

1
2
3
4
5
6
7
8
9
10
11
12
13
14
15
16
17
18
19
20
21

Effects of inert gas jet on the transition from deflagration to detonation in a stoichiometric methane-oxygen mixture

Jun Cheng¹, Bo Zhang^{1†}, Hoi Dick Ng², Hong Liu¹, Fuxing Wang¹

¹Shanghai Jiao Tong University
School of Aeronautics and Astronautics, Shanghai, 200240, China

²Concordia University
Department of Mechanical, Industrial and Aerospace Engineering
Montréal, QC, H3G 1M8, Canada

[†]Corresponding Author

E-mail: bozhang@sjtu.edu.cn

Abstract

22
23
24 Detonation is an energetic combustion mode augmenting high flow momentum and thermodynamic
25 efficiency, it has been applied in detonation engines, such as pulse detonation engines (PDEs) and
26 rotating detonation engines (RDEs), they have become potential aerospace propulsion equipment.
27 Recently, fluidic jet-in-cross flow (JICF) has been demonstrated experimentally and numerically
28 that can accelerate the deflagration-to-detonation transition (DDT) process. Nonetheless, most of
29 previous studies focused on the jets using combustible mixture or oxygen, which may bring
30 additional risk for turbulence-generated system in detonation engines. In this study, a more safe and
31 controllable inert gas (i.e., Ar) is applied for JICF, experiments are carried out to investigate effects
32 of argon jet as an enhancement method on promoting the DDT in a stoichiometric methane-oxygen
33 mixture. The effects of local argon concentration, turbulence intensity and injection position on the
34 DDT process are systematically examined. Two-dimensional numerical simulations are also
35 performed to elucidate the details of the injection evolution. The experimental results show that
36 turbulence generated by the argon injection can promote flame acceleration and the onset of
37 detonation only in the fast deflagration regime. The enhancing effect is more prominent at higher
38 turbulence intensity by increasing jet injection pressure and shorter injection time. Too long
39 injection duration increases argon local concentration that leads to an adverse effect prohibiting the
40 DDT occurrence. During the initial laminar flame acceleration, referred to as the slow deflagration
41 regime, no enhancement by the argon jet on DDT can be observed. By looking numerically at the
42 flow structure of the argon jet, the vortical features enhance the transport and mixing between
43 reactants and products. The interaction between the reactive travelling wave and the jet structure
44 further induces turbulence and thus accelerates the chemical reaction rate. With the time elapsed,
45 the injected argon entrains largely and dilutes the ambient combustible mixture, and restrains the

46 DDT. Furthermore, a novel dimensionless criterion and a characteristic parameter Tur_c are proposed,
47 quantitatively analyzing the dominate mechanism in flame propagation and the initial stage of DDT
48 as inert jet is introduced.

49 **Keywords:** Jet flow; Inert gas; Turbulence; Detonation; DDT

50

51 **1. Introduction**

52 In recent years, there is an increasing interest in developing detonation-based engines, such as
53 Pulsed or Rotating Detonation Engines (PDEs or RDEs) for hypersonic propulsion applications
54 [1-3]. One of the major challenges in designing these propulsive systems is the capability to initiate
55 a detonation in a chamber of limited size [4-6], wherein the detonation propagation limits are a key
56 as well as the fundamental problem for maintaining the propagation of detonations without failure
57 to sustain their propulsion thrust [7-9], this topic has been widely investigated as the detonations
58 propagate through obstacles in recent years by Zhang et al. [10-12], Cao et al. [13] and Gao et al.
59 [14-16].

60 A detonation can be initiated by direct or indirect ways [17-19]. Direct initiation refers to an
61 instantaneous detonation formation using a large energy deposition into the explosion [20, 21]. Such
62 initiation method is impractical for engineering applications in most industrial settings.
63 Alternatively, a detonation can be formed by indirect way, referring to the
64 deflagration-to-detonation transition (DDT) [22, 23]. It requires a weak ignition source and DDT is
65 achieved after different stages of flame acceleration from slow burning to high speed turbulent
66 deflagration, and eventually the detonation onset for various physical mechanisms involved [24-26].
67 In smooth tubes, the distance from the ignition to the onset of detonation, named as the DDT run-up
68 distance, can be very long. A wealth of experimental and numerical studies thus focus on finding
69 most effective flame acceleration configurations to achieve DDT in short run-up distance. Since the
70 pioneer work of Shchelkin [27], turbulence is known to play a significant role on the flame
71 acceleration. Hence, over the past decades, the use of repeated obstacles to generate turbulence in
72 the path of flame propagation to promote DDT has been extensively researched [28-30].

73 Although physical obstacles can prominently promote DDT, but they also induce significant

74 pressures losses in the process. Cooper *et al.* [31] pointed out that the obstacles can reduce the
75 impulse of a single-cycle PDE by up to 25%. Recently, Knox *et al.* [32] introduced a fluidic jet
76 obstacle as an alternative to conventional DDT enhancement devices for turbulence generation.
77 They compared the relative performance of fluidic and physical obstacles on DDT, and found that
78 both the intense turbulent mixing characteristics inherent of a high-velocity jet and the blockage
79 created by the virtual obstacle can significantly facilitate flame acceleration and transition to
80 detonation.

81 To systematically investigate turbulence-induced DDT, Chambers & Ahmed [33] focused
82 experimentally on the flame acceleration regime in a highly turbulent environment. They looked at
83 experimentally and classified characteristics of turbulent flame dynamics and fast flame
84 propagation modes at various regimes. The critical stage for turbulence driven deflagration to
85 detonation of fast flames was closely examined. McGarry & Ahmed [34, 35] and later Chambers &
86 Ahmed [36] and Tarrant *et al.* [37] examined interaction mechanisms of the propagating flame with
87 turbulence induced by a fluidic jet, specifically analyzing the resulting flame-turbulence interaction
88 modes and their influence on the flame propagation dynamics. Recently, Peng *et al.* [38]
89 investigated the effects of fluidic jet-in-crossflow (JICF) on flame acceleration and DDT. They used
90 a reactive transverse CH₄-O₂ mixture as jet in crossflow, showing promising enhancement on flame
91 acceleration.

92 JICF has been proven promising as a practical, efficient method to accelerate flame
93 propagation and promote DDT with less overall pressure losses [39-41]. It is noteworthy that most
94 JICF enhancers used combustible gases. Such use may bring more uncontrollable factors and
95 increase the risk for the turbulence-generated system in detonation-based propulsion devices.
96 Up-to-date, studies using inert gas for JICF and investigating its effect on flame acceleration and

97 DDT behavior are scarce. Hence, in this work, the effects of inert argon jet with different injection
 98 conditions on DDT are explored via experiments and numerical simulations. Piezoelectric probes
 99 are used to measure the time-of-arrival of the leading travelling wave and such velocity results are
 100 then used to characterize different flame propagation regimes. The unsteady argon jet structures are
 101 analyzed using two-dimensional numerical simulations. The present results can thus advance
 102 understanding of the influencing mechanism of the inert gas JICF on flame acceleration and DDT.

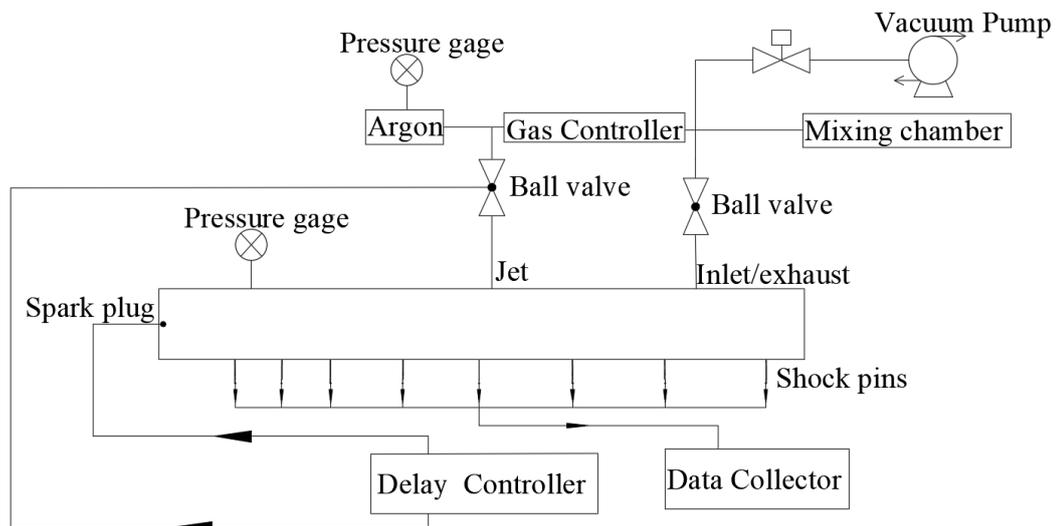
103

104 2. Experimental Details and Numerical Methodology

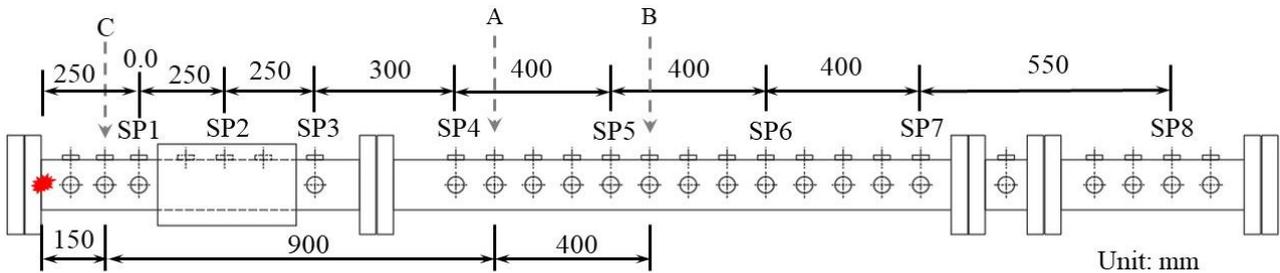
105 2.1 Experiments

106

107



(a)



(b)

108

109

110

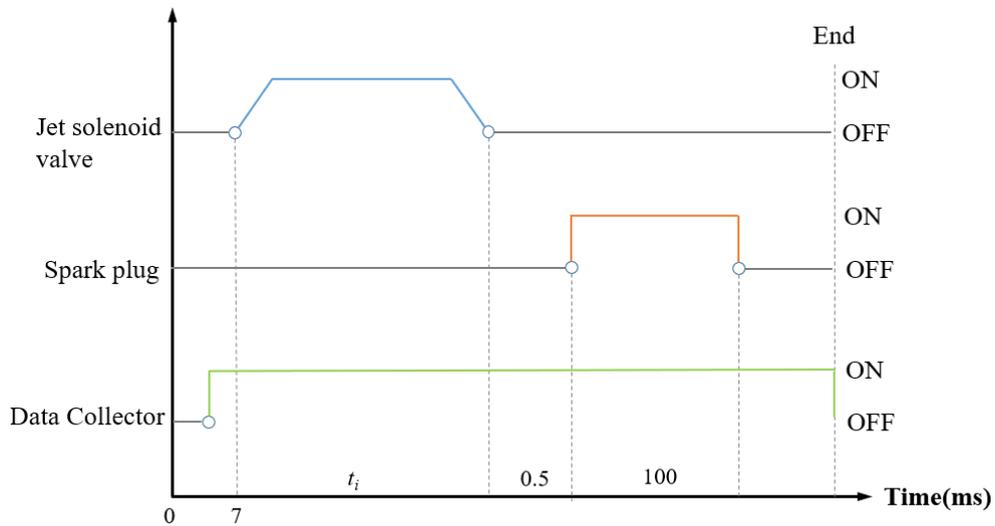
111 **Fig. 1** Sketch of the experimental setup. a) A schematic of the experiment system; and b) shock pins and
 112 jet positions (A, B and C).

113

114 Figure 1a shows a schematic of the experimental facility which consisted of the fuel/oxygen
 115 mixing system, argon gas supply, ignition system, delay control system, data acquisition, gas

116 controller and the shock tube, more details can be found in authors' previous literature [42, 43]. The
 117 methane fuel and oxygen was homogeneously mixed at the stoichiometric ratio for 24 h in a 180-L
 118 mixing chamber. The spark plug was placed at the center of the left end wall. The delay control
 119 system is composed of ARM-STM32F103CB development boards and an Ingenex-H3MB-052D
 120 solid-state relay; the arrows from the delay controller in Fig. 1a denote the signals sent to the
 121 ignition spark and the solenoid valve.

122 The stainless steel tube has a length of 2950-mm long and a 100 mm×100 mm cross-section. It
 123 was divided into four parts connected by flanges. Eight Dynasen shock pins (CA-1134) were used
 124 to capture the time-of-arrival (TOA) of the leading shock wave, and the shock pins arrangement is
 125 shown as SP1 to SP8 in Fig. 1b. The argon supply system includes a solenoid valve (AM230C), a
 126 one-way valve and a 2.4-L tank. The solenoid valve response time is 7 ms. Three jet positions (A, B,
 127 C) were chosen to investigate the effect of jet location on DDT. The distance relative to the position
 128 of SP1 was 0.9, 1.3 and -0.1 m for A, B and C jet position, respectively. The mixture was ignited by
 129 a 40-J spark. The mixture initial pressure was monitored by OMEGA digital gauges (PXM00710V,
 130 0-700 kPa with an accuracy of $\pm 0.25\%$ full scale).



131 **Fig. 2** Time sequence of the control signals.

132
 133
 134
 135 Before each experiment, the tube was evacuated to about 0.1 kPa (absolute pressure), and then

136 filled with the combustible mixture from the mixing chamber, controlled by the gas controller to a
137 desired initial pressure, P_0 . The 2.4-L storage bottle was filled with argon at a high initial injection
138 pressure, P_i . Figure 2 shows the time sequence of the signals sent to the solenoid valve and the
139 spark igniter. The argon injection was triggered first by opening the solenoid valve for a duration t_i .
140 The igniter was then fired after 0.5 ms. The whole process was recorded by the data acquisition
141 system.

142 In this study, the initial mixture pressure P_0 was kept at 25 kPa and the initial temperature 300 K.
143 Various injection pressures and durations were used to investigate the influence of the generated
144 turbulence characteristics and local argon concentration on DDT. Four groups of experiments were
145 designed to examine effects of each individual factor.

146

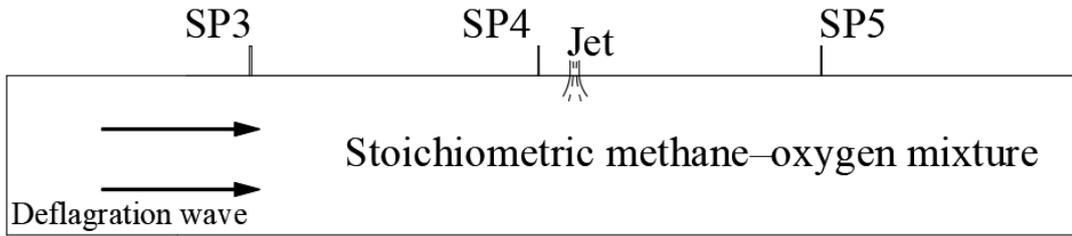
147 **2.2 Numerical simulations**

148 As the turbulence structure induced by the jet had fully developed before the ignition was
149 triggered, therefore a series of large eddy simulations were carried to investigate the turbulence
150 characteristic under different injection conditions. The process of Ar injected into the tube was
151 simulated by two dimensional (2D) Navier-Stokes equations. A Roe Riemann solver was utilized to
152 construct numerical upwind fluxes, and the Minmod limiter with MUSCL reconstruction was
153 applied to construct a third-order method in space. The time integration was advanced using a
154 fourth-order explicit Runge–Kutta algorithm. Two different injection positions were chosen to
155 compare the turbulence structure evolution while Ar was being injected into the tube, two
156 schematics are shown in Fig. 3. To improve the efficiency of calculation, the length of the domain
157 was reduced to 1.5 m and 0.8 m respectively for diagram (a) and (b). The left and right boundary
158 conditions were set as pressure far-field, the upper and bottom sides were set as adiabatic walls. The
159 jet velocity was given by an average value (approximately 100 m/s) in the process of injection,
160 based on the calculation of injection pressure (P_j) was 200 kPa. The initial pressure P_0 was 25 kPa
161 and the temperature T_0 was 300 K. In order to keep consistent with the experimental conditions, the

162 jet was shut down for 0.5 ms when the injection time t_i was 15 ms, 25 ms, 50 ms, 75 ms and 100 ms.
 163 The mesh was set adaptively refined with the gradient of the concentration of Ar. The base size of
 164 the mesh was 0.5 mm, and the minimum mesh was refined to 0.0625 mm.

165

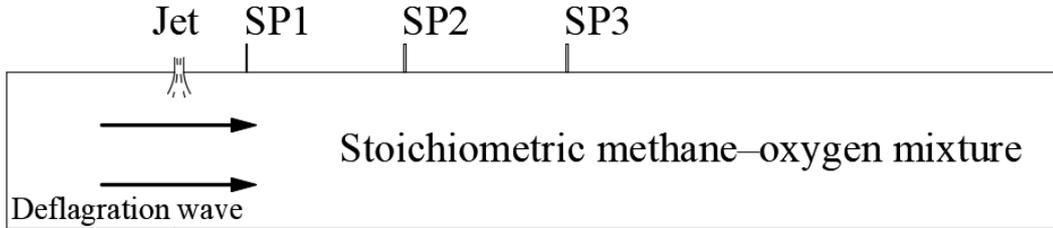
166



167

168

(a)



169

170

(b)

Fig. 3 A schematic of the simulation setup with different jet positions.

171

172

173 3. Results and Discussion

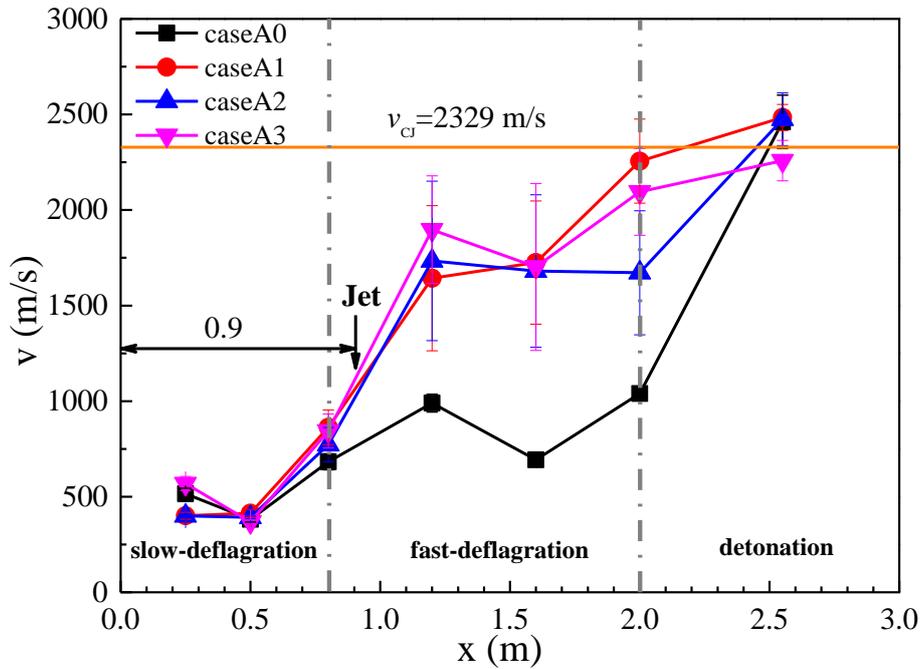
174 3.1 The effect of jet injection pressure

175 To distinguish the impact of P_i on DDT, four experimental cases are designed (Table 1). Case
 176 A0 represents the flame propagation in the tube without argon jet and used as the baseline case.
 177 Cases A1 to A3 consider different injection pressure but at a fixed injection duration and position.

178 **Table 1** Injection parameters for experiments with varying injection pressure.

Case #	P_i /kPa	t_i /ms	Position
A0	0	0	N/A
A1	100	15	A
A2	150	15	A
A3	200	15	A

179



180
181 **Fig. 4** Velocity of the lead wave with different argon injection pressure P_i .
182

183 Figure 4 shows the leading wave velocity behavior for the cases without (case A0) and with
184 argon injection (cases A1 to A3). The data points correspond the shock pins location, starting with
185 SP2 as the first point. The y-axis is the average velocity calculated between the adjacent shock pins.
186 From the baseline case A0 velocity results, one can deduce correspondingly the wave propagation
187 dynamics into three stages. The first is related to the slow deflagration burning from the ignition
188 point to the location of SP4 ($x = 0.8$ m), and the wave velocity remains approximately 500 m/s.
189 When the flame accelerates, the leading shock strengthens to almost a velocity about 1000 m/s. This
190 second stage is referred to as the fast deflagration in this study, from SP4 to SP7 ($x = 2.0$ m) along
191 the tube. Subsequently, DDT occurs near the end of the tube from SP7 to the SP8 ($x = 2.55$ m). In
192 this last stage, the wave propagation velocity reaches the theoretical C-J detonation velocity.
193 Therefore, under the condition of quiescent state (case A0), a slow deflagration is first ignited by the
194 spark, and then a turbulent deflagration accelerates to half the CJ velocity (~ 1000 m/s), which is the
195 typical condition observed in DDT prior to the onset of detonation [4].

196 For cases A1 to A3, a vertical argon jet is introduced at a location near SP4 prior to the arrival
197 of the reactive wave complex and the results are also given in Fig. 4. The velocity behavior shows

198 noticeably a significant difference compared with the baseline case A0. The argon jet leads to an
199 abrupt wave acceleration. The results show that, although P_i is different for each case, all starts to
200 accelerate abruptly from $x = 0.8$ m to an average velocity about 1860 m/s and an earlier DDT. The
201 error bars shown on the plot represent the variations between repeated shots (more than 5) with the
202 same initial condition. With the argon jet, the flow field experiences different levels of turbulence
203 perturbation, causing a larger velocity variation between shots in the fast deflagration regime
204 perturbed by the argon jet.

205 When the injection forms certain turbulent structure in the tube, it changes the local
206 combustible mixture concentration near the jet. Therefore, it is necessary to define how dynamically
207 the injection accelerates the wave propagation. From the fluid dynamic point-of-view, the
208 turbulence effect can be divided into two simple physics problems [44]. One is the flame interaction
209 with the vortex introduced by the injection due to the Kelvin-Helmholtz (K-H) instability, the other
210 is the Richtmyer–Meshkov (R-M) instability when the lead shock passed through the mixture with
211 density gradient. Both turbulence generation mechanisms may contribute to the wave acceleration.
212 It is noteworthy that as an inert gas, the injected argon can have a negative effect on the combustion
213 process due to the dilution and reduce the reactivity of the mixture and hence, slow down the flame
214 propagation. Nonetheless, Fig. 4 demonstrates that under the specific conditions given in Table 1,
215 the turbulence mechanism dominates, providing a positive effect on the flame acceleration.
216 However, a competitive effect exists between the level of argon concentration and turbulence
217 generation effect by the argon jet.

218

219 **3.2 The effect of local argon concentration**

220 The second set of experiments were performed to look at the argon concentration effect on the
221 wave acceleration and DDT. Experiments with a fixed P_i but variable t_i at location A were
222 considered (Table 2). The case B0 is the baseline without any argon injection. Note that the local
223 argon concentration in the tube increases as t_i increases. However, as P_i is kept constant, the

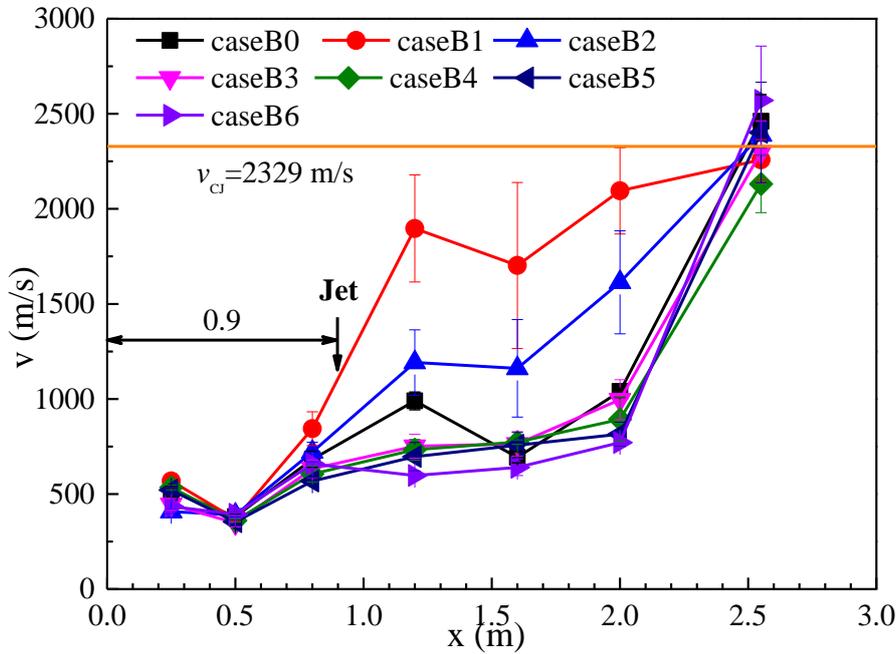
224 injection velocity is almost the same for the cases B1 to B6, which ensures the turbulence effect due
 225 to jet injection remains at the same level.

226
 227

Table 2 Injection parameters for experiments with varying local argon concentration.

Case #	P_i /kPa	t_i /ms	Position
B0	0	0	N/A
B1	200	15	A
B2	200	25	A
B3	200	50	A
B4	200	100	A
B5	200	200	A
B6	200	400	A

228



229

Fig. 5 Velocity of the lead wave with different time duration of argon jet injection, t_i .

230

231

232

233

234

235

236

237

238

Figure 5 shows the wave propagation velocity for different cases. Compared with the baseline case B0, the case with an injection $t_i = 25$ ms (Case B2) still shows a noticeable wave acceleration in the fast deflagration regime. However, increasing further the injection time (case B3), the velocity behavior approaches to the baseline trend without demonstrating anymore acceleration effect. For cases B4 to B6, the argon jet appears to have an adverse effect weakening the wave propagation. Hence, argon injection with a short injection duration (e.g., $t_i = 15$ ms, case B1) has an

239 enhancement effect on the wave propagation. On the contrary, by introducing high pressure argon
 240 jet with long injection time, the local argon concentration therewith increases and as a result,
 241 prohibiting the flame acceleration.

242 3.3 The effect of turbulence intensity

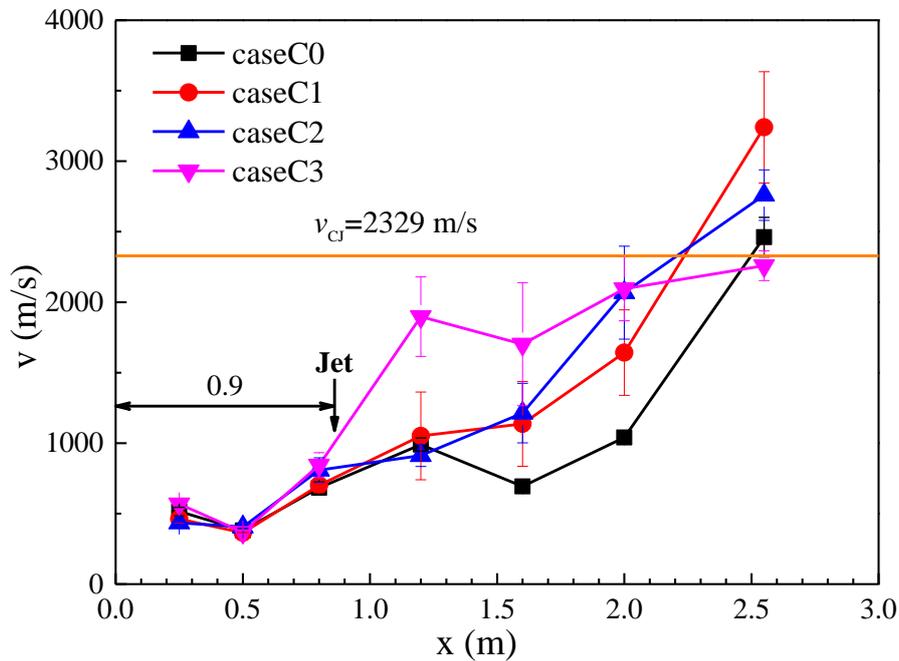
243 The experimental conditions in Table 3 are designed to examine the turbulence intensity effect.
 244 P_i and t_i are set as variables to ensure the volume of argon injected into the tube remains the same.
 245 To minimize the influence of local concentration, the amount of argon injected into the tube is
 246 maintained with a small value in these experiments.

247
 248

Table 3 Injection parameters for experiments with varying turbulence intensities.

Case #	P_i /kPa	t_i /ms	Position
C0	0	0	N/A
C1	100	50	A
C2	150	21	A
C3	200	15	A

249



250
 251
 252

Fig. 6 Velocity of the leading wave with different P_i and t_i for varying turbulence intensity.

253 Figure 6 shows the wave propagation velocity with different turbulence intensities. The jet
 254 location is also highlighted. The propagation velocities of cases C1 to C3 accelerate apparently by

255 comparing with the baseline case C0. In the slow deflagration region where the flow does not
 256 experience turbulence generated by the jet, the propagation velocities of all four cases are very close,
 257 again about ~500 m/s. After the injection at position A, $x = 0.8$ m, the velocity for case C3
 258 accelerates rapidly, and this acceleration performance is better than cases C1 and C2. Wave
 259 acceleration can only be seen from SP6 ($x = 1.6$ m) for cases C1 and C2, reaching a velocity of
 260 1570 m/s and 2045 m/s, respectively, at the position of SP7. By comparing with the baseline case,
 261 the results demonstrate again that higher turbulence intensity enhances the acceleration of the flame
 262 propagation. Therefore, when the concentration of the argon gas is at a low level, the turbulence
 263 dominates the wave propagation and DDT process.

264

265 3.4 The effect of injection position

266 Three jet positions are chosen to examine the injection position effect on DDT and wave
 267 propagation behavior. The distance between position C and SP1 is 10 cm (here 0 in the x -axis
 268 represents the first shock pin location, hence $x = -0.1$ m for position C). The other details regarding
 269 the jet positions are shown in Fig. 2. The jet parameters of this group are shown in Table 4.

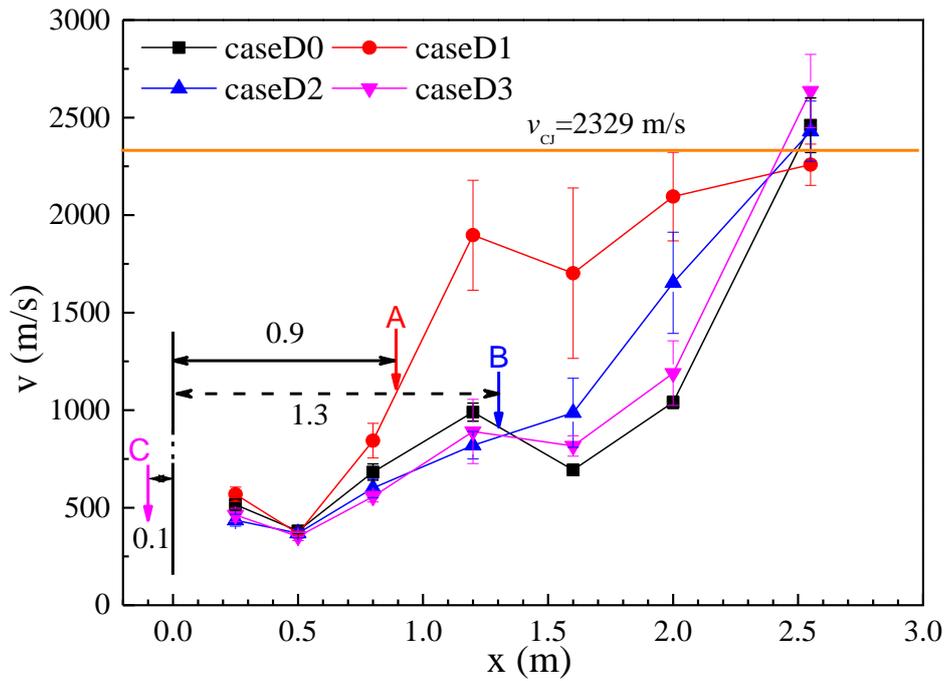
270

271

Table 4 Injection parameters for experiments with varying injection location.

Case #	P_i /kPa	t_i /ms	Position	x /m
D0	0	0	N/A	N/A
D1	200	15	A	0.9
D2	200	15	B	1.3
D3	200	15	C	-0.1

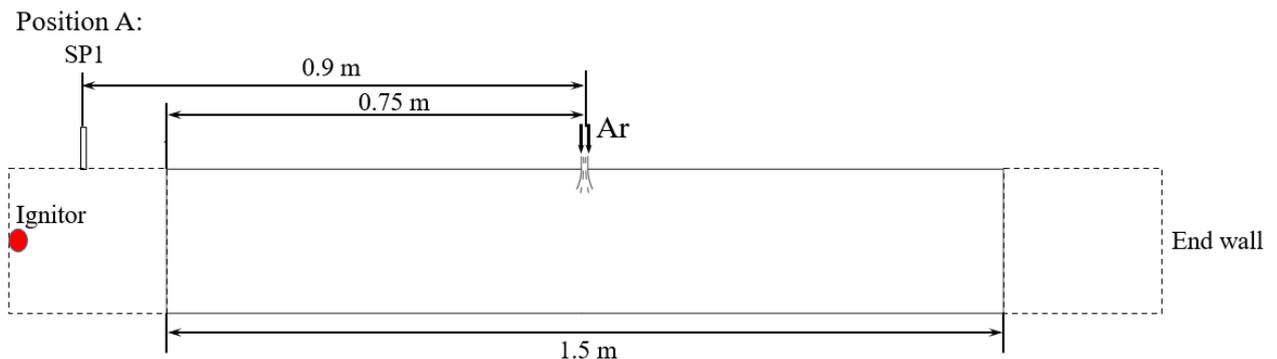
272



273
274 **Fig. 7** Velocity of the leading wave with different injection location.
275

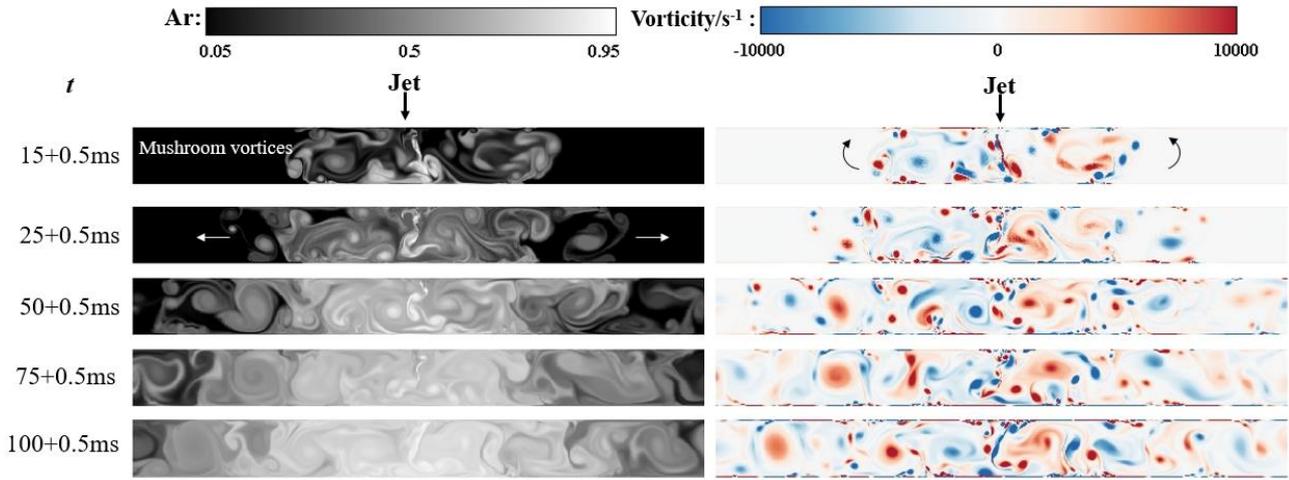
276 Figure 7 shows the wave velocity behavior during DDT with the argon jet placed at different
277 positions. For the case D1, an enhanced acceleration of the propagation velocity is clearly observed.
278 For the injection position located at the middle of the tube (case D2), only a small enhancement is
279 seen on the wave acceleration. At this position B, the baseline case already shows the wave
280 propagating at a high velocity about 700 m/s supported possibly by a high level of turbulence
281 originated from the evolving flame acceleration process. The added turbulence by the argon jet thus
282 has less prominent effect on such wave condition. For the last case D3 where the jet is introduced in
283 the slow deflagration stage, there is also no significant influence on the wave propagation.
284

285 **3.5 Effect of the turbulence evolution on the wave acceleration**



286

(a)



288

289

(b)

290

Fig. 8 The evolution of the distribution of Ar and the vortices formed at position A

291

292

293

294

295

296

297

298

299

300

The two-dimensional turbulence flow structures generated by argon jet with two different positions are simulated. Fig. 8(b) shows the distribution of Ar and vorticity at corresponding time to the experiments B1~B4 mentioned in Section 3.2 which placed the jet at position A, as shown in Fig. 8(a). It can be seen that the jet has already collided with the bottom wall when t_i is 15 ms, forming mushroom vortices. The vortices formed on the both sides of the jet inversely roll up. The vortices formed earlier are pushed away by the following jet due to the increasing mass of Ar. It is obvious that peripheral vortices gradually dissipate during their paths to the upstream/downstream of the jet, which is due to the collisions from vortex-to-vortex and vortex-to-wall. Note that both the mass of Ar and the vorticity increase with the extending of t_i , hereby a criterion is proposed to evaluate the dominant factor of the concentration and the turbulence on the propagation of flame.

301

302

303

304

As Ar was considered as an inert gas suppressing the combustion, while the turbulence structure especially the vortices have a positive influence on flame propagation, both the concentration of Ar and the vorticity have been nondimensionalized. The concentration Ar is nondimensionalized with the initial density of mixture:

305

$$C_{Ar} = \int_A \frac{\rho Y_{Ar}}{\rho_0} dA \quad (1)$$

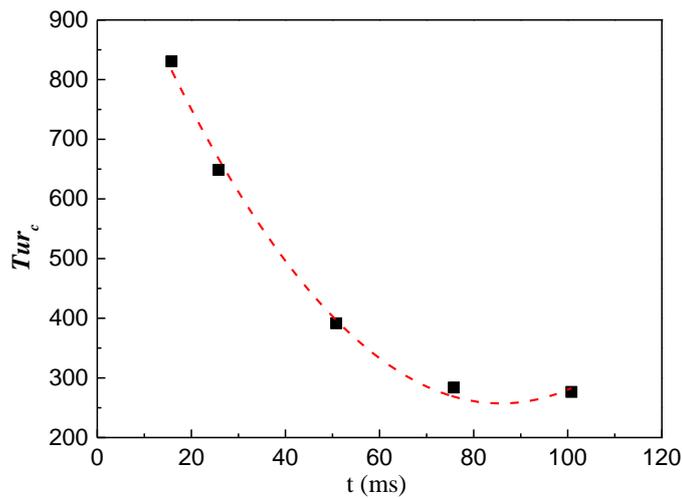
306 Where ρ , ρ_0 represent the current mixture density and the initial mixture density respectively;
 307 Y_{Ar} is the current mass fraction of Ar; A is the area of the computation domain. The dimensionless
 308 treatment of vorticity (Vor) can be described as follows:

$$309 \quad Vor = \int_A \left| \frac{\partial(v/v_j)}{\partial(x/l)} - \frac{\partial(u/v_j)}{\partial(y/h)} \right| dA \quad (2)$$

310 Where u , v and v_j are the x velocity component, y velocity component and jet velocity,
 311 respectively; l represents the length of the domain, the h represents the height; A is the same as the
 312 Equation 1. Thus, a dimensionless criterion factor can be obtained by calculating the ratio between
 313 two factors, i.e., Tur_c :

$$314 \quad Tur_c = \frac{Vor}{C_{Ar}} \quad (3)$$

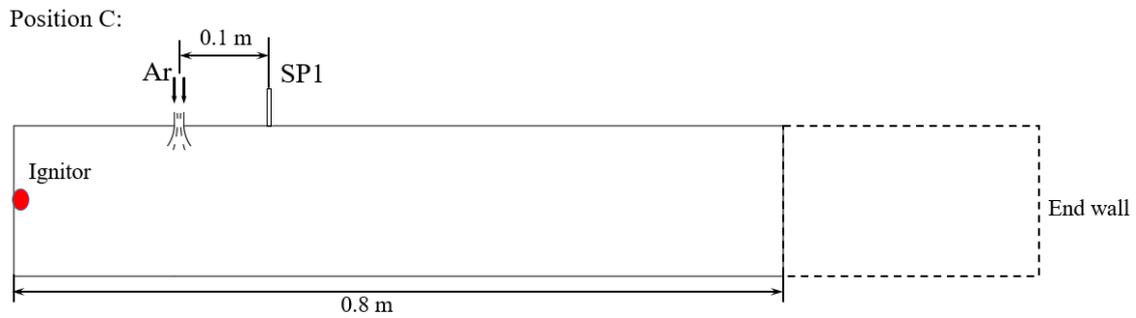
315 Fig. 9 shows Tur_c evolves with the jet duration time. Apparently, the curve drops dramatically
 316 while jet extending from 15 ms to 50 ms, and it keeps at a relatively stable value from 75 ms to 100
 317 ms. This is in good agreement with the results observed from experiments B1~B4. The results
 318 confirm that the positive effect of the turbulence is offset with the development of time, and even be
 319 oppressed by the combustion inhibition effect due to the increasing mass of Ar, as the jet duration
 320 time is consistently extended.



321

322

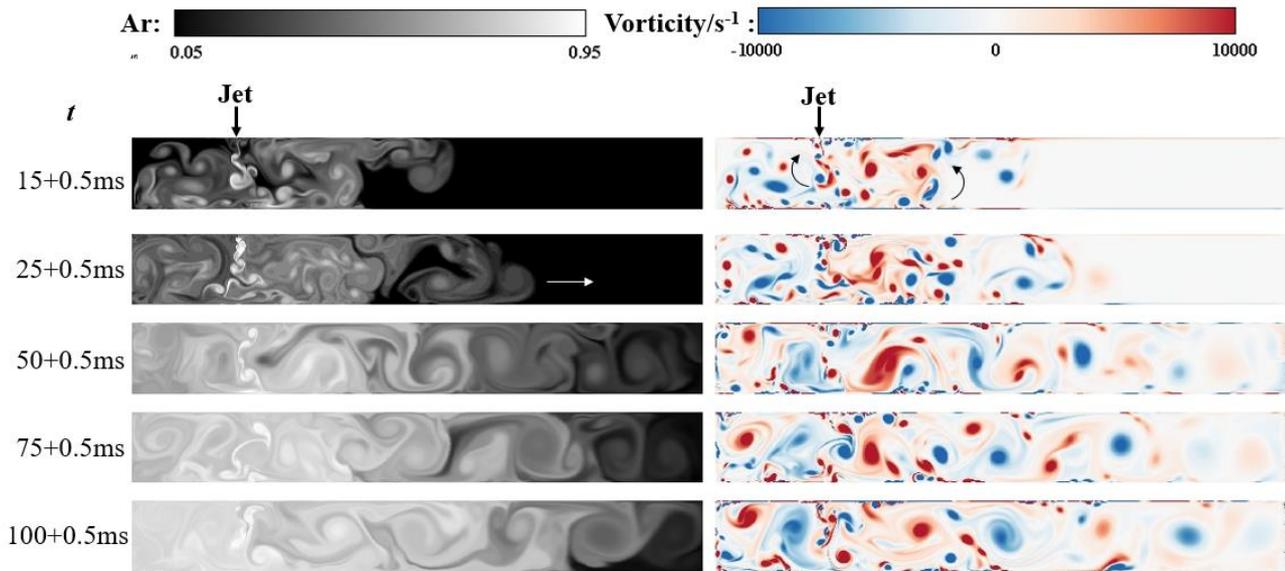
Fig. 9 The tendency of Vor/C varied with jet duration time



323

324

(a)



325

326

(b)

327

Fig. 10 The evolution of the distribution of Ar and the vortices formed with jet placed at position C

328

329

330

331

332

333

334

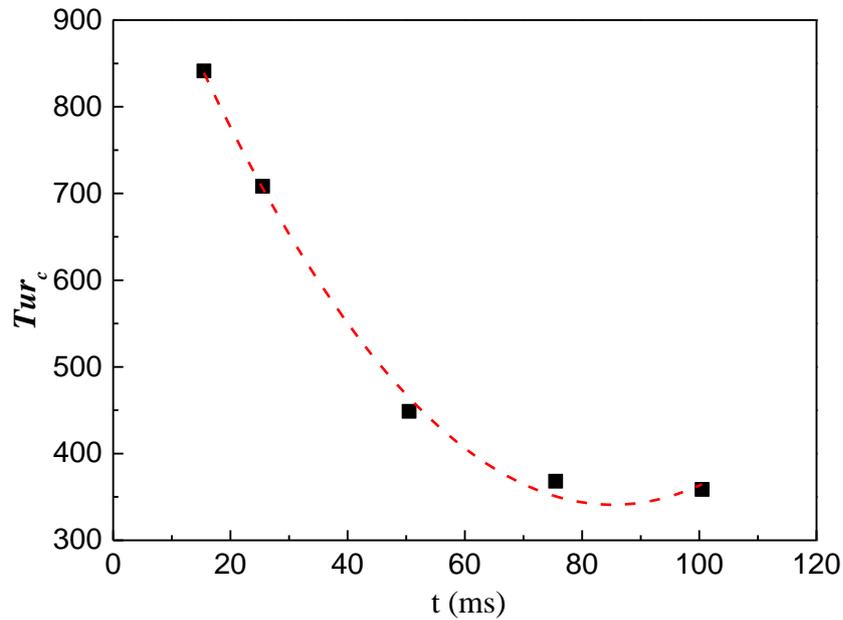
335

336

337

Fig. 10(b) depicts the Ar distribution and the vortices evolution for the case that with the jet placed at position C (as shown in Fig. 10(a)). The Ar concentration and vortices evolution processes are similar to the case in which the jet placed at position A. The difference between those two positions is that the Ar concentration in position C increases to approximately 1 in the region on the left side of the jet as the jet is placed to the location adjacent to the ignition end wall. This scenario occurred mainly because of the expansion of Ar concentration is constrained by the left wall. However, the Ar injection has a positive effect on the formation of the vortices near the left end wall (clearly shown by the right column of Fig.10(b)). The vortices propagate to the downstream and gradually dissipate, which are similar as the case shown in Fig.8. To verify the competition effect between the negative influence of Ar dilution and the positive influence of the turbulence enhancing

338 the vortices, the dimensionless criterion Tur_c is then calculated and shown in Fig. 11. It can be seen
339 from Fig. 11 that, with the increasing of injection time t_i , the tendency of Tur_c curve presents a
340 monotonous decrease, indicating the dominant factor affecting the DDT behavior is the Ar
341 concentration, rather than the stretch effect of the vortices. It is noteworthy that Ar jet is placed at
342 position C near the ignitor, and therefore the increased Ar concentration with longer t_i greatly
343 inhibits the ignition progress, resulting in a slower flame propagation velocity than the one with the
344 jet placed at position A.



345
346 Fig. 11 The tendency of Vor/C varied with jet duration time for the jet placed at position C

347
348 By comparing the Ar distribution and vortices evolution of the cases with different jet positions,
349 it is obvious that as a jet having a shorter distance to the ignitor, resulting more Ar concentrate on
350 the left side of the jet contacting the flame front primarily. The suppression effect from the
351 increasing inert gas mass is more prominent than the flame-acceleration effect from vortices. The
352 dimensionless criterion Tur_c is verified to be an adequate parameter to estimate the dominate
353 mechanism in flame propagation and the initial stage of DDT as inert jet is applied for detonation
354 enhancement.

355

356 **4. Conclusion**

357 With the increasing interest in employing detonation as fast combustion mode for advanced
358 propulsion systems, controlled initiation and rapid onset of detonation from DDT become desirable.
359 In this study, we explore the use of an inert argon jet as an enhancement method by its turbulence
360 generation to promote further flame acceleration and strengthen the lead shock wave for DDT. A
361 series of experiments and numerical simulations were conducted to investigate effects of various
362 injection parameters on DDT, including the initial injection pressure, duration time and jet position.
363 The results show that turbulence generated by argon injection accelerates the wave propagation
364 velocity when the wave is at the fast deflagration regime with velocity above 500 m/s. At such
365 condition, the enhanced DDT process by the jet has no noticeable variation with increasing P_i .
366 However, either a long injection time or an exceeding injection pressure will increase the local
367 argon concentration, and prohibit the wave acceleration. The present study has proved that higher
368 turbulence intensity induced by higher P_i and shorter t_i has better performance on enhancing the
369 DDT. In the experiments, different injection positions also are tested and found have impacted the
370 DDT process. Only an appropriate injection position can make a positive effect on the DDT process.
371 Additionally, by analyzing the experiments and numerical simulations of the cases with different jet
372 positions, the dimensionless criterion Tur_c proposed has demonstrated the dominant factor of the
373 flame propagation. The results show that the suppression effect by Ar dilution is more prominent
374 than the stretch dynamics on the flame acceleration by vortices with the increasing of t_i .

375

376 **Acknowledgment**

377 This work is supported by the National Natural Science Foundation of China – China (Grant Nos.:
378 91741114 and 11772199).

379
380
381
382
383
384
385
386
387
388
389
390
391
392
393
394
395
396
397
398
399
400
401
402

References:

- [1] Roy G D, Frolov S M, Borisov A A, Netzer D W, Pulse detonation propulsion: challenges, current status, and future perspective. *Prog Energy Combust* 2004;30 (6): 545-672.
- [2] Kailasanath K, Recent developments in the research on pulse detonation engines. *AIAA Journal* 2003;41 (2): 145-59.
- [3] Xiang G, Li H, Cao R, Chen X, Study of the features of oblique detonation induced by a finite wedge in hydrogen-air mixtures with varying equivalence ratios. *Fuel* 2020;264: 116854.
- [4] Xiang G X, Gao X, Tang W J, Numerical study on transition structures of oblique detonations with expansion wave from finite-length cowl. *Phys Fluids* 2020;29 (5): 56108.
- [5] Zhou R, Wang J P, Numerical investigation of flow particle paths and thermodynamic performance of continuously rotating detonation engines. *Combust Flame* 2012;159 (12): 3632-45.
- [6] Wolański P, Detonative propulsion. *Proc Combust Inst* 2013;34 (1): 125-58.
- [7] Zhang B, Liu H, Yan B J, Velocity behavior downstream of perforated plates with large blockage ratio for unstable and stable detonations. *Aerosp Sci Technol* 2019;86: 236-43.
- [8] Zhang B, Liu H, Wang C, Yan B J, An experimental study on the detonability of gaseous hydrocarbon fuel - oxygen mixtures in narrow channels. *Aerosp Sci Technol* 2017;69: 193-200.
- [9] Zhang B, Liu H, Theoretical prediction model and experimental investigation of detonation limits in combustible gaseous mixtures. *Fuel* 2019;258: 116132.
- [10] Zhang B, Liu H, Yan B, Ng H D, Experimental study of detonation limits in methane-oxygen mixtures: Determining tube scale and initial pressure effects. *Fuel* 2020;259: 116220.

- 403 [11] Zhang B, Detonation limits in methane-hydrogen-oxygen mixtures: Dominant effect of
404 induction length. *Int J Hydrogen Energ* 2019;44 (41): 23532-7.
- 405 [12] Zhang B, Liu H, Wang C, On the detonation propagation behavior in hydrogen-oxygen
406 mixture under the effect of spiral obstacles. *Int J Hydrogen Energ* 2017;42: 21392-402.
- 407 [13] Cao W, Gao D, Ng H D, Lee J H S, Experimental investigation of near-limit gaseous
408 detonations in small diameter spiral tubing. *Proc Combust Inst* 2019;37 (3): 3555-63.
- 409 [14] Gao Y, Ng H, Lee J H S, Near-limit propagation of gaseous detonations in thin annular
410 channels. *Shock Waves* 2017;27 (2): 199-207.
- 411 [15] Gao Y, Zhang B, Ng H D, Lee J H S, An experimental investigation of detonation limits in
412 hydrogen-oxygen-argon mixtures. *Int J Hydrogen Energ* 2016;41: 6076-83.
- 413 [16] Gao Y, Lee J H S, Ng H D, Velocity fluctuations near the detonation limits. *Combust Flame*
414 2014;161 (11): 2982-90.
- 415 [17] Lee J H S, *The detonation phenomenon*, Cambridge University Press, 2008.
- 416 [18] Zhang B, Mehrjoo N, Ng H D, Lee J H S, Bai C H, On the dynamic detonation parameters in
417 acetylene-oxygen mixtures with varying amount of argon dilution. *Combust Flame* 2014;161
418 (5): 1390-7.
- 419 [19] Zhang B, Kamenskihs V, Ng H D, Lee J H S, Direct blast initiation of spherical gaseous
420 detonations in highly argon diluted mixtures. *Proc Combust Inst* 2011;33 (2): 2265-71.
- 421 [20] Zhang B, Ng H D, Lee J H S, Measurement and relationship between critical tube diameter and
422 critical energy for direct blast initiation of gaseous detonations. *J Loss Prevent Proc* 2013;26:
423 1293-9.
- 424 [21] Zhang B, Ng H D, Lee J H S, Measurement and scaling analysis of critical energy for direct
425 initiation of gaseous detonations. *Shock Waves* 2012;22: 275-9.

- 426 [22] Bai C H, Zhang B, Xiu G L, Liu Q M, Chen M, Deflagration to detonation transition and
427 detonation structure in diethyl ether mist/aluminum dust /air mixtures. *Fuel* 2013;107: 400-8.
- 428 [23] Wang C, Zhao Y Y, Zhang B, Numerical simulation of flame acceleration and
429 deflagration-to-detonation transition of ethylene in channels. *J Loss Prevent Proc* 2016;43:
430 120-6.
- 431 [24] Wu M, Kuo W, Accelerative expansion and DDT of stoichiometric ethylene/oxygen flame
432 rings in micro-gaps. *Proc Combust Inst* 2013;34 (2): 2017-24.
- 433 [25] Zhang B, Liu H, The effects of large scale perturbation-generating obstacles on the propagation
434 of detonation filled with methane-oxygen mixture. *Combust Flame* 2017;182: 279-87.
- 435 [26] Zhang B, The influence of wall roughness on detonation limits in hydrogen-oxygen mixture.
436 *Combust Flame* 2016;169: 333-9.
- 437 [27] Shchelkin K I, Influence of tube roughness on the formation and detonation propagation in gas.
438 *J Exp Theor Phys* 1940;(10): 823-7.
- 439 [28] Ciccarelli G, Dorofeev S, Flame acceleration and transition to detonation in ducts. *Prog Energ*
440 *Combust* 2008;34 (4): 499-550.
- 441 [29] Dorofeev S B, Flame acceleration and explosion safety applications. *Proc Combust Inst*
442 2011;33 (2): 2161-75.
- 443 [30] Goodwin G B, Houim R W, Oran E S, Shock transition to detonation in channels with
444 obstacles. *Proc Combust Inst* 2017;36 (2): 2717-24.
- 445 [31] Cooper M, Jackson S, Austin J, Wintenberger E, Shepherd J E, Direct Experimental Impulse
446 Measurements for Detonations and Deflagrations. *J Propul Power* 2001;18 (5): 1033-41.
- 447 [32] Knox B W, Forliti D J, Stevens C A, Hoke J L, Schauer F R, A comparison of fluidic and
448 physical obstacles for deflagration-to-detonation transition, 49th AIAA Aerospace Sciences

- 449 Meeting including the New Horizons Forum and Aerospace Exposition, Orlando, Florida,
450 2011.
- 451 [33] Chambers J M, Ahmed K A, Investigation of Flame Regimes for Flame Acceleration to
452 Detonation, 2018 Joint Propulsion Conference, Cincinnati, Ohio, 2018.
- 453 [34] McGarry J P, Ahmed K A, Flame-turbulence interaction of laminar premixed deflagrated
454 flames. *Combust Flame* 2017;176: 439-50.
- 455 [35] McGarry J P, Ahmed K A, Laminar Deflagrated Flame Interaction with a Fluidic Jet Flow for
456 Deflagration-to-Detonation Flame Acceleration, 51st AIAA/SAE/ASEE Joint Propulsion
457 Conference, Orlando, FL, 2015.
- 458 [36] Chambers J, Ahmed K, Turbulent flame augmentation using a fluidic jet for
459 Deflagration-to-Detonation. *Fuel* 2017;199: 616-26.
- 460 [37] Tarrant D J, Chambers J M, Ahmed K A, Fast flame interaction with high pressure turbulent
461 jets for turbulent deflagration-to-detonation transition, 2018 Joint Propulsion Conference,
462 Cincinnati, Ohio, 2018. DOI: 10.2514/6.2018-4780
- 463 [38] Peng H, Huang Y, Deiterding R, Luan Z, Xing F, You Y, Effects of jet in crossflow on flame
464 acceleration and deflagration to detonation transition in methane-oxygen mixture. *Combust
465 Flame* 2018;198: 69-80.
- 466 [39] Zhang B, Chang X Y, Bai C H, End-wall ignition of methane-air mixtures under the effects of
467 CO₂/Ar/N₂ fluidic jets. *Fuel* 2020;270: 117485.
- 468 [40] Bai C H, Chang X Y, Zhang B, Impacts of turbulence on explosion characteristics of
469 methane-air mixtures with different fuel concentration. *Fuel* 2020;271: 117610.
- 470 [41] Chang X Y, Zhang B, Ng H D, Bai C H, The effects of pre-ignition turbulence by gas jets on
471 the explosion behavior of methane-oxygen mixtures. *Fuel* 2020;277: 118190.

- 472 [42] Cheng J, Zhang B, Liu H, Wang F, The precursor shock wave and flame propagation
473 enhancement by CO₂ injection in a methane-oxygen mixture. *Fuel* 2021;283: 118917.
- 474 [43] Cheng J, Zhang B, Liu H, Wang F, Experimental study on the effects of different fluidic jets on
475 the acceleration of deflagration prior its transition to detonation. *Aerosp Sci Technol* 2020: In
476 Press.
- 477 [44] Ding J, Liang Y, Chen M, Zhai Z, Si T, Luo X, nteraction of planar shock wave with
478 three-dimensional heavy cylindrical bubble. *Phys Fluids* 2018;30: 106109.
- 479
- 480
- 481

Effects of inert gas jet on the transition from deflagration to detonation in a stoichiometric methane-oxygen mixture

Jun Cheng¹, Bo Zhang^{1†}, Hoi Dick Ng², Hong Liu¹, Fuxing Wang¹

¹Shanghai Jiao Tong University
School of Aeronautics and Astronautics, Shanghai, 200240, China

²Concordia University
Department of Mechanical, Industrial and Aerospace Engineering
Montréal, QC, H3G 1M8, Canada

Declaration of interests

The authors declare that they have no known competing financial interests or personal relationships that could have appeared to influence the work reported in this paper.

The authors declare the following financial interests/personal relationships which may be considered as potential competing interests:

CRedit authorship contribution statement

Jun Cheng: Investigation, Data collection & analysis , Methodology.

Bo Zhang: Investigation, Validation, Writing - review & editing

Hong Liu: Supervision

Fuxing Wang: Supervision