.57	44.096	600.00	CYLINDRICAL	0.00

RE-08

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224	170.58	172.50	600.00	0.00	CYLINDRICAL	0.00
225	168.92	180.00	600.00	0.00	CYLINDRICAL	0.00
226	170.38	187.50	600.00	0.00	CYLINDRICAL	0.00
227	168.42	195.00	600.00	0.00	CYLINDRICAL	0.00
228	168.57	202.50	600.00	0.00	CYLINDRICAL	0.00
229	170.38	210.00	600.00	0.00	CYLINDRICAL	0.00
230	174.41	217.50	600.00	0.00	CYLINDRICAL	0.00
231	174.55	225.00	600.00	0.00	CYLINDRICAL	0.00
232	170.38	247.50	600.00	0.00	CYLINDRICAL	0.00
233	170.38	270.00	600.00	0.00	CYLINDRICAL	0.00
234	170.38	292.50	600.00	0.00	CYLINDRICAL	0.00
235	170.38	315.00	600.00	0.00	CYLINDRICAL	0.00
236	170.38	337.50	600.00	0.00	CYLINDRICAL	0.00
237	169.42	35.00	684.00	0.00	CYLINDRICAL	0.00
238	169.42	142.50	684.00	0.00	CYLINDRICAL	0.00
239	169.42	150.00	684.00	0.00	CYLINDRICAL	0.00
240	169.42	157.50	684.00	0.00	CYLINDRICAL	0.00
241	169.42	165.00	684.00	0.00	CYLINDRICAL	0.00
242	169.42	172.50	684.00	0.00	CYLINDRICAL	0.00
243	169.42	180.00	684.00	0.00	CYLINDRICAL	0.00
244	169.42	187.50	684.00	0.00	CYLINDRICAL	0.00
245	169.42	195.00	684.00	0.00	CYLINDRICAL	0.00
246	169.42	202.50	684.00	0.00	CYLINDRICAL	0.00
247	169.42	210.00	684.00	0.00	CYLINDRICAL	0.00
248	169.42	217.50	684.00	0.00	CYLINDRICAL	0.00
249	169.42	225.00	684.00	0.00	CYLINDRICAL	0.00
250	167.58	0.00	768.00	0.00	CYLINDRICAL	0.00
251	167.58	0.00	768.00	0.00	CYLINDRICAL	0.00
252	155.08	22.50	768.00	0.00	CYLINDRICAL	0.00
253	167.58	45.00	768.00	0.00	CYLINDRICAL	0.00
254	167.58	67.50	768.00	0.00	CYLINDRICAL	0.00
255	167.58	90.00	768.00	0.00	CYLINDRICAL	0.00
256	167.58	112.50	768.00	0.00	CYLINDRICAL	0.00
257	167.58	135.00	768.00	0.00	CYLINDRICAL	0.00
258	167.58	0.00	768.00	0.00	CYLINDRICAL	0.00
259	167.58	0.00	768.00	0.00	CYLINDRICAL	0.00
260	167.58	0.00	768.00	0.00	CYLINDRICAL	0.00
261	167.58	0.00	768.00	0.00	CYLINDRICAL	0.00
262	167.58	0.00	768.00	0.00	CYLINDRICAL	0.00
263	167.58	0.00	768.00	0.00	CYLINDRICAL	0.00
264	167.58	0.00	768.00	0.00	CYLINDRICAL	0.00
265	167.58	0.00	768.00	0.00	CYLINDRICAL	0.00
266	167.58	0.00	768.00	0.00	CYLINDRICAL	0.00
267	167.58	0.00	768.00	0.00	CYLINDRICAL	0.00
268	167.58	0.00	768.00	0.00	CYLINDRICAL	0.00
269	167.58	0.00	768.00	0.00	CYLINDRICAL	0.00
270	167.58	0.00	768.00	0.00	CYLINDRICAL	0.00
271	167.58	0.00	768.00	0.00	CYLINDRICAL	0.00
272	167.58	0.00	768.00	0.00	CYLINDRICAL	0.00
273	167.58	0.00	768.00	0.00	CYLINDRICAL	0.00
274	167.58	0.00	768.00	0.00	CYLINDRICAL	0.00
275	167.58	0.00	768.00	0.00	CYLINDRICAL	0.00
276	167.58	0.00	768.00	0.00	CYLINDRICAL	0.00
277	167.58	0.00	768.00	0.00	CYLINDRICAL	0.00
278	167.58	0.00	768.00	0.00	CYLINDRICAL	0.00
279	167.58	0.00	768.00	0.00	CYLINDRICAL	0.00
280	167.58	0.00	768.00	0.00	CYLINDRICAL	0.00
281	167.58	0.00	768.00	0.00	CYLINDRICAL	0.00
282	167.58	0.00	768.00	0.00	CYLINDRICAL	0.00
283	167.58	0.00	768.00	0.00	CYLINDRICAL	0.00
284	167.58	0.00	768.00	0.00	CYLINDRICAL	0.00
285	167.58	0.00	768.00	0.00	CYLINDRICAL	0.00
286	167.58	0.00	768.00	0.00	CYLINDRICAL	0.00
287	167.58	0.00	768.00	0.00	CYLINDRICAL	0.00
288	167.58	0.00	768.00	0.00	CYLINDRICAL	0.00
289	167.58	0.00	768.00	0.00	CYLINDRICAL	0.00
290	167.58	0.00	768.00	0.00	CYLINDRICAL	0.00
291	167.58	0.00	768.00	0.00	CYLINDRICAL	0.00
292	167.58	0.00	768.00	0.00	CYLINDRICAL	0.00
293	167.58	0.00	768.00	0.00	CYLINDRICAL	0.00
294	167.58	0.00	768.00	0.00	CYLINDRICAL	0.00
295	167.58	0.00	768.00	0.00	CYLINDRICAL	0.00
296	167.58	0.00	768.00	0.00	CYLINDRICAL	0.00
297	167.58	0.00	768.00	0.00	CYLINDRICAL	0.00
298	167.58	0.00	768.00	0.00	CYLINDRICAL	0.00
299	167.58	0.00	768.00	0.00	CYLINDRICAL	0.00
300	167.58	0.00	768.00	0.00	CYLINDRICAL	0.00

257	167.00	142.50	768.00	0.00	CYLINDRICAL
258	137.47	0.00	768.00	0.00	CYLINDRICAL
259	187.97	0.00	768.00	0.00	CYLINDRICAL
260	157.08	0.00	768.00	0.00	CYLINDRICAL
261	167.44	0.00	768.00	0.00	CYLINDRICAL
262	167.44	0.00	768.00	0.00	CYLINDRICAL
263	167.44	0.00	768.00	0.00	CYLINDRICAL
264	167.44	0.00	768.00	0.00	CYLINDRICAL
265	167.44	0.00	768.00	0.00	CYLINDRICAL
266	167.44	0.00	768.00	0.00	CYLINDRICAL
267	167.44	0.00	768.00	0.00	CYLINDRICAL
268	167.44	0.00	768.00	0.00	CYLINDRICAL
269	167.44	0.00	768.00	0.00	CYLINDRICAL
270	167.44	0.00	768.00	0.00	CYLINDRICAL
271	167.44	0.00	768.00	0.00	CYLINDRICAL
272	167.44	0.00	768.00	0.00	CYLINDRICAL
273	167.44	0.00	768.00	0.00	CYLINDRICAL
274	167.44	0.00	768.00	0.00	CYLINDRICAL
275	167.44	0.00	768.00	0.00	CYLINDRICAL
276	167.44	0.00	768.00	0.00	CYLINDRICAL
277	167.44	0.00	768.00	0.00	CYLINDRICAL
278	167.44	0.00	768.00	0.00	CYLINDRICAL
279	167.44	0.00	768.00	0.00	CYLINDRICAL
280	167.44	0.00	768.00	0.00	CYLINDRICAL
281	167.44	0.00	768.00	0.00	CYLINDRICAL
282	167.44	0.00	768.00	0.00	CYLINDRICAL
283	167.44	0.00	768.00	0.00	CYLINDRICAL
284	167.44	0.00	768.00	0.00	CYLINDRICAL
285	167.44	0.00	768.00	0.00	CYLINDRICAL
286	167.44	0.00	768.00	0.00	CYLINDRICAL
287	167.44	0.00	768.00	0.00	CYLINDRICAL
288	167.44	0.00	768.00	0.00	CYLINDRICAL
289	167.44	0.00	768.00	0.00	CYLINDRICAL
289	167.44	0.00	768.00	0.00	CYLINDRICAL

290	64.94	67.50	960.00	0.00	CYLINDRICAL
291	64.94	90.00	960.00	0.00	CYLINDRICAL
292	64.94	112.50	960.00	0.00	CYLINDRICAL
293	64.94	135.00	960.00	0.00	CYLINDRICAL
294	64.94	142.50	960.00	0.00	CYLINDRICAL
295	64.94	150.00	960.00	0.00	CYLINDRICAL
296	64.94	157.50	960.00	0.00	CYLINDRICAL
297	64.94	165.00	960.00	0.00	CYLINDRICAL
298	64.94	172.50	960.00	0.00	CYLINDRICAL
299	64.94	180.00	960.00	0.00	CYLINDRICAL
300	64.94	187.50	960.00	0.00	CYLINDRICAL
301	64.94	195.00	960.00	0.00	CYLINDRICAL
302	64.94	202.50	960.00	0.00	CYLINDRICAL
303	64.94	210.00	960.00	0.00	CYLINDRICAL
304	64.94	217.50	960.00	0.00	CYLINDRICAL
305	64.94	225.00	960.00	0.00	CYLINDRICAL
306	64.94	247.50	960.00	0.00	CYLINDRICAL
307	64.94	270.00	960.00	0.00	CYLINDRICAL
308	64.94	292.50	960.00	0.00	CYLINDRICAL
309	64.94	315.00	960.00	0.00	CYLINDRICAL
310	64.94	337.50	960.00	0.00	CYLINDRICAL
311	64.94	350.00	1080.00	0.00	CYLINDRICAL
312	64.94	350.00	1080.00	0.00	CYLINDRICAL
313	64.94	415.50	1080.00	0.00	CYLINDRICAL
314	64.94	480.00	1080.00	0.00	CYLINDRICAL
315	64.94	495.00	1080.00	0.00	CYLINDRICAL
316	64.94	210.00	1080.00	0.00	CYLINDRICAL
317	64.94	225.00	1080.00	0.00	CYLINDRICAL
318	64.94	0.00	1200.00	0.00	CYLINDRICAL
319	64.94	22.50	1200.00	0.00	CYLINDRICAL
320	64.94	45.00	1200.00	0.00	CYLINDRICAL
321	64.94	67.50	1200.00	0.00	CYLINDRICAL
322	64.94	90.00	1200.00	0.00	CYLINDRICAL

164.94E-08  
164.94E-08

64.94E-04  
64.94E-08



356	154.14	109.02	135.00	0	CYLINDRICAL	0
357	142.44	107.50	137.50	0	CYLINDRICAL	0
358	154.14	109.02	140.00	0	CYLINDRICAL	0
359	142.44	107.50	160.00	0	CYLINDRICAL	0
360	154.14	109.02	202.50	0	CYLINDRICAL	0
361	154.14	109.02	225.00	0	CYLINDRICAL	0
362	154.14	109.02	247.50	0	CYLINDRICAL	0
363	154.14	109.02	270.00	0	CYLINDRICAL	0
364	154.14	109.02	292.50	0	CYLINDRICAL	0
365	154.14	109.02	315.00	0	CYLINDRICAL	0
366	154.14	109.02	337.50	0	CYLINDRICAL	0
367	154.14	109.02	0	0	CYLINDRICAL	0
368	154.14	109.02	0	0	CYLINDRICAL	0
369	154.14	109.02	0	0	CYLINDRICAL	0
370	154.14	109.02	0	0	CYLINDRICAL	0
371	154.14	109.02	0	0	CYLINDRICAL	0
372	154.14	109.02	0	0	CYLINDRICAL	0
373	154.14	109.02	0	0	CYLINDRICAL	0
374	154.14	109.02	0	0	CYLINDRICAL	0
375	154.14	109.02	0	0	CYLINDRICAL	0
376	154.14	109.02	0	0	CYLINDRICAL	0
377	154.14	109.02	0	0	CYLINDRICAL	0
378	154.14	109.02	0	0	CYLINDRICAL	0
379	154.14	109.02	0	0	CYLINDRICAL	0
380	154.14	109.02	0	0	CYLINDRICAL	0
381	154.14	109.02	0	0	CYLINDRICAL	0
382	154.14	109.02	0	0	CYLINDRICAL	0
383	154.14	109.02	0	0	CYLINDRICAL	0
384	154.14	109.02	0	0	CYLINDRICAL	0
385	154.14	109.02	0	0	CYLINDRICAL	0
386	154.14	109.02	0	0	CYLINDRICAL	0
387	154.14	109.02	0	0	CYLINDRICAL	0
388	154.14	109.02	0	0	CYLINDRICAL	0

389	146.74	157.50	2160.0	0.0	0.0	CYLINDRICAL
390	-135.74	50.245	1600.0	0.0	0.0	CYLINDRICAL
391	-146.94	-150.00	1600.0	0.0	0.0	CYLINDRICAL
392	-146.94	-202.245	1600.0	0.0	0.0	CYLINDRICAL
393	-146.94	-225.00	1600.0	0.0	0.0	CYLINDRICAL
394	-146.94	-247.50	1600.0	0.0	0.0	CYLINDRICAL
395	122.47E-08	-270.00	1600.0	0.0	0.0	CYLINDRICAL
396	146.94	-292.50	1600.0	0.0	0.0	CYLINDRICAL
397	146.94	-315.00	1600.0	0.0	0.0	CYLINDRICAL
398	146.94	-337.50	1600.0	0.0	0.0	CYLINDRICAL
399	143.38	0.0	2400.0	0.0	0.0	CYLINDRICAL
400	143.38	2.500	2400.0	0.0	0.0	CYLINDRICAL
401	143.38	45.000	2400.0	0.0	0.0	CYLINDRICAL
402	143.38	67.500	2400.0	0.0	0.0	CYLINDRICAL
403	143.38	90.000	2400.0	0.0	0.0	CYLINDRICAL
404	143.38	112.500	2400.0	0.0	0.0	CYLINDRICAL
405	143.38	135.000	2400.0	0.0	0.0	CYLINDRICAL
406	143.38	157.500	2400.0	0.0	0.0	CYLINDRICAL
407	143.38	180.000	2400.0	0.0	0.0	CYLINDRICAL
408	143.38	146.34E-08	2400.0	0.0	0.0	CYLINDRICAL
409	143.38	202.50	2400.0	0.0	0.0	CYLINDRICAL
410	143.38	225.00	2400.0	0.0	0.0	CYLINDRICAL
411	143.38	247.50	2400.0	0.0	0.0	CYLINDRICAL
412	143.38	270.00	2400.0	0.0	0.0	CYLINDRICAL
413	143.38	292.50	2400.0	0.0	0.0	CYLINDRICAL
		315.00	2400.0	0.0	0.0	CYLINDRICAL
		337.50	2400.0	0.0	0.0	CYLINDRICAL

XMIN= -179.4 XMAX= 179.4 YMIN= -179.4 YMAX= 179.4 ZMIN= 0.0 ZMAX= 240.0

\*\*\*\*\* MATERIAL PROPERTIES (CARD H) \*\*\*\*\*

MATERIAL 1

EX = 2400000E+08  
 ALFA = 0  
 MUAY = .300000  
 BENS = .280000

SPANSON ANALYSIS SYSTEMS, INC. ENGINEERING ANALYSIS SYSTEM UPI88 REV 2 04/17/72  
 ELIZABETH, PENNSYLVANIA 15037 PHONE (412) 751-1940  
 LOAD COMBINATION, DL + 9.2PSF E TO W LATERAL WIND CP = 4.7853 5/11/77  
 LOAD STEP NUMBER = 1 PP# 0.000

\*\*\*\*\* LOAD STEP OPTIONS (CARD L AND M) \*\*\*\*\*

VALUE	VARIABLE NAME	COLUMNS
1	KOIS	2-4
0	KYCRP	5-6
1	NITIER	7-9
1	NPRINT	10-12
-0.	TIME	13-14
1	NOPRNT	15-17
1	TITLE	78
-0.	COORDINATE ACCELERATIONS	1.0000



\*\*\*\*\* SPECIFIED DISPLACEMENTS (CARD N) \*\*\*\*\*

NO.	NODE	DIRECTION	VALUE
1	1	UX	0.
2	2	UX	0.
3	3	UX	0.
4	4	UX	0.
5	5	UX	0.
6	6	UX	0.
7	7	UX	0.
8	8	UX	0.
9	9	UX	0.
10	10	UX	0.
11	11	UX	0.
12	12	UX	0.
13	13	UX	0.
14	14	UX	0.
15	15	UX	0.
16	16	UX	0.
17	17	UX	0.
18	18	UX	0.
19	19	UX	0.
20	20	UX	0.
21	21	UX	0.
22	22	UX	0.
23	23	UX	0.
24	24	UX	0.
1	1	UY	0.
2	2	UY	0.
3	3	UY	0.
4	4	UY	0.
5	5	UY	0.
6	6	UY	0.
7	7	UY	0.
8	8	UY	0.
9	9	UY	0.
10	10	UY	0.
11	11	UY	0.
12	12	UY	0.
13	13	UY	0.







19	671.000	FZ	257
20	894.000	FZ	220
21	901.000	FZ	214
22	1021.000	FZ	216
23	2302.000	FZ	217
24	2486.000	FZ	220
25	1340.000	FZ	222
26	423.000	FZ	223
27	321.000	FZ	224
28	1114.000	FZ	227
29	1121.000	FZ	228
30	2321.000	FZ	229
31	42.000	FZ	230
32	51.000	FZ	231
33	140.000	FZ	232
34	302.000	FZ	233
35	248.000	FZ	235
36	1781.000	FZ	236
37	1022.000	FZ	237
38	5199.000	FZ	238
39	4795.000	FZ	239
40	3670.000	FZ	240
41	1986.000	FZ	241
42	1986.000	FZ	242
43	1986.000	FZ	243
44	1986.000	FZ	244
45	1986.000	FZ	245
46	1986.000	FZ	246
47	1986.000	FZ	247
48	1986.000	FZ	248
49	1986.000	FZ	249
50	1986.000	FZ	250
51	1986.000	FZ	251
52	1986.000	FZ	252
53	1986.000	FZ	253
54	1986.000	FZ	254
55	1986.000	FZ	255
56	1986.000	FZ	256
57	1986.000	FZ	257
58	1986.000	FZ	258
59	1986.000	FZ	259
60	1986.000	FZ	260
61	1986.000	FZ	261
62	1986.000	FZ	262
63	1986.000	FZ	263
64	1986.000	FZ	264
65	1986.000	FZ	265
66	1986.000	FZ	266
67	1986.000	FZ	267
68	1986.000	FZ	268
69	1986.000	FZ	269
70	1986.000	FZ	270
71	1986.000	FZ	271
72	1986.000	FZ	272
73	1986.000	FZ	273
74	1986.000	FZ	274
75	1986.000	FZ	275
76	1986.000	FZ	276
77	1986.000	FZ	277
78	1986.000	FZ	278
79	1986.000	FZ	279
80	1986.000	FZ	280
81	1986.000	FZ	281
82	1986.000	FZ	282
83	1986.000	FZ	283
84	1986.000	FZ	284



151	411	FZ	54275.0
152	402	FZ	-56405.0
153	410	FZ	-56405.0
154	403	FZ	-171085.0
155	404	FZ	-171085.0
156	404	FZ	-266611.0
157	408	FZ	-330439.0
158	405	FZ	-330439.0
159	407	FZ	-352253.0
160	406	FZ	94338.00
161	399	FY	17125.00
162	405	FY	17125.00
163	410	FY	17125.00
164	404	FY	221338.00
165	401	FY	221338.00
166	403	FY	23009.00
167	402	FY	23009.00
168	407	FY	-9858.00
169	413	FY	-9858.00
170	404	FY	-17125.00
171	412	FY	-17125.00
172	409	FY	-231338.00
173	411	FY	-231338.00
174	410	FY	-231338.00

\*\*\*\*\* LOAD SUMMARY = 144 DISPLACEMENTS 174 FORCES 0 PRESSURES \*\*\*\*\*

MAXIMUM STIFFNESS = .480896E+09 MINIMUM STIFFNESS = .167016E+08

MAXIMUM WAVE FRONT (EQUATIONS) USED 162  
MAXIMUM WAVE FRONT FOR THIS CORE SIZE 235

LCM CORE REQUIRED(CRITICAL)00043563

LCM CORE REQUIRED FOR RANDOM NODE ORDER INPUT (CRITICAL) 00006626

TIMES AT START OF BACK-SUBSTITUTION CP= 69.971 PP= 0.000 STEP= 1 ITERATIONS 1



52	697122-02	103727E-01	106976E-02	214274E-03	750002E-03
53	977002-01	501297E-02	155412E-02	115231E-02	214835E-02
54	134874	429486E-01	407525E-01	225921E-02	115041E-02
55	136744	419147E-01	705777E-01	343322E-02	415047E-02
56	741443E-01	269141E-01	180002E-02	415047E-02	415047E-02
57	122414	599069E-02	444099E-02	235822E-02	415047E-02
58	122414	627716E-02	627683E-02	235822E-02	415047E-02
59	122414	599069E-02	444099E-02	235822E-02	415047E-02
60	122414	627716E-02	627683E-02	235822E-02	415047E-02
61	625599E-01	294584E-01	444110E-02	215804E-02	415047E-02
62	625599E-01	294584E-01	444110E-02	215804E-02	415047E-02
63	134874	228722E-01	103908E-03	243322E-02	415047E-02
64	134874	228722E-01	103908E-03	243322E-02	415047E-02
65	134874	228722E-01	103908E-03	243322E-02	415047E-02
66	134874	228722E-01	103908E-03	243322E-02	415047E-02
67	134874	228722E-01	103908E-03	243322E-02	415047E-02
68	134874	228722E-01	103908E-03	243322E-02	415047E-02
69	134874	228722E-01	103908E-03	243322E-02	415047E-02
70	134874	228722E-01	103908E-03	243322E-02	415047E-02
71	134874	228722E-01	103908E-03	243322E-02	415047E-02
72	134874	228722E-01	103908E-03	243322E-02	415047E-02
73	134874	228722E-01	103908E-03	243322E-02	415047E-02
74	134874	228722E-01	103908E-03	243322E-02	415047E-02
75	134874	228722E-01	103908E-03	243322E-02	415047E-02
76	134874	228722E-01	103908E-03	243322E-02	415047E-02
77	134874	228722E-01	103908E-03	243322E-02	415047E-02
78	134874	228722E-01	103908E-03	243322E-02	415047E-02
79	134874	228722E-01	103908E-03	243322E-02	415047E-02
80	134874	228722E-01	103908E-03	243322E-02	415047E-02
81	134874	228722E-01	103908E-03	243322E-02	415047E-02
82	134874	228722E-01	103908E-03	243322E-02	415047E-02
83	134874	228722E-01	103908E-03	243322E-02	415047E-02
84	134874	228722E-01	103908E-03	243322E-02	415047E-02
85	134874	228722E-01	103908E-03	243322E-02	415047E-02
86	134874	228722E-01	103908E-03	243322E-02	415047E-02
87	134874	228722E-01	103908E-03	243322E-02	415047E-02
88	134874	228722E-01	103908E-03	243322E-02	415047E-02
89	134874	228722E-01	103908E-03	243322E-02	415047E-02
90	134874	228722E-01	103908E-03	243322E-02	415047E-02
91	134874	228722E-01	103908E-03	243322E-02	415047E-02
92	134874	228722E-01	103908E-03	243322E-02	415047E-02
93	134874	228722E-01	103908E-03	243322E-02	415047E-02
94	134874	228722E-01	103908E-03	243322E-02	415047E-02
95	134874	228722E-01	103908E-03	243322E-02	415047E-02
96	134874	228722E-01	103908E-03	243322E-02	415047E-02
97	134874	228722E-01	103908E-03	243322E-02	415047E-02
98	134874	228722E-01	103908E-03	243322E-02	415047E-02
99	134874	228722E-01	103908E-03	243322E-02	415047E-02
100	134874	228722E-01	103908E-03	243322E-02	415047E-02
101	134874	228722E-01	103908E-03	243322E-02	415047E-02
102	134874	228722E-01	103908E-03	243322E-02	415047E-02
103	134874	228722E-01	103908E-03	243322E-02	415047E-02
104	134874	228722E-01	103908E-03	243322E-02	415047E-02
105	134874	228722E-01	103908E-03	243322E-02	415047E-02
106	134874	228722E-01	103908E-03	243322E-02	415047E-02
107	134874	228722E-01	103908E-03	243322E-02	415047E-02
108	134874	228722E-01	103908E-03	243322E-02	415047E-02
109	134874	228722E-01	103908E-03	243322E-02	415047E-02
110	134874	228722E-01	103908E-03	243322E-02	415047E-02
111	134874	228722E-01	103908E-03	243322E-02	415047E-02
112	134874	228722E-01	103908E-03	243322E-02	415047E-02
113	134874	228722E-01	103908E-03	243322E-02	415047E-02
114	134874	228722E-01	103908E-03	243322E-02	415047E-02
115	134874	228722E-01	103908E-03	243322E-02	415047E-02
116	134874	228722E-01	103908E-03	243322E-02	415047E-02
117	134874	228722E-01	103908E-03	243322E-02	415047E-02







04 718E 11  
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 11443357E 002  
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 15530507E 002  
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33	33	10982		33	33	60522		33	33	44077		33	33	159494E-02		33	33	189983E-02		33	33	189983E-02	
34	34	10982		34	34	60522		34	34	44077		34	34	159494E-02		34	34	189983E-02		34	34	189983E-02	
35	35	10982		35	35	60522		35	35	44077		35	35	159494E-02		35	35	189983E-02		35	35	189983E-02	
36	36	10982		36	36	60522		36	36	44077		36	36	159494E-02		36	36	189983E-02		36	36	189983E-02	
37	37	10982		37	37	60522		37	37	44077		37	37	159494E-02		37	37	189983E-02		37	37	189983E-02	
38	38	10982		38	38	60522		38	38	44077		38	38	159494E-02		38	38	189983E-02		38	38	189983E-02	
39	39	10982		39	39	60522		39	39	44077		39	39	159494E-02		39	39	189983E-02		39	39	189983E-02	
40	40	10982		40	40	60522		40	40	44077		40	40	159494E-02		40	40	189983E-02		40	40	189983E-02	



387	1.50510	9.18520E-10	3.64394	1.133004E-08	3.45353E-02	2.76730E-10
388	1.37461	1.38552	3.271211	1.153268E-02	3.15069E-02	3.08355E-02
389	1.61601	1.52925	3.60069E-01	3.18380E-02	2.77743E-02	4.28099E-03
390	1.41642	9.73302	1.18171	3.16494E-02	1.15127E-03	3.03799E-02
391	1.79584	1.42477	4.40118	2.77039E-02	2.10840E-02	4.02755E-02
392	3.42804	1.81601E-08	5.34067	1.15951E-08	2.87182E-02	2.73417E-09
393	4.39184	1.295195	5.68477	1.57098E-02	3.13686E-02	1.46334E-02
394	3.92804	1.95161	5.34067	1.25951E-02	2.87182E-02	2.73417E-02
395	2.79454	2.22477	4.90118	1.82598E-02	2.10840E-02	3.03799E-03
396	1.41642	3.93302	2.95175	1.82598E-02	2.10840E-02	3.03799E-03
397	1.41193	3.4975	1.8171	3.18380E-02	1.15127E-03	4.28099E-02
398	1.79454	1.81193	5.40069E-01	2.67743E-02	2.13163E-02	2.80452E-02
399	1.37461	1.38552	3.271211	1.153268E-02	3.15069E-02	3.08355E-02
400	1.61601	1.52925	3.60069E-01	3.16494E-02	2.77039E-02	4.02755E-02
401	1.41642	1.42477	4.40118	2.77039E-02	2.10840E-02	4.02755E-02
402	1.79584	1.81601E-08	5.34067	1.15951E-08	2.87182E-02	1.46334E-02
403	3.42804	1.295195	5.68477	1.57098E-02	2.10840E-02	3.03799E-03
404	4.39184	2.22477	4.90118	1.82598E-02	2.10840E-02	3.03799E-03
405	3.92804	3.93302	2.95175	1.82598E-02	2.10840E-02	3.03799E-03
406	2.79454	1.41642	3.4975	3.18380E-02	1.15127E-03	4.28099E-02
407	1.41193	1.38552	3.271211	1.153268E-02	3.15069E-02	3.08355E-02
408	1.79454	1.81193	5.40069E-01	2.67743E-02	2.13163E-02	2.80452E-02
409	1.37461	1.38552	3.271211	1.153268E-02	3.15069E-02	3.08355E-02
410	1.61601	1.52925	3.60069E-01	3.16494E-02	2.77039E-02	4.02755E-02
411	1.41642	1.42477	4.40118	2.77039E-02	2.10840E-02	4.02755E-02
412	1.79584	1.81601E-08	5.34067	1.15951E-08	2.87182E-02	1.46334E-02
413	3.42804	1.295195	5.68477	1.57098E-02	2.10840E-02	3.03799E-03

MAXIMUM VALU 406  
 MINRES 406  
 DISPI 5.10004  
 48386 MUMS WRITTEN TO BLOCKS 1 AND 2  
 56397 MUMS WRITTEN TO BLOCK 3  
 2334 ACTIVE DEGREES OF FREEDOM  
 135.0 H.M.S. HAYEFRONT

MATRIX SOLUTION TIME ESTIMATE (CDC7600) = 42.53 SECONDS.

PER

SWANSON ANALYSIS SYSTEMS, INC. ELIZABETH, PENNSYLVANIA 15037. PHONE (412) 751-1940

LOAD COMBINATION, DL + 9.2PSF L TO W LATERAL WIND CP = 10.804/ 5/11/77 PP = 0.000

\*\*\*\*\* ELEMENT STRESSES \*\*\*\*\* TIME = 0. LOAD STEP = 1 ITERATION = 1 CUM. ILLK. = 1

LINE ELEMENT STRESSES ASSUME WEIGHT IS CONCENTRATED AT NODAL POINTS

HEAT ITOP FORCES AND ELEMENT FORCES ASSUME DISTRIBUTED WEIGHTS

Table with columns for ELEMENT, NODES, MATERIAL, AREA, SMIN, SMIN, TAUMAX, TAUMAX, XC, YC, ZC, ITOP, THOI, PRESS, and VME. It lists stress data for 10 elements across various nodes and materials.

ELEM	10	NDDESE	10	11	29	28	MATERIAL =	1	AREA =	843.50	1TOP,1BOI =	70.	70.	PMESSE =	0.
TOP SIR	1574	MX,MY,MXY =	33	30	SMAX,SMIN,IAUMAXE	30	XC, YC, ZC =	-169.2	57.45	18.00					
MID SIR	-690.	SMAX,SMIN,IAUMAXE	204	33	SMAX,SMIN,IAUMAXE	33	SMAX,SMIN,IAUMAXE	2432.	AE = 89.6	VME	5174.				
BOT SIR	-807.	SMAX,SMIN,IAUMAXE	375	375	SMAX,SMIN,IAUMAXE	375	SMAX,SMIN,IAUMAXE	-5392.	AE = 87.5	VME	5046.				
			-5331.					-776.	AE = 85.3	VME	5019.				
ELEM	11	NDDESE	11	12	30	29	MATERIAL =	1	AREA =	843.50	1TOP,1BOI =	70.	70.	PMESSE =	0.
TOP SIR	61.	MX,MY,MXY =	37	38	SMAX,SMIN,IAUMAXE	37	XC, YC, ZC =	-175.3	490.	18.00	490.				
MID SIR	-206.	SMAX,SMIN,IAUMAXE	38	38	SMAX,SMIN,IAUMAXE	38	SMAX,SMIN,IAUMAXE	-1042.	AE = 87.4	VME	954.				
BOT SIR	-352.	SMAX,SMIN,IAUMAXE	280	280	SMAX,SMIN,IAUMAXE	280	SMAX,SMIN,IAUMAXE	-1409.	AE = 87.2	VME	1293.				
			-1335.					-278.	AE = 75.2	VME					
ELEM	12	NDDESE	12	13	31	30	MATERIAL =	1	AREA =	843.50	1TOP,1BOI =	70.	70.	PMESSE =	0.
TOP SIR	211.	MX,MY,MXY =	290	44	SMAX,SMIN,IAUMAXE	290	XC, YC, ZC =	-178.3	1083.	18.00	1083.				
MID SIR	398.	SMAX,SMIN,IAUMAXE	150	44	SMAX,SMIN,IAUMAXE	150	SMAX,SMIN,IAUMAXE	201.	AE = 7.8	VME	2154.				
BOT SIR			344.	344	SMAX,SMIN,IAUMAXE	344	SMAX,SMIN,IAUMAXE	394.	AE = 4.9	VME	1528.				
									AE = 4.0	VME	975.				
ELEM	13	NDDESE	13	14	32	31	MATERIAL =	1	AREA =	843.50	1TOP,1BOI =	70.	70.	PMESSE =	0.
TOP SIR	211.	MX,MY,MXY =	290	44	SMAX,SMIN,IAUMAXE	290	XC, YC, ZC =	-178.3	1083.	18.00	1083.				
MID SIR	398.	SMAX,SMIN,IAUMAXE	150	44	SMAX,SMIN,IAUMAXE	150	SMAX,SMIN,IAUMAXE	201.	AE = 7.8	VME	2154.				
BOT SIR			344.	344	SMAX,SMIN,IAUMAXE	344	SMAX,SMIN,IAUMAXE	394.	AE = 4.9	VME	1528.				
									AE = 4.0	VME	975.				
ELEM	14	NDDESE	14	15	33	32	MATERIAL =	1	AREA =	843.50	1TOP,1BOI =	70.	70.	PMESSE =	0.
TOP SIR	206.	MX,MY,MXY =	397	38	SMAX,SMIN,IAUMAXE	397	XC, YC, ZC =	-895.	490.	18.00	490.				
MID SIR	-352.	SMAX,SMIN,IAUMAXE	34	34	SMAX,SMIN,IAUMAXE	34	SMAX,SMIN,IAUMAXE	-1044.	AE = 87.7	VME	954.				
BOT SIR			-280.	-280	SMAX,SMIN,IAUMAXE	-280	SMAX,SMIN,IAUMAXE	-278.	AE = -75.2	VME	1293.				
ELEM	15	NDDESE	15	16	34	33	MATERIAL =	1	AREA =	843.50	1TOP,1BOI =	70.	70.	PMESSE =	0.
TOP SIR	574.	MX,MY,MXY =	33	30	SMAX,SMIN,IAUMAXE	33	XC, YC, ZC =	-169.2	-57.45	18.00					
MID SIR	-690.	SMAX,SMIN,IAUMAXE	204	33	SMAX,SMIN,IAUMAXE	204	SMAX,SMIN,IAUMAXE	2432.	AE = 89.6	VME	5174.				
BOT SIR	-807.	SMAX,SMIN,IAUMAXE	375	375	SMAX,SMIN,IAUMAXE	375	SMAX,SMIN,IAUMAXE	-5392.	AE = 87.5	VME	5046.				
			-5331.					-776.	AE = 85.3	VME	5019.				
ELEM	16	NDDESE	16	17	35	34	MATERIAL =	1	AREA =	843.50	1TOP,1BOI =	70.	70.	PMESSE =	0.
TOP SIR	1182.	MX,MY,MXY =	175	57	SMAX,SMIN,IAUMAXE	175	XC, YC, ZC =	-160.3	4496.	18.00	4496.				
MID SIR	-1162.	SMAX,SMIN,IAUMAXE	77	77	SMAX,SMIN,IAUMAXE	77	SMAX,SMIN,IAUMAXE	-9407.	AE = 80.9	VME	9754.				
BOT SIR	-943.	SMAX,SMIN,IAUMAXE	160	160	SMAX,SMIN,IAUMAXE	160	SMAX,SMIN,IAUMAXE	-940.	AE = 90.0	VME	4360.				
									AE = -88.9	VME	4016.				
ELEM	17	NDDESE	17	18	36	35	MATERIAL =	1	AREA =	843.50	1TOP,1BOI =	70.	70.	PMESSE =	0.
TOP SIR	211.	MX,MY,MXY =	802	55	SMAX,SMIN,IAUMAXE	802	XC, YC, ZC =	-148.6	6042.	18.00	6042.				
MID SIR	-1515.	SMAX,SMIN,IAUMAXE	650	55	SMAX,SMIN,IAUMAXE	650	SMAX,SMIN,IAUMAXE	-13744.	AE = 65.9	VME	12494.				
BOT SIR	-1509.	SMAX,SMIN,IAUMAXE	438	438	SMAX,SMIN,IAUMAXE	438	SMAX,SMIN,IAUMAXE	-13004.	AE = 86.8	VME	12332.				
									AE = 87.7	VME	11679.				
ELEM	18	NDDESE	18	19	37	36	MATERIAL =	1	AREA =	843.50	1TOP,1BOI =	70.	70.	PMESSE =	0.
TOP SIR	3247.	MX,MY,MXY =	532	40	SMAX,SMIN,IAUMAXE	532	XC, YC, ZC =	-144.4	6726.	18.00	6726.				
MID SIR	-1699.	SMAX,SMIN,IAUMAXE	412	40	SMAX,SMIN,IAUMAXE	412	SMAX,SMIN,IAUMAXE	-14394.	AE = 87.7	VME	12494.				
BOT SIR	-2051.	SMAX,SMIN,IAUMAXE	377	377	SMAX,SMIN,IAUMAXE	377	SMAX,SMIN,IAUMAXE	-13609.	AE = 88.8	VME	12714.				
									AE = 88.8	VME	12714.				
ELEM	19	NDDESE	19	20	52	51	MATERIAL =	1	AREA =	3772.2	1TOP,1BOI =	70.	70.	PMESSE =	0.
TOP SIR	1683.	MX,MY,MXY =	779	28	SMAX,SMIN,IAUMAXE	779	XC, YC, ZC =	-97.54	3745.	27.00	3745.				
MID SIR	-1576.	SMAX,SMIN,IAUMAXE	580	28	SMAX,SMIN,IAUMAXE	580	SMAX,SMIN,IAUMAXE	-9091.	AE = 84.0	VME	14405.				
BOT SIR	-1470.	SMAX,SMIN,IAUMAXE	380	380	SMAX,SMIN,IAUMAXE	380	SMAX,SMIN,IAUMAXE	-8775.	AE = 88.8	VME	8117.				
									AE = -88.8	VME	7841.				
ELEM	20	NDDESE	20	21	61	60	MATERIAL =	1	AREA =	7111.8	1TOP,1BOI =	70.	70.	PMESSE =	0.
TOP SIR	1177.	MX,MY,MXY =	789	44	SMAX,SMIN,IAUMAXE	789	XC, YC, ZC =	-34.22	1137.	51.00	1137.				
MID SIR	-1008.	SMAX,SMIN,IAUMAXE	793	44	SMAX,SMIN,IAUMAXE	793	SMAX,SMIN,IAUMAXE	-5137.	AE = 68.2	VME	2607.				
BOT SIR	-834.	SMAX,SMIN,IAUMAXE	721	721	SMAX,SMIN,IAUMAXE	721	SMAX,SMIN,IAUMAXE	-3090.	AE = 70.1	VME	2795.				
									AE = -71.9	VME	2745.				

21 02 81 MATERIAL= 1 AHEAD= 9188.5 1TOP, TBO1= 70.70. PMPSS= 0.0.  
 ELEM 21 MMODESE 0.0. XC, YC, ZC= 34.13  
 TOP SIK SX, SY, SKXE -300. SHAX, SHIN, TAUMAXE 360. VME 66.00  
 MID SIK SX, SY, SKXE 528. SHAX, SHIN, TAUMAXE 1171.6 VME 704.  
 BOT SIK SX, SY, SKXE 58. -318. SHAX, SHIN, TAUMAXE 401. VME 725.  
 22 23 82 MATERIAL= 1 AHEAD= 9188.5 1TOP, TBO1= 70.70. PMPSS= 0.0.  
 ELEM 22 MMODESE 0.0. XC, YC, ZC= 97.20  
 TOP SIK SX, SY, SKXE 396. SHAX, SHIN, TAUMAXE 1472. VME 5143.  
 MID SIK SX, SY, SKXE 370. SHAX, SHIN, TAUMAXE 348. VME 3114.  
 BOT SIK SX, SY, SKXE 350. SHAX, SHIN, TAUMAXE 329. VME 3094.  
 23 24 83 MATERIAL= 1 AHEAD= 9188.5 1TOP, TBO1= 70.70. PMPSS= 0.0.  
 ELEM 23 MMODESE 0.0. XC, YC, ZC= 145.5  
 TOP SIK SX, SY, SKXE 65. SHAX, SHIN, TAUMAXE 2417. VME 5169.  
 MID SIK SX, SY, SKXE 61. SHAX, SHIN, TAUMAXE 609. VME 5131.  
 BOT SIK SX, SY, SKXE 60. SHAX, SHIN, TAUMAXE 597. VME 5093.  
 24 1 84 MATERIAL= 1 AHEAD= 9188.5 1TOP, TBO1= 70.70. PMPSS= 0.0.  
 ELEM 24 MMODESE 0.0. XC, YC, ZC= 71.6  
 TOP SIK SX, SY, SKXE 751. SHAX, SHIN, TAUMAXE 2945. VME 6238.  
 MID SIK SX, SY, SKXE 729. SHAX, SHIN, TAUMAXE 727. VME 6238.  
 BOT SIK SX, SY, SKXE 730. SHAX, SHIN, TAUMAXE 718. VME 6191.  
 25 26 85 MATERIAL= 1 AHEAD= 840.96 1TOP, TBO1= 70.70. PMPSS= 0.0.  
 ELEM 25 MMODESE 0.0. XC, YC, ZC= 111.8  
 TOP SIK SX, SY, SKXE 81. SHAX, SHIN, TAUMAXE 5350. VME 10456.  
 MID SIK SX, SY, SKXE 498. SHAX, SHIN, TAUMAXE 5463. VME 10705.  
 BOT SIK SX, SY, SKXE 410. SHAX, SHIN, TAUMAXE 413. VME 10472.  
 26 27 86 MATERIAL= 1 AHEAD= 840.96 1TOP, TBO1= 70.70. PMPSS= 0.0.  
 ELEM 26 MMODESE 0.0. XC, YC, ZC= 174.5  
 TOP SIK SX, SY, SKXE 115. SHAX, SHIN, TAUMAXE 6678. VME 12628.  
 MID SIK SX, SY, SKXE 105. SHAX, SHIN, TAUMAXE 161. VME 12721.  
 BOT SIK SX, SY, SKXE 415. SHAX, SHIN, TAUMAXE 342. VME 12721.  
 27 28 87 MATERIAL= 1 AHEAD= 840.96 1TOP, TBO1= 70.70. PMPSS= 0.0.  
 ELEM 27 MMODESE 0.0. XC, YC, ZC= 159.8  
 TOP SIK SX, SY, SKXE 87. SHAX, SHIN, TAUMAXE 7410. VME 14369.  
 MID SIK SX, SY, SKXE 80. SHAX, SHIN, TAUMAXE 451. VME 14149.  
 BOT SIK SX, SY, SKXE -31. SHAX, SHIN, TAUMAXE -1. VME 13920.  
 28 29 88 MATERIAL= 1 AHEAD= 840.96 1TOP, TBO1= 70.70. PMPSS= 0.0.  
 ELEM 28 MMODESE 0.0. XC, YC, ZC= 93.3  
 TOP SIK SX, SY, SKXE 330. SHAX, SHIN, TAUMAXE 5747. VME 11394.  
 MID SIK SX, SY, SKXE 329. SHAX, SHIN, TAUMAXE 93.3. VME 11020.  
 BOT SIK SX, SY, SKXE 329. SHAX, SHIN, TAUMAXE -184. VME 11020.  
 29 30 89 MATERIAL= 1 AHEAD= 840.96 1TOP, TBO1= 70.70. PMPSS= 0.0.  
 ELEM 29 MMODESE 0.0. XC, YC, ZC= 174.8  
 TOP SIK SX, SY, SKXE 311. SHAX, SHIN, TAUMAXE 2445. VME 4802.  
 MID SIK SX, SY, SKXE 165. SHAX, SHIN, TAUMAXE 181. VME 5094.  
 BOT SIK SX, SY, SKXE -115. SHAX, SHIN, TAUMAXE -84. VME 5634.  
 30 31 90 MATERIAL= 1 AHEAD= 840.96 1TOP, TBO1= 70.70. PMPSS= 0.0.  
 ELEM 30 MMODESE 0.0. XC, YC, ZC= 606.  
 TOP SIK SX, SY, SKXE -420. SHAX, SHIN, TAUMAXE 740. VME 1298.  
 MID SIK SX, SY, SKXE -201. SHAX, SHIN, TAUMAXE 351. VME 877.  
 BOT SIK SX, SY, SKXE 11. SHAX, SHIN, TAUMAXE 17. VME 867.  
 31 32 91 MATERIAL= 1 AHEAD= 840.96 1TOP, TBO1= 70.70. PMPSS= 0.0.  
 ELEM 31 MMODESE 0.0. XC, YC, ZC= 711.8  
 TOP SIK SX, SY, SKXE -639. SHAX, SHIN, TAUMAXE 1004. VME 1704.  
 MID SIK SX, SY, SKXE 1254. SHAX, SHIN, TAUMAXE 624. VME 1227.  
 BOT SIK SX, SY, SKXE 1933. SHAX, SHIN, TAUMAXE -44. VME 1090.  
 32 33 92 MATERIAL= 1 AHEAD= 840.96 1TOP, TBO1= 70.70. PMPSS= 0.0.  
 ELEM 32 MMODESE 0.0. XC, YC, ZC= 985.  
 TOP SIK SX, SY, SKXE 811. SHAX, SHIN, TAUMAXE 485. VME 1090.

ELEM	32	MODESE	31	32	46	45	MATERIAL=	1	AREA=	840.96	XC, YC, ZC=	1109.70	IBOIF=	70	70.	PRESS=	0.
MX, MY	-29.	MX, MY, MXY=	-188.	SMAX, SMIN, TAUMAX=	375.	SMAX, SMIN, TAUMAX=	1294.	XC, YC, ZC=	-177.8	1004.	AE=	11.0	VME=	54.00	11.64.	11.64.	11.64.
TOP SIK	81.	SMAX, SMIN, TAUMAX=	376.	SMAX, SMIN, TAUMAX=	1279.	SMAX, SMIN, TAUMAX=	1249.	XC, YC, ZC=	-44.	824.	AE=	18.5	VME=	1227.	1227.	1227.	1227.
MID SIK	801.	SMAX, SMIN, TAUMAX=	376.	SMAX, SMIN, TAUMAX=	933.	SMAX, SMIN, TAUMAX=	485.	XC, YC, ZC=	485.	382.	AE=	40.1	VME=	1090.	1090.	1090.	1090.
ELEM	33	MODESE	32	33	47	46	MATERIAL=	1	AREA=	840.96	XC, YC, ZC=	1109.70	IBOIF=	70	70.	PRESS=	0.
MX, MY	-115.	MX, MY, MXY=	-57.	SMAX, SMIN, TAUMAX=	642.	SMAX, SMIN, TAUMAX=	174.8	XC, YC, ZC=	-87.4	740.	AE=	33.7	VME=	54.00	1298.	1298.	1298.
TOP SIK	147.	SMAX, SMIN, TAUMAX=	348.	SMAX, SMIN, TAUMAX=	625.	SMAX, SMIN, TAUMAX=	79.	XC, YC, ZC=	79.	554.	AE=	51.2	VME=	671.	671.	671.	671.
MID SIK	147.	SMAX, SMIN, TAUMAX=	348.	SMAX, SMIN, TAUMAX=	14.	SMAX, SMIN, TAUMAX=	17.	XC, YC, ZC=	-858.	438.	AE=	89.1	VME=	867.	867.	867.	867.
ELEM	34	MODESE	33	34	48	47	MATERIAL=	1	AREA=	840.96	XC, YC, ZC=	1109.70	IBOIF=	70	70.	PRESS=	0.
MX, MY	-115.	MX, MY, MXY=	-73.	SMAX, SMIN, TAUMAX=	277.	SMAX, SMIN, TAUMAX=	174.8	XC, YC, ZC=	-168.7	2445.	AE=	89.7	VME=	54.00	54.00	54.00	54.00
TOP SIK	147.	SMAX, SMIN, TAUMAX=	277.	SMAX, SMIN, TAUMAX=	105.	SMAX, SMIN, TAUMAX=	18.	XC, YC, ZC=	18.	2632.	AE=	85.5	VME=	4892.	4892.	4892.	4892.
MID SIK	147.	SMAX, SMIN, TAUMAX=	277.	SMAX, SMIN, TAUMAX=	105.	SMAX, SMIN, TAUMAX=	18.	XC, YC, ZC=	18.	2632.	AE=	85.5	VME=	4892.	4892.	4892.	4892.
ELEM	35	MODESE	34	35	49	48	MATERIAL=	1	AREA=	840.96	XC, YC, ZC=	1109.70	IBOIF=	70	70.	PRESS=	0.
MX, MY	-9.	MX, MY, MXY=	70.	SMAX, SMIN, TAUMAX=	488.	SMAX, SMIN, TAUMAX=	151.	XC, YC, ZC=	-159.8	5747.	AE=	87.6	VME=	11194.	11194.	11194.	11194.
TOP SIK	40.	SMAX, SMIN, TAUMAX=	70.	SMAX, SMIN, TAUMAX=	113.	SMAX, SMIN, TAUMAX=	336.	XC, YC, ZC=	336.	5617.	AE=	84.9	VME=	14149.	14149.	14149.	14149.
MID SIK	40.	SMAX, SMIN, TAUMAX=	70.	SMAX, SMIN, TAUMAX=	113.	SMAX, SMIN, TAUMAX=	336.	XC, YC, ZC=	336.	5617.	AE=	84.9	VME=	14149.	14149.	14149.	14149.
ELEM	36	MODESE	35	36	50	49	MATERIAL=	1	AREA=	840.96	XC, YC, ZC=	1109.70	IBOIF=	70	70.	PRESS=	0.
MX, MY	-115.	MX, MY, MXY=	116.	SMAX, SMIN, TAUMAX=	608.	SMAX, SMIN, TAUMAX=	21.	XC, YC, ZC=	-148.2	7410.	AE=	89.9	VME=	14369.	14369.	14369.	14369.
TOP SIK	40.	SMAX, SMIN, TAUMAX=	116.	SMAX, SMIN, TAUMAX=	725.	SMAX, SMIN, TAUMAX=	451.	XC, YC, ZC=	451.	7185.	AE=	87.4	VME=	13920.	13920.	13920.	13920.
MID SIK	40.	SMAX, SMIN, TAUMAX=	116.	SMAX, SMIN, TAUMAX=	725.	SMAX, SMIN, TAUMAX=	451.	XC, YC, ZC=	451.	7185.	AE=	87.4	VME=	13920.	13920.	13920.	13920.
ELEM	37	MODESE	36	37	51	50	MATERIAL=	1	AREA=	840.96	XC, YC, ZC=	1109.70	IBOIF=	70	70.	PRESS=	0.
MX, MY	-115.	MX, MY, MXY=	116.	SMAX, SMIN, TAUMAX=	338.	SMAX, SMIN, TAUMAX=	69.	XC, YC, ZC=	-134.0	6678.	AE=	88.6	VME=	12628.	12628.	12628.	12628.
TOP SIK	40.	SMAX, SMIN, TAUMAX=	116.	SMAX, SMIN, TAUMAX=	604.	SMAX, SMIN, TAUMAX=	1241.	XC, YC, ZC=	1241.	6491.	AE=	87.3	VME=	12721.	12721.	12721.	12721.
MID SIK	40.	SMAX, SMIN, TAUMAX=	116.	SMAX, SMIN, TAUMAX=	604.	SMAX, SMIN, TAUMAX=	1241.	XC, YC, ZC=	1241.	6491.	AE=	87.3	VME=	12721.	12721.	12721.	12721.
ELEM	38	MODESE	37	38	37	37	MATERIAL=	1	AREA=	1252.3	XC, YC, ZC=	1109.70	IBOIF=	70	70.	PRESS=	0.
MX, MY	-281.	MX, MY, MXY=	12.	SMAX, SMIN, TAUMAX=	59.	SMAX, SMIN, TAUMAX=	59.	XC, YC, ZC=	-111.8	2350.	AE=	88.5	VME=	34.00	34.00	34.00	34.00
TOP SIK	40.	SMAX, SMIN, TAUMAX=	12.	SMAX, SMIN, TAUMAX=	-56.	SMAX, SMIN, TAUMAX=	453.	XC, YC, ZC=	-1047.2	3283.	AE=	89.7	VME=	4890.	4890.	4890.	4890.
MID SIK	40.	SMAX, SMIN, TAUMAX=	12.	SMAX, SMIN, TAUMAX=	-56.	SMAX, SMIN, TAUMAX=	453.	XC, YC, ZC=	-1047.2	3283.	AE=	89.7	VME=	4890.	4890.	4890.	4890.
ELEM	39	MODESE	38	39	38	38	MATERIAL=	1	AREA=	2076.7	XC, YC, ZC=	1109.70	IBOIF=	70	70.	PRESS=	0.
MX, MY	-84.	MX, MY, MXY=	95.	SMAX, SMIN, TAUMAX=	26.	SMAX, SMIN, TAUMAX=	26.	XC, YC, ZC=	-51.09	2280.	AE=	89.1	VME=	102.0	102.0	102.0	102.0
TOP SIK	40.	SMAX, SMIN, TAUMAX=	95.	SMAX, SMIN, TAUMAX=	-69.	SMAX, SMIN, TAUMAX=	496.	XC, YC, ZC=	-4065.	2623.	AE=	89.7	VME=	4534.	4534.	4534.	4534.
MID SIK	40.	SMAX, SMIN, TAUMAX=	95.	SMAX, SMIN, TAUMAX=	-69.	SMAX, SMIN, TAUMAX=	496.	XC, YC, ZC=	-4065.	2623.	AE=	89.7	VME=	4890.	4890.	4890.	4890.
ELEM	40	MODESE	39	40	53	52	MATERIAL=	1	AREA=	2917.9	XC, YC, ZC=	1109.70	IBOIF=	70	70.	PRESS=	0.
MX, MY	-80.	MX, MY, MXY=	95.	SMAX, SMIN, TAUMAX=	18.	SMAX, SMIN, TAUMAX=	18.	XC, YC, ZC=	-97.00	3752.	AE=	85.0	VME=	7462.	7462.	7462.	7462.
TOP SIK	40.	SMAX, SMIN, TAUMAX=	95.	SMAX, SMIN, TAUMAX=	661.	SMAX, SMIN, TAUMAX=	327.	XC, YC, ZC=	-742.8	3667.	AE=	85.0	VME=	7454.	7454.	7454.	7454.
MID SIK	40.	SMAX, SMIN, TAUMAX=	95.	SMAX, SMIN, TAUMAX=	661.	SMAX, SMIN, TAUMAX=	327.	XC, YC, ZC=	-742.8	3667.	AE=	85.0	VME=	7454.	7454.	7454.	7454.
ELEM	41	MODESE	40	41	54	53	MATERIAL=	1	AREA=	559.23	XC, YC, ZC=	1109.70	IBOIF=	70	70.	PRESS=	0.
MX, MY	-206.	MX, MY, MXY=	16.	SMAX, SMIN, TAUMAX=	18.	SMAX, SMIN, TAUMAX=	18.	XC, YC, ZC=	-131.6	1020.	AE=	89.9	VME=	13858.	13858.	13858.	13858.
TOP SIK	40.	SMAX, SMIN, TAUMAX=	16.	SMAX, SMIN, TAUMAX=	-709.	SMAX, SMIN, TAUMAX=	-350.	XC, YC, ZC=	-1311.	6899.	AE=	89.9	VME=	13100.	13100.	13100.	13100.
MID SIK	40.	SMAX, SMIN, TAUMAX=	16.	SMAX, SMIN, TAUMAX=	-709.	SMAX, SMIN, TAUMAX=	-350.	XC, YC, ZC=	-1311.	6899.	AE=	89.9	VME=	13100.	13100.	13100.	13100.
ELEM	42	MODESE	41	42	54	54	MATERIAL=	1	AREA=	559.23	XC, YC, ZC=	1109.70	IBOIF=	70	70.	PRESS=	0.
MX, MY	-321.	MX, MY, MXY=	116.	SMAX, SMIN, TAUMAX=	-116.	SMAX, SMIN, TAUMAX=	-116.	XC, YC, ZC=	-147.8	7933.	AE=	87.6	VME=	15222.	15222.	15222.	15222.
TOP SIK	40.	SMAX, SMIN, TAUMAX=	116.	SMAX, SMIN, TAUMAX=	-116.	SMAX, SMIN, TAUMAX=	-116.	XC, YC, ZC=	-147.8	7933.	AE=	87.6	VME=	15222.	15222.	15222.	15222.
MID SIK	40.	SMAX, SMIN, TAUMAX=	116.	SMAX, SMIN, TAUMAX=	-116.	SMAX, SMIN, TAUMAX=	-116.	XC, YC, ZC=	-147.8	7933.	AE=	87.6	VME=	15222.	15222.	15222.	15222.









80	131	102	7445B	70	186.0	0.
ITEM	MX MY	MX MY	MATERIAL = 1	TOP, THOI =	PRESSE =	
TOP SIK	13363	112	-111	-33.84	186.0	
MID SIK	3184	-727	SMAX,SMIN,TAUMAXE	-1579.	VME	3219.
BOT SIK	3005	-770	SMAX,SMIN,TAUMAXE	-3404.	VME	3189.
		-613	SMAX,SMIN,TAUMAXE	-3229.	VME	3204.
81	132	131	49075	70	204.0	0.
ITEM	MX MY	MX MY	MATERIAL = 1	TOP, THOI =	PRESSE =	
TOP SIK	349	46	-16	303	204.0	584.
MID SIK	279	-269	SMAX,SMIN,TAUMAXE	-247.	VME	671.
BOT SIK	212	-350	SMAX,SMIN,TAUMAXE	-408.	VME	794.
		-411	SMAX,SMIN,TAUMAXE			
82	133	132	99075	70	204.0	0.
ITEM	MX MY	MX MY	MATERIAL = 1	TOP, THOI =	PRESSE =	
TOP SIK	3261	21	-3	1649.	204.0	4376.
MID SIK	3316	-295	SMAX,SMIN,TAUMAXE	-13.	VME	4346.
BOT SIK	3171	-356	SMAX,SMIN,TAUMAXE	-200.	VME	3297.
		-271	SMAX,SMIN,TAUMAXE			
83	134	133	99075	70	204.0	0.
ITEM	MX MY	MX MY	MATERIAL = 1	TOP, THOI =	PRESSE =	
TOP SIK	5306	-2	-0	143.8	204.0	0.
MID SIK	5314	-180	SMAX,SMIN,TAUMAXE	-170.	VME	5399.
BOT SIK		-181	SMAX,SMIN,TAUMAXE	-162.	VME	5399.
		-181	SMAX,SMIN,TAUMAXE	-155.	VME	5399.
84	134	134	99075	70	204.0	0.
ITEM	MX MY	MX MY	MATERIAL = 1	TOP, THOI =	PRESSE =	
TOP SIK	6428	-6	0	169.6	204.0	0.
MID SIK	6438	-65	SMAX,SMIN,TAUMAXE	-208.	VME	6516.
BOT SIK		-65	SMAX,SMIN,TAUMAXE	-193.	VME	6536.
		-65	SMAX,SMIN,TAUMAXE	-177.	VME	6536.
94	85	85	1284	70	186.0	0.
ITEM	MX MY	MX MY	MATERIAL = 1	TOP, THOI =	PRESSE =	
TOP SIK	8099	29	-34	134.4	186.0	0.
MID SIK	8147	643	SMAX,SMIN,TAUMAXE	-110.6	VME	7048.
BOT SIK		-8099	SMAX,SMIN,TAUMAXE	4091.	VME	8153.
		1163	SMAX,SMIN,TAUMAXE	4061.	VME	8796.
85	95	98	83164	70	186.0	0.
ITEM	MX MY	MX MY	MATERIAL = 1	TOP, THOI =	PRESSE =	
TOP SIK	10072	140	0	116.2	186.0	0.
MID SIK	10074	-476	SMAX,SMIN,TAUMAXE	-9916.5	VME	9757.
BOT SIK		-725	SMAX,SMIN,TAUMAXE	-9029.	VME	10067.
			SMAX,SMIN,TAUMAXE	-1025.	VME	10464.
86	96	95	83164	70	186.0	0.
ITEM	MX MY	MX MY	MATERIAL = 1	TOP, THOI =	PRESSE =	
TOP SIK	13197	152	-145	97.90	186.0	0.
MID SIK	13816	-7	SMAX,SMIN,TAUMAXE	-146.5	VME	3504.
BOT SIK		-576	SMAX,SMIN,TAUMAXE	-13839.	VME	13855.
			SMAX,SMIN,TAUMAXE	-485.	VME	14284.
87	97	96	83164	70	186.0	0.
ITEM	MX MY	MX MY	MATERIAL = 1	TOP, THOI =	PRESSE =	
TOP SIK	16591	49	-135	158.0	186.0	0.
MID SIK	19356	70	SMAX,SMIN,TAUMAXE	-18591.	VME	16080.
BOT SIK		-20121	SMAX,SMIN,TAUMAXE	-1.	VME	19366.
			SMAX,SMIN,TAUMAXE	-153.	VME	20091.
89	99	98	83164	70	186.0	0.
ITEM	MX MY	MX MY	MATERIAL = 1	TOP, THOI =	PRESSE =	
TOP SIK	16591	49	-135	158.0	186.0	0.
MID SIK	19356	70	SMAX,SMIN,TAUMAXE	-18591.	VME	16080.
BOT SIK		-20121	SMAX,SMIN,TAUMAXE	-1.	VME	19366.
			SMAX,SMIN,TAUMAXE	-153.	VME	20091.
90	100	99	83164	70	186.0	0.
ITEM	MX MY	MX MY	MATERIAL = 1	TOP, THOI =	PRESSE =	
TOP SIK	13197	152	-145	146.5	186.0	0.
MID SIK	14435	77	SMAX,SMIN,TAUMAXE	-13197.	VME	1504.
BOT SIK		-576	SMAX,SMIN,TAUMAXE	-13839.	VME	14355.
			SMAX,SMIN,TAUMAXE	-485.	VME	14284.

ELEM	07	01	02	101	100	MATERIALS	1	AREA=	631.64	70.	70.	PRESS=	0.	
MAX	SY	SY	SY	MX	MY	140.	14	129.	XC	-132.5	-116.2	185.0		
TOP	SIX	SX	SY	463	4572	-267.	SMAX	SMIN	TAUMAX	4759.	AE=-88.4	VME	9757.	
MID	SIX	SX	SY	70.	10024	229.	SMAX	SMIN	TAUMAX	5052.	AE	88.7	VME	10067.
BOT	SIX	SX	SY	608.	-10076	725.	SMAX	SMIN	TAUMAX	657.	AE	88.1	VME	10467.
ELEM	08	09	101	102	92	MATERIALS	1	AREA=	1238.4	70.	70.	PRESS=	0.	
MAX	SY	SY	SY	MX	MY	293.	13	54.	XC	-110.6	-134.4	186.0		
TOP	SIX	SX	SY	1487	4051	-693.	SMAX	SMIN	TAUMAX	8110.	AE=-84.6	VME	7469.	
MID	SIX	SX	SY	70.	8094	497.	SMAX	SMIN	TAUMAX	61.	AE=-86.9	VME	8153.	
BOT	SIX	SX	SY	1163.	-8147	1231.	SMAX	SMIN	TAUMAX	1169.	AE=-88.6	VME	8796.	
ELEM	09	115	93	93	93	MATERIALS	1	AREA=	2401.7	70.	70.	PRESS=	0.	
MAX	SY	SY	SY	MX	MY	145.	37.	37.	XC	56.50	105.7	240.0		
TOP	SIX	SX	SY	156.	4571.	-57.	SMAX	SMIN	TAUMAX	151.	AE=-89.3	VME	9854.	
MID	SIX	SX	SY	712.	14755.	253.	SMAX	SMIN	TAUMAX	1276.	AE	87.1	VME	5147.
BOT	SIX	SX	SY	1268.	-4922.	848.	SMAX	SMIN	TAUMAX	1276.	AE	87.1	VME	5681.
ELEM	09	103	116	103	116	MATERIALS	1	AREA=	3707.	70.	70.	PRESS=	0.	
MAX	SY	SY	SY	MX	MY	334.	83	8.	XC	-95.87	143.4	231.0		
TOP	SIX	SX	SY	1266.	6800.	513.	SMAX	SMIN	TAUMAX	1219.	AE	84.8	VME	6529.
MID	SIX	SX	SY	14.	448.	480.	SMAX	SMIN	TAUMAX	50.	AE	85.8	VME	6543.
BOT	SIX	SX	SY	1295.	-6168.	848.	SMAX	SMIN	TAUMAX	1322.	AE	86.6	VME	6947.
ELEM	09	104	103	104	103	MATERIALS	1	AREA=	829.09	70.	70.	PRESS=	0.	
MAX	SY	SY	SY	MX	MY	218.	31	-69.	XC	-132.1	115.6	222.0		
TOP	SIX	SX	SY	1376.	6795.	254.	SMAX	SMIN	TAUMAX	1219.	AE	84.4	VME	9401.
MID	SIX	SX	SY	1190.	-9916.	12.	SMAX	SMIN	TAUMAX	33.	AE=-84.9	VME	9900.	
BOT	SIX	SX	SY	1694.	-14528.	760.	SMAX	SMIN	TAUMAX	811.	AE=-88.5	VME	10473.	
ELEM	09	105	104	105	104	MATERIALS	1	AREA=	829.09	70.	70.	PRESS=	0.	
MAX	SY	SY	SY	MX	MY	359.	33	-15.	XC	-146.1	107.7	222.0		
TOP	SIX	SX	SY	1376.	6795.	694.	SMAX	SMIN	TAUMAX	1459.	AE=-87.2	VME	12002.	
MID	SIX	SX	SY	1190.	-9916.	127.	SMAX	SMIN	TAUMAX	1489.	AE=-86.9	VME	13187.	
BOT	SIX	SX	SY	1694.	-14528.	342.	SMAX	SMIN	TAUMAX	1218.	AE=-86.6	VME	13616.	
ELEM	09	106	105	106	105	MATERIALS	1	AREA=	829.09	70.	70.	PRESS=	0.	
MAX	SY	SY	SY	MX	MY	339.	64	204.	XC	-175.6	107.7	222.0		
TOP	SIX	SX	SY	1376.	6795.	1237.	SMAX	SMIN	TAUMAX	1459.	AE=-88.4	VME	10580.	
MID	SIX	SX	SY	1190.	-9916.	127.	SMAX	SMIN	TAUMAX	1489.	AE=-87.1	VME	22621.	
BOT	SIX	SX	SY	1694.	-14528.	342.	SMAX	SMIN	TAUMAX	1218.	AE=-84.9	VME	10473.	
ELEM	09	107	107	107	107	MATERIALS	1	AREA=	829.09	70.	70.	PRESS=	0.	
MAX	SY	SY	SY	MX	MY	339.	64	204.	XC	-175.6	107.7	222.0		
TOP	SIX	SX	SY	1376.	6795.	1237.	SMAX	SMIN	TAUMAX	1459.	AE=-88.4	VME	10580.	
MID	SIX	SX	SY	1190.	-9916.	127.	SMAX	SMIN	TAUMAX	1489.	AE=-87.1	VME	22621.	
BOT	SIX	SX	SY	1694.	-14528.	342.	SMAX	SMIN	TAUMAX	1218.	AE=-84.9	VME	10473.	
ELEM	09	109	108	109	108	MATERIALS	1	AREA=	829.09	70.	70.	PRESS=	0.	
MAX	SY	SY	SY	MX	MY	359.	33	-15.	XC	-146.1	107.7	222.0		
TOP	SIX	SX	SY	1376.	6795.	694.	SMAX	SMIN	TAUMAX	1459.	AE=-87.2	VME	12002.	
MID	SIX	SX	SY	1190.	-9916.	127.	SMAX	SMIN	TAUMAX	1489.	AE=-86.9	VME	13187.	
BOT	SIX	SX	SY	1694.	-14528.	342.	SMAX	SMIN	TAUMAX	1218.	AE=-86.6	VME	13616.	
ELEM	09	110	109	110	109	MATERIALS	1	AREA=	829.09	70.	70.	PRESS=	0.	
MAX	SY	SY	SY	MX	MY	359.	31	69.	XC	-132.1	115.6	222.0		
TOP	SIX	SX	SY	1376.	6795.	694.	SMAX	SMIN	TAUMAX	1459.	AE=-88.4	VME	9401.	
MID	SIX	SX	SY	1190.	-9916.	127.	SMAX	SMIN	TAUMAX	1489.	AE=-84.9	VME	9900.	
BOT	SIX	SX	SY	1694.	-14528.	342.	SMAX	SMIN	TAUMAX	811.	AE=-88.5	VME	10473.	
ELEM	09	101	101	101	101	MATERIALS	1	AREA=	3707.	70.	70.	PRESS=	0.	
MAX	SY	SY	SY	MX	MY	334.	83	8.	XC	-95.87	143.4	231.0		
TOP	SIX	SX	SY	1266.	6800.	513.	SMAX	SMIN	TAUMAX	1219.	AE	84.8	VME	6529.
MID	SIX	SX	SY	14.	448.	480.	SMAX	SMIN	TAUMAX	50.	AE	85.8	VME	6543.
BOT	SIX	SX	SY	1295.	-6168.	848.	SMAX	SMIN	TAUMAX	1322.	AE=-86.6	VME	6947.	



63	109	RUDESE	113	114	179	178	MATERIAL=	1	AKLAE	1059P	1059P	70	70	PRESSE	0
	MAX,MY	SK,SY,SKYE	237	237	MX,MY	MXY	24	24	SKPB	XC,YC,ZC	94.85	141.9	141.9	354.0	0
	TOP SIK	SK,SY,SKYE	237	237	MAX,MY	MXY	54	54	SMAX,SMIN,IAUMAXE	XC,YC,ZC	217	1517	1517	4.6	3149
	MID SIK	SK,SY,SKYE	228	228	3136	3044	295	295	SMAX,SMIN,IAUMAXE	XC,YC,ZC	4	1642	1642	5.4	3163
	BOT SIK	SK,SY,SKYE	-181	-181	3044		306	306	SMAX,SMIN,IAUMAXE	XC,YC,ZC	-210				
63	110	RUDESE	114	115	180	179	MATERIAL=	1	AKLAE	1059P	1059P	70	70	PRESSE	0
	MAX,MY	SK,SY,SKYE	172	172	MX,MY	MXY	51	51	SMAX,SMIN,IAUMAXE	XC,YC,ZC	33.31	167.4	167.4	354.0	0
	TOP SIK	SK,SY,SKYE	172	172	325	313	325	325	SMAX,SMIN,IAUMAXE	XC,YC,ZC	-225	331	331	50.7	504
	MID SIK	SK,SY,SKYE	-22	-22	41	41	375	375	SMAX,SMIN,IAUMAXE	XC,YC,ZC	567	435	435	45.0	650
	BOT SIK	SK,SY,SKYE	-216	-216	-87		425	425	SMAX,SMIN,IAUMAXE	XC,YC,ZC	-581				760
63	111	RUDESE	115	116	148	180	MATERIAL=	1	AKLAE	7357	7357	70	70	PRESSE	0
	MAX,MY	SK,SY,SKYE	-9	-9	MX,MY	MXY	150	150	SMAX,SMIN,IAUMAXE	XC,YC,ZC	33.84	167.6	167.6	330.0	0
	TOP SIK	SK,SY,SKYE	-1276	-1276	3495	872	872	872	SMAX,SMIN,IAUMAXE	XC,YC,ZC	-1796	1411	1411	70.9	545
	MID SIK	SK,SY,SKYE	-778	-778	3316	898	924	924	SMAX,SMIN,IAUMAXE	XC,YC,ZC	-3601	1554	1554	72.4	3362
	BOT SIK	SK,SY,SKYE	-281	-281	-3137		924	924	SMAX,SMIN,IAUMAXE	XC,YC,ZC	-3409	1701	1701	73.6	3405
63	112	RUDESE	116	117	135	148	MATERIAL=	1	AKLAE	2867	2867	70	70	PRESSE	0
	MAX,MY	SK,SY,SKYE	-62	-62	MX,MY	MXY	73	73	SMAX,SMIN,IAUMAXE	XC,YC,ZC	-95.33	142.6	142.6	297.0	0
	TOP SIK	SK,SY,SKYE	-892	-892	122	122	122	122	SMAX,SMIN,IAUMAXE	XC,YC,ZC	-7113	3112	3112	86.9	6712
	MID SIK	SK,SY,SKYE	109	109	-6830	320	517	517	SMAX,SMIN,IAUMAXE	XC,YC,ZC	-6844	3484	3484	87.4	6907
	BOT SIK	SK,SY,SKYE	1109	1109	-6549		517	517	SMAX,SMIN,IAUMAXE	XC,YC,ZC	-6584	3864	3864	86.2	7224
63	113	RUDESE	117	118	136	135	MATERIAL=	1	AKLAE	549	549	70	70	PRESSE	0
	MAX,MY	SK,SY,SKYE	30	30	MX,MY	MXY	9	9	SMAX,SMIN,IAUMAXE	XC,YC,ZC	-131.3	115.2	115.2	288.0	0
	TOP SIK	SK,SY,SKYE	30	30	-6047	281	281	281	SMAX,SMIN,IAUMAXE	XC,YC,ZC	-9040	4576	4576	86.6	9118
	MID SIK	SK,SY,SKYE	-31	-31	-9329	1109	1109	1109	SMAX,SMIN,IAUMAXE	XC,YC,ZC	-9337	4675	4675	85.3	9343
	BOT SIK	SK,SY,SKYE			-9610				SMAX,SMIN,IAUMAXE	XC,YC,ZC	-9737	4916	4916	85.5	9785
63	114	RUDESE	118	119	137	136	MATERIAL=	1	AKLAE	549	549	70	70	PRESSE	0
	MAX,MY	SK,SY,SKYE	205	205	MX,MY	MXY	174	174	SMAX,SMIN,IAUMAXE	XC,YC,ZC	-145.2	97.05	97.05	288.0	0
	TOP SIK	SK,SY,SKYE	205	205	-1498	205	205	205	SMAX,SMIN,IAUMAXE	XC,YC,ZC	-11468	7163	7163	86.7	1332
	MID SIK	SK,SY,SKYE	-3090	-3090	-1498	208	208	208	SMAX,SMIN,IAUMAXE	XC,YC,ZC	-13536	6337	6337	86.3	12505
	BOT SIK	SK,SY,SKYE			-15498				SMAX,SMIN,IAUMAXE	XC,YC,ZC	-13870	5577	5577	79.4	12731
63	115	RUDESE	119	120	138	137	MATERIAL=	1	AKLAE	549	549	70	70	PRESSE	0
	MAX,MY	SK,SY,SKYE	521	521	MX,MY	MXY	150	150	SMAX,SMIN,IAUMAXE	XC,YC,ZC	-156.7	77.26	77.26	288.0	0
	TOP SIK	SK,SY,SKYE	521	521	-14824	657	657	657	SMAX,SMIN,IAUMAXE	XC,YC,ZC	-14842	10194	10194	88.2	18254
	MID SIK	SK,SY,SKYE	-48	-48	-14928	894	894	894	SMAX,SMIN,IAUMAXE	XC,YC,ZC	-14945	7458	7458	88.3	1929
	BOT SIK	SK,SY,SKYE	-5618	-5618	-15033		331	331	SMAX,SMIN,IAUMAXE	XC,YC,ZC	-15044	4719	4719	86.0	13164
63	116	RUDESE	120	121	139	138	MATERIAL=	1	AKLAE	549	549	70	70	PRESSE	0
	MAX,MY	SK,SY,SKYE	250	250	MX,MY	MXY	165	165	SMAX,SMIN,IAUMAXE	XC,YC,ZC	-165.4	50.15	50.15	288.0	0
	TOP SIK	SK,SY,SKYE	250	250	-6476	1242	1242	1242	SMAX,SMIN,IAUMAXE	XC,YC,ZC	-9603	6243	6243	84.3	10819
	MID SIK	SK,SY,SKYE	-250	-250	-7038	2576	2576	2576	SMAX,SMIN,IAUMAXE	XC,YC,ZC	-7904	4261	4261	71.4	8230
	BOT SIK	SK,SY,SKYE			-6258		6393	6393	SMAX,SMIN,IAUMAXE	XC,YC,ZC	-13356	6428	6428	48.0	13113
63	117	RUDESE	121	122	140	139	MATERIAL=	1	AKLAE	549	549	70	70	PRESSE	0
	MAX,MY	SK,SY,SKYE	321	321	MX,MY	MXY	168	168	SMAX,SMIN,IAUMAXE	XC,YC,ZC	-171.3	34.08	34.08	288.0	0
	TOP SIK	SK,SY,SKYE	321	321	614	3212	3212	3212	SMAX,SMIN,IAUMAXE	XC,YC,ZC	-4534	3576	3576	52.0	6268
	MID SIK	SK,SY,SKYE	384	384	642	837	837	837	SMAX,SMIN,IAUMAXE	XC,YC,ZC	-37	456	456	51.9	1693
	BOT SIK	SK,SY,SKYE	7394	7394	1150	209	209	209	SMAX,SMIN,IAUMAXE	XC,YC,ZC	132	2464	2464	70.7	4995
63	118	RUDESE	122	123	141	140	MATERIAL=	1	AKLAE	549	549	70	70	PRESSE	0
	MAX,MY	SK,SY,SKYE	627	627	MX,MY	MXY	186	186	SMAX,SMIN,IAUMAXE	XC,YC,ZC	-174.3	11.42	11.42	266.0	0
	TOP SIK	SK,SY,SKYE	627	627	-711	841	841	841	SMAX,SMIN,IAUMAXE	XC,YC,ZC	-6744	5075	5075	7.9	646
	MID SIK	SK,SY,SKYE	384	384	219	316	316	316	SMAX,SMIN,IAUMAXE	XC,YC,ZC	-25	327	327	52.3	641
	BOT SIK	SK,SY,SKYE	7394	7394	1150	209	209	209	SMAX,SMIN,IAUMAXE	XC,YC,ZC	1143	3129	3129	86.1	6901
63	119	RUDESE	123	124	142	141	MATERIAL=	1	AKLAE	549	549	70	70	PRESSE	0
	MAX,MY	SK,SY,SKYE	607	607	MX,MY	MXY	186	186	SMAX,SMIN,IAUMAXE	XC,YC,ZC	-174.3	-11.42	-11.42	288.0	0
	TOP SIK	SK,SY,SKYE	607	607	-41	841	841	841	SMAX,SMIN,IAUMAXE	XC,YC,ZC	-6744	5075	5075	7.9	646
	MID SIK	SK,SY,SKYE	184	184	-210	316	316	316	SMAX,SMIN,IAUMAXE	XC,YC,ZC	-25	327	327	52.3	641
	BOT SIK	SK,SY,SKYE	7394	7394	1150	209	209	209	SMAX,SMIN,IAUMAXE	XC,YC,ZC	1143	3129	3129	86.1	6901



ELEM	131	NUDESE	149	148	135	135	MATERIAL=	1	AREA=	1224.5	ITOP,IBOI=	70,	PMESSE=	0.
TOP	MP,MY	SA,SY,SAYE	-31	147.	SMAX,SMIN,IAUMAXE	76.	XC,YC,ZC=	-109.3	315.0	70.132.9	VME	7003.		
MID	SMX,SY,SAYE	302.	10.	SMAX,SMIN,IAUMAXE	302.	XC,YC,ZC=	-7765.	3948.	AE=89.9	7779.				
BOI	SMX,SY,SAYE	-7642.	302.	SMAX,SMIN,IAUMAXE	302.	XC,YC,ZC=	-240.	4201.	AE=87.8	8017.				
ELEM	132	NUDESE	155	149	150	149	MATERIAL=	1	AREA=	822.31	ITOP,IBOI=	70,	PMESSE=	0.
TOP	MP,MY	SA,SY,SAYE	792.	173.	SMAX,SMIN,IAUMAXE	792.	XC,YC,ZC=	-9179.	318.0	70.114.9	VME	95M4.		
MID	SMX,SY,SAYE	-9432.	173.	SMAX,SMIN,IAUMAXE	173.	XC,YC,ZC=	-9440.	4760.	AE=86.9	5490.				
BOI	SMX,SY,SAYE	-9687.	173.	SMAX,SMIN,IAUMAXE	173.	XC,YC,ZC=	-9846.	4653.	AE=82.5	9587.				
ELEM	133	NUDESE	156	137	150	150	MATERIAL=	1	AREA=	822.31	ITOP,IBOI=	70,	PMESSE=	0.
TOP	MP,MY	SA,SY,SAYE	1777.	42.	SMAX,SMIN,IAUMAXE	42.	XC,YC,ZC=	-12577.	318.0	70.90.80.	VME	13250.		
MID	SMX,SY,SAYE	-12242.	42.	SMAX,SMIN,IAUMAXE	42.	XC,YC,ZC=	-12577.	1017.	AE=89.6	12201.				
BOI	SMX,SY,SAYE	-1787.	42.	SMAX,SMIN,IAUMAXE	42.	XC,YC,ZC=	-12537.	5329.	AE=84.2	11589.				
ELEM	134	NUDESE	137	158	152	151	MATERIAL=	1	AREA=	822.31	ITOP,IBOI=	70,	PMESSE=	0.
TOP	MP,MY	SA,SY,SAYE	20.	698.	SMAX,SMIN,IAUMAXE	698.	XC,YC,ZC=	-156.3	318.0	70.77.06	VME	14033.		
MID	SMX,SY,SAYE	154.	150.	SMAX,SMIN,IAUMAXE	150.	XC,YC,ZC=	-1339.	7617.	AE=89.4	12400.				
BOI	SMX,SY,SAYE	-2526.	150.	SMAX,SMIN,IAUMAXE	150.	XC,YC,ZC=	-12312.	4965.	AE=83.3	11317.				
ELEM	135	NUDESE	138	139	153	152	MATERIAL=	1	AREA=	822.31	ITOP,IBOI=	70,	PMESSE=	0.
TOP	MP,MY	SA,SY,SAYE	2575.	167.	SMAX,SMIN,IAUMAXE	167.	XC,YC,ZC=	-165.0	318.0	70.50.00	VME	6592.		
MID	SMX,SY,SAYE	-243.	167.	SMAX,SMIN,IAUMAXE	167.	XC,YC,ZC=	-7890.	3975.	AE=78.8	7920.				
BOI	SMX,SY,SAYE	-3061.	167.	SMAX,SMIN,IAUMAXE	167.	XC,YC,ZC=	-10107.	4461.	AE=82.7	9569.				
ELEM	136	NUDESE	139	140	154	153	MATERIAL=	1	AREA=	822.31	ITOP,IBOI=	70,	PMESSE=	0.
TOP	MP,MY	SA,SY,SAYE	157.	170.	SMAX,SMIN,IAUMAXE	170.	XC,YC,ZC=	-170.9	318.0	70.33.99	VME	4427.		
MID	SMX,SY,SAYE	405.	1644.	SMAX,SMIN,IAUMAXE	1644.	XC,YC,ZC=	-2923.	1657.	AE=48.6	824.				
BOI	SMX,SY,SAYE	247.	1644.	SMAX,SMIN,IAUMAXE	1644.	XC,YC,ZC=	-183.	827.	AE=64.5	5160.				
ELEM	137	NUDESE	140	141	154	154	MATERIAL=	1	AREA=	822.31	ITOP,IBOI=	70,	PMESSE=	0.
TOP	MP,MY	SA,SY,SAYE	4057.	118.	SMAX,SMIN,IAUMAXE	118.	XC,YC,ZC=	-173.9	318.0	70.11.40	VME	4427.		
MID	SMX,SY,SAYE	375.	939.	SMAX,SMIN,IAUMAXE	939.	XC,YC,ZC=	-4258.	2289.	AE=12.6	824.				
BOI	SMX,SY,SAYE	4806.	1674.	SMAX,SMIN,IAUMAXE	1674.	XC,YC,ZC=	157.1	1683.	AE=79.2	4359.				
ELEM	138	NUDESE	141	142	155	155	MATERIAL=	1	AREA=	822.31	ITOP,IBOI=	70,	PMESSE=	0.
TOP	MP,MY	SA,SY,SAYE	4057.	33.	SMAX,SMIN,IAUMAXE	33.	XC,YC,ZC=	-173.9	318.0	70.11.40	VME	4427.		
MID	SMX,SY,SAYE	375.	939.	SMAX,SMIN,IAUMAXE	939.	XC,YC,ZC=	-4258.	2289.	AE=12.6	824.				
BOI	SMX,SY,SAYE	4806.	1674.	SMAX,SMIN,IAUMAXE	1674.	XC,YC,ZC=	157.1	1683.	AE=79.2	4359.				
ELEM	139	NUDESE	142	143	157	156	MATERIAL=	1	AREA=	822.31	ITOP,IBOI=	70,	PMESSE=	0.
TOP	MP,MY	SA,SY,SAYE	157.	15.	SMAX,SMIN,IAUMAXE	15.	XC,YC,ZC=	-170.9	318.0	70.33.99	VME	3137.		
MID	SMX,SY,SAYE	405.	1475.	SMAX,SMIN,IAUMAXE	1475.	XC,YC,ZC=	-2923.	1657.	AE=48.6	824.				
BOI	SMX,SY,SAYE	247.	1475.	SMAX,SMIN,IAUMAXE	1475.	XC,YC,ZC=	-183.	827.	AE=64.5	5160.				
ELEM	140	NUDESE	143	144	157	157	MATERIAL=	1	AREA=	822.31	ITOP,IBOI=	70,	PMESSE=	0.
TOP	MP,MY	SA,SY,SAYE	2575.	20.	SMAX,SMIN,IAUMAXE	20.	XC,YC,ZC=	-165.0	318.0	70.50.00	VME	6592.		
MID	SMX,SY,SAYE	-243.	1548.	SMAX,SMIN,IAUMAXE	1548.	XC,YC,ZC=	-7890.	3975.	AE=78.8	7920.				
BOI	SMX,SY,SAYE	-3061.	1548.	SMAX,SMIN,IAUMAXE	1548.	XC,YC,ZC=	-10107.	4461.	AE=82.7	9569.				
ELEM	141	NUDESE	144	145	158	158	MATERIAL=	1	AREA=	822.31	ITOP,IBOI=	70,	PMESSE=	0.
TOP	MP,MY	SA,SY,SAYE	20.	698.	SMAX,SMIN,IAUMAXE	698.	XC,YC,ZC=	-156.3	318.0	70.77.06	VME	14033.		
MID	SMX,SY,SAYE	154.	150.	SMAX,SMIN,IAUMAXE	150.	XC,YC,ZC=	-1339.	7617.	AE=89.4	12400.				
BOI	SMX,SY,SAYE	-2526.	150.	SMAX,SMIN,IAUMAXE	150.	XC,YC,ZC=	-12312.	4965.	AE=83.3	11317.				



63	142	RUWSE	145	146	159	MATERIAL	1	AREA	822.31	XC, YC, ZC	70.	70.	PRESSE	0.
		AX, NY		160	160	159	MATERIAL	1	AREA	-127.	XC, YC, ZC	-96.80	318.0	
		TOP SIR	177.	177.	177.	SMAX, SMIN, IAUMAXE	464.	SMAX, SMIN, IAUMAXE	177.8	7017.	A=-89.6	VME	13230.	
		MID SIR	-5.	-5.	-5.	SMAX, SMIN, IAUMAXE	-580.	SMAX, SMIN, IAUMAXE	6140.	6140.	A=-87.3	VME	12361.	
		BOT SIR	-1767.	-1767.	-1767.	SMAX, SMIN, IAUMAXE	-1068.	SMAX, SMIN, IAUMAXE	-1679.	5329.	A=-84.2	VME	11589.	
63	145	RUWSE	146	147	160	MATERIAL	1	AREA	822.31	XC, YC, ZC	70.	70.	PRESSE	0.
		AX, NY		161	160	MATERIAL	1	AREA	-179.	XC, YC, ZC	-114.9	318.0		
		TOP SIR	792.	792.	792.	SMAX, SMIN, IAUMAXE	173.	SMAX, SMIN, IAUMAXE	4972.	4972.	A=-89.0	VME	9584.	
		MID SIR	-32.	-32.	-32.	SMAX, SMIN, IAUMAXE	-943.	SMAX, SMIN, IAUMAXE	60.	4760.	A=-80.9	VME	9490.	
		BOT SIR	-698.	-698.	-698.	SMAX, SMIN, IAUMAXE	-1203.	SMAX, SMIN, IAUMAXE	-540.	4653.	A=-82.5	VME	9587.	
63	149	RUWSE	162	147	147	MATERIAL	1	AREA	1224.5	XC, YC, ZC	70.	70.	PRESSE	0.
		AX, NY		147	147	MATERIAL	1	AREA	-76.	XC, YC, ZC	-109.3	318.0		
		TOP SIR	335.	335.	335.	SMAX, SMIN, IAUMAXE	-10.	SMAX, SMIN, IAUMAXE	3715.	3715.	A=-89.9	VME	7603.	
		MID SIR	228.	228.	228.	SMAX, SMIN, IAUMAXE	-302.	SMAX, SMIN, IAUMAXE	240.	3948.	A=-87.8	VME	7779.	
		BOT SIR	791.	791.	791.	SMAX, SMIN, IAUMAXE	-595.	SMAX, SMIN, IAUMAXE	834.	4201.	A=-85.9	VME	8017.	
63	145	RUWSE	181	148	148	MATERIAL	1	AREA	3238.4	XC, YC, ZC	70.	70.	PRESSE	0.
		AX, NY		148	148	MATERIAL	1	AREA	40.	XC, YC, ZC	-86.90	314.0		
		TOP SIR	340.	340.	340.	SMAX, SMIN, IAUMAXE	-181.	SMAX, SMIN, IAUMAXE	-332.	2028.	A=-87.4	VME	4232.	
		MID SIR	29.	29.	29.	SMAX, SMIN, IAUMAXE	-27.	SMAX, SMIN, IAUMAXE	-29.	2220.	A=-89.0	VME	4426.	
		BOT SIR	397.	397.	397.	SMAX, SMIN, IAUMAXE	126.	SMAX, SMIN, IAUMAXE	401.	2423.	A=-88.5	VME	4660.	
63	146	RUWSE	144	149	181	MATERIAL	1	AREA	4479.8	XC, YC, ZC	70.	70.	PRESSE	0.
		AX, NY		149	181	MATERIAL	1	AREA	70.	XC, YC, ZC	141.7	309.0		
		TOP SIR	-815.	-815.	-815.	SMAX, SMIN, IAUMAXE	294.	SMAX, SMIN, IAUMAXE	-801.	2773.	A=-87.3	VME	5987.	
		MID SIR	-394.	-394.	-394.	SMAX, SMIN, IAUMAXE	576.	SMAX, SMIN, IAUMAXE	-337.	2929.	A=-84.3	VME	6031.	
		BOT SIR	28.	28.	28.	SMAX, SMIN, IAUMAXE	868.	SMAX, SMIN, IAUMAXE	152.	3109.	A=-81.9	VME	6143.	
63	147	RUWSE	149	150	163	MATERIAL	1	AREA	819.77	XC, YC, ZC	70.	70.	PRESSE	0.
		AX, NY		150	163	MATERIAL	1	AREA	77.	XC, YC, ZC	-130.6	354.0		
		TOP SIR	416.	416.	416.	SMAX, SMIN, IAUMAXE	208.	SMAX, SMIN, IAUMAXE	424.	5490.	A=-88.4	VME	10774.	
		MID SIR	-195.	-195.	-195.	SMAX, SMIN, IAUMAXE	52.	SMAX, SMIN, IAUMAXE	-151.	5190.	A=-86.7	VME	10401.	
		BOT SIR	-807.	-807.	-807.	SMAX, SMIN, IAUMAXE	886.	SMAX, SMIN, IAUMAXE	-728.	4911.	A=-84.8	VME	10203.	
63	148	RUWSE	150	151	164	MATERIAL	1	AREA	819.77	XC, YC, ZC	70.	70.	PRESSE	0.
		AX, NY		151	164	MATERIAL	1	AREA	-25.	XC, YC, ZC	-194.4	354.0		
		TOP SIR	536.	536.	536.	SMAX, SMIN, IAUMAXE	391.	SMAX, SMIN, IAUMAXE	549.	5990.	A=-88.1	VME	11710.	
		MID SIR	25.	25.	25.	SMAX, SMIN, IAUMAXE	266.	SMAX, SMIN, IAUMAXE	12.	5616.	A=-88.5	VME	11215.	
		BOT SIR	-467.	-467.	-467.	SMAX, SMIN, IAUMAXE	-199.	SMAX, SMIN, IAUMAXE	-403.	5242.	A=-88.9	VME	10734.	
63	149	RUWSE	151	152	165	MATERIAL	1	AREA	819.77	XC, YC, ZC	70.	70.	PRESSE	0.
		AX, NY		152	165	MATERIAL	1	AREA	-109.	XC, YC, ZC	-155.8	354.0		
		TOP SIR	485.	485.	485.	SMAX, SMIN, IAUMAXE	195.	SMAX, SMIN, IAUMAXE	488.	5711.	A=-89.0	VME	11186.	
		MID SIR	42.	42.	42.	SMAX, SMIN, IAUMAXE	-455.	SMAX, SMIN, IAUMAXE	81.	5357.	A=-87.6	VME	10684.	
		BOT SIR	-401.	-401.	-401.	SMAX, SMIN, IAUMAXE	-1105.	SMAX, SMIN, IAUMAXE	-280.	5089.	A=-83.7	VME	10322.	
63	150	RUWSE	153	154	166	MATERIAL	1	AREA	819.77	XC, YC, ZC	70.	70.	PRESSE	0.
		AX, NY		154	166	MATERIAL	1	AREA	-315.	XC, YC, ZC	-164.5	354.0		
		TOP SIR	-884.	-884.	-884.	SMAX, SMIN, IAUMAXE	172.	SMAX, SMIN, IAUMAXE	887.	4131.	A=-88.8	VME	7850.	
		MID SIR	-38.	-38.	-38.	SMAX, SMIN, IAUMAXE	-1037.	SMAX, SMIN, IAUMAXE	106.	3807.	A=-82.1	VME	7502.	
		BOT SIR	-960.	-960.	-960.	SMAX, SMIN, IAUMAXE	-2245.	SMAX, SMIN, IAUMAXE	-250.	3908.	A=-72.5	VME	7945.	
63	151	RUWSE	154	166	167	MATERIAL	1	AREA	819.77	XC, YC, ZC	70.	70.	PRESSE	0.
		AX, NY		166	167	MATERIAL	1	AREA	-397.	XC, YC, ZC	-170.4	354.0		
		TOP SIR	529.	529.	529.	SMAX, SMIN, IAUMAXE	578.	SMAX, SMIN, IAUMAXE	636.	1575.	A=-74.2	VME	2884.	
		MID SIR	-49.	-49.	-49.	SMAX, SMIN, IAUMAXE	-945.	SMAX, SMIN, IAUMAXE	235.	1712.	A=-59.2	VME	3312.	
		BOT SIR	-827.	-827.	-827.	SMAX, SMIN, IAUMAXE	-2468.	SMAX, SMIN, IAUMAXE	816.	2832.	A=-73.7	VME	3504.	
63	152	RUWSE	154	169	168	MATERIAL	1	AREA	819.77	XC, YC, ZC	70.	70.	PRESSE	0.
		AX, NY		168	168	MATERIAL	1	AREA	-140.	XC, YC, ZC	-173.3	354.0		
		TOP SIR	-25.	-25.	-25.	SMAX, SMIN, IAUMAXE	387.	SMAX, SMIN, IAUMAXE	792.	924.	A=-12.4	VME	1600.	
		MID SIR	50.	50.	50.	SMAX, SMIN, IAUMAXE	-381.	SMAX, SMIN, IAUMAXE	511.	372.	A=-33.2	VME	675.	
		BOT SIR	1071.	1071.	1071.	SMAX, SMIN, IAUMAXE	-212.	SMAX, SMIN, IAUMAXE	1729.	1200.	A=-58.4	VME	2149.	

ELEM 153 NUMBSE 155 170 169 MATERIAL= 1 AREA= 819.77 ITOP,THUF= 70. PRESSE= 0.  
 MX,MY= 25. XC, YL,ZC= -173.3 928. AE=-12.4 VME 354.0  
 TOP SIK SX,SY,SXAY= -970. SMAX,SMIN,IAUMAXE 190. -1056. AE=33.2 VME 1604.  
 MID SIK SX,SY,SXAY= 50. 347. SMAX,SMIN,IAUMAXE 571. -174. AE=58.4 VME 875.  
 BOT SIK SX,SY,SXAY= 1071. -12. 1070. SMAX,SMIN,IAUMAXE 1729. -670. AE=58.4 VME 2144.

ELEM 154 NUMBSE 156 171 170 MATERIAL= 1 AREA= 819.77 ITOP,THUF= 70. PRESSE= 0.  
 MX,MY= 526. XC, YL,ZC= -170.4 1575. AE=-79.2 VME 2804.  
 TOP SIK SX,SY,SXAY= 526. -2401. SMAX,SMIN,IAUMAXE 819. -2511. AE=73.2 VME 3112.  
 MID SIK SX,SY,SXAY= -45. -2903. SMAX,SMIN,IAUMAXE 235. -2188. AE=73.2 VME 3112.  
 BOT SIK SX,SY,SXAY= -627. -2903. SMAX,SMIN,IAUMAXE 816. -4848. AE=59.7 VME 3304.

ELEM 155 NUMBSE 157 172 171 MATERIAL= 1 AREA= 819.77 ITOP,THUF= 70. PRESSE= 0.  
 MX,MY= 486. XC, YL,ZC= -164.5 4131. AE=-83.8 VME 354.0  
 TOP SIK SX,SY,SXAY= -48. -538. SMAX,SMIN,IAUMAXE 515. -7375. AE=82.1 VME 7856.  
 MID SIK SX,SY,SXAY= -38. -7365. SMAX,SMIN,IAUMAXE 106. -2509. AE=82.1 VME 7562.  
 BOT SIK SX,SY,SXAY= -986. -7358. SMAX,SMIN,IAUMAXE 2245. -8067. AE=72.5 VME 7945.

ELEM 156 NUMBSE 158 173 172 MATERIAL= 1 AREA= 819.77 ITOP,THUF= 70. PRESSE= 0.  
 MX,MY= 482. XC, YL,ZC= -155.8 5711. AE=-82.2 VME 354.0  
 TOP SIK SX,SY,SXAY= 482. -10430. SMAX,SMIN,IAUMAXE 169. -10934. AE=89.0 VME 1186.  
 MID SIK SX,SY,SXAY= 42. -10634. SMAX,SMIN,IAUMAXE 455. -10653. AE=87.6 VME 10684.  
 BOT SIK SX,SY,SXAY= -401. -10337. SMAX,SMIN,IAUMAXE 1105. -10459. AE=83.7 VME 10322.

ELEM 157 NUMBSE 159 174 173 MATERIAL= 1 AREA= 819.77 ITOP,THUF= 70. PRESSE= 0.  
 MX,MY= 536. XC, YL,ZC= -144.4 5990. AE=-88.1 VME 354.0  
 TOP SIK SX,SY,SXAY= 536. -11419. SMAX,SMIN,IAUMAXE 25. -11432. AE=88.1 VME 11716.  
 MID SIK SX,SY,SXAY= 25. -11191. SMAX,SMIN,IAUMAXE 32. -11199. AE=88.1 VME 11215.  
 BOT SIK SX,SY,SXAY= -487. -10963. SMAX,SMIN,IAUMAXE -199. -10967. AE=88.9 VME 10734.

ELEM 158 NUMBSE 160 175 174 MATERIAL= 1 AREA= 819.77 ITOP,THUF= 70. PRESSE= 0.  
 MX,MY= 416. XC, YL,ZC= -130.6 5490. AE=-88.4 VME 354.0  
 TOP SIK SX,SY,SXAY= 416. -10346. SMAX,SMIN,IAUMAXE -77. -144.4. AE=88.4 VME 10774.  
 MID SIK SX,SY,SXAY= -195. -10507. SMAX,SMIN,IAUMAXE 424. -10536. AE=86.7 VME 10461.  
 BOT SIK SX,SY,SXAY= -807. -10406. SMAX,SMIN,IAUMAXE -866. -10547. AE=84.8 VME 10203.

ELEM 159 NUMBSE 161 162 195 MATERIAL= 1 AREA= 4479.8 ITOP,THUF= 70. PRESSE= 0.  
 MX,MY= 28. XC, YL,ZC= -94.76 2773. AE=-87.1 VME 369.0  
 TOP SIK SX,SY,SXAY= 28. -6333. SMAX,SMIN,IAUMAXE -284. -6347. AE=84.3 VME 5987.  
 MID SIK SX,SY,SXAY= -394. -5437. SMAX,SMIN,IAUMAXE -578. -8195. AE=84.3 VME 6033.  
 BOT SIK SX,SY,SXAY= 28. -5422. SMAX,SMIN,IAUMAXE 152. -8066. AE=81.9 VME 6143.

ELEM 160 NUMBSE 196 162 162 MATERIAL= 1 AREA= 5238.4 ITOP,THUF= 70. PRESSE= 0.  
 MX,MY= 26. XC, YL,ZC= -49.90 2628. AE=87.4 VME 369.0  
 TOP SIK SX,SY,SXAY= 26. -4361. SMAX,SMIN,IAUMAXE -40. -4389. AE=87.4 VME 4232.  
 MID SIK SX,SY,SXAY= 29. -4412. SMAX,SMIN,IAUMAXE 29. -4412. AE=89.6 VME 4426.  
 BOT SIK SX,SY,SXAY= 397. -4403. SMAX,SMIN,IAUMAXE 401. -4446. AE=88.5 VME 4660.

ELEM 161 NUMBSE 182 183 163 MATERIAL= 1 AREA= 2024.0 ITOP,THUF= 70. PRESSE= 0.  
 MX,MY= 3. XC, YL,ZC= -108.6 4254. AE=89.6 VME 402.0  
 TOP SIK SX,SY,SXAY= 3. -8091. SMAX,SMIN,IAUMAXE 17. -8292. AE=89.6 VME 8362.  
 MID SIK SX,SY,SXAY= 714. -8091. SMAX,SMIN,IAUMAXE 716. -8404. AE=89.2 VME 8474.  
 BOT SIK SX,SY,SXAY= 1251. -7889. SMAX,SMIN,IAUMAXE 1255. -7895. AE=88.6 VME 8592.

ELEM 162 NUMBSE 163 184 183 MATERIAL= 1 AREA= 1360.6 ITOP,THUF= 70. PRESSE= 0.  
 MX,MY= 125. XC, YL,ZC= -130.0 4850. AE=86.7 VME 402.0  
 TOP SIK SX,SY,SXAY= 125. -9348. SMAX,SMIN,IAUMAXE 558. -9544. AE=86.7 VME 9623.  
 MID SIK SX,SY,SXAY= 65. -9348. SMAX,SMIN,IAUMAXE 552. -9416. AE=86.7 VME 9475.  
 BOT SIK SX,SY,SXAY= 45. -9257. SMAX,SMIN,IAUMAXE 545. -9289. AE=86.7 VME 9328.

ELEM 163 NUMBSE 164 184 183 MATERIAL= 1 AREA= 1360.6 ITOP,THUF= 70. PRESSE= 0.  
 MX,MY= 682. XC, YL,ZC= -143.8 5101. AE=87.0 VME 402.0  
 TOP SIK SX,SY,SXAY= 682. -10906. SMAX,SMIN,IAUMAXE -169. -10918. AE=87.0 VME 10324.  
 MID SIK SX,SY,SXAY= 160. -10309. SMAX,SMIN,IAUMAXE 18. -10311. AE=89.1 VME 10324.  
 BOT SIK SX,SY,SXAY= 684. -9818. SMAX,SMIN,IAUMAXE -107. -9819. AE=89.4 VME 10168.

ELM 104	NUMSE	105	106	105	104	MATERIAL=	1	AREA=	1300.6	XC, YC, ZC=	70.50	70.	PRESSE	0.
TOP SIK	SA, SY, SAYE	-049.	-049.	MX, MY, MXY=	-106.	-128	SMAX, SMIN, TAUMAXE	-147	-155.1	-9728.	4541.	402.0	9421.	
MID SIK	SA, SY, SAYE	782.	782.	MX, MY, MXY=	155.	155	SMAX, SMIN, TAUMAXE	84	84	884.	4881.	402.0	9421.	
BOT SIK	SA, SY, SAYE			MX, MY, MXY=	-973.	-973	SMAX, SMIN, TAUMAXE	880.	880.	-8842.	4881.	402.0	9421.	
ELM 105	NUMSE	106	107	105	104	MATERIAL=	1	AREA=	1300.6	XC, YC, ZC=	70.55	70.	PRESSE	0.
TOP SIK	SA, SY, SAYE	129.	129.	MX, MY, MXY=	26.	26	SMAX, SMIN, TAUMAXE	-203	-163.8	-6863.	3497.	402.0	9421.	
MID SIK	SA, SY, SAYE	29.	29.	MX, MY, MXY=	-89.	-89	SMAX, SMIN, TAUMAXE	130.	130.	-6863.	3497.	402.0	9421.	
BOT SIK	SA, SY, SAYE	-71.	-71.	MX, MY, MXY=	-168.	-168	SMAX, SMIN, TAUMAXE	315.	315.	-7096.	3706.	402.0	9421.	
ELM 106	NUMSE	167	168	107	106	MATERIAL=	1	AREA=	1300.6	XC, YC, ZC=	70.35	70.	PRESSE	0.
TOP SIK	SA, SY, SAYE	1066.	1066.	MX, MY, MXY=	111	111	SMAX, SMIN, TAUMAXE	-202	-169.7	-3121.	2096.	402.0	9421.	
MID SIK	SA, SY, SAYE	-19.	-19.	MX, MY, MXY=	-139.	-139	SMAX, SMIN, TAUMAXE	1071.	1071.	-3121.	2096.	402.0	9421.	
BOT SIK	SA, SY, SAYE	-1109.	-1109.	MX, MY, MXY=	-913.	-913	SMAX, SMIN, TAUMAXE	204.	204.	-4750.	2213.	402.0	9421.	
ELM 107	NUMSE	168	169	108	107	MATERIAL=	1	AREA=	1300.6	XC, YC, ZC=	70.11	70.	PRESSE	0.
TOP SIK	SA, SY, SAYE	1629.	1629.	MX, MY, MXY=	435.	435	SMAX, SMIN, TAUMAXE	-89	-172.6	-157.	894.	402.0	9421.	
MID SIK	SA, SY, SAYE	-40.	-40.	MX, MY, MXY=	-57.	-57	SMAX, SMIN, TAUMAXE	87.	87.	-1291.	609.	402.0	9421.	
BOT SIK	SA, SY, SAYE	-1709.	-1709.	MX, MY, MXY=	-742.	-742	SMAX, SMIN, TAUMAXE	-1164.	-1164.	-2717.	777.	402.0	9421.	
ELM 108	NUMSE	169	170	109	108	MATERIAL=	1	AREA=	1300.6	XC, YC, ZC=	70.11	70.	PRESSE	0.
TOP SIK	SA, SY, SAYE	1620.	1620.	MX, MY, MXY=	455.	455	SMAX, SMIN, TAUMAXE	84	172.6	-157.	894.	402.0	9421.	
MID SIK	SA, SY, SAYE	-184.	-184.	MX, MY, MXY=	194.	194	SMAX, SMIN, TAUMAXE	163.	163.	-1291.	609.	402.0	9421.	
BOT SIK	SA, SY, SAYE	-1704.	-1704.	MX, MY, MXY=	-1172.	-1172	SMAX, SMIN, TAUMAXE	-1164.	-1164.	-2717.	777.	402.0	9421.	
ELM 109	NUMSE	170	171	110	109	MATERIAL=	1	AREA=	1300.6	XC, YC, ZC=	70.11	70.	PRESSE	0.
TOP SIK	SA, SY, SAYE	1086.	1086.	MX, MY, MXY=	138.	138	SMAX, SMIN, TAUMAXE	202	-169.7	-3121.	2096.	402.0	9421.	
MID SIK	SA, SY, SAYE	-19.	-19.	MX, MY, MXY=	413.	413	SMAX, SMIN, TAUMAXE	1071.	1071.	-3121.	2096.	402.0	9421.	
BOT SIK	SA, SY, SAYE	-1104.	-1104.	MX, MY, MXY=	-1687.	-1687	SMAX, SMIN, TAUMAXE	-323.	-323.	-4750.	2213.	402.0	9421.	
ELM 110	NUMSE	171	172	111	110	MATERIAL=	1	AREA=	1300.6	XC, YC, ZC=	70.11	70.	PRESSE	0.
TOP SIK	SA, SY, SAYE	159.	159.	MX, MY, MXY=	86.	86	SMAX, SMIN, TAUMAXE	130.	-6863.	3497.	402.0	9421.		
MID SIK	SA, SY, SAYE	-71.	-71.	MX, MY, MXY=	-867.	-867	SMAX, SMIN, TAUMAXE	130.	130.	-6863.	3497.	402.0	9421.	
BOT SIK	SA, SY, SAYE			MX, MY, MXY=	1648.	1648	SMAX, SMIN, TAUMAXE	315.	315.	-7096.	3706.	402.0	9421.	
ELM 111	NUMSE	172	173	112	111	MATERIAL=	1	AREA=	1300.6	XC, YC, ZC=	70.11	70.	PRESSE	0.
TOP SIK	SA, SY, SAYE	-649.	-649.	MX, MY, MXY=	-186.	-186	SMAX, SMIN, TAUMAXE	147.	-155.1	-9728.	4541.	402.0	9421.	
MID SIK	SA, SY, SAYE	782.	782.	MX, MY, MXY=	155.	155	SMAX, SMIN, TAUMAXE	84	84	884.	4881.	402.0	9421.	
BOT SIK	SA, SY, SAYE			MX, MY, MXY=	-9734.	-9734	SMAX, SMIN, TAUMAXE	880.	880.	-8842.	4881.	402.0	9421.	
ELM 112	NUMSE	173	174	113	112	MATERIAL=	1	AREA=	1300.6	XC, YC, ZC=	70.11	70.	PRESSE	0.
TOP SIK	SA, SY, SAYE	4.	4.	MX, MY, MXY=	-169.	-169	SMAX, SMIN, TAUMAXE	69	-143.8	-10416.	5101.	402.0	9421.	
MID SIK	SA, SY, SAYE	-82.	-82.	MX, MY, MXY=	-426.	-426	SMAX, SMIN, TAUMAXE	117.	117.	-9218.	4668.	402.0	9421.	
BOT SIK	SA, SY, SAYE	664.	664.	MX, MY, MXY=	107.	107	SMAX, SMIN, TAUMAXE	685.	685.	-19819.	5242.	402.0	9421.	
ELM 113	NUMSE	174	175	114	113	MATERIAL=	1	AREA=	1300.6	XC, YC, ZC=	70.11	70.	PRESSE	0.
TOP SIK	SA, SY, SAYE	125.	125.	MX, MY, MXY=	10.	10	SMAX, SMIN, TAUMAXE	2	-130.0	-114.6	4850.	402.0	9421.	
MID SIK	SA, SY, SAYE	85.	85.	MX, MY, MXY=	-558.	-558	SMAX, SMIN, TAUMAXE	117.	117.	-9218.	4668.	402.0	9421.	
BOT SIK	SA, SY, SAYE	45.	45.	MX, MY, MXY=	-545.	-545	SMAX, SMIN, TAUMAXE	77.	77.	-9218.	4668.	402.0	9421.	
ELM 114	NUMSE	195	194	175	175	MATERIAL=	1	AREA=	2024.0	XC, YC, ZC=	70.13	70.	PRESSE	0.
TOP SIK	SA, SY, SAYE	176.	176.	MX, MY, MXY=	-140.	-140	SMAX, SMIN, TAUMAXE	177.	-826.8	4234.	402.0	9421.		
MID SIK	SA, SY, SAYE	174.	174.	MX, MY, MXY=	-65.	-65	SMAX, SMIN, TAUMAXE	176.	176.	-826.8	4234.	402.0	9421.	
BOT SIK	SA, SY, SAYE	1751.	1751.	MX, MY, MXY=	-193.	-193	SMAX, SMIN, TAUMAXE	1255.	-7895.	4575.	402.0	9421.		



ELEM 175	NUMDSE 170	177	214	213	MATERIALS 1	AREA=	11252.	XC,	YC,ZC=	165.1	70	32.84	70.	PRESS=	0.
MAX,MY=	SA,SY,SAVE	0.	MX,MY,MXY=	64.	SMAX,SMIN,TAUMAX=	0.	XC,	YC,ZC=	165.1	3068.	AE=	6.	VME	516.0	6127.
MID SIK	SA,SY,SAVE	-16.	6116.	66.	SMAX,SMIN,TAUMAX=	6116.	6116.	-10.	3063.	AE=	6.	VME	6121.	6116.	6116.
ROT SIK	SA,SY,SAVE	-0.	6115.	67.	SMAX,SMIN,TAUMAX=	6115.	6115.	-1.	3058.	AE=	6.	VME	6116.	6116.	6116.
ELEM 176	NUMDSE 171	178	215	214	MATERIALS 1	AREA=	11252.	XC,	YC,ZC=	140.0	70	93.52	70.	PRESS=	0.
MAX,MY=	SA,SY,SAVE	87.	MX,MY,MXY=	171.	SMAX,SMIN,TAUMAX=	4.	XC,	YC,ZC=	140.0	2516.	AE=	2.0	VME	516.0	5072.
MID SIK	SA,SY,SAVE	-6.	3069.	166.	SMAX,SMIN,TAUMAX=	3035.	3035.	-13.	2534.	AE=	2.1	VME	5082.	5082.	5082.
ROT SIK	SA,SY,SAVE	-99.	5031.	200.	SMAX,SMIN,TAUMAX=	5036.	5036.	-107.	2573.	AE=	2.2	VME	5093.	5093.	5093.
ELEM 177	NUMDSE 178	179	216	215	MATERIALS 1	AREA=	11252.	XC,	YC,ZC=	93.52	70	140.0	70.	PRESS=	0.
MAX,MY=	SA,SY,SAVE	261.	MX,MY,MXY=	263.	SMAX,SMIN,TAUMAX=	10.	XC,	YC,ZC=	93.52	1413.	AE=	5.4	VME	516.0	2951.
MID SIK	SA,SY,SAVE	-12.	2941.	301.	SMAX,SMIN,TAUMAX=	2972.	2972.	-62.	1507.	AE=	5.8	VME	2993.	2993.	2993.
ROT SIK	SA,SY,SAVE	-284.	2846.	339.	SMAX,SMIN,TAUMAX=	2862.	2862.	-320.	1601.	AE=	6.1	VME	3055.	3055.	3055.
ELEM 178	NUMDSE 179	180	217	216	MATERIALS 1	AREA=	11252.	XC,	YC,ZC=	32.84	70	165.1	70.	PRESS=	0.
MAX,MY=	SA,SY,SAVE	215.	MX,MY,MXY=	374.	SMAX,SMIN,TAUMAX=	204.	XC,	YC,ZC=	32.84	344.	AE=	48.6	VME	516.0	614.
MID SIK	SA,SY,SAVE	-192.	150.	378.	SMAX,SMIN,TAUMAX=	404.	404.	-344.	344.	AE=	43.5	VME	649.	649.	649.
ROT SIK	SA,SY,SAVE	-192.	-27.	409.	SMAX,SMIN,TAUMAX=	308.	308.	-527.	417.	AE=	39.3	VME	731.	731.	731.
ELEM 179	NUMDSE 180	181	218	217	MATERIALS 1	AREA=	11252.	XC,	YC,ZC=	165.1	70	165.1	70.	PRESS=	0.
MAX,MY=	SA,SY,SAVE	-193.	MX,MY,MXY=	362.	SMAX,SMIN,TAUMAX=	24.	XC,	YC,ZC=	165.1	1729.	AE=	44.0	VME	516.0	3537.
MID SIK	SA,SY,SAVE	413.	-3493.	453.	SMAX,SMIN,TAUMAX=	189.	189.	-3549.	1859.	AE=	82.9	VME	3937.	3937.	3937.
ROT SIK	SA,SY,SAVE	413.	-3414.	545.	SMAX,SMIN,TAUMAX=	495.	495.	-3490.	1992.	AE=	82.1	VME	3762.	3762.	3762.
ELEM 180	NUMDSE 181	182	200	218	MATERIALS 1	AREA=	18459.0	XC,	YC,ZL=	140.2	70	140.2	70.	PRESS=	0.
MAX,MY=	SA,SY,SAVE	-1571.	MX,MY,MXY=	636.	SMAX,SMIN,TAUMAX=	12.	XC,	YC,ZL=	140.2	5021.	AE=	63.9	VME	495.0	4920.
MID SIK	SA,SY,SAVE	-189.	-7483.	681.	SMAX,SMIN,TAUMAX=	159.	159.	-3549.	1859.	AE=	83.9	VME	6692.	6692.	6692.
ROT SIK	SA,SY,SAVE	-109.	-6863.	727.	SMAX,SMIN,TAUMAX=	-31.	-31.	-6940.	3454.	AE=	83.9	VME	8924.	8924.	8924.
ELEM 181	NUMDSE 182	183	201	200	MATERIALS 1	AREA=	1893.0	XC,	YC,ZC=	113.3	70	113.3	70.	PRESS=	0.
MAX,MY=	SA,SY,SAVE	-966.	MX,MY,MXY=	480.	SMAX,SMIN,TAUMAX=	-60.	XC,	YC,ZC=	113.3	4426.	AE=	88.9	VME	9368.	9368.
MID SIK	SA,SY,SAVE	-821.	-9371.	251.	SMAX,SMIN,TAUMAX=	876.	876.	-9378.	4651.	AE=	88.5	VME	9340.	9340.	9340.
ROT SIK	SA,SY,SAVE	821.	-8957.	21.	SMAX,SMIN,TAUMAX=	821.	821.	-8957.	4883.	AE=	89.9	VME	9394.	9394.	9394.
ELEM 182	NUMDSE 183	184	202	201	MATERIALS 1	AREA=	1893.0	XC,	YC,ZC=	142.9	70	142.9	70.	PRESS=	0.
MAX,MY=	SA,SY,SAVE	-1358.	MX,MY,MXY=	366.	SMAX,SMIN,TAUMAX=	-100.	XC,	YC,ZC=	142.9	4038.	AE=	87.4	VME	474.0	474.0.
MID SIK	SA,SY,SAVE	-168.	-8800.	-19.	SMAX,SMIN,TAUMAX=	88.	88.	-8541.	4865.	AE=	89.4	VME	8623.	8623.	8623.
ROT SIK	SA,SY,SAVE	1534.	-8281.	-403.	SMAX,SMIN,TAUMAX=	1531.	1531.	-8296.	4924.	AE=	87.7	VME	4172.	4172.	4172.
ELEM 183	NUMDSE 184	185	203	202	MATERIALS 1	AREA=	1893.0	XC,	YC,ZC=	154.2	70	154.2	70.	PRESS=	0.
MAX,MY=	SA,SY,SAVE	-1026.	MX,MY,MXY=	25.	SMAX,SMIN,TAUMAX=	-109.	XC,	YC,ZC=	154.2	3644.	AE=	89.8	VME	474.0	474.0.
MID SIK	SA,SY,SAVE	-89.	-7097.	-394.	SMAX,SMIN,TAUMAX=	109.	109.	-7326.	4017.	AE=	87.2	VME	7843.	7843.	7843.
ROT SIK	SA,SY,SAVE	1274.	-7430.	-813.	SMAX,SMIN,TAUMAX=	1349.	1349.	-7505.	4427.	AE=	84.7	VME	8243.	8243.	8243.
ELEM 184	NUMDSE 185	186	204	203	MATERIALS 1	AREA=	1893.0	XC,	YC,ZC=	162.8	70	162.8	70.	PRESS=	0.
MAX,MY=	SA,SY,SAVE	158.	MX,MY,MXY=	-290.	SMAX,SMIN,TAUMAX=	-95.	XC,	YC,ZC=	162.8	3160.	AE=	87.0	VME	474.0	474.0.
MID SIK	SA,SY,SAVE	-51.	-6148.	-123.	SMAX,SMIN,TAUMAX=	123.	123.	-6149.	4169.	AE=	84.0	VME	6236.	6236.	6236.
ROT SIK	SA,SY,SAVE	-51.	-6161.	-1021.	SMAX,SMIN,TAUMAX=	115.	115.	-6327.	3221.	AE=	80.8	VME	6365.	6365.	6365.
ELEM 185	NUMDSE 186	187	205	204	MATERIALS 1	AREA=	1893.0	XC,	YC,ZC=	168.6	70	168.6	70.	PRESS=	0.
MAX,MY=	SA,SY,SAVE	1933.	MX,MY,MXY=	-375.	SMAX,SMIN,TAUMAX=	-64.	XC,	YC,ZC=	168.6	2725.	AE=	86.0	VME	474.0	474.0.
MID SIK	SA,SY,SAVE	-147.	-4268.	-821.	SMAX,SMIN,TAUMAX=	49.	49.	-4355.	2202.	AE=	81.8	VME	4380.	4380.	4380.
ROT SIK	SA,SY,SAVE	-1453.	-5041.	-866.	SMAX,SMIN,TAUMAX=	-1755.	-1755.	-5269.	1757.	AE=	75.2	VME	4647.	4647.	4647.



ELEM	197	MODE SE	198	199	236	235	MATERIAL = 1	AREA = 11252.	XC = -4.	YC, ZC =	TTOP, THOT =	70.52	70.	PRESSE = 0.
MAX, MY =	0.	MAX, MY, MXY =	24.	24.	SMAX, SMIN, TAUMAX =	181.	SMAX, SMIN, TAUMAX =	5075.	5075.	140.0	82.	2516.	2.0	VME 516.0
TOP SIR	SX, SY, SAY =	87.	-176.	-176.	SMAX, SMIN, TAUMAX =	5031.	SMAX, SMIN, TAUMAX =	5038.	5038.	-13.	-13.	2544.	2.0	VME 5082.
MID SIR	SX, SY, SAY =	-99.	-200.	-200.	MATERIAL = 1	AREA = -2.	XC = -0.	YC, ZC =	TTOP, THOT =	70.52	70.	PRESSE = 0.		
BOT SIR	SX, SY, SAY =				MAX, MY =	0.	MAX, MY, MXY =	6117.	6117.	165.1	-10.	3068.	0.6	VME 6127.
ELEM	198	MODE SE	199	200	213	213	MATERIAL = 1	AREA = -0.	XC = -0.	YC, ZC =	TTOP, THOT =	70.52	70.	PRESSE = 0.
MAX, MY =	0.	MAX, MY, MXY =	16.	16.	SMAX, SMIN, TAUMAX =	6116.	SMAX, SMIN, TAUMAX =	6116.	6116.	-10.	-10.	3068.	0.6	VME 6127.
TOP SIR	SX, SY, SAY =	-19.	-69.	-69.	SMAX, SMIN, TAUMAX =	6115.	SMAX, SMIN, TAUMAX =	6116.	6116.	-10.	-10.	3068.	0.6	VME 6127.
MID SIR	SX, SY, SAY =	-0.	-0.	-0.	MATERIAL = 1	AREA = -0.	XC = -0.	YC, ZC =	TTOP, THOT =	70.52	70.	PRESSE = 0.		
BOT SIR	SX, SY, SAY =				MAX, MY =	200.	MAX, MY, MXY =	2792.3	2792.3	107.1	107.1	3004.	0.	VME 7390.
ELEM	199	MODE SE	219	219	200	200	MATERIAL = 1	AREA = -0.	XC = -0.	YC, ZC =	TTOP, THOT =	70.52	70.	PRESSE = 0.
MAX, MY =	0.	MAX, MY, MXY =	-1052.	-1052.	SMAX, SMIN, TAUMAX =	-728.	SMAX, SMIN, TAUMAX =	507.	507.	-4068.	-4068.	4068.	0.	VME 7844.
TOP SIR	SX, SY, SAY =	507.	-7397.	-7397.	SMAX, SMIN, TAUMAX =	2067.	SMAX, SMIN, TAUMAX =	2067.	2067.	-7397.	-7397.	4732.	0.	VME 8619.
MID SIR	SX, SY, SAY =	2067.			MATERIAL = 1	AREA = 1879.2	XC = 1879.2	YC, ZC =	TTOP, THOT =	70.52	70.	PRESSE = 0.		
BOT SIR	SX, SY, SAY =				MAX, MY =	219	MAX, MY, MXY =	1879.2	1879.2	128.3	128.3	3542.	0.	VME 8038.
ELEM	200	MODE SE	200	201	220	219	MATERIAL = 1	AREA = -70.	XC = -70.	YC, ZC =	TTOP, THOT =	70.52	70.	PRESSE = 0.
MAX, MY =	0.	MAX, MY, MXY =	-1.	-1.	SMAX, SMIN, TAUMAX =	366.	SMAX, SMIN, TAUMAX =	1050.	1050.	128.3	128.3	3542.	0.	VME 8038.
TOP SIR	SX, SY, SAY =	-1076.	-171.	-171.	SMAX, SMIN, TAUMAX =	99.	SMAX, SMIN, TAUMAX =	109.	109.	8081.	8081.	4095.	0.	VME 8156.
MID SIR	SX, SY, SAY =	108.	-7444.	-7444.	MATERIAL = 1	AREA = -428.	XC = -428.	YC, ZC =	TTOP, THOT =	70.52	70.	PRESSE = 0.		
BOT SIR	SX, SY, SAY =	1885.			MAX, MY =	220	MAX, MY, MXY =	1879.2	1879.2	141.9	141.9	3488.	0.	VME 7990.
ELEM	201	MODE SE	201	202	221	220	MATERIAL = 1	AREA = -81.	XC = -81.	YC, ZC =	TTOP, THOT =	70.52	70.	PRESSE = 0.
MAX, MY =	0.	MAX, MY, MXY =	-1606.	-1606.	SMAX, SMIN, TAUMAX =	-152.	SMAX, SMIN, TAUMAX =	48.	48.	-6273.	-6273.	4007.	0.	VME 8471.
TOP SIR	SX, SY, SAY =	1606.	-729.	-729.	SMAX, SMIN, TAUMAX =	869.	SMAX, SMIN, TAUMAX =	912.	912.	-6756.	-6756.	3834.	0.	VME 7255.
MID SIR	SX, SY, SAY =	45.			MATERIAL = 1	AREA = -0.	XC = -0.	YC, ZC =	TTOP, THOT =	70.52	70.	PRESSE = 0.		
BOT SIR	SX, SY, SAY =	1690.			MAX, MY =	221	MAX, MY, MXY =	1879.2	1879.2	153.0	153.0	3244.	0.	VME 6011.
ELEM	202	MODE SE	202	203	222	221	MATERIAL = 1	AREA = -0.	XC = -0.	YC, ZC =	TTOP, THOT =	70.52	70.	PRESSE = 0.
MAX, MY =	0.	MAX, MY, MXY =	-729.	-729.	SMAX, SMIN, TAUMAX =	-329.	SMAX, SMIN, TAUMAX =	85.	85.	-6216.	-6216.	3533.	0.	VME 7023.
TOP SIR	SX, SY, SAY =	729.	-6714.	-6714.	SMAX, SMIN, TAUMAX =	869.	SMAX, SMIN, TAUMAX =	912.	912.	-6756.	-6756.	3834.	0.	VME 7255.
MID SIR	SX, SY, SAY =	70.			MATERIAL = 1	AREA = -30.	XC = -30.	YC, ZC =	TTOP, THOT =	70.52	70.	PRESSE = 0.		
BOT SIR	SX, SY, SAY =	869.			MAX, MY =	222	MAX, MY, MXY =	1879.2	1879.2	161.6	161.6	3208.	0.	VME 8060.
ELEM	203	MODE SE	203	204	223	222	MATERIAL = 1	AREA = -30.	XC = -30.	YC, ZC =	TTOP, THOT =	70.52	70.	PRESSE = 0.
MAX, MY =	0.	MAX, MY, MXY =	759.	759.	SMAX, SMIN, TAUMAX =	-421.	SMAX, SMIN, TAUMAX =	48.	48.	-5642.	-5642.	2967.	0.	VME 5951.
TOP SIR	SX, SY, SAY =	18.	-5697.	-5697.	SMAX, SMIN, TAUMAX =	-723.	SMAX, SMIN, TAUMAX =	-671.	-671.	-6219.	-6219.	2774.	0.	VME 5913.
MID SIR	SX, SY, SAY =	18.			MATERIAL = 1	AREA = 1879.2	XC = 1879.2	YC, ZC =	TTOP, THOT =	70.52	70.	PRESSE = 0.		
BOT SIR	SX, SY, SAY =	869.			MAX, MY =	223	MAX, MY, MXY =	1879.2	1879.2	167.4	167.4	3325.	0.	VME 5824.
ELEM	204	MODE SE	204	205	224	223	MATERIAL = 1	AREA = -4.	XC = -4.	YC, ZC =	TTOP, THOT =	70.52	70.	PRESSE = 0.
MAX, MY =	0.	MAX, MY, MXY =	2907.	2907.	SMAX, SMIN, TAUMAX =	-332.	SMAX, SMIN, TAUMAX =	2418.	2418.	-4273.	-4273.	4473.	0.	VME 4863.
TOP SIR	SX, SY, SAY =	2907.	-2511.	-2511.	SMAX, SMIN, TAUMAX =	-367.	SMAX, SMIN, TAUMAX =	-2469.	-2469.	-5735.	-5735.	1632.	0.	VME 4981.
MID SIR	SX, SY, SAY =	2511.			MATERIAL = 1	AREA = 1879.2	XC = 1879.2	YC, ZC =	TTOP, THOT =	70.52	70.	PRESSE = 0.		
BOT SIR	SX, SY, SAY =	2907.			MAX, MY =	224	MAX, MY, MXY =	1879.2	1879.2	180.3	180.3	3358.	0.	VME 5819.
ELEM	205	MODE SE	205	206	225	224	MATERIAL = 1	AREA = 2	XC = 2	YC, ZC =	TTOP, THOT =	70.52	70.	PRESSE = 0.
MAX, MY =	0.	MAX, MY, MXY =	3570.	3570.	SMAX, SMIN, TAUMAX =	-145.	SMAX, SMIN, TAUMAX =	106.	106.	-5142.	-5142.	5358.	0.	VME 5819.
TOP SIR	SX, SY, SAY =	3570.	-4419.	-4419.	SMAX, SMIN, TAUMAX =	-139.	SMAX, SMIN, TAUMAX =	106.	106.	-5142.	-5142.	2159.	0.	VME 4372.
MID SIR	SX, SY, SAY =	110.	-5699.	-5699.	MATERIAL = 1	AREA = -0.	XC = -0.	YC, ZC =	TTOP, THOT =	70.52	70.	PRESSE = 0.		
BOT SIR	SX, SY, SAY =	3790.			MAX, MY =	225	MAX, MY, MXY =	1879.2	1879.2	170.3	170.3	3358.	0.	VME 5819.
ELEM	206	MODE SE	206	207	226	225	MATERIAL = 1	AREA = -2	XC = -2	YC, ZC =	TTOP, THOT =	70.52	70.	PRESSE = 0.
MAX, MY =	0.	MAX, MY, MXY =	-13.	-13.	SMAX, SMIN, TAUMAX =	145.	SMAX, SMIN, TAUMAX =	573.	573.	-5170.	-5170.	3358.	0.	VME 5819.
TOP SIR	SX, SY, SAY =	3570.	-4419.	-4419.	SMAX, SMIN, TAUMAX =	139.	SMAX, SMIN, TAUMAX =	106.	106.	-5142.	-5142.	2159.	0.	VME 4372.
MID SIR	SX, SY, SAY =	110.	-5699.	-5699.	MATERIAL = 1	AREA = 4	XC = 4	YC, ZC =	TTOP, THOT =	70.52	70.	PRESSE = 0.		
BOT SIR	SX, SY, SAY =	3790.			MAX, MY =	226	MAX, MY, MXY =	1879.2	1879.2	170.3	170.3	3358.	0.	VME 5819.
ELEM	207	MODE SE	207	208	227	226	MATERIAL = 1	AREA = 4	XC = 4	YC, ZC =	TTOP, THOT =	70.52	70.	PRESSE = 0.
MAX, MY =	0.	MAX, MY, MXY =	4215.	4215.	SMAX, SMIN, TAUMAX =	332.	SMAX, SMIN, TAUMAX =	2418.	2418.	-4273.	-4273.	4473.	0.	VME 4863.
TOP SIR	SX, SY, SAY =	4215.	-2511.	-2511.	SMAX, SMIN, TAUMAX =	347.	SMAX, SMIN, TAUMAX =	-2469.	-2469.	-5735.	-5735.	1632.	0.	VME 4981.
MID SIR	SX, SY, SAY =	2511.			MATERIAL = 1	AREA = 1879.2	XC = 1879.2	YC, ZC =	TTOP, THOT =	70.52	70.	PRESSE = 0.		
BOT SIR	SX, SY, SAY =	4215.			MAX, MY =	227	MAX, MY, MXY =	1879.2	1879.2	170.4	170.4	3358.	0.	VME 5819.



ELN	219	NUMBS	214	220	238	MAT	189	1865.3	70.	70.	PRESSE	0.
TOP	219	NUMBS	214	220	238	MAT	189	1865.3	70.	70.	PRESSE	0.
MID	219	NUMBS	214	220	238	MAT	189	1865.3	70.	70.	PRESSE	0.
BOT	219	NUMBS	214	220	238	MAT	189	1865.3	70.	70.	PRESSE	0.
ELN	220	NUMBS	220	221	239	MAT	189	1865.3	70.	70.	PRESSE	0.
TOP	220	NUMBS	220	221	239	MAT	189	1865.3	70.	70.	PRESSE	0.
MID	220	NUMBS	220	221	239	MAT	189	1865.3	70.	70.	PRESSE	0.
BOT	220	NUMBS	220	221	239	MAT	189	1865.3	70.	70.	PRESSE	0.
ELN	221	NUMBS	221	222	240	MAT	189	1865.3	70.	70.	PRESSE	0.
TOP	221	NUMBS	221	222	240	MAT	189	1865.3	70.	70.	PRESSE	0.
MID	221	NUMBS	221	222	240	MAT	189	1865.3	70.	70.	PRESSE	0.
BOT	221	NUMBS	221	222	240	MAT	189	1865.3	70.	70.	PRESSE	0.
ELN	222	NUMBS	222	223	241	MAT	189	1865.3	70.	70.	PRESSE	0.
TOP	222	NUMBS	222	223	241	MAT	189	1865.3	70.	70.	PRESSE	0.
MID	222	NUMBS	222	223	241	MAT	189	1865.3	70.	70.	PRESSE	0.
BOT	222	NUMBS	222	223	241	MAT	189	1865.3	70.	70.	PRESSE	0.
ELN	223	NUMBS	223	224	242	MAT	189	1865.3	70.	70.	PRESSE	0.
TOP	223	NUMBS	223	224	242	MAT	189	1865.3	70.	70.	PRESSE	0.
MID	223	NUMBS	223	224	242	MAT	189	1865.3	70.	70.	PRESSE	0.
BOT	223	NUMBS	223	224	242	MAT	189	1865.3	70.	70.	PRESSE	0.
ELN	224	NUMBS	224	225	243	MAT	189	1865.3	70.	70.	PRESSE	0.
TOP	224	NUMBS	224	225	243	MAT	189	1865.3	70.	70.	PRESSE	0.
MID	224	NUMBS	224	225	243	MAT	189	1865.3	70.	70.	PRESSE	0.
BOT	224	NUMBS	224	225	243	MAT	189	1865.3	70.	70.	PRESSE	0.
ELN	225	NUMBS	225	226	244	MAT	189	1865.3	70.	70.	PRESSE	0.
TOP	225	NUMBS	225	226	244	MAT	189	1865.3	70.	70.	PRESSE	0.
MID	225	NUMBS	225	226	244	MAT	189	1865.3	70.	70.	PRESSE	0.
BOT	225	NUMBS	225	226	244	MAT	189	1865.3	70.	70.	PRESSE	0.
ELN	226	NUMBS	226	227	245	MAT	189	1865.3	70.	70.	PRESSE	0.
TOP	226	NUMBS	226	227	245	MAT	189	1865.3	70.	70.	PRESSE	0.
MID	226	NUMBS	226	227	245	MAT	189	1865.3	70.	70.	PRESSE	0.
BOT	226	NUMBS	226	227	245	MAT	189	1865.3	70.	70.	PRESSE	0.
ELN	227	NUMBS	227	228	246	MAT	189	1865.3	70.	70.	PRESSE	0.
TOP	227	NUMBS	227	228	246	MAT	189	1865.3	70.	70.	PRESSE	0.
MID	227	NUMBS	227	228	246	MAT	189	1865.3	70.	70.	PRESSE	0.
BOT	227	NUMBS	227	228	246	MAT	189	1865.3	70.	70.	PRESSE	0.
ELN	228	NUMBS	228	229	247	MAT	189	1865.3	70.	70.	PRESSE	0.
TOP	228	NUMBS	228	229	247	MAT	189	1865.3	70.	70.	PRESSE	0.
MID	228	NUMBS	228	229	247	MAT	189	1865.3	70.	70.	PRESSE	0.
BOT	228	NUMBS	228	229	247	MAT	189	1865.3	70.	70.	PRESSE	0.
ELN	229	NUMBS	229	230	248	MAT	189	1865.3	70.	70.	PRESSE	0.
TOP	229	NUMBS	229	230	248	MAT	189	1865.3	70.	70.	PRESSE	0.
MID	229	NUMBS	229	230	248	MAT	189	1865.3	70.	70.	PRESSE	0.
BOT	229	NUMBS	229	230	248	MAT	189	1865.3	70.	70.	PRESSE	0.









ELEM 263	MUUSE=	262	263	281	280	MATERIAL=	1	AHEA=	2099.0	XC,ZC=	2099.0	TTOP,TRUF=	70.91	70.	PRESSE=	0.
MAXMY=	-5.0	-13.	3013.	MX,MY,MXY=	1797.	210.	SMAX,SMIN,TAUMAX=	-8.	-186.4	-186.4	9064.	AE=89.7	VME	816.0	7126.	
TOP SIK	SX,SY,SXY=	3013.	46.	SMAX,SMIN,TAUMAX=	47.	SMAX,SMIN,TAUMAX=	3013.	2454.	2454.	2454.	AE=89.8	VME	2931.	5861.		
MID SIK	SX,SY,SXY=	-46.	-17.	SMAX,SMIN,TAUMAX=	17.	SMAX,SMIN,TAUMAX=	-3105.	1839.	1839.	1839.	AE=-89.8	VME	5861.			
HOT SIK	SX,SY,SXY=	-3105.	-6783.		-12.											
ELEM 264	MUUSE=	263	264	282	281	MATERIAL=	1	AHEA=	2099.0	XC,ZC=	2099.0	TTOP,TRUF=	70.	70.	PRESSE=	0.
MAXMY=	-15.	-2598.	5153.	MX,MY,MXY=	686.	136.	SMAX,SMIN,TAUMAX=	-19.	-163.6	-163.6	3879.	AE=88.9	VME	816.0	6839.	
TOP SIK	SX,SY,SXY=	2598.	156.	SMAX,SMIN,TAUMAX=	156.	SMAX,SMIN,TAUMAX=	2062.	3013.	3013.	3013.	AE=89.2	VME	6044.			
MID SIK	SX,SY,SXY=	-17.	-6060.	SMAX,SMIN,TAUMAX=	18.	SMAX,SMIN,TAUMAX=	-2672.	2148.	2148.	2148.	AE=89.8	VME	6008.			
HOT SIK	SX,SY,SXY=	-2672.	-6963.													
ELEM 265	MUUSE=	264	265	283	282	MATERIAL=	1	AHEA=	2099.0	XC,ZC=	2099.0	TTOP,TRUF=	70.	70.	PRESSE=	0.
MAXMY=	-22.	-0.	5748.	MX,MY,MXY=	434.	126.	SMAX,SMIN,TAUMAX=	-23.	-157.9	-157.9	1684.	AE=88.4	VME	816.0	6720.	
TOP SIK	SX,SY,SXY=	0.	1635.	SMAX,SMIN,TAUMAX=	204.	SMAX,SMIN,TAUMAX=	1691.	3409.	3409.	3409.	AE=88.5	VME	6410.			
MID SIK	SX,SY,SXY=	-32.	-6229.	SMAX,SMIN,TAUMAX=	117.	SMAX,SMIN,TAUMAX=	-30.	3202.	3202.	3202.	AE=88.8	VME	6213.			
HOT SIK	SX,SY,SXY=	-1700.	-6709.		30.	SMAX,SMIN,TAUMAX=	-1700.	3005.	3005.	3005.	AE=89.3	VME	6282.			
ELEM 266	MUUSE=	265	266	284	283	MATERIAL=	1	AHEA=	2099.0	XC,ZC=	2099.0	TTOP,TRUF=	70.	70.	PRESSE=	0.
MAXMY=	-27.	18.	6306.	MX,MY,MXY=	130.	28.	SMAX,SMIN,TAUMAX=	-14.	-149.6	-149.6	2978.	AE=89.7	VME	816.0	6375.	
TOP SIK	SX,SY,SXY=	27.	485.	SMAX,SMIN,TAUMAX=	184.	SMAX,SMIN,TAUMAX=	490.	3409.	3409.	3409.	AE=89.7	VME	6496.			
MID SIK	SX,SY,SXY=	-14.	-6413.	SMAX,SMIN,TAUMAX=	130.	SMAX,SMIN,TAUMAX=	-11.	3254.	3254.	3254.	AE=89.6	VME	6090.			
HOT SIK	SX,SY,SXY=	-513.	-6521.		76.	SMAX,SMIN,TAUMAX=	-512.	3005.	3005.	3005.	AE=89.3	VME	6282.			
ELEM 267	MUUSE=	266	267	285	284	MATERIAL=	1	AHEA=	2099.0	XC,ZC=	2099.0	TTOP,TRUF=	70.	70.	PRESSE=	0.
MAXMY=	-25.	70.	6725.	MX,MY,MXY=	207.	63.	SMAX,SMIN,TAUMAX=	-23.	-138.7	-138.7	2978.	AE=89.7	VME	816.0	6375.	
TOP SIK	SX,SY,SXY=	25.	24.	SMAX,SMIN,TAUMAX=	42.	SMAX,SMIN,TAUMAX=	24.	3254.	3254.	3254.	AE=89.6	VME	6496.			
MID SIK	SX,SY,SXY=	818.	-6243.	SMAX,SMIN,TAUMAX=	48.	SMAX,SMIN,TAUMAX=	819.	3531.	3531.	3531.	AE=89.6	VME	6090.			
HOT SIK	SX,SY,SXY=	1756.	-6061.													
ELEM 268	MUUSE=	267	268	286	285	MATERIAL=	1	AHEA=	2099.0	XC,ZC=	2099.0	TTOP,TRUF=	70.	70.	PRESSE=	0.
MAXMY=	-17.	11.	7208.	MX,MY,MXY=	458.	149.	SMAX,SMIN,TAUMAX=	-14.	-125.4	-125.4	2725.	AE=89.3	VME	816.0	6511.	
TOP SIK	SX,SY,SXY=	17.	1761.	SMAX,SMIN,TAUMAX=	69.	SMAX,SMIN,TAUMAX=	1769.	3617.	3617.	3617.	AE=89.3	VME	6614.			
MID SIK	SX,SY,SXY=	-2.	-6833.	SMAX,SMIN,TAUMAX=	7.	SMAX,SMIN,TAUMAX=	-6833.	3904.	3904.	3904.	AE=89.4	VME	7105.			
HOT SIK	SX,SY,SXY=	1756.	-6061.		81.	SMAX,SMIN,TAUMAX=	1757.	3904.	3904.	3904.	AE=89.4	VME	7105.			
ELEM 269	MUUSE=	268	269	306	286	MATERIAL=	1	AHEA=	9378.2	XC,ZC=	9378.2	TTOP,TRUF=	70.	70.	PRESSE=	0.
MAXMY=	12.	0.	5386.	MX,MY,MXY=	466.	162.	SMAX,SMIN,TAUMAX=	25.	-90.93	-90.93	2146.	AE=83.7	VME	840.0	5648.	
TOP SIK	SX,SY,SXY=	12.	-2159.	SMAX,SMIN,TAUMAX=	808.	SMAX,SMIN,TAUMAX=	-2106.	2704.	2704.	2704.	AE=86.1	VME	5567.			
MID SIK	SX,SY,SXY=	-368.	-368.	SMAX,SMIN,TAUMAX=	-372.	SMAX,SMIN,TAUMAX=	1434.	3274.	3274.	3274.	AE=-87.6	VME	5962.			
HOT SIK	SX,SY,SXY=	1423.	-5103.													
ELEM 270	MUUSE=	269	270	307	306	MATERIAL=	1	AHEA=	12468.	XC,ZC=	12468.	TTOP,TRUF=	70.	70.	PRESSE=	0.
MAXMY=	10.	10.	3745.	MX,MY,MXY=	186.	54.	SMAX,SMIN,TAUMAX=	9.	-31.84	-31.84	1569.	AE=81.5	VME	868.0	3505.	
TOP SIK	SX,SY,SXY=	10.	-712.	SMAX,SMIN,TAUMAX=	474.	SMAX,SMIN,TAUMAX=	52.	1796.	1796.	1796.	AE=81.5	VME	3567.			
MID SIK	SX,SY,SXY=	715.	-3265.	SMAX,SMIN,TAUMAX=	391.	SMAX,SMIN,TAUMAX=	754.	2026.	2026.	2026.	AE=-84.4	VME	3737.			
HOT SIK	SX,SY,SXY=	-226.	-371.													
ELEM 271	MUUSE=	270	271	308	307	MATERIAL=	1	AHEA=	12468.	XC,ZC=	12468.	TTOP,TRUF=	70.	70.	PRESSE=	0.
MAXMY=	3.	0.	240.	MX,MY,MXY=	57.	35.	SMAX,SMIN,TAUMAX=	-13.	-90.68	-90.68	1215.	AE=-8.9	VME	864.0	2625.	
TOP SIK	SX,SY,SXY=	3.	211.	SMAX,SMIN,TAUMAX=	359.	SMAX,SMIN,TAUMAX=	2784.	359.	359.	359.	AE=-8.9	VME	2674.			
MID SIK	SX,SY,SXY=	-8.	-2624.	SMAX,SMIN,TAUMAX=	-303.	SMAX,SMIN,TAUMAX=	2658.	1474.	1474.	1474.	AE=-8.9	VME	2764.			
HOT SIK	SX,SY,SXY=	-226.	-371.		352.	SMAX,SMIN,TAUMAX=	-416.	2418.	2418.	2418.	AE=-2.0	VME	4676.			
ELEM 272	MUUSE=	271	272	309	308	MATERIAL=	1	AHEA=	12468.	XC,ZC=	12468.	TTOP,TRUF=	70.	70.	PRESSE=	0.
MAXMY=	-1.	0.	4754.	MX,MY,MXY=	98.	22.	SMAX,SMIN,TAUMAX=	-8.	-135.7	-135.7	2259.	AE=-2.0	VME	4676.	4718.	
TOP SIK	SX,SY,SXY=	-1.	248.	SMAX,SMIN,TAUMAX=	157.	SMAX,SMIN,TAUMAX=	4754.	242.	242.	242.	AE=-2.0	VME	4676.			
MID SIK	SX,SY,SXY=	248.	4663.	SMAX,SMIN,TAUMAX=	100.	SMAX,SMIN,TAUMAX=	4632.	2418.	2418.	2418.	AE=-2.0	VME	4718.			
HOT SIK	SX,SY,SXY=	-234.	-4581.		322.	SMAX,SMIN,TAUMAX=	-4582.	2418.	2418.	2418.	AE=-2.0	VME	4718.			

ELEM 274	NUMER= 273	250	287	310	MATERIAL= 1	AKLAE	12408	XC	YC	ZC=	110.1	70	70	PRESSE= 0.	63
MX,MY	111.	0.	5757.	5757.	-53.	SMAX,SMIN,TAUMAXE	-1.	100.1	2824.	AE	-31.84	AE	-31.84	664.0	
MID SIR	SA,SY,SKAYE	111.	5757.	5757.	-53.	SMAX,SMIN,TAUMAXE	-1.	110.	2824.	AE	-31.84	AE	-31.84	5703.	
BOT SIR	SA,SY,SKAYE	-93.	5684.	5684.	-68.	SMAX,SMIN,TAUMAXE	5685.	-94.	2890.	AE	-31.84	AE	-31.84	5717.	
ELEM 275	NUMER= 293	292	274	274	MATERIAL= 1	AKLAE	3090.9	XC	YC	ZC=	103.8	70	70	PRESSE= 0.	63
MX,MY	159.6	159.6	6094.	6094.	-533.	SMAX,SMIN,TAUMAXE	-22.	103.8	2250.	AE	88.7	AE	88.7	5476.	
MID SIR	SA,SY,SKAYE	159.6	6094.	6094.	-533.	SMAX,SMIN,TAUMAXE	-22.	109.0	2250.	AE	88.7	AE	88.7	5476.	
BOT SIR	SA,SY,SKAYE	2497.	5370.	5370.	-60.	SMAX,SMIN,TAUMAXE	2498.	-5370.	3434.	AE	-89.6	AE	-89.6	6968.	
ELEM 276	NUMER= 274	275	294	294	MATERIAL= 1	AKLAE	2080.9	XC	YC	ZC=	124.3	70	70	PRESSE= 0.	63
MX,MY	147.6	147.6	6044.	6044.	-193.	SMAX,SMIN,TAUMAXE	-147.1	124.3	2526.	AE	87.8	AE	87.8	5927.	
MID SIR	SA,SY,SKAYE	147.6	6044.	6044.	-193.	SMAX,SMIN,TAUMAXE	-147.1	124.3	2526.	AE	87.8	AE	87.8	5927.	
BOT SIR	SA,SY,SKAYE	1600.	5572.	5572.	-82.	SMAX,SMIN,TAUMAXE	1601.	-8048.	3587.	AE	89.3	AE	89.3	6523.	
ELEM 277	NUMER= 275	276	295	294	MATERIAL= 1	AKLAE	2080.9	XC	YC	ZC=	137.5	70	70	PRESSE= 0.	63
MX,MY	12.	12.	6368.	6368.	-27.	SMAX,SMIN,TAUMAXE	7.	137.5	4051.	AE	-89.4	AE	-89.4	6239.	
MID SIR	SA,SY,SKAYE	12.	6368.	6368.	-27.	SMAX,SMIN,TAUMAXE	7.	137.5	4051.	AE	-89.4	AE	-89.4	6239.	
BOT SIR	SA,SY,SKAYE	249.	6132.	6132.	0.	SMAX,SMIN,TAUMAXE	298.	-6250.	3133.	AE	90.0	AE	90.0	6250.	
ELEM 278	NUMER= 276	277	296	295	MATERIAL= 1	AKLAE	2080.9	XC	YC	ZC=	148.3	70	70	PRESSE= 0.	63
MX,MY	75.0	75.0	6482.	6482.	-110.	SMAX,SMIN,TAUMAXE	23.	148.3	3374.	AE	-89.1	AE	-89.1	6402.	
MID SIR	SA,SY,SKAYE	75.0	6482.	6482.	-110.	SMAX,SMIN,TAUMAXE	23.	148.3	3374.	AE	-89.1	AE	-89.1	6402.	
BOT SIR	SA,SY,SKAYE	-755.	6482.	6482.	-63.	SMAX,SMIN,TAUMAXE	-754.	-6483.	2860.	AE	89.4	AE	89.4	6141.	
ELEM 279	NUMER= 277	278	297	296	MATERIAL= 1	AKLAE	2080.9	XC	YC	ZC=	156.8	70	70	PRESSE= 0.	63
MX,MY	15.	15.	5687.	5687.	-179.	SMAX,SMIN,TAUMAXE	1814.	156.8	3133.	AE	-88.6	AE	-88.6	6783.	
MID SIR	SA,SY,SKAYE	15.	5687.	5687.	-179.	SMAX,SMIN,TAUMAXE	1814.	156.8	3133.	AE	-88.6	AE	-88.6	6783.	
BOT SIR	SA,SY,SKAYE	-1810.	6799.	6799.	-1.	SMAX,SMIN,TAUMAXE	-1810.	-6799.	2495.	AE	-90.0	AE	-90.0	6094.	
ELEM 280	NUMER= 278	279	298	297	MATERIAL= 1	AKLAE	2080.9	XC	YC	ZC=	162.2	70	70	PRESSE= 0.	63
MX,MY	10.	10.	6250.	6250.	-199.	SMAX,SMIN,TAUMAXE	-20.	162.2	4015.	AE	-89.3	AE	-89.3	6240.	
MID SIR	SA,SY,SKAYE	10.	6250.	6250.	-199.	SMAX,SMIN,TAUMAXE	-20.	162.2	4015.	AE	-89.3	AE	-89.3	6240.	
BOT SIR	SA,SY,SKAYE	-2402.	6842.	6842.	41.	SMAX,SMIN,TAUMAXE	-2401.	-6843.	2221.	AE	89.5	AE	89.5	6013.	
ELEM 281	NUMER= 279	280	299	298	MATERIAL= 1	AKLAE	2080.9	XC	YC	ZC=	165.0	70	70	PRESSE= 0.	63
MX,MY	11.	11.	6417.	6417.	-80.	SMAX,SMIN,TAUMAXE	2732.	165.0	4025.	AE	-89.4	AE	-89.4	7091.	
MID SIR	SA,SY,SKAYE	11.	6417.	6417.	-80.	SMAX,SMIN,TAUMAXE	2732.	165.0	4025.	AE	-89.4	AE	-89.4	7091.	
BOT SIR	SA,SY,SKAYE	-2704.	7190.	7190.	-25.	SMAX,SMIN,TAUMAXE	-2704.	-7190.	2241.	AE	89.7	AE	89.7	6284.	
ELEM 282	NUMER= 280	281	300	299	MATERIAL= 1	AKLAE	2080.9	XC	YC	ZC=	165.0	70	70	PRESSE= 0.	63
MX,MY	11.	11.	6250.	6250.	-199.	SMAX,SMIN,TAUMAXE	-20.	165.0	4015.	AE	-89.3	AE	-89.3	6240.	
MID SIR	SA,SY,SKAYE	11.	6250.	6250.	-199.	SMAX,SMIN,TAUMAXE	-20.	165.0	4015.	AE	-89.3	AE	-89.3	6240.	
BOT SIR	SA,SY,SKAYE	-2709.	6842.	6842.	25.	SMAX,SMIN,TAUMAXE	-2709.	-6843.	2241.	AE	89.7	AE	89.7	6284.	
ELEM 283	NUMER= 281	282	301	300	MATERIAL= 1	AKLAE	2080.9	XC	YC	ZC=	162.2	70	70	PRESSE= 0.	63
MX,MY	11.	11.	6250.	6250.	-199.	SMAX,SMIN,TAUMAXE	-20.	162.2	4015.	AE	88.6	AE	88.6	7144.	
MID SIR	SA,SY,SKAYE	11.	6250.	6250.	-199.	SMAX,SMIN,TAUMAXE	-20.	162.2	4015.	AE	88.6	AE	88.6	7144.	
BOT SIR	SA,SY,SKAYE	-2402.	6842.	6842.	-41.	SMAX,SMIN,TAUMAXE	-2401.	-6843.	2221.	AE	-89.5	AE	-89.5	6013.	
ELEM 284	NUMER= 282	283	302	301	MATERIAL= 1	AKLAE	2080.9	XC	YC	ZC=	156.6	70	70	PRESSE= 0.	63
MX,MY	109.	109.	6243.	6243.	179.	SMAX,SMIN,TAUMAXE	1814.	156.6	3753.	AE	88.6	AE	88.6	6783.	
MID SIR	SA,SY,SKAYE	109.	6243.	6243.	179.	SMAX,SMIN,TAUMAXE	1814.	156.6	3753.	AE	88.6	AE	88.6	6783.	
BOT SIR	SA,SY,SKAYE	-1810.	6799.	6799.	1.	SMAX,SMIN,TAUMAXE	-1810.	-6799.	2495.	AE	90.0	AE	90.0	6094.	

ELEM	285	MODESE	283	284	303	302	MATERIAL=	1	AKLA=	2080.9	XC,YC,ZC=	70.13	70.13	PRESSE	0.
MX,MY	-2.5	0.	0.	0.	MX,MY	MXY=	64	-23.	-140.5	-73.13	AE=	89.1	VME	912.0	
TOP SIR	SX,SY,SAYE	756.	756.	756.	SMAX,SMIN,IAUMAXE	110.	SMAX,SMIN,IAUMAXE	750.	5374.	3114.	AE=	89.8	VME	6402.	
MID SIR	SX,SY,SAYE	-755.	-755.	-755.	SMAX,SMIN,IAUMAXE	-63.	SMAX,SMIN,IAUMAXE	-754.	2864.	2864.	AE=	-89.4	VME	6141.	
BOT SIR	SX,SY,SAYE														
ELEM	286	MODESE	284	285	304	303	MATERIAL=	1	ANEAE=	2080.9	XC,YC,ZC=	70.06	70.06	PRESSE	0.
MX,MY	-2.5	0.	0.	0.	MX,MY	MXY=	27.	-73.	-137.5	-61.	AE=	89.8	VME	6239.	
TOP SIR	SX,SY,SAYE	-266.	-266.	-266.	SMAX,SMIN,IAUMAXE	27.	SMAX,SMIN,IAUMAXE	266.	3051.	3051.	AE=	89.8	VME	6239.	
MID SIR	SX,SY,SAYE	298.	298.	298.	SMAX,SMIN,IAUMAXE	-9.	SMAX,SMIN,IAUMAXE	16.	3133.	3133.	AE=	-90.0	VME	6258.	
BOT SIR	SX,SY,SAYE								298.	298.	AE=	-89.8	VME	6286.	
ELEM	287	MODESE	285	286	304	304	MATERIAL=	1	ANEAE=	2080.9	XC,YC,ZC=	70.109	70.109	PRESSE	0.
MX,MY	-2.5	0.	0.	0.	MX,MY	MXY=	-123.	15.	-128.3	-109.0	AE=	87.8	VME	5927.	
TOP SIR	SX,SY,SAYE	-1474.	-1474.	-1474.	SMAX,SMIN,IAUMAXE	-195.	SMAX,SMIN,IAUMAXE	-1471.	2526.	2526.	AE=	87.8	VME	6080.	
MID SIR	SX,SY,SAYE	1600.	1600.	1600.	SMAX,SMIN,IAUMAXE	-139.	SMAX,SMIN,IAUMAXE	64.	3056.	3056.	AE=	-89.3	VME	6523.	
BOT SIR	SX,SY,SAYE								1601.	1601.	AE=	-89.3	VME	6523.	
ELEM	288	MODESE	286	287	286	286	MATERIAL=	1	ANEAE=	3090.1	XC,YC,ZC=	70.126	70.126	PRESSE	0.
MX,MY	-2.5	0.	0.	0.	MX,MY	MXY=	-531.	22.	-163.8	-126.1	AE=	86.7	VME	5476.	
TOP SIR	SX,SY,SAYE	-1598.	-1598.	-1598.	SMAX,SMIN,IAUMAXE	-105.	SMAX,SMIN,IAUMAXE	-1597.	2250.	2250.	AE=	89.6	VME	5969.	
MID SIR	SX,SY,SAYE	2400.	2400.	2400.	SMAX,SMIN,IAUMAXE	-23.	SMAX,SMIN,IAUMAXE	489.	3091.	3091.	AE=	-89.6	VME	6064.	
BOT SIR	SX,SY,SAYE								2497.	2497.	AE=	-89.6	VME	6064.	
ELEM	289	MODESE	287	288	319	318	MATERIAL=	1	ANEAE=	15282	XC,YC,ZC=	70.31	70.31	PRESSE	0.
MX,MY	-2.5	0.	0.	0.	MX,MY	MXY=	68.	4.	157.0	31.22	AE=	5	VME	5422.	
TOP SIR	SX,SY,SAYE	244.	244.	244.	SMAX,SMIN,IAUMAXE	45.	SMAX,SMIN,IAUMAXE	3540.	243.	2644.	AE=	5	VME	5422.	
MID SIR	SX,SY,SAYE	-244.	-244.	-244.	SMAX,SMIN,IAUMAXE	61.	SMAX,SMIN,IAUMAXE	3421.	-1.	2711.	AE=	8	VME	5429.	
BOT SIR	SX,SY,SAYE								-245.	2774.	AE=	8	VME	5429.	
ELEM	290	MODESE	288	289	320	319	MATERIAL=	1	ANEAE=	15282	XC,YC,ZC=	70.88	70.88	PRESSE	0.
MX,MY	-2.5	0.	0.	0.	MX,MY	MXY=	93.	11.	133.1	88.91	AE=	1.9	VME	4354.	
TOP SIR	SX,SY,SAYE	356.	356.	356.	SMAX,SMIN,IAUMAXE	138.	SMAX,SMIN,IAUMAXE	4518.	351.	2084.	AE=	2.4	VME	4384.	
MID SIR	SX,SY,SAYE	-356.	-356.	-356.	SMAX,SMIN,IAUMAXE	180.	SMAX,SMIN,IAUMAXE	4379.	-9.	2194.	AE=	2.8	VME	4437.	
BOT SIR	SX,SY,SAYE								-369.	2305.	AE=	2.8	VME	4437.	
ELEM	291	MODESE	289	290	321	320	MATERIAL=	1	ANEAE=	15282	XC,YC,ZC=	70.133	70.133	PRESSE	0.
MX,MY	-2.5	0.	0.	0.	MX,MY	MXY=	99.	14.	84.91	133.1	AE=	6.6	VME	1080.	
TOP SIR	SX,SY,SAYE	374.	374.	374.	SMAX,SMIN,IAUMAXE	297.	SMAX,SMIN,IAUMAXE	2370.	349.	1073.	AE=	7.1	VME	2330.	
MID SIR	SX,SY,SAYE	-374.	-374.	-374.	SMAX,SMIN,IAUMAXE	349.	SMAX,SMIN,IAUMAXE	2250.	-43.	1206.	AE=	7.6	VME	2495.	
BOT SIR	SX,SY,SAYE								-431.	1340.	AE=	7.6	VME	2495.	
ELEM	292	MODESE	290	291	322	321	MATERIAL=	1	ANEAE=	15282	XC,YC,ZC=	70.157	70.157	PRESSE	0.
MX,MY	-2.5	0.	0.	0.	MX,MY	MXY=	-8.	2.	31.22	157.0	AE=	58.1	VME	1080.	
TOP SIR	SX,SY,SAYE	-85.	-85.	-85.	SMAX,SMIN,IAUMAXE	373.	SMAX,SMIN,IAUMAXE	148.	416.	416.	AE=	59.1	VME	768.	
MID SIR	SX,SY,SAYE	68.	68.	68.	SMAX,SMIN,IAUMAXE	381.	SMAX,SMIN,IAUMAXE	220.	-647.	433.	AE=	60.1	VME	797.	
BOT SIR	SX,SY,SAYE								-647.	451.	AE=	60.1	VME	797.	
ELEM	293	MODESE	291	292	323	322	MATERIAL=	1	ANEAE=	15282	XC,YC,ZC=	70.157	70.157	PRESSE	0.
MX,MY	-2.5	0.	0.	0.	MX,MY	MXY=	-98.	-21.	-31.22	157.0	AE=	80.5	VME	3251.	
TOP SIR	SX,SY,SAYE	-1154.	-1154.	-1154.	SMAX,SMIN,IAUMAXE	417.	SMAX,SMIN,IAUMAXE	-1085.	-3655.	1285.	AE=	84.1	VME	3255.	
MID SIR	SX,SY,SAYE	1124.	1124.	1124.	SMAX,SMIN,IAUMAXE	337.	SMAX,SMIN,IAUMAXE	1146.	-3244.	1631.	AE=	86.3	VME	3562.	
BOT SIR	SX,SY,SAYE								-2848.	1997.	AE=	86.3	VME	3562.	
ELEM	294	MODESE	292	293	311	323	MATERIAL=	1	ANEAE=	11504	XC,YC,ZC=	70.133	70.133	PRESSE	0.
MX,MY	-2.5	0.	0.	0.	MX,MY	MXY=	-408.	-16.	-69.23	133.4	AE=	85.8	VME	1050.	
TOP SIR	SX,SY,SAYE	-2975.	-2975.	-2975.	SMAX,SMIN,IAUMAXE	395.	SMAX,SMIN,IAUMAXE	-500.	-5715.	1842.	AE=	85.9	VME	5017.	
MID SIR	SX,SY,SAYE	1027.	1027.	1027.	SMAX,SMIN,IAUMAXE	311.	SMAX,SMIN,IAUMAXE	1040.	-5146.	2323.	AE=	87.2	VME	4915.	
BOT SIR	SX,SY,SAYE								-4584.	2812.	AE=	87.2	VME	5163.	
ELEM	295	MODESE	293	294	311	311	MATERIAL=	1	ANEAE=	12949	XC,YC,ZC=	70.112	70.112	PRESSE	0.
MX,MY	-2.5	0.	0.	0.	MX,MY	MXY=	-55.	8.	-119.6	112.0	AE=	89.6	VME	1020.	
TOP SIR	SX,SY,SAYE	-671.	-671.	-671.	SMAX,SMIN,IAUMAXE	37.	SMAX,SMIN,IAUMAXE	36.	-8094.	2722.	AE=	89.6	VME	5901.	
MID SIR	SX,SY,SAYE	721.	721.	721.	SMAX,SMIN,IAUMAXE	98.	SMAX,SMIN,IAUMAXE	723.	-5883.	2959.	AE=	89.1	VME	5065.	
BOT SIR	SX,SY,SAYE								-5871.	3197.	AE=	89.1	VME	5065.	

ELEM 296	NUMSE = 312	311	294	MATERIAL = 1	AREAE = 2556.7	110P, IBOI = 70	70	70	PRESSE = 0	63
MAX, MY	-20	-133	-29	SMAX, SMIN, TAUMAX	-3	XC, YC, ZC = -129.6	99.46	1020		
TOP SIK	SK, SY, SX	-16	-6076	SMAX, SMIN, TAUMAX	-3	XC, YC, ZC = -6077	2720	8	5785	
MID SIK	SK, SY, SX	-126	-5967	SMAX, SMIN, TAUMAX	-3	XC, YC, ZC = -125	2921	7	5905	
BOT SIK	SK, SY, SX	385	-5657	SMAX, SMIN, TAUMAX	-3	XC, YC, ZC = 365	3121	6	6059	
ELEM 297	NUMSE = 294	295	312	MATERIAL = 1	AREAE = 1294.9	110P, IBOI = 70	70	70	PRESSE = 0	63
MAX, MY	11	-14	-7	SMAX, SMIN, TAUMAX	22	XC, YC, ZC = -139.1	3015	8	6016	
TOP SIK	SK, SY, SX	25	-6004	SMAX, SMIN, TAUMAX	22	XC, YC, ZC = -6004	3015	8	6016	
MID SIK	SK, SY, SX	80	-5978	SMAX, SMIN, TAUMAX	22	XC, YC, ZC = 82	3031	0	6021	
BOT SIK	SK, SY, SX	135	-5952	SMAX, SMIN, TAUMAX	22	XC, YC, ZC = 180	3049	2	6030	
ELEM 298	NUMSE = 295	296	312	MATERIAL = 1	AREAE = 1294.9	110P, IBOI = 70	70	70	PRESSE = 0	63
MAX, MY	10	-132	-7	SMAX, SMIN, TAUMAX	24	XC, YC, ZC = -144.5	3211	3	6161	
TOP SIK	SK, SY, SX	50	-5861	SMAX, SMIN, TAUMAX	24	XC, YC, ZC = 581	3220	3	6174	
MID SIK	SK, SY, SX	153	-5812	SMAX, SMIN, TAUMAX	24	XC, YC, ZC = 58	3234	3	6185	
BOT SIK	SK, SY, SX	453	-6112	SMAX, SMIN, TAUMAX	24	XC, YC, ZC = -449	2634	5	5945	
ELEM 299	NUMSE = 313	312	296	MATERIAL = 1	AREAE = 2556.7	110P, IBOI = 70	70	70	PRESSE = 0	63
MAX, MY	13	408	12	SMAX, SMIN, TAUMAX	12	XC, YC, ZC = -150.9	3670	5	6677	
TOP SIK	SK, SY, SX	1621	-5706	SMAX, SMIN, TAUMAX	12	XC, YC, ZC = 1626	3686	5	6703	
MID SIK	SK, SY, SX	56	-6069	SMAX, SMIN, TAUMAX	12	XC, YC, ZC = 60	3686	8	6703	
BOT SIK	SK, SY, SX	1509	-6429	SMAX, SMIN, TAUMAX	12	XC, YC, ZC = -1507	2462	8	5826	
ELEM 300	NUMSE = 296	297	313	MATERIAL = 1	AREAE = 1294.9	110P, IBOI = 70	70	70	PRESSE = 0	63
MAX, MY	14	247	60	SMAX, SMIN, TAUMAX	27	XC, YC, ZC = -156.8	3472	5	6496	
TOP SIK	SK, SY, SX	1013	-6020	SMAX, SMIN, TAUMAX	27	XC, YC, ZC = 1014	3472	5	6496	
MID SIK	SK, SY, SX	61	-6159	SMAX, SMIN, TAUMAX	27	XC, YC, ZC = 61	3472	5	6496	
BOT SIK	SK, SY, SX	887	-6389	SMAX, SMIN, TAUMAX	27	XC, YC, ZC = -638	2755	5	6000	
ELEM 301	NUMSE = 297	298	313	MATERIAL = 1	AREAE = 1294.9	110P, IBOI = 70	70	70	PRESSE = 0	63
MAX, MY	14	332	15	SMAX, SMIN, TAUMAX	23	XC, YC, ZC = -163.7	3514	3	6483	
TOP SIK	SK, SY, SX	1310	-5727	SMAX, SMIN, TAUMAX	23	XC, YC, ZC = 1311	3514	3	6483	
MID SIK	SK, SY, SX	17	-6167	SMAX, SMIN, TAUMAX	23	XC, YC, ZC = 17	3102	3	6185	
BOT SIK	SK, SY, SX	1237	-6606	SMAX, SMIN, TAUMAX	23	XC, YC, ZC = -1235	2687	3	6087	
ELEM 302	NUMSE = 314	315	298	MATERIAL = 1	AREAE = 2556.7	110P, IBOI = 70	70	70	PRESSE = 0	63
MAX, MY	1	556	139	SMAX, SMIN, TAUMAX	-10	XC, YC, ZC = -162.0	3981	4	7119	
TOP SIK	SK, SY, SX	200	-5749	SMAX, SMIN, TAUMAX	-10	XC, YC, ZC = 2206	3981	4	7119	
MID SIK	SK, SY, SX	63	-6243	SMAX, SMIN, TAUMAX	-10	XC, YC, ZC = 63	3184	4	6332	
BOT SIK	SK, SY, SX	2074	-6818	SMAX, SMIN, TAUMAX	-10	XC, YC, ZC = -2054	2391	4	6077	
ELEM 303	NUMSE = 298	299	314	MATERIAL = 1	AREAE = 1294.9	110P, IBOI = 70	70	70	PRESSE = 0	63
MAX, MY	1	283	43	SMAX, SMIN, TAUMAX	7	XC, YC, ZC = -163.7	3699	3	6846	
TOP SIK	SK, SY, SX	147	-6250	SMAX, SMIN, TAUMAX	7	XC, YC, ZC = 147	3239	3	6948	
MID SIK	SK, SY, SX	62	-6519	SMAX, SMIN, TAUMAX	7	XC, YC, ZC = 62	2780	3	6135	
BOT SIK	SK, SY, SX	1024	-6583	SMAX, SMIN, TAUMAX	7	XC, YC, ZC = -1024	2780	3	6135	
ELEM 304	NUMSE = 299	300	314	MATERIAL = 1	AREAE = 1294.9	110P, IBOI = 70	70	70	PRESSE = 0	63
MAX, MY	1	203	43	SMAX, SMIN, TAUMAX	-7	XC, YC, ZC = -163.7	3694	3	6846	
TOP SIK	SK, SY, SX	147	-6250	SMAX, SMIN, TAUMAX	-7	XC, YC, ZC = 147	3239	3	6948	
MID SIK	SK, SY, SX	62	-6419	SMAX, SMIN, TAUMAX	-7	XC, YC, ZC = 62	2780	3	6135	
BOT SIK	SK, SY, SX	1024	-6583	SMAX, SMIN, TAUMAX	-7	XC, YC, ZC = -1024	2780	3	6135	
ELEM 305	NUMSE = 315	314	300	MATERIAL = 1	AREAE = 2556.7	110P, IBOI = 70	70	70	PRESSE = 0	63
MAX, MY	3	556	139	SMAX, SMIN, TAUMAX	10	XC, YC, ZC = -162.0	3981	3	7119	
TOP SIK	SK, SY, SX	200	-5749	SMAX, SMIN, TAUMAX	10	XC, YC, ZC = 2206	3981	3	7119	
MID SIK	SK, SY, SX	93	-6283	SMAX, SMIN, TAUMAX	10	XC, YC, ZC = 93	3184	3	6332	
BOT SIK	SK, SY, SX	2074	-6818	SMAX, SMIN, TAUMAX	10	XC, YC, ZC = -2054	2391	3	6077	
ELEM 306	NUMSE = 500	501	315	MATERIAL = 1	AREAE = 1294.9	110P, IBOI = 70	70	70	PRESSE = 0	63
MAX, MY	4	332	15	SMAX, SMIN, TAUMAX	-23	XC, YC, ZC = -150.5	3519	3	6483	
TOP SIK	SK, SY, SX	5	-5726	SMAX, SMIN, TAUMAX	-23	XC, YC, ZC = 5	3519	3	6483	
MID SIK	SK, SY, SX	37	-6167	SMAX, SMIN, TAUMAX	-23	XC, YC, ZC = 37	3102	3	6185	
BOT SIK	SK, SY, SX	1237	-6606	SMAX, SMIN, TAUMAX	-23	XC, YC, ZC = -1235	2687	3	6087	

63	ELEM 307	MUDELSE	301	302	315	315	MATERIAL=	1	AREAE	1294.9	110P,THOT=	70	70	PRESSE	0.
	MX,MY	SA,SY,SX,SZ	1013	5426	65	247	SMAX,SMIN,TAUMAXE	60	SMIN,TAUMAXE	1014	XC,YC,ZC=	-156.8	-47.57	1020	0.
	MID SIR	SA,SY,SX,SZ	887	6139	-148	40	SMAX,SMIN,TAUMAXE	149	SMIN,TAUMAXE	884	XC,YC,ZC=	3472	AE=84.3	6496	6496
	ROT SIR	SA,SY,SX,SZ	-887	-6139	-148	40	SMAX,SMIN,TAUMAXE	149	SMIN,TAUMAXE	884	XC,YC,ZC=	3111	AE=-84.3	6191	6191
												2755	AE=-88.5	6006	6006
63	ELEM 308	MUDELSE	316	315	302	302	MATERIAL=	1	AREAE	2556.7	110P,THOT=	70	70	PRESSE	0.
	MX,MY	SA,SY,SX,SZ	1023	5799	193	406	SMAX,SMIN,TAUMAXE	94	SMIN,TAUMAXE	1256	XC,YC,ZC=	-150.9	-66.52	1020	0.
	MID SIR	SA,SY,SX,SZ	36	6066	-149	149	SMAX,SMIN,TAUMAXE	104	SMIN,TAUMAXE	60	XC,YC,ZC=	3670	AE=88.5	6077	6077
	ROT SIR	SA,SY,SX,SZ	-1509	-6829	104	149	SMAX,SMIN,TAUMAXE	104	SMIN,TAUMAXE	60	XC,YC,ZC=	3066	AE=88.5	6103	6103
												-6431	AE=88.8	5826	5826
63	ELEM 309	MUDELSE	302	303	316	316	MATERIAL=	1	AREAE	1294.9	110P,THOT=	70	70	PRESSE	0.
	MX,MY	SA,SY,SX,SZ	560	560	74	132	SMAX,SMIN,TAUMAXE	33	SMIN,TAUMAXE	561	XC,YC,ZC=	-144.5	-77.20	1020	0.
	MID SIR	SA,SY,SX,SZ	451	5986	-136	136	SMAX,SMIN,TAUMAXE	150	SMIN,TAUMAXE	54	XC,YC,ZC=	3211	AE=89.3	6161	6161
	ROT SIR	SA,SY,SX,SZ	-451	-5986	-136	136	SMAX,SMIN,TAUMAXE	150	SMIN,TAUMAXE	54	XC,YC,ZC=	3020	AE=-89.3	6014	6014
												-6116	AE=-88.5	5905	5905
63	ELEM 310	MUDELSE	303	304	316	316	MATERIAL=	1	AREAE	1294.9	110P,THOT=	70	70	PRESSE	0.
	MX,MY	SA,SY,SX,SZ	11	24	-20	14	SMAX,SMIN,TAUMAXE	-7	SMIN,TAUMAXE	-134.1	XC,YC,ZC=	3015	AE=89.8	6010	6010
	MID SIR	SA,SY,SX,SZ	60	5978	-191	191	SMAX,SMIN,TAUMAXE	101	SMIN,TAUMAXE	82	XC,YC,ZC=	3031	AE=-89.8	6021	6021
	ROT SIR	SA,SY,SX,SZ	135	-5978	-191	191	SMAX,SMIN,TAUMAXE	101	SMIN,TAUMAXE	82	XC,YC,ZC=	3049	AE=-88.2	6030	6030
												-5958	AE=-88.2	6030	6030
63	ELEM 311	MUDELSE	317	316	309	309	MATERIAL=	1	AREAE	2556.7	110P,THOT=	70	70	PRESSE	0.
	MX,MY	SA,SY,SX,SZ	816	816	20	133	SMAX,SMIN,TAUMAXE	29	SMIN,TAUMAXE	817	XC,YC,ZC=	-139.6	-94.46	1020	0.
	MID SIR	SA,SY,SX,SZ	16	5976	-136	136	SMAX,SMIN,TAUMAXE	59	SMIN,TAUMAXE	616	XC,YC,ZC=	2720	AE=89.8	5785	5785
	ROT SIR	SA,SY,SX,SZ	163	-5976	-136	136	SMAX,SMIN,TAUMAXE	59	SMIN,TAUMAXE	616	XC,YC,ZC=	2921	AE=89.7	5905	5905
												-5857	AE=89.6	6059	6059
63	ELEM 312	MUDELSE	304	305	317	317	MATERIAL=	1	AREAE	1294.9	110P,THOT=	70	70	PRESSE	0.
	MX,MY	SA,SY,SX,SZ	811	811	-35	179	SMAX,SMIN,TAUMAXE	-15	SMIN,TAUMAXE	-651	XC,YC,ZC=	6094	AE=89.4	5790	5790
	MID SIR	SA,SY,SX,SZ	651	5982	-67	67	SMAX,SMIN,TAUMAXE	98	SMIN,TAUMAXE	36	XC,YC,ZC=	2722	AE=-89.4	5901	5901
	ROT SIR	SA,SY,SX,SZ	721	-5982	-67	67	SMAX,SMIN,TAUMAXE	98	SMIN,TAUMAXE	36	XC,YC,ZC=	2959	AE=-89.4	5901	5901
												-5671	AE=-89.1	6065	6065
63	ELEM 313	MUDELSE	305	306	320	317	MATERIAL=	1	AREAE	11504	110P,THOT=	70	70	PRESSE	0.
	MX,MY	SA,SY,SX,SZ	115	115	-395	404	SMAX,SMIN,TAUMAXE	143	SMIN,TAUMAXE	16	XC,YC,ZC=	-89.23	-13.4	1050	0.
	MID SIR	SA,SY,SX,SZ	524	524	-331	331	SMAX,SMIN,TAUMAXE	271	SMIN,TAUMAXE	500	XC,YC,ZC=	5146	AE=83.8	5017	5017
	ROT SIR	SA,SY,SX,SZ	1027	-4571	-271	271	SMAX,SMIN,TAUMAXE	271	SMIN,TAUMAXE	1040	XC,YC,ZC=	8323	AE=-83.8	4415	4415
												-4584	AE=-87.2	5185	5185
63	ELEM 314	MUDELSE	306	307	330	329	MATERIAL=	1	AREAE	15282	110P,THOT=	70	70	PRESSE	0.
	MX,MY	SA,SY,SX,SZ	0	0	-417	297	SMAX,SMIN,TAUMAXE	-98	SMIN,TAUMAXE	21	XC,YC,ZC=	-51.22	-157.0	1080	0.
	MID SIR	SA,SY,SX,SZ	1124	1124	-329	329	SMAX,SMIN,TAUMAXE	329	SMIN,TAUMAXE	1085	XC,YC,ZC=	1655	AE=80.5	3251	3251
	ROT SIR	SA,SY,SX,SZ	1129	-2832	-256	256	SMAX,SMIN,TAUMAXE	256	SMIN,TAUMAXE	1146	XC,YC,ZC=	1633	AE=-84.1	3255	3255
												-2848	AE=-86.3	3562	3562
63	ELEM 315	MUDELSE	307	308	331	330	MATERIAL=	1	AREAE	15282	110P,THOT=	70	70	PRESSE	0.
	MX,MY	SA,SY,SX,SZ	0	0	-419	20	SMAX,SMIN,TAUMAXE	-8	SMIN,TAUMAXE	-2	XC,YC,ZC=	31.22	-157.0	1080	0.
	MID SIR	SA,SY,SX,SZ	85	85	-419	173	SMAX,SMIN,TAUMAXE	173	SMIN,TAUMAXE	220	XC,YC,ZC=	410	AE=58.1	788	788
	ROT SIR	SA,SY,SX,SZ	68	-386	-190	190	SMAX,SMIN,TAUMAXE	190	SMIN,TAUMAXE	292	XC,YC,ZC=	451	AE=-50.1	797	797
												-848	AE=-50.1	797	797
63	ELEM 316	MUDELSE	308	309	332	331	MATERIAL=	1	AREAE	15282	110P,THOT=	70	70	PRESSE	0.
	MX,MY	SA,SY,SX,SZ	374	374	-297	99	SMAX,SMIN,TAUMAXE	39	SMIN,TAUMAXE	249.2	XC,YC,ZC=	86.91	-13.1	1080	0.
	MID SIR	SA,SY,SX,SZ	36	2333	-297	297	SMAX,SMIN,TAUMAXE	36	SMIN,TAUMAXE	2370	XC,YC,ZC=	1073	AE=-7.6	2330	2330
	ROT SIR	SA,SY,SX,SZ	385	-2303	-349	349	SMAX,SMIN,TAUMAXE	349	SMIN,TAUMAXE	2250	XC,YC,ZC=	1206	AE=-7.6	2392	2392
												-431	AE=-7.6	2493	2493
63	ELEM 317	MUDELSE	309	310	333	332	MATERIAL=	1	AREAE	15282	110P,THOT=	70	70	PRESSE	0.
	MX,MY	SA,SY,SX,SZ	556	556	-138	93	SMAX,SMIN,TAUMAXE	37	SMIN,TAUMAXE	451.6	XC,YC,ZC=	351	AE=-1.9	1060	1060
	MID SIR	SA,SY,SX,SZ	4312	4312	-159	159	SMAX,SMIN,TAUMAXE	159	SMIN,TAUMAXE	4279	XC,YC,ZC=	5084	AE=-1.4	4353	4353
	ROT SIR	SA,SY,SX,SZ	431	-4230	-223	223	SMAX,SMIN,TAUMAXE	223	SMIN,TAUMAXE	4241	XC,YC,ZC=	369	AE=-2.6	4437	4437



ELEM 318	MATERIAL= 1	AREA= 15282.	XC, YC, ZC=	157.0	70.	PRESSE= 0.
MX, MY, MXY=	31	-9.		-31.22	70.	1080.
TOP SIR	SMAX, SMIN, TAUMAXE	5539.	5421.	2648.	AE= -5	5422.
MID SIR	SMAX, SMIN, TAUMAXE	5420.	5421.	-1.	AE= -8	5422.
BOT SIR	SMAX, SMIN, TAUMAXE	5302.	5303.	-245.	AE= -8	5429.
ELEM 319	MATERIAL= 1	AREA= 3778.3	XC, YC, ZC=	2110	70.	PRESSE= 0.
MX, MY, MXY=	-80	-101.7		123.5	70.	1140.
TOP SIR	SMAX, SMIN, TAUMAXE	5623.	5624.	2648.	AE= -87.1	5077.
MID SIR	SMAX, SMIN, TAUMAXE	555.	5527.	2642.	AE= -87.5	5305.
BOT SIR	SMAX, SMIN, TAUMAXE	1514.	1326.	3173.	AE= -87.5	5798.
ELEM 320	MATERIAL= 1	AREA= 2556.1	XC, YC, ZC=	106.3	70.	PRESSE= 0.
MX, MY, MXY=	97	-243		106.3	70.	1140.
TOP SIR	SMAX, SMIN, TAUMAXE	5105.	5106.	2572.	AE= 89.1	5269.
MID SIR	SMAX, SMIN, TAUMAXE	482.	488.	2684.	AE= 89.1	5324.
BOT SIR	SMAX, SMIN, TAUMAXE	409.	423.	2800.	AE= 87.1	5401.
ELEM 321	MATERIAL= 1	AREA= 3780.6	XC, YC, ZC=	160.9	70.	PRESSE= 0.
MX, MY, MXY=	11	-121.2		160.9	70.	1140.
TOP SIR	SMAX, SMIN, TAUMAXE	420.	421.	2999.	AE= -86.0	5901.
MID SIR	SMAX, SMIN, TAUMAXE	3811.	3835.	5912.	AE= -86.7	5429.
BOT SIR	SMAX, SMIN, TAUMAXE	3652.	3871.	2825.	AE= -86.7	5764.
ELEM 322	MATERIAL= 1	AREA= 2555.8	XC, YC, ZC=	149.3	70.	PRESSE= 0.
MX, MY, MXY=	106	-149.3		149.3	70.	1140.
TOP SIR	SMAX, SMIN, TAUMAXE	1123.	1123.	3580.	AE= 89.1	6275.
MID SIR	SMAX, SMIN, TAUMAXE	119.	119.	3058.	AE= 85.4	6083.
BOT SIR	SMAX, SMIN, TAUMAXE	699.	698.	2738.	AE= 87.6	6029.
ELEM 323	MATERIAL= 1	AREA= 3780.6	XC, YC, ZC=	156.4	70.	PRESSE= 0.
MX, MY, MXY=	132	-156.4		156.4	70.	1140.
TOP SIR	SMAX, SMIN, TAUMAXE	79.	79.	3749.	AE= -89.4	6774.
MID SIR	SMAX, SMIN, TAUMAXE	6189.	6185.	3086.	AE= 89.1	6159.
BOT SIR	SMAX, SMIN, TAUMAXE	6693.	6695.	2384.	AE= -86.8	5970.
ELEM 324	MATERIAL= 1	AREA= 2556.1	XC, YC, ZC=	160.9	70.	PRESSE= 0.
MX, MY, MXY=	140	-160.9		160.9	70.	1140.
TOP SIR	SMAX, SMIN, TAUMAXE	269.	268.	5809.	AE= -85.0	6941.
MID SIR	SMAX, SMIN, TAUMAXE	233.	233.	3255.	AE= 87.9	6505.
BOT SIR	SMAX, SMIN, TAUMAXE	196.	196.	2700.	AE= 87.9	6378.
ELEM 325	MATERIAL= 1	AREA= 3780.6	XC, YC, ZC=	160.9	70.	PRESSE= 0.
MX, MY, MXY=	140	-160.9		160.9	70.	1140.
TOP SIR	SMAX, SMIN, TAUMAXE	269.	268.	5809.	AE= -85.0	6941.
MID SIR	SMAX, SMIN, TAUMAXE	233.	233.	3255.	AE= 87.9	6505.
BOT SIR	SMAX, SMIN, TAUMAXE	196.	196.	2700.	AE= 87.9	6378.
ELEM 326	MATERIAL= 1	AREA= 3780.6	XC, YC, ZC=	160.9	70.	PRESSE= 0.
MX, MY, MXY=	140	-160.9		160.9	70.	1140.
TOP SIR	SMAX, SMIN, TAUMAXE	269.	268.	5809.	AE= -85.0	6941.
MID SIR	SMAX, SMIN, TAUMAXE	233.	233.	3255.	AE= 87.9	6505.
BOT SIR	SMAX, SMIN, TAUMAXE	196.	196.	2700.	AE= 87.9	6378.
ELEM 327	MATERIAL= 1	AREA= 2555.8	XC, YC, ZC=	149.3	70.	PRESSE= 0.
MX, MY, MXY=	106	-149.3		149.3	70.	1140.
TOP SIR	SMAX, SMIN, TAUMAXE	1123.	1123.	3380.	AE= -89.1	6275.
MID SIR	SMAX, SMIN, TAUMAXE	169.	168.	3058.	AE= -85.4	6083.
BOT SIR	SMAX, SMIN, TAUMAXE	6452.	6461.	2738.	AE= -87.6	6029.
ELEM 328	MATERIAL= 1	AREA= 3780.6	XC, YC, ZC=	136.5	70.	PRESSE= 0.
MX, MY, MXY=	11	-121.2		136.5	70.	1140.
TOP SIR	SMAX, SMIN, TAUMAXE	420.	421.	2999.	AE= 86.0	5901.
MID SIR	SMAX, SMIN, TAUMAXE	373.	373.	5811.	AE= 86.3	5429.
BOT SIR	SMAX, SMIN, TAUMAXE	325.	325.	2825.	AE= 86.7	5764.

ELEM 329	NUDESE 310	317	328	MATEJIAL=	1 29	AHEA=	2550.1	70.3	70.3	PMESSE=	U.
TOP SIK	SK,SY,SKYE	-244.	MX,MY,MYE	-85.	SMAX,SMIN,TAUMAXE	-27.	XC, YC, ZC=	-121.2	-100.3	1140.	5269.
MID SIK	SK,SY,SKYE	82.	-5274.	-77.	SMAX,SMIN,TAUMAXE	88.	88.	2572.	A=-89.1	VME	5131.
BOT SIK	SK,SY,SKYE	409.	-5162.	-180.	SMAX,SMIN,TAUMAXE	423.	423.	2664.	A=-87.1	VME	5401.
ELEM 330	NUDESE 329	328	337	MATEJIAL=	1 80	AHEA=	3778.3	70.5	70.5	PMESSE=	U.
TOP SIK	SK,SY,SKYE	-1423.	MX,MY,MYE	-156.	SMAX,SMIN,TAUMAXE	-1413.	XC, YC, ZC=	-161.7	-123.5	1180.	5077.
MID SIK	SK,SY,SKYE	-55.	-5316.	211.	SMAX,SMIN,TAUMAXE	-44.	-44.	2110.	A= 87.1	VME	5305.
BOT SIK	SK,SY,SKYE	1314.	-5009.	272.	SMAX,SMIN,TAUMAXE	1526.	1526.	3173.	A= 87.5	VME	5744.
ELEM 331	NUDESE 318	319	335	MATEJIAL=	1 35	AHEA=	14945.	70.53	70.53	PMESSE=	U.
TOP SIK	SK,SY,SKYE	431.	MX,MY,MYE	111.	SMAX,SMIN,TAUMAXE	5269.	XC, YC, ZC=	153.5	153.5	1320.	6004.
MID SIK	SK,SY,SKYE	6.	5133.	47.	SMAX,SMIN,TAUMAXE	4998.	4998.	419.	A= 7	VME	5131.
BOT SIK	SK,SY,SKYE	-419.	-4997.	67.	SMAX,SMIN,TAUMAXE	4998.	4998.	5709.	A= 7	VME	5221.
ELEM 332	NUDESE 319	320	336	MATEJIAL=	1 40	AHEA=	14945.	70.95	70.95	PMESSE=	U.
TOP SIK	SK,SY,SKYE	0.	MX,MY,MYE	121.	SMAX,SMIN,TAUMAXE	13.	XC, YC, ZC=	130.1	86.95	1320.	4014.
MID SIK	SK,SY,SKYE	465.	4210.	153.	SMAX,SMIN,TAUMAXE	4224.	4224.	459.	A= 2.3	VME	4074.
BOT SIK	SK,SY,SKYE	-465.	3912.	250.	SMAX,SMIN,TAUMAXE	3927.	3927.	-8.	A= 3.3	VME	4165.
ELEM 333	NUDESE 320	321	337	MATEJIAL=	1 22	AHEA=	14945.	70.130.1	70.130.1	PMESSE=	U.
TOP SIK	SK,SY,SKYE	242.	MX,MY,MYE	64.	SMAX,SMIN,TAUMAXE	2167.	XC, YC, ZC=	200.	993.	1320.	2094.
MID SIK	SK,SY,SKYE	-5.	2059.	284.	SMAX,SMIN,TAUMAXE	2108.	2108.	-54.	A= 8.3	VME	4074.
BOT SIK	SK,SY,SKYE	-252.	1973.	323.	SMAX,SMIN,TAUMAXE	2030.	2030.	-309.	A= 9.0	VME	2200.
ELEM 334	NUDESE 321	322	338	MATEJIAL=	1 39	AHEA=	14945.	70.153.53	70.153.53	PMESSE=	U.
TOP SIK	SK,SY,SKYE	-513.	MX,MY,MYE	492.	SMAX,SMIN,TAUMAXE	-77.	XC, YC, ZC=	145.53	403.	1320.	688.
MID SIK	SK,SY,SKYE	-6.	-432.	376.	SMAX,SMIN,TAUMAXE	213.	213.	-652.	A= 59.8	VME	781.
BOT SIK	SK,SY,SKYE	-501.	-282.	351.	SMAX,SMIN,TAUMAXE	636.	636.	-416.	A= 69.1	VME	918.
ELEM 335	NUDESE 322	323	339	MATEJIAL=	1 19	AHEA=	14945.	70.153.53	70.153.53	PMESSE=	U.
TOP SIK	SK,SY,SKYE	-1275.	MX,MY,MYE	-318.	SMAX,SMIN,TAUMAXE	-19.	XC, YC, ZC=	-30.53	927.	1320.	6652.
MID SIK	SK,SY,SKYE	1323.	-2598.	364.	SMAX,SMIN,TAUMAXE	1201.	1201.	-3055.	A= 78.5	VME	6072.
BOT SIK	SK,SY,SKYE	-1323.	-2208.	221.	SMAX,SMIN,TAUMAXE	1337.	1337.	-2222.	A= 86.4	VME	3114.
ELEM 336	NUDESE 323	324	340	MATEJIAL=	1 106	AHEA=	14945.	70.150.1	70.150.1	PMESSE=	U.
TOP SIK	SK,SY,SKYE	-1065.	MX,MY,MYE	-297.	SMAX,SMIN,TAUMAXE	-4.	XC, YC, ZC=	-86.95	1840.	1320.	4303.
MID SIK	SK,SY,SKYE	1218.	-6716.	234.	SMAX,SMIN,TAUMAXE	1050.	1050.	-4731.	A= 87.1	VME	4362.
BOT SIK	SK,SY,SKYE	1218.	-3898.	260.	SMAX,SMIN,TAUMAXE	1226.	1226.	-3607.	A= 87.7	VME	4643.
ELEM 337	NUDESE 324	325	341	MATEJIAL=	1 9	AHEA=	14945.	70.80.95	70.80.95	PMESSE=	U.
TOP SIK	SK,SY,SKYE	147.	MX,MY,MYE	46.	SMAX,SMIN,TAUMAXE	17.	XC, YC, ZC=	-130.1	2433.	1320.	5743.
MID SIK	SK,SY,SKYE	-30.	-5717.	100.	SMAX,SMIN,TAUMAXE	-28.	28.	-5754.	A= 89.0	VME	5740.
BOT SIK	SK,SY,SKYE	-207.	-5787.	164.	SMAX,SMIN,TAUMAXE	-202.	-202.	-5792.	A= 88.5	VME	5643.
ELEM 338	NUDESE 325	326	342	MATEJIAL=	1 19	AHEA=	14945.	70.153.53	70.153.53	PMESSE=	U.
TOP SIK	SK,SY,SKYE	1577.	MX,MY,MYE	-19.	SMAX,SMIN,TAUMAXE	1549.	XC, YC, ZC=	1549.	3778.	1320.	6913.
MID SIK	SK,SY,SKYE	-1703.	-6507.	47.	SMAX,SMIN,TAUMAXE	-1703.	-1703.	-6507.	A= 89.9	VME	6469.
BOT SIK	SK,SY,SKYE	-1703.	-7006.	47.	SMAX,SMIN,TAUMAXE	-1703.	-1703.	-7007.	A= 89.5	VME	6350.
ELEM 339	NUDESE 326	327	343	MATEJIAL=	1 30	AHEA=	14945.	70.153.53	70.153.53	PMESSE=	U.
TOP SIK	SK,SY,SKYE	1546.	MX,MY,MYE	423.	SMAX,SMIN,TAUMAXE	-7.	XC, YC, ZC=	-153.5	3778.	1320.	6913.
MID SIK	SK,SY,SKYE	-1777.	-6507.	19.	SMAX,SMIN,TAUMAXE	1549.	1549.	-6507.	A= 89.8	VME	6469.
BOT SIK	SK,SY,SKYE	-1703.	-7006.	-47.	SMAX,SMIN,TAUMAXE	-1703.	-1703.	-7007.	A= 89.5	VME	6350.

SWANSON ANALYSIS SYSTEMS, INC. ENGINEERING ANALYSIS SYSTEM UPTON BLVD 2 BIRMINGHAM ALA 35202

JAN 1 1972 PHONE (412) 751-1940

LOAD COMBINATION, DL + 9.2PSF F TO W LATERAL WIND CP = 101.819 5/11/77 RP = 0.000

ITERATION = 1

LOAD STEP = 1

U. TIME =

\*\*\*\*\* REAL REACTION FORCES \*\*\*\*\*

NODE - REACTION FORCES ARE IN THE NODAL COORDINATE SYSTEM

NO.	NODE	DIRECTION	VALUE
1	1	FX	42673.02
2	1	FY	42637.03
3	2	FX	30753.00
4	2	FY	13632.03
5	3	FX	-18550.01
6	3	FY	-34096.07
7	4	FX	-34238.09
8	4	FY	-18550.01
9	5	FX	-10784.03
10	5	FY	-7659.05
11	6	FX	-4034.06
12	6	FY	191695
13	7	FX	2415.24
14	7	FY	191695
15	8	FX	4034.06
16	8	FY	-7659.05
17	9	FX	-10784.03
18	9	FY	-14554.04
19	10	FX	-34238.09
20	10	FY	-34096.07
21	11	FX	-9850.01
22	11	FY	13632.03
23	12	FX	10784.03
24	12	FY	-7659.05
25	13	FX	-89152E-04
26	13	FY	80152E-04
27	14	FX	16019.07
28	14	FY	16019.07
29	15	FX	31036.07
30	15	FY	31036.07
31	16	FX	-16724.11
32	16	FY	-7458.11
33	17	FX	-2561.42
34	17	FY	-20081.5
35	18	FX	-19470.02
36	18	FY	-9961.00
37	19	FX	-87555E-03
38	19	FY	9961.00
39	20	FX	19470.02
40	20	FY	20081.5
41	21	FX	2561.42
42	21	FY	16724.11
43	22	FX	7458.11
44	22	FY	31036.07
45	23	FX	20081.5
46	23	FY	19470.02
47	24	FX	9961.00
48	24	FY	87555E-03
49	25	FX	-38699.0
50	25	FY	-55841.0
51	26	FX	-189361.5
52	26	FY	-66462.5

54	ZZ	07	2015463.
55	ZZ	09	200077.
56	ZZ	11	407023.
57	ZZ	13	326140.
58	ZZ	15	220952.
59	ZZ	17	97274.1
60	ZZ	19	12021.0
61	ZZ	21	54800.1
62	ZZ	23	72071.1
63	ZZ	25	97274.1
64	ZZ	27	200652.
65	ZZ	29	369140.
66	ZZ	31	707077.
67	ZZ	33	503463.5
68	ZZ	35	66462.5
69	ZZ	37	889361.
70	ZZ	39	1006699.
71	ZZ	41	300417.
72	ZZ	43	800658E-05
73	MX	45	2199630
74	MX	47	14.1031
75	MX	49	140.235
76	MX	51	458.775
77	MX	53	244.600
78	MX	55	237.032
79	MX	57	30.032
80	MX	59	0.016
81	MX	61	65.7475
82	MX	63	770.000
83	MX	65	100.920E-03
84	MX	67	110.607
85	MX	69	59.7872
86	MX	71	68.0150
87	MX	73	92.6012
88	MX	75	93.0220
89	MX	77	3376.500
90	MX	79	244.600
91	MX	81	458.775
92	MX	83	140.235
93	MX	85	14.1031
94	MX	87	14.1031
95	MX	89	14.1031
96	MX	91	14.1031
97	MX	93	14.1031
98	MX	95	14.1031
99	MX	97	14.1031
100	MX	99	14.1031
101	MX	01	3651.29
102	MX	03	1885.47
103	MX	05	402.142
104	MX	07	8065.12
105	MX	09	2637.9
106	MX	11	50673.3
107	MX	13	10128.1
108	MX	15	10348.8
109	MX	17	664.86
110	MX	19	2338.29
111	MX	21	8339.85
112	MX	23	2338.29
113	MX	25	7084.87
114	MX	27	10548.3
115	MX	29	14150.3
116	MX	31	20673.9
117	MX	33	32319.4
118	MX	35	8085.12
119	MX	37	1485.47
120	MX	39	3051.29

140	MZ	4176.21
141	MZ	-423.3318
142	MZ	-123.524
143	MZ	-418.080
144	MZ	-30.0589
145	MZ	7243.63
146	MZ	477.486
147	MZ	92.0836
148	MZ	-683.440
149	MZ	-133.422
150	MZ	-476.362
151	MZ	-432.641-05
152	MZ	-474.383
153	MZ	13.4990
154	MZ	853.679
155	MZ	-92.8725
156	MZ	-477.450
157	MZ	-7243.62
158	MZ	-8526.72
159	MZ	30.0589
140	MZ	416.081
141	MZ	123.524
142	MZ	42.3318

STEP 1 ITERATION 1 COMPLETE TIME= 0. NOISE 1 AVERAGE 0

\*\*\*\*\* PROBLEM COMPLETED \*\*\*\*\* CP = 101.686 PP = 0.000

\*\*\*\*\* RUN COMPLETED \*\*\*\*\* CP = 101.687 PP = 0.000



