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**Effects of inert gas jet on the transition from deflagration to detonation in  
a stoichiometric methane-oxygen mixture**

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## Abstract

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24 Detonation is an energetic combustion mode augmenting high flow momentum and thermodynamic  
25 efficiency, it has been applied in detonation engines, such as pulse detonation engines (PDEs) and  
26 rotating detonation engines (RDEs), they have become potential aerospace propulsion equipment.  
27 Recently, fluidic jet-in-cross flow (JICF) has been demonstrated experimentally and numerically  
28 that can accelerate the deflagration-to-detonation transition (DDT) process. Nonetheless, most of  
29 previous studies focused on the jets using combustible mixture or oxygen, which may bring  
30 additional risk for turbulence-generated system in detonation engines. In this study, a more safe and  
31 controllable inert gas (i.e., Ar) is applied for JICF, experiments are carried out to investigate effects  
32 of argon jet as an enhancement method on promoting the DDT in a stoichiometric methane-oxygen  
33 mixture. The effects of local argon concentration, turbulence intensity and injection position on the  
34 DDT process are systematically examined. Two-dimensional numerical simulations are also  
35 performed to elucidate the details of the injection evolution. The experimental results show that  
36 turbulence generated by the argon injection can promote flame acceleration and the onset of  
37 detonation only in the fast deflagration regime. The enhancing effect is more prominent at higher  
38 turbulence intensity by increasing jet injection pressure and shorter injection time. Too long  
39 injection duration increases argon local concentration that leads to an adverse effect prohibiting the  
40 DDT occurrence. During the initial laminar flame acceleration, referred to as the slow deflagration  
41 regime, no enhancement by the argon jet on DDT can be observed. By looking numerically at the  
42 flow structure of the argon jet, the vortical features enhance the transport and mixing between  
43 reactants and products. The interaction between the reactive travelling wave and the jet structure  
44 further induces turbulence and thus accelerates the chemical reaction rate. With the time elapsed,  
45 the injected argon entrains largely and dilutes the ambient combustible mixture, and restrains the

46 DDT. Furthermore, a novel dimensionless criterion and a characteristic parameter  $Tur_c$  are proposed,  
47 quantitatively analyzing the dominate mechanism in flame propagation and the initial stage of DDT  
48 as inert jet is introduced.

49 **Keywords:** Jet flow; Inert gas; Turbulence; Detonation; DDT

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## 51 **1. Introduction**

52 In recent years, there is an increasing interest in developing detonation-based engines, such as  
53 Pulsed or Rotating Detonation Engines (PDEs or RDEs) for hypersonic propulsion applications  
54 [1-3]. One of the major challenges in designing these propulsive systems is the capability to initiate  
55 a detonation in a chamber of limited size [4-6], wherein the detonation propagation limits are a key  
56 as well as the fundamental problem for maintaining the propagation of detonations without failure  
57 to sustain their propulsion thrust [7-9], this topic has been widely investigated as the detonations  
58 propagate through obstacles in recent years by Zhang et al. [10-12], Cao et al. [13] and Gao et al.  
59 [14-16].

60 A detonation can be initiated by direct or indirect ways [17-19]. Direct initiation refers to an  
61 instantaneous detonation formation using a large energy deposition into the explosion [20, 21]. Such  
62 initiation method is impractical for engineering applications in most industrial settings.  
63 Alternatively, a detonation can be formed by indirect way, referring to the  
64 deflagration-to-detonation transition (DDT) [22, 23]. It requires a weak ignition source and DDT is  
65 achieved after different stages of flame acceleration from slow burning to high speed turbulent  
66 deflagration, and eventually the detonation onset for various physical mechanisms involved [24-26].  
67 In smooth tubes, the distance from the ignition to the onset of detonation, named as the DDT run-up  
68 distance, can be very long. A wealth of experimental and numerical studies thus focus on finding  
69 most effective flame acceleration configurations to achieve DDT in short run-up distance. Since the  
70 pioneer work of Shchelkin [27], turbulence is known to play a significant role on the flame  
71 acceleration. Hence, over the past decades, the use of repeated obstacles to generate turbulence in  
72 the path of flame propagation to promote DDT has been extensively researched [28-30].

73 Although physical obstacles can prominently promote DDT, but they also induce significant

74 pressures losses in the process. Cooper *et al.* [31] pointed out that the obstacles can reduce the  
75 impulse of a single-cycle PDE by up to 25%. Recently, Knox *et al.* [32] introduced a fluidic jet  
76 obstacle as an alternative to conventional DDT enhancement devices for turbulence generation.  
77 They compared the relative performance of fluidic and physical obstacles on DDT, and found that  
78 both the intense turbulent mixing characteristics inherent of a high-velocity jet and the blockage  
79 created by the virtual obstacle can significantly facilitate flame acceleration and transition to  
80 detonation.

81 To systematically investigate turbulence-induced DDT, Chambers & Ahmed [33] focused  
82 experimentally on the flame acceleration regime in a highly turbulent environment. They looked at  
83 experimentally and classified characteristics of turbulent flame dynamics and fast flame  
84 propagation modes at various regimes. The critical stage for turbulence driven deflagration to  
85 detonation of fast flames was closely examined. McGarry & Ahmed [34, 35] and later Chambers &  
86 Ahmed [36] and Tarrant *et al.* [37] examined interaction mechanisms of the propagating flame with  
87 turbulence induced by a fluidic jet, specifically analyzing the resulting flame-turbulence interaction  
88 modes and their influence on the flame propagation dynamics. Recently, Peng *et al.* [38]  
89 investigated the effects of fluidic jet-in-crossflow (JICF) on flame acceleration and DDT. They used  
90 a reactive transverse CH<sub>4</sub>-O<sub>2</sub> mixture as jet in crossflow, showing promising enhancement on flame  
91 acceleration.

92 JICF has been proven promising as a practical, efficient method to accelerate flame  
93 propagation and promote DDT with less overall pressure losses [39-41]. It is noteworthy that most  
94 JICF enhancers used combustible gases. Such use may bring more uncontrollable factors and  
95 increase the risk for the turbulence-generated system in detonation-based propulsion devices.  
96 Up-to-date, studies using inert gas for JICF and investigating its effect on flame acceleration and

97 DDT behavior are scarce. Hence, in this work, the effects of inert argon jet with different injection  
 98 conditions on DDT are explored via experiments and numerical simulations. Piezoelectric probes  
 99 are used to measure the time-of-arrival of the leading travelling wave and such velocity results are  
 100 then used to characterize different flame propagation regimes. The unsteady argon jet structures are  
 101 analyzed using two-dimensional numerical simulations. The present results can thus advance  
 102 understanding of the influencing mechanism of the inert gas JICF on flame acceleration and DDT.

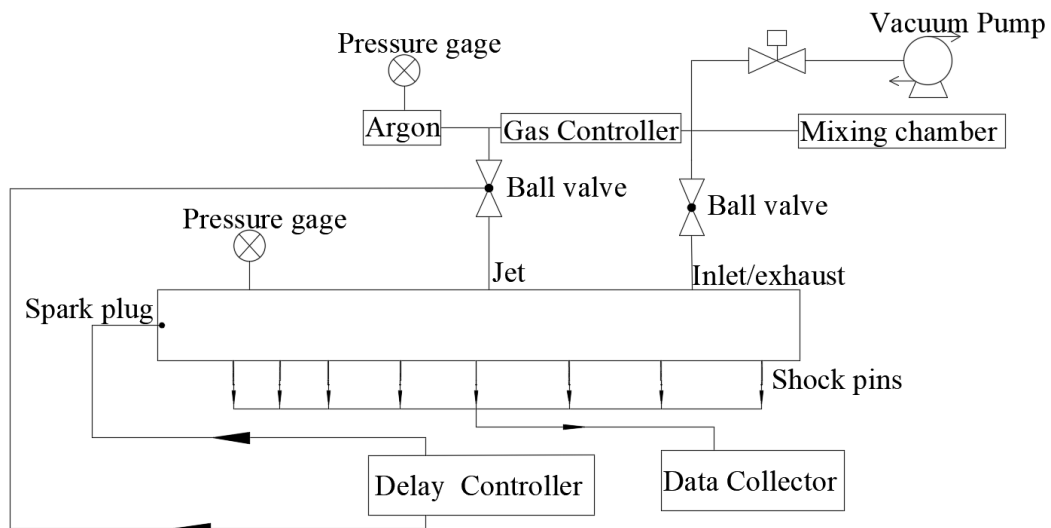
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## 104 2. Experimental Details and Numerical Methodology

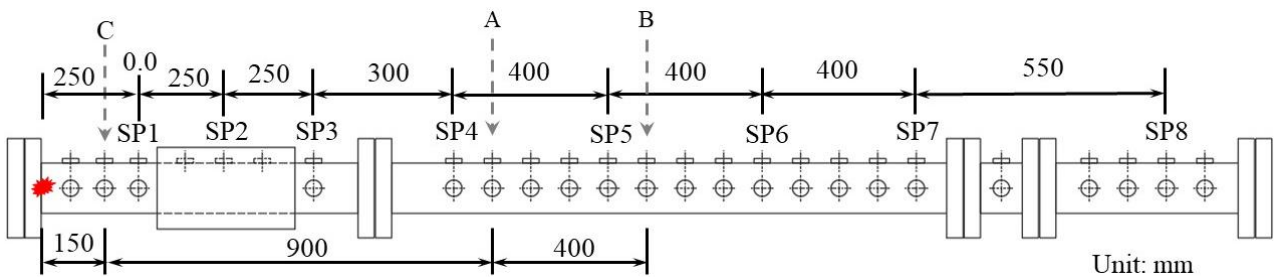
### 105 2.1 Experiments

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(b)

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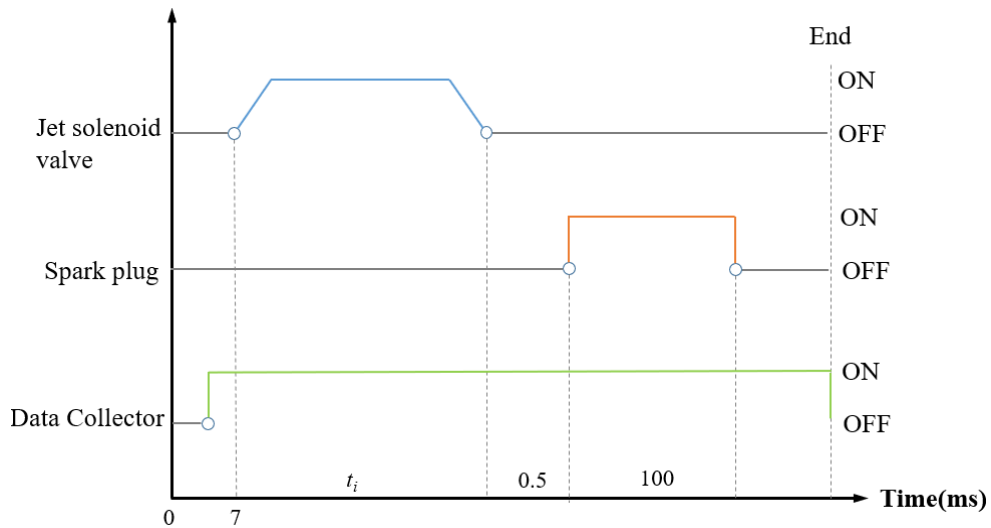
111 **Fig. 1** Sketch of the experimental setup. a) A schematic of the experiment system; and b) shock pins and  
 112 jet positions (A, B and C).

113

114 Figure 1a shows a schematic of the experimental facility which consisted of the fuel/oxygen  
 115 mixing system, argon gas supply, ignition system, delay control system, data acquisition, gas

116 controller and the shock tube, more details can be found in authors' previous literature [42, 43]. The  
 117 methane fuel and oxygen was homogeneously mixed at the stoichiometric ratio for 24 h in a 180-L  
 118 mixing chamber. The spark plug was placed at the center of the left end wall. The delay control  
 119 system is composed of ARM-STM32F103CB development boards and an Ingenex-H3MB-052D  
 120 solid-state relay; the arrows from the delay controller in Fig. 1a denote the signals sent to the  
 121 ignition spark and the solenoid valve.

122 The stainless steel tube has a length of 2950-mm long and a 100 mm×100 mm cross-section. It  
 123 was divided into four parts connected by flanges. Eight Dynasen shock pins (CA-1134) were used  
 124 to capture the time-of-arrival (TOA) of the leading shock wave, and the shock pins arrangement is  
 125 shown as SP1 to SP8 in Fig. 1b. The argon supply system includes a solenoid valve (AM230C), a  
 126 one-way valve and a 2.4-L tank. The solenoid valve response time is 7 ms. Three jet positions (A, B,  
 127 C) were chosen to investigate the effect of jet location on DDT. The distance relative to the position  
 128 of SP1 was 0.9, 1.3 and -0.1 m for A, B and C jet position, respectively. The mixture was ignited by  
 129 a 40-J spark. The mixture initial pressure was monitored by OMEGA digital gauges (PXM00710V,  
 130 0-700 kPa with an accuracy of  $\pm 0.25\%$  full scale).



131 **Fig. 2** Time sequence of the control signals.

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 135 Before each experiment, the tube was evacuated to about 0.1 kPa (absolute pressure), and then

136 filled with the combustible mixture from the mixing chamber, controlled by the gas controller to a  
137 desired initial pressure,  $P_0$ . The 2.4-L storage bottle was filled with argon at a high initial injection  
138 pressure,  $P_i$ . Figure 2 shows the time sequence of the signals sent to the solenoid valve and the  
139 spark igniter. The argon injection was triggered first by opening the solenoid valve for a duration  $t_i$ .  
140 The igniter was then fired after 0.5 ms. The whole process was recorded by the data acquisition  
141 system.

142 In this study, the initial mixture pressure  $P_0$  was kept at 25 kPa and the initial temperature 300 K.  
143 Various injection pressures and durations were used to investigate the influence of the generated  
144 turbulence characteristics and local argon concentration on DDT. Four groups of experiments were  
145 designed to examine effects of each individual factor.

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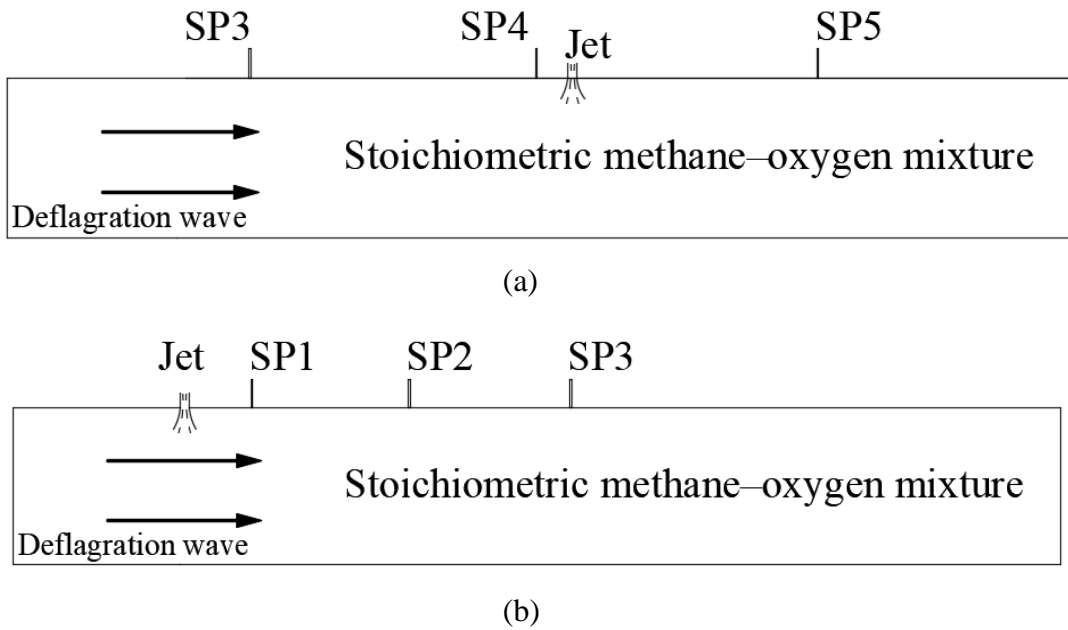
## 147 **2.2 Numerical simulations**

148 As the turbulence structure induced by the jet had fully developed before the ignition was  
149 triggered, therefore a series of large eddy simulations were carried to investigate the turbulence  
150 characteristic under different injection conditions. The process of Ar injected into the tube was  
151 simulated by two dimensional (2D) Navier-Stokes equations. A Roe Riemann solver was utilized to  
152 construct numerical upwind fluxes, and the Minmod limiter with MUSCL reconstruction was  
153 applied to construct a third-order method in space. The time integration was advanced using a  
154 fourth-order explicit Runge–Kutta algorithm. Two different injection positions were chosen to  
155 compare the turbulence structure evolution while Ar was being injected into the tube, two  
156 schematics are shown in Fig. 3. To improve the efficiency of calculation, the length of the domain  
157 was reduced to 1.5 m and 0.8 m respectively for diagram (a) and (b). The left and right boundary  
158 conditions were set as pressure far-field, the upper and bottom sides were set as adiabatic walls. The  
159 jet velocity was given by an average value (approximately 100 m/s) in the process of injection,  
160 based on the calculation of injection pressure ( $P_j$ ) was 200 kPa. The initial pressure  $P_0$  was 25 kPa  
161 and the temperature  $T_0$  was 300 K. In order to keep consistent with the experimental conditions, the



162 jet was shut down for 0.5 ms when the injection time  $t_i$  was 15 ms, 25 ms, 50 ms, 75 ms and 100 ms.  
 163 The mesh was set adaptively refined with the gradient of the concentration of Ar. The base size of  
 164 the mesh was 0.5 mm, and the minimum mesh was refined to 0.0625 mm.

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**Fig. 3** A schematic of the simulation setup with different jet positions.

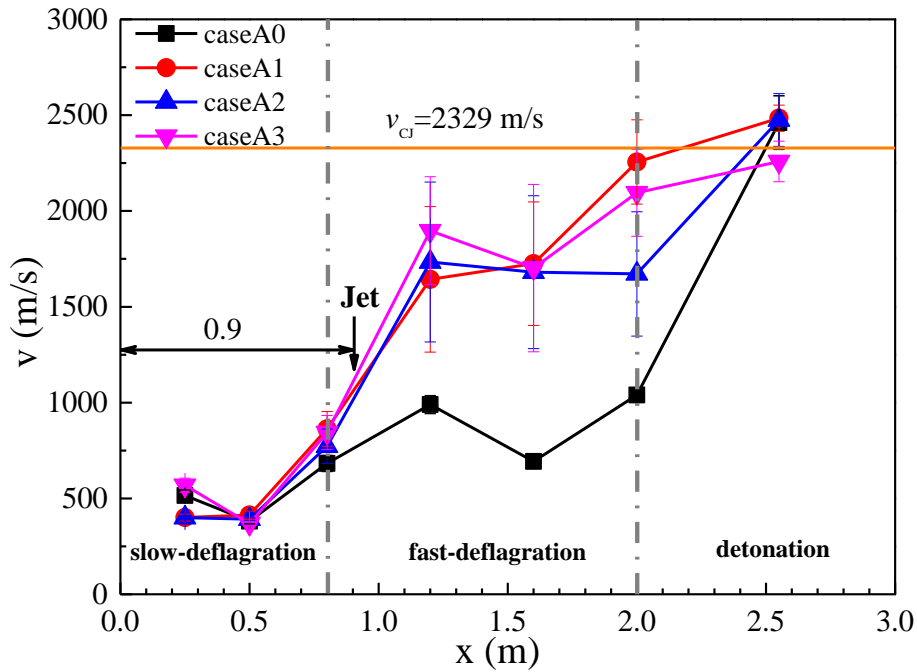
### 3. Results and Discussion

#### 3.1 The effect of jet injection pressure

To distinguish the impact of  $P_i$  on DDT, four experimental cases are designed (Table 1). Case A0 represents the flame propagation in the tube without argon jet and used as the baseline case. Cases A1 to A3 consider different injection pressure but at a fixed injection duration and position.

**Table 1** Injection parameters for experiments with varying injection pressure.

Case #	$P_i$ /kPa	$t_i$ /ms	Position
A0	0	0	N/A
A1	100	15	A
A2	150	15	A
A3	200	15	A



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181 **Fig. 4** Velocity of the lead wave with different argon injection pressure  $P_i$ .  
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183 Figure 4 shows the leading wave velocity behavior for the cases without (case A0) and with  
184 argon injection (cases A1 to A3). The data points correspond the shock pins location, starting with  
185 SP2 as the first point. The y-axis is the average velocity calculated between the adjacent shock pins.  
186 From the baseline case A0 velocity results, one can deduce correspondingly the wave propagation  
187 dynamics into three stages. The first is related to the slow deflagration burning from the ignition  
188 point to the location of SP4 ( $x = 0.8$  m), and the wave velocity remains approximately 500 m/s.  
189 When the flame accelerates, the leading shock strengthens to almost a velocity about 1000 m/s. This  
190 second stage is referred to as the fast deflagration in this study, from SP4 to SP7 ( $x = 2.0$  m) along  
191 the tube. Subsequently, DDT occurs near the end of the tube from SP7 to the SP8 ( $x = 2.55$  m). In  
192 this last stage, the wave propagation velocity reaches the theoretical C-J detonation velocity.  
193 Therefore, under the condition of quiescent state (case A0), a slow deflagration is first ignited by the  
194 spark, and then a turbulent deflagration accelerates to half the CJ velocity ( $\sim 1000$  m/s), which is the  
195 typical condition observed in DDT prior to the onset of detonation [4].

196 For cases A1 to A3, a vertical argon jet is introduced at a location near SP4 prior to the arrival  
197 of the reactive wave complex and the results are also given in Fig. 4. The velocity behavior shows

198 noticeably a significant difference compared with the baseline case A0. The argon jet leads to an  
199 abrupt wave acceleration. The results show that, although  $P_i$  is different for each case, all starts to  
200 accelerate abruptly from  $x = 0.8$  m to an average velocity about 1860 m/s and an earlier DDT. The  
201 error bars shown on the plot represent the variations between repeated shots (more than 5) with the  
202 same initial condition. With the argon jet, the flow field experiences different levels of turbulence  
203 perturbation, causing a larger velocity variation between shots in the fast deflagration regime  
204 perturbed by the argon jet.

205       When the injection forms certain turbulent structure in the tube, it changes the local  
206 combustible mixture concentration near the jet. Therefore, it is necessary to define how dynamically  
207 the injection accelerates the wave propagation. From the fluid dynamic point-of-view, the  
208 turbulence effect can be divided into two simple physics problems [44]. One is the flame interaction  
209 with the vortex introduced by the injection due to the Kelvin-Helmholtz (K-H) instability, the other  
210 is the Richtmyer–Meshkov (R-M) instability when the lead shock passed through the mixture with  
211 density gradient. Both turbulence generation mechanisms may contribute to the wave acceleration.  
212 It is noteworthy that as an inert gas, the injected argon can have a negative effect on the combustion  
213 process due to the dilution and reduce the reactivity of the mixture and hence, slow down the flame  
214 propagation. Nonetheless, Fig. 4 demonstrates that under the specific conditions given in Table 1,  
215 the turbulence mechanism dominates, providing a positive effect on the flame acceleration.  
216 However, a competitive effect exists between the level of argon concentration and turbulence  
217 generation effect by the argon jet.

218

### 219 **3.2 The effect of local argon concentration**

220       The second set of experiments were performed to look at the argon concentration effect on the  
221 wave acceleration and DDT. Experiments with a fixed  $P_i$  but variable  $t_i$  at location A were  
222 considered (Table 2). The case B0 is the baseline without any argon injection. Note that the local  
223 argon concentration in the tube increases as  $t_i$  increases. However, as  $P_i$  is kept constant, the

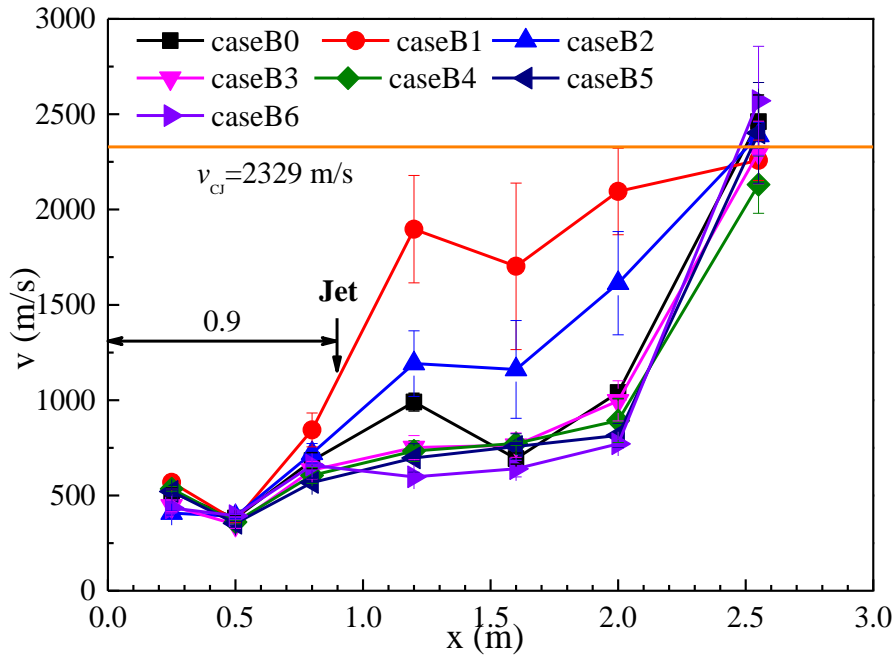
224 injection velocity is almost the same for the cases B1 to B6, which ensures the turbulence effect due  
 225 to jet injection remains at the same level.

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**Table 2** Injection parameters for experiments with varying local argon concentration.

Case #	$P_i$ /kPa	$t_i$ /ms	Position
B0	0	0	N/A
B1	200	15	A
B2	200	25	A
B3	200	50	A
B4	200	100	A
B5	200	200	A
B6	200	400	A

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**Fig. 5** Velocity of the lead wave with different time duration of argon jet injection,  $t_i$ .

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Figure 5 shows the wave propagation velocity for different cases. Compared with the baseline case B0, the case with an injection  $t_i = 25$  ms (Case B2) still shows a noticeable wave acceleration in the fast deflagration regime. However, increasing further the injection time (case B3), the velocity behavior approaches to the baseline trend without demonstrating anymore acceleration effect. For cases B4 to B6, the argon jet appears to have an adverse effect weakening the wave propagation. Hence, argon injection with a short injection duration (e.g.,  $t_i = 15$  ms, case B1) has an

239 enhancement effect on the wave propagation. On the contrary, by introducing high pressure argon  
 240 jet with long injection time, the local argon concentration therewith increases and as a result,  
 241 prohibiting the flame acceleration.

### 242 3.3 The effect of turbulence intensity

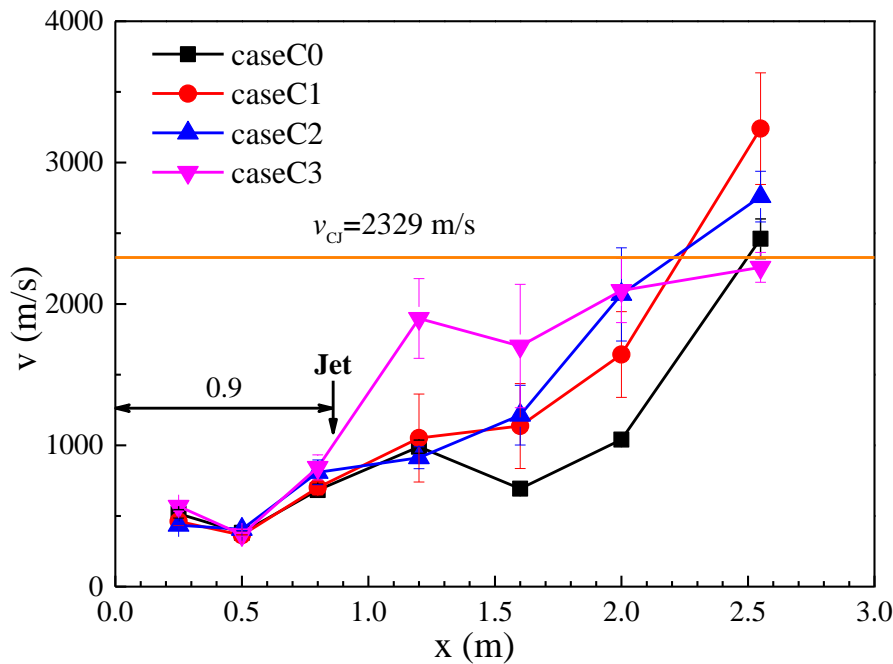
243 The experimental conditions in Table 3 are designed to examine the turbulence intensity effect.  
 244  $P_i$  and  $t_i$  are set as variables to ensure the volume of argon injected into the tube remains the same.  
 245 To minimize the influence of local concentration, the amount of argon injected into the tube is  
 246 maintained with a small value in these experiments.

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**Table 3** Injection parameters for experiments with varying turbulence intensities.

Case #	$P_i$ /kPa	$t_i$ /ms	Position
C0	0	0	N/A
C1	100	50	A
C2	150	21	A
C3	200	15	A

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**Fig. 6** Velocity of the leading wave with different  $P_i$  and  $t_i$  for varying turbulence intensity.

253 Figure 6 shows the wave propagation velocity with different turbulence intensities. The jet  
 254 location is also highlighted. The propagation velocities of cases C1 to C3 accelerate apparently by

255 comparing with the baseline case C0. In the slow deflagration region where the flow does not  
 256 experience turbulence generated by the jet, the propagation velocities of all four cases are very close,  
 257 again about ~500 m/s. After the injection at position A,  $x = 0.8$  m, the velocity for case C3  
 258 accelerates rapidly, and this acceleration performance is better than cases C1 and C2. Wave  
 259 acceleration can only be seen from SP6 ( $x = 1.6$  m) for cases C1 and C2, reaching a velocity of  
 260 1570 m/s and 2045 m/s, respectively, at the position of SP7. By comparing with the baseline case,  
 261 the results demonstrate again that higher turbulence intensity enhances the acceleration of the flame  
 262 propagation. Therefore, when the concentration of the argon gas is at a low level, the turbulence  
 263 dominates the wave propagation and DDT process.

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### 265 3.4 The effect of injection position

266 Three jet positions are chosen to examine the injection position effect on DDT and wave  
 267 propagation behavior. The distance between position C and SP1 is 10 cm (here 0 in the  $x$ -axis  
 268 represents the first shock pin location, hence  $x = -0.1$  m for position C). The other details regarding  
 269 the jet positions are shown in Fig. 2. The jet parameters of this group are shown in Table 4.

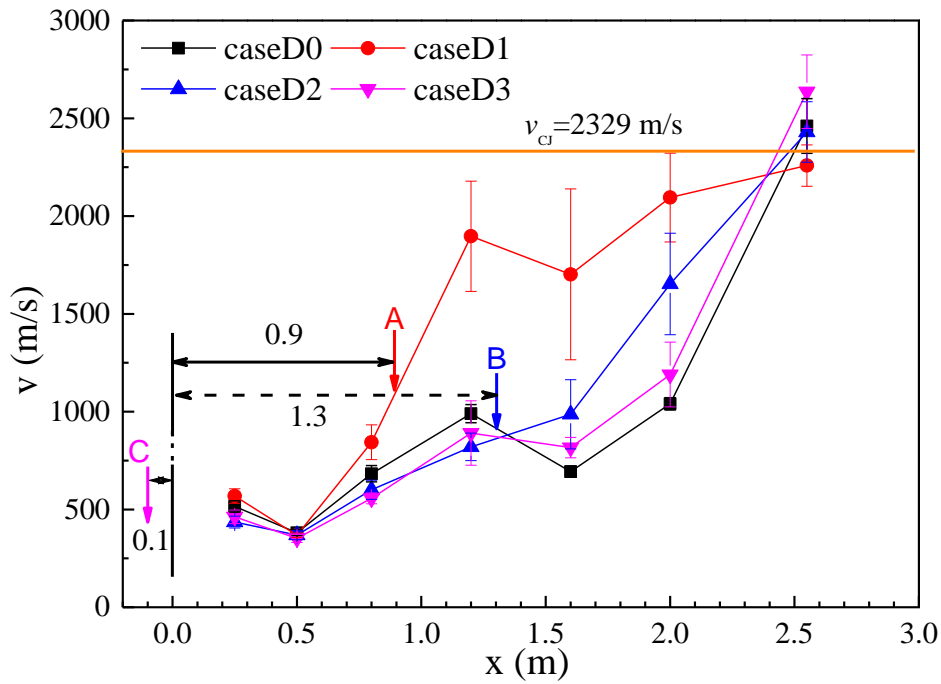
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**Table 4** Injection parameters for experiments with varying injection location.

Case #	$P_i$ /kPa	$t_i$ /ms	Position	$x/m$
D0	0	0	N/A	N/A
D1	200	15	A	0.9
D2	200	15	B	1.3
D3	200	15	C	-0.1

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274 **Fig. 7** Velocity of the leading wave with different injection location.

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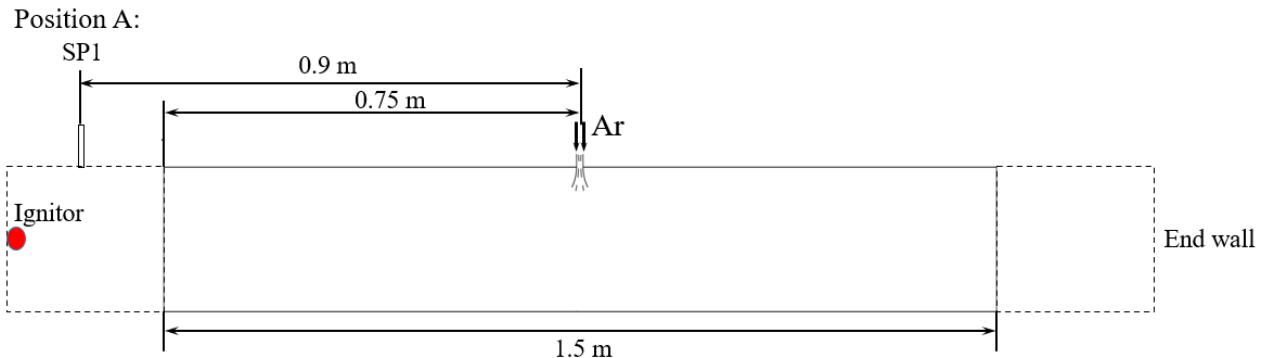
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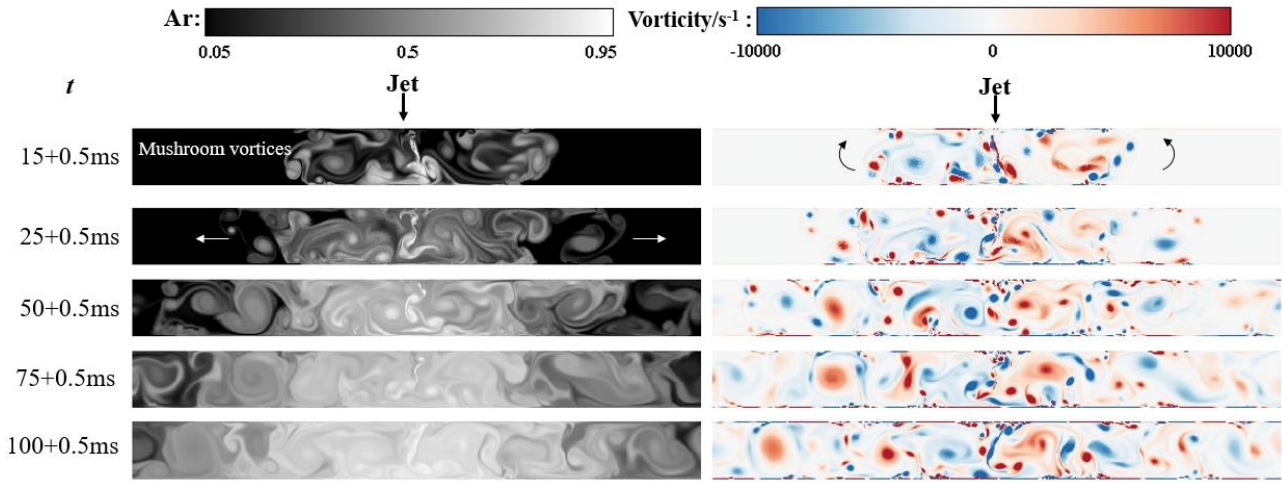
Figure 7 shows the wave velocity behavior during DDT with the argon jet placed at different positions. For the case D1, an enhanced acceleration of the propagation velocity is clearly observed. For the injection position located at the middle of the tube (case D2), only a small enhancement is seen on the wave acceleration. At this position B, the baseline case already shows the wave propagating at a high velocity about 700 m/s supported possibly by a high level of turbulence originated from the evolving flame acceleration process. The added turbulence by the argon jet thus has less prominent effect on such wave condition. For the last case D3 where the jet is introduced in the slow deflagration stage, there is also no significant influence on the wave propagation.

### 3.5 Effect of the turbulence evolution on the wave acceleration



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(a)



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(b)

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**Fig. 8** The evolution of the distribution of Ar and the vortices formed at position A

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The two-dimensional turbulence flow structures generated by argon jet with two different positions are simulated. Fig. 8(b) shows the distribution of Ar and vorticity at corresponding time to the experiments B1~B4 mentioned in Section 3.2 which placed the jet at position A, as shown in Fig. 8(a). It can be seen that the jet has already collided with the bottom wall when  $t_i$  is 15 ms, forming mushroom vortices. The vortices formed on the both sides of the jet inversely roll up. The vortices formed earlier are pushed away by the following jet due to the increasing mass of Ar. It is obvious that peripheral vortices gradually dissipate during their paths to the upstream/downstream of the jet, which is due to the collisions from vortex-to-vortex and vortex-to-wall. Note that both the mass of Ar and the vorticity increase with the extending of  $t_i$ , hereby a criterion is proposed to evaluate the dominant factor of the concentration and the turbulence on the propagation of flame.

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As Ar was considered as an inert gas suppressing the combustion, while the turbulence structure especially the vortices have a positive influence on flame propagation, both the concentration of Ar and the vorticity have been nondimensionalized. The concentration Ar is nondimensionalized with the initial density of mixture:

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$$C_{Ar} = \int_A \frac{\rho Y_{Ar}}{\rho_0} dA \quad (1)$$



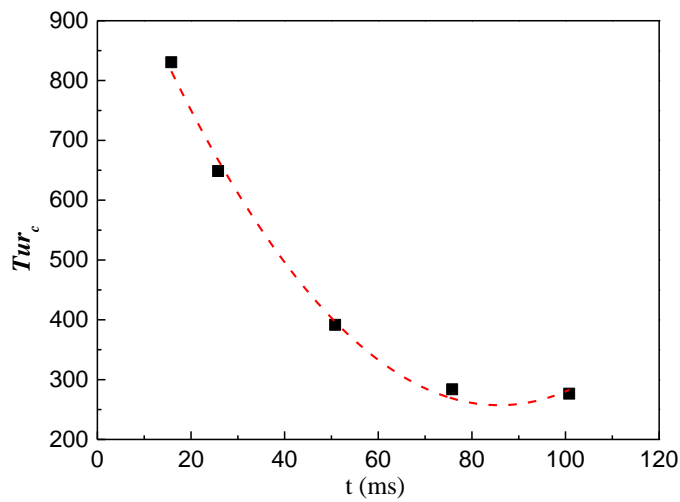
306 Where  $\rho$ ,  $\rho_0$  represent the current mixture density and the initial mixture density respectively;  
 307  $Y_{Ar}$  is the current mass fraction of Ar;  $A$  is the area of the computation domain. The dimensionless  
 308 treatment of vorticity ( $Vor$ ) can be described as follows:

$$309 \quad Vor = \int_A \left| \frac{\partial(v/v_j)}{\partial(x/l)} - \frac{\partial(u/v_j)}{\partial(y/h)} \right| dA \quad (2)$$

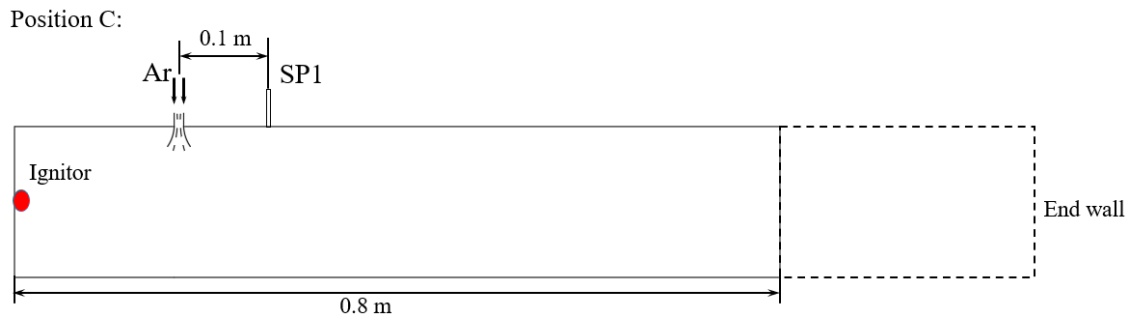
310 Where  $u$ ,  $v$  and  $v_j$  are the  $x$  velocity component,  $y$  velocity component and jet velocity,  
 311 respectively;  $l$  represents the length of the domain, the  $h$  represents the height;  $A$  is the same as the  
 312 Equation 1. Thus, a dimensionless criterion factor can be obtained by calculating the ratio between  
 313 two factors, i.e.,  $Tur_c$ :

$$314 \quad Tur_c = \frac{Vor}{C_{Ar}} \quad (3)$$

315 Fig. 9 shows  $Tur_c$  evolves with the jet duration time. Apparently, the curve drops dramatically  
 316 while jet extending from 15 ms to 50 ms, and it keeps at a relatively stable value from 75 ms to 100  
 317 ms. This is in good agreement with the results observed from experiments B1~B4. The results  
 318 confirm that the positive effect of the turbulence is offset with the development of time, and even be  
 319 oppressed by the combustion inhibition effect due to the increasing mass of Ar, as the jet duration  
 320 time is consistently extended.



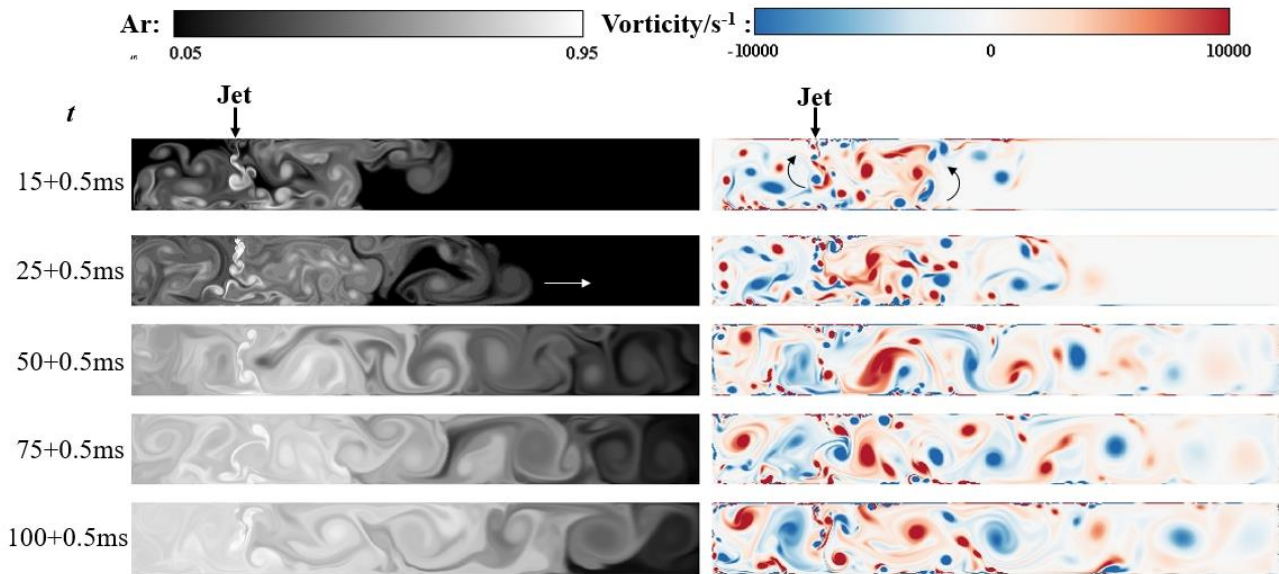
321  
 322 **Fig. 9** The tendency of  $Vor/C$  varied with jet duration time



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(a)



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(b)

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Fig. 10 The evolution of the distribution of Ar and the vortices formed with jet placed at position C

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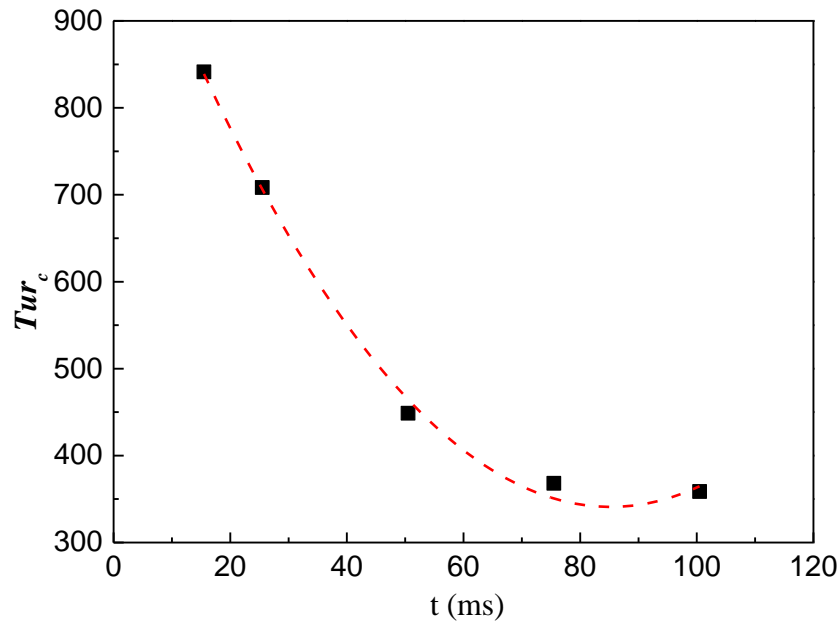
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Fig. 10(b) depicts the Ar distribution and the vortices evolution for the case that with the jet placed at position C (as shown in Fig. 10(a)). The Ar concentration and vortices evolution processes are similar to the case in which the jet placed at position A. The difference between those two positions is that the Ar concentration in position C increases to approximately 1 in the region on the left side of the jet as the jet is placed to the location adjacent to the ignition end wall. This scenario occurred mainly because of the expansion of Ar concentration is constrained by the left wall. However, the Ar injection has a positive effect on the formation of the vortices near the left end wall (clearly shown by the right column of Fig.10(b)). The vortices propagate to the downstream and gradually dissipate, which are similar as the case shown in Fig.8. To verify the competition effect between the negative influence of Ar dilution and the positive influence of the turbulence enhancing

338 the vortices, the dimensionless criterion  $Tur_c$  is then calculated and shown in Fig. 11. It can be seen  
339 from Fig. 11 that, with the increasing of injection time  $t_i$ , the tendency of  $Tur_c$  curve presents a  
340 monotonous decrease, indicating the dominant factor affecting the DDT behavior is the Ar  
341 concentration, rather than the stretch effect of the vortices. It is noteworthy that Ar jet is placed at  
342 position C near the ignitor, and therefore the increased Ar concentration with longer  $t_i$  greatly  
343 inhibits the ignition progress, resulting in a slower flame propagation velocity than the one with the  
344 jet placed at position A.



345  
346 Fig. 11 The tendency of  $Vor/C$  varied with jet duration time for the jet placed at position C

347  
348 By comparing the Ar distribution and vortices evolution of the cases with different jet positions,  
349 it is obvious that as a jet having a shorter distance to the ignitor, resulting more Ar concentrate on  
350 the left side of the jet contacting the flame front primarily. The suppression effect from the  
351 increasing inert gas mass is more prominent than the flame-acceleration effect from vortices. The  
352 dimensionless criterion  $Tur_c$  is verified to be an adequate parameter to estimate the dominate  
353 mechanism in flame propagation and the initial stage of DDT as inert jet is applied for detonation  
354 enhancement.

355

#### 356 **4. Conclusion**

357 With the increasing interest in employing detonation as fast combustion mode for advanced  
358 propulsion systems, controlled initiation and rapid onset of detonation from DDT become desirable.  
359 In this study, we explore the use of an inert argon jet as an enhancement method by its turbulence  
360 generation to promote further flame acceleration and strengthen the lead shock wave for DDT. A  
361 series of experiments and numerical simulations were conducted to investigate effects of various  
362 injection parameters on DDT, including the initial injection pressure, duration time and jet position.  
363 The results show that turbulence generated by argon injection accelerates the wave propagation  
364 velocity when the wave is at the fast deflagration regime with velocity above 500 m/s. At such  
365 condition, the enhanced DDT process by the jet has no noticeable variation with increasing  $P_i$ .  
366 However, either a long injection time or an exceeding injection pressure will increase the local  
367 argon concentration, and prohibit the wave acceleration. The present study has proved that higher  
368 turbulence intensity induced by higher  $P_i$  and shorter  $t_i$  has better performance on enhancing the  
369 DDT. In the experiments, different injection positions also are tested and found have impacted the  
370 DDT process. Only an appropriate injection position can make a positive effect on the DDT process.  
371 Additionally, by analyzing the experiments and numerical simulations of the cases with different jet  
372 positions, the dimensionless criterion  $Tur_c$  proposed has demonstrated the dominant factor of the  
373 flame propagation. The results show that the suppression effect by Ar dilution is more prominent  
374 than the stretch dynamics on the flame acceleration by vortices with the increasing of  $t_i$ .

375

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# Effects of inert gas jet on the transition from deflagration to detonation in a stoichiometric methane-oxygen mixture

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## Declaration of interests

The authors declare that they have no known competing financial interests or personal relationships that could have appeared to influence the work reported in this paper.

The authors declare the following financial interests/personal relationships which may be considered as potential competing interests:

## CRedit authorship contribution statement

Jun Cheng: Investigation, Data collection & analysis , Methodology.

Bo Zhang: Investigation, Validation, Writing - review & editing

Hong Liu: Supervision

Fuxing Wang: Supervision